



Wind Turbine Design Guideline DG03: Yaw and Pitch Rolling Bearing Life

T. Harris
J.H. Rumbarger
C.P. Butterfield

Technical Report
NREL/TP-500-42362
December 2009



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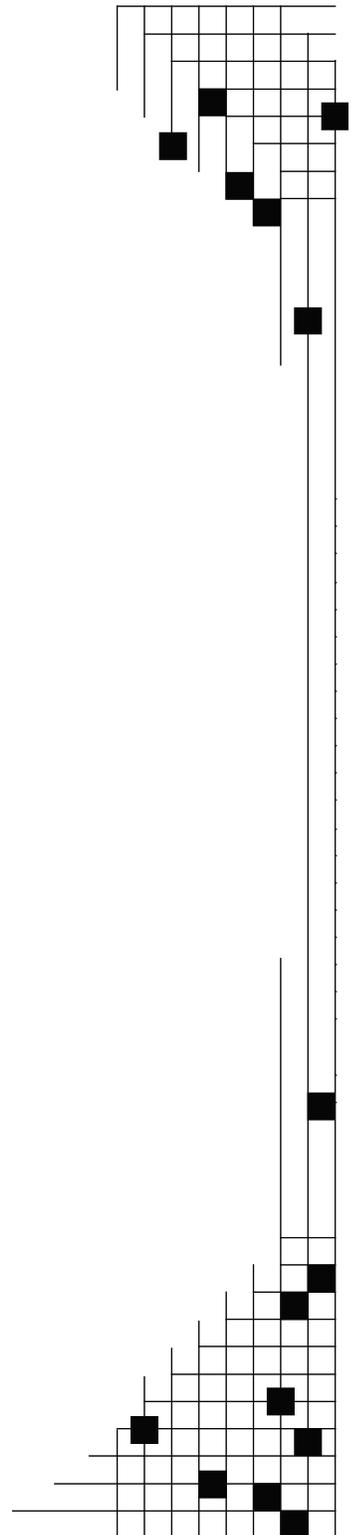


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Prepared under Task No. WE101131

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Foreword

This document is part of a series developed by the National Renewable Energy Laboratory, National Wind Technology Center. These guidelines are intended to be a design aid for wind turbine designers. The complete list of guidelines is provided below.

- DG01 Loads Analysis
- DG02 Strength Analysis
- DG03 Yaw and Pitch Rolling Bearing Life Design Guideline
- DG04 Gearbox Specification
- DG05 Control and Protection System Specification and Design

All of these documents—with the exception of the present document (DG03)—are in draft form and are available only by request sent to NREL, Sandy.Butterfield@NREL.Gov.

Modern wind turbines use large turntable bearings at the root of each blade to enable pitch angle changes and thus aerodynamic performance and load control. Yaw bearings are used for angular realignment of the nacelle into the predominant wind direction. These applications require long periods in nearly stationary positions with large stochastic loads. Due to this demanding load environment and the fact that bearings exist in the critical load path, their design becomes critical to the safety and reliability of most turbine designs.

This document attempts to introduce modern bearing-design practice and its relation to the unique requirements of wind turbine applications. The fundamental theory presented here is available in other textbooks referenced in this document, and included in the References section found at the end of this report.

First drafted in 1999 by John Rumbarger, this document was reviewed and updated extensively by Tedric Harris in 2008, and later in 2008 it was reviewed and edited by Bob Errichello.

Sandy Butterfield,
Chief Engineer
NREL Wind Program

November 10, 2008

List of Acronyms

DEH	Distortion Energy Hypothesis
FEA	finite element analysis
kNm	Kilo Newton-Meters
MPa	Mega Pascals
opm	oscillations per minute
rms	root-mean-square
rpm	rotations per minute
TDC	thin dense chrome

Contents

List of Figures.....	v
List of Tables	v
1. Introduction.....	1
2. Yaw and Pitch Rolling Bearing Design Types.....	1
3. Summary of Calculation Methods.....	3
3.1. General.....	3
3.2. American National Standard Method	3
3.3. International Standard Method	4
3.4. Stress-Life Method.....	4
3.5. Bearing Design Criteria	4
4. Calculation of Bearing Fatigue Life	5
4.1. Lundberg-Palmgren Theory.....	5
4.2. Applied Loading	5
4.3. Basic Dynamic Axial Load Rating	5
4.4. Dynamic Equivalent Axial Load	7
4.5. Bearing L_{10} Life	8
4.5.1. Basic L_{10} Life Equations.....	8
4.5.2. ANSI/ABMA Standard Modified Rating Life	9
4.5.2.1. Life Modification Factor for Reliability	9
4.5.2.2. Life Modification Factor for Material.....	10
4.5.2.3. Life Modification Factor for Lubrication	10
4.5.2.4. Life Modification Factor for Flexible Supporting Structure.....	11
4.5.2.5. Life Calculation for Oscillating Bearings.....	12
4.5.2.6. False Brinelling and Fretting Corrosion	13
4.5.3. ISO Standard Life.....	15
4.5.4. Stress-Life Method	16
5. Bearing Static Capacity.....	18
5.1. Permanent Deformation	18
5.2. Maximum Ball or Roller Load.....	18
5.3. Maximum Contact Stress.....	18
5.4. Contact Dimensions.....	19
5.4.1. Point Contact	19
5.4.2. Line Contact	20
5.5. Static Load Factor.....	22
5.6. Effect of Surface Hardness on Static Capacity	23
6. Estimation of the Case-Core Interface Depth	23
6.1. Design Stresses	23
6.2. Allowable Shear Stresses for Yield and Fatigue.....	24
6.3. Calculation of Subsurface Shear Stress	25
6.4. Evaluation of the Case-Core Interface.....	26
7. Effective Lubrication	27
8. Bearing Friction Torque.....	28

9. Miscellaneous Design Considerations	28
9.1. General.....	28
9.2. External Bolting.....	28
9.3. Cages or Separators.....	28
9.4. Integral Seals.....	29
Nomenclature	30
References.....	32
Appendix A. Wind Turbine Yaw Bearing.....	33
A.1. Operating Conditions and Bearing Dimensions.....	33
A.2. Dynamic Equivalent Loads.....	34
A.3. Basic Dynamic Axial Load Rating.....	35
A.4. Basic L_{10} Life.....	35
A.5. Modified L_{10} Life According to ANSI/ABMA Standard 11.....	35
A.6. Modified L_{10} Life According to ISO Standard 281.....	36
A.7. Modified L_{10} Life According to Stress-Life Method.....	37
A.8. Maximum Roller Load.....	38
A.9. Contact Semi-Width, Hertz Stress, and SF.....	38
A.10. Case-Core Interface.....	39
A.11. Evaluation for Potential Fretting Corrosion.....	40
A.12. Lubricant and Lubrication.....	40
A.13. Bearing Friction Torque.....	40
A.14. Design Summary.....	41
Appendix B. Wind Turbine Pitch Bearing.....	42
B.1. Operating Conditions and Bearing Dimensions.....	42
B.2. Bearing Dynamic Equivalent Loads and Speeds.....	43
B.3. Bearing Basic Dynamic Axial Load Rating for Rotation.....	44
B.4. Basic Dynamic Axial Load Rating for the Oscillating Bearing.....	45
B.5. Basic L_{10} Life.....	45
B.6. Modified L_{10} Life According to ANSI/ABMA Standard 11.....	45
B.7. Modified L_{10} Life According to ISO Standard 281.....	47
B.8. Modified L_{10} Life According to Stress-Life Method.....	48
B.9. Maximum Ball Load.....	48
B.10. Contact Ellipse Dimensions.....	49
B.11. Hertz Contact Stress and Static Load Factor.....	50
B.12. Case–Core Interface.....	50
B.13. Hertz Stress Limits for Avoidance of Fretting Corrosion.....	51
B.14. Dither Angle.....	52
B.15. Lubrication.....	52
B.16. Friction Torque.....	52
B.17. Miscellaneous.....	53
B.18. Summary.....	53

List of Figures

Figure 1. Yaw and pitch bearing types (eight-point contact ball bearing, four-point contact ball bearing, cross roller (x) bearing)	2
Figure 2. Pitch bearing coordinates and loads	3
Figure 3. Application of loads for a single-bearing system	7
Figure 4. Amplitude angle for one cycle of oscillation	12
Figure 5. Schematic of two solid elastic curved bodies (Body a, Body b) compressed by load Q	19
Figure 6. Radii of curvature for a four-point contact ball bearing	21
Figure 7. Typical subsurface shear stress curve	24
Figure 8. Allowable subsurface shear stresses	25
Figure 9. Subsurface shear stress parameter versus z/b and b/a (Curve A: $b/a = 0$; B: $b/a = 0.1$; C: $b/a = 0.2$; D: $b/a = 0.3$; E: $b/a = 0.4$; F: $b/a = 0.5$)	26
Figure A-1. Yaw bearing coordinates and loads	33
Figure B-1. Pitch bearing coordinates and loads	42
Figure B-2. Four-point contact ball bearing	43
Figure B-3. Eight-point contact ball bearing	46

List of Tables

Table 1. Values of f_{cm} for Yaw and Pitch Ball Bearings $f_i = f_o = 0.53$	6
Table 2. Values of f_{cm} for $\alpha = 45^\circ$ Yaw and Pitch Roller Bearings	7
Table 3. Life Modification Factor a_1 for Reliability	9
Table 4. Life Modification Factor a_2 for Material Hardness	10
Table 5. Life Modification Factor a_4 for Flexible Support Structure	12
Table 6. Constants and Exponents for Thrust Bearings (Using Equation 24)	16
Table 7. Constants c_1 and c_2 Versus Contamination Level	16
Table 8. Mises Fatigue Limit Stress ($\Phi_{VM,lim}$) for Various Steels	17
Table 9. Maximum Hertz Stress (Permanent Deformation = $0.0001D$)	18
Table 10. Values a^* and b^* Versus $F(\rho)$	21
Table 11. Curvatures for a Four-Point Contact Ball Bearing	22
Table 12. Curvatures for a Cylindrical Cross Roller Bearing	22
Table 13. Hardness Reduction Factor f_s Versus Hardness	23
Table 14. Allowable Subsurface Shear Stresses	25
Table 15. Subsurface Shear Stress Parameter Versus z/b and b/a	26
Table 16. Hertz Stress Limits to Minimize Fretting Corrosion Failures in Yaw and Pitch Bearings	27
Table 17. Bearing Friction Coefficient μ Versus Bearing Type	28
Table A-1. Yaw Bearing Load Duty Cycle	33
Table A-2. Yaw Bearing Dimensions and Material Properties	34
Table A-3. Yaw Bearing Dynamic Equivalent Axial Loading for the Duty Cycle	34
Table A-4. Q_{max} for Equivalent and Limit Loads	38
Table A-5. Q_{max} for Equivalent and Limit Loads	39

Table A-6. Comparison of Calculated Hertz Stresses with Recommended Limit Stresses to Avoid Fretting in Bearing Raceways	40
Table A-7. Bearing Friction Torque for Average and Limit Loads.....	40
Table B-1. Bearing Duty Cycle Speeds and Loads.....	42
Table B-2. Four-Point Ball Bearing Dimensions and Material Properties	43
Table B-3. Bearing Dynamic Equivalent Axial Loads over the Duty Cycle.....	43
Table B-4. Eight-Point Ball Bearing Dimensions and Material Properties	46
Table B-5. Q_{\max} for Equivalent and Limit Loads.....	49
Table B-6. Comparison of Calculated Hertz Stresses with Recommended Limit Stresses to Avoid Fretting in Bearing Raceways	51
Table B-7. Bearing Friction Torque for Operating and Limit Loads	52

1. Introduction

This report describes the design criteria, calculation methods, and applicable standards recommended for use in performance and life analyses of ball and roller (rolling) bearings for yaw and pitch motion support in wind turbine applications. The formulae presented here for rolling bearing analytical methods and bearing-life ratings are consistent with methods in current use by wind turbine designers and rolling-bearing manufacturers. Methods for determining the loads and moments to be used for rating yaw and pitch bearings are covered in a separate design guide, available upon request from Sandy.Butterfield@NREL.Gov.

National Renewable Energy Laboratory (NREL) Guideline DG03 can be used with relevant standards and guidelines either for designing yaw and pitch rolling bearings for wind turbines or for verifying a design. This guideline provides a method for estimating the life of yaw and pitch rolling bearings used in wind turbines and gives general recommendations for enhancing bearing life in these applications.

2. Yaw and Pitch Rolling Bearing Design Types

Large wind turbines (those rated at more than 250 kW) use ball or roller bearings with special configurations for blade retention pitch bearing and yaw bearing locations. Three common bearing configurations are shown in Figure 1. The bearings consist of two ring-rolled forgings forming the outer and inner raceways and a complement of either balls or rollers. The inner and outer continuous-ring forgings have mounting holes that allow the bearing to be bolted directly to the supporting structures. The balls or rollers are inserted into the bearing through a radial cylindrical hole in one of the rings. The hole then is closed using a removable loading plug contoured to the ball path or roller path surface. The three bearing types shown in Figure 1 have two distinct thrust load paths and can carry simultaneous radial, axial, and overturning moment loads.

It is common practice to cut a spur gear integral with one of the bearing rings, especially for the yaw bearing application. A detailed cross-sectional sketch and cutaway sketch of a four-point contact ball bearing is shown in Figure 1. Individual plastic spacers or thin section cage arc segments are used to separate the balls. The spacers are cylindrical with a hemispherical end (not shown in Figure 1). The individual rollers in the cross-roller bearing are separated by plastic, saddle-shaped spacers. The rollers in the cross-roller bearing alternate in their orientation to carry load.

The inner and outer rings are hardened from 250 to 300 Brinell hardness (HB). This is referred to as the “core hardness” of the ring. This core hardness should provide adequate core yield and fatigue strength, yet remain at a hardness low enough to facilitate machining of the rings, gear teeth, and mounting-bolt holes. The actual ball or roller path (rolling contact surface) is induction heated, quenched, and tempered to provide a hard surface or “case.” The surface hardness of the raceway is a minimum of 58 HRC (Rockwell C scale hardness). The depth of the hardened case is defined as the depth to a hardness of 50 HRC.

The four-point contact ball bearing type shown in Figure 1 consists of a single row of balls separated by cage segments or plastic spacers. The ball groove configuration is that of a gothic arch, which provides two distinct thrust load paths for each ball. The eight-point contact ball bearing type shown in Figure 1 is similar, but it has two rows of balls in gothic arch ball grooves.

The two-row, eight-point contact ball bearing type is more costly to manufacture than the single-row, four-point contact ball bearing. In addition to having a second row of balls and separators, the two-row bearing must be repeatedly assembled and disassembled during manufacture to accurately measure and match the internal diametral clearance or preload of the two ball rows.

The main advantages of the eight-point contact ball bearing, as compared to the four-point contact ball bearing, are:

- Lower ball loads;
- Lower Hertz stresses;
- Less required case depth; and
- Increased fatigue life.

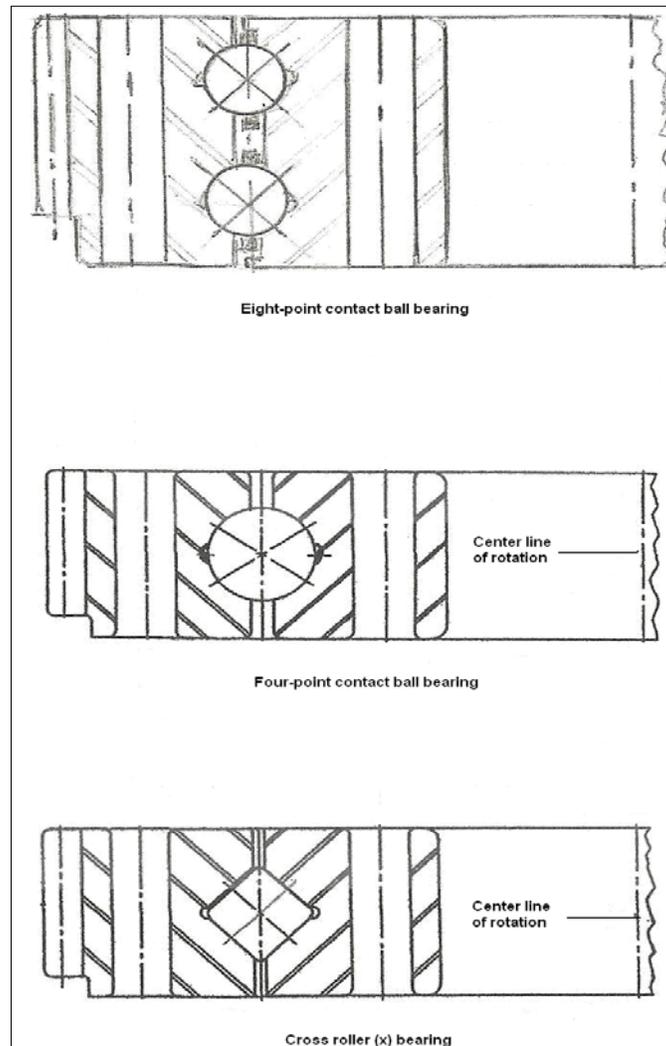


Figure 1. Yaw and pitch bearing types (eight-point contact ball bearing, four-point contact ball bearing, cross roller (x) bearing)

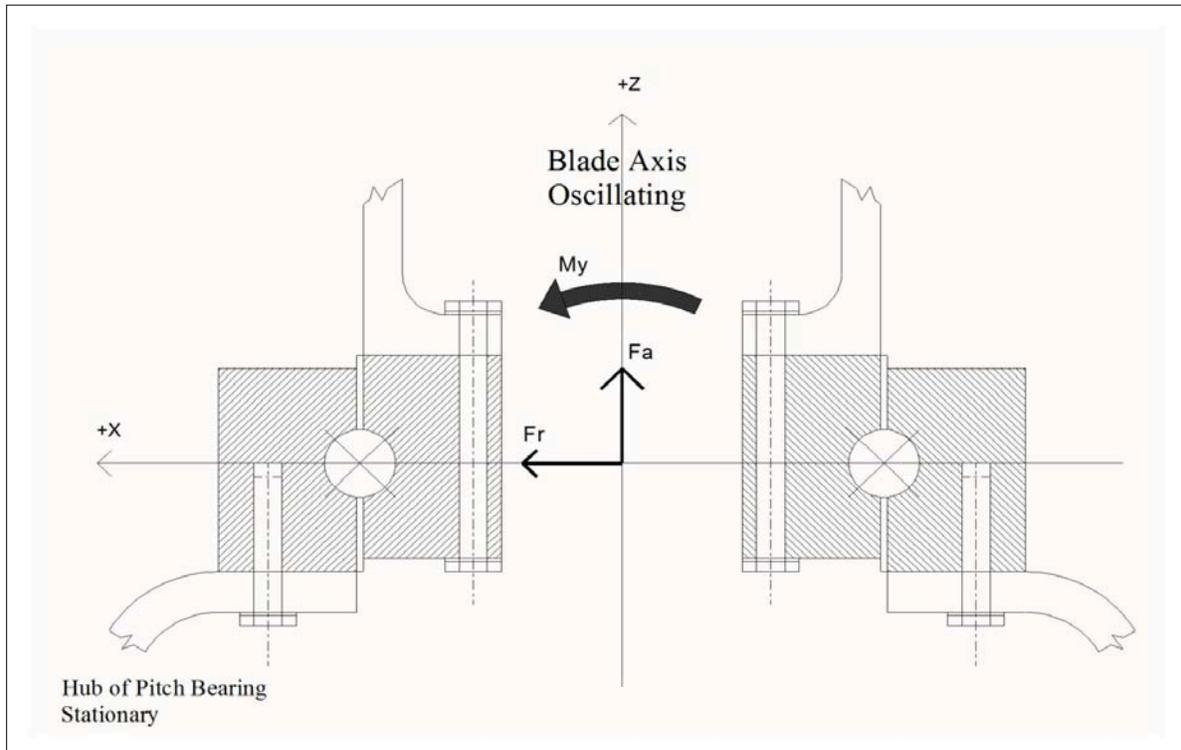


Figure 2. Pitch bearing coordinates and loads

3. Summary of Calculation Methods

3.1. General

The method for estimating rolling bearing fatigue life in wind turbine yaw and pitch applications is based on—and similar to—the equations and procedures provided in the International Standards and American National Standards (1–4) for load and life ratings. It basically involves five steps:

1. Calculating the dynamic load rating for the bearing;
2. Calculating life adjustment factors for the application;
3. Determining the rolling element–raceway load at each rolling element angular location;
4. Calculating the dynamic equivalent load; and
5. Using the values to calculate the bearing fatigue life.

3.2. American National Standard Method

The dynamic load rating for the bearing can be obtained from a bearing manufacturer’s catalog or it can be calculated using the equations presented in this guide; these equations are examined in detail by Harris and Kotzalas [5]. Fatigue life then is calculated according to American National Standards Institute/American Bearing Manufacturers Association (ANSI/ABMA) Standard 9 [3] for a ball bearing or ANSI/ABMA Standard 11 [4] for a roller bearing. The standard life rating model then is modified for application to oscillating bearings. This modification also is supplied in Harris & Kotzalas [5].

Life adjustment factors for bearing material, lubrication, and contamination, described by Zaretsky [6] and given in Harris & Kotzalas [5], can be applied to wind turbine yaw and pitch bearings discussed in this guide. For a bearing subjected to combined radial, axial, and moment loading, as indicated in Figure 2, a dynamic equivalent load must be used to determine life. Further, if the applied loads and speeds vary over the length of time that the bearing operates, then an equivalent load also must be determined. Equations to calculate equivalent loads are presented in Harris & Kotzalas [5] and also are supplied in this guide.

The foregoing method is presented here. It is expected, however, that this eventually will be superseded by the International Standard method.

3.3. International Standard Method

After the dynamic load rating for the bearing is obtained from the bearing manufacturer's catalog or calculated using the equations provided in this report, fatigue life then is calculated according to ISO Standard 281 [1] for either a ball or roller bearing. A single, integrated life adjustment factor, designated a_{ISO} , then is calculated according to the methods shown in ISO Standard 281 [1] and provided in detail in Harris & Kotzalas [5]. The use of this method requires the input of a fatigue limit load (P_w); this parameter generally is given in bearing manufacturers' catalogs.

3.4. Stress-Life Method

The stress-life method of fatigue life calculation is similar to the ISO Standard 281 method in that it employs an integrated life adjustment factor, a_{SL} . Included in the determination of a_{SL} is a fatigue limit stress $\Phi_{VM,lim}$ based on the Von Mises stress criterion. This is a different stress than that used by the other methods, which are based on the maximum orthogonal shear stress τ_0 as the failure criterion. In determining a_{SL} , all conditions that influence fatigue endurance are converted to stresses and added as vectors to create the Von Mises stress at each point in the stressed material. The Von Mises stress at each subsurface and surface point is compared against $\Phi_{VM,lim}$ to determine the possibility and probability of fatigue failure. The stress-life method is explained in detail in Harris and Kotzalas [5] and also in Barnsby et al. [7].

3.5. Bearing Design Criteria

The proper design of a yaw or pitch bearing must satisfy five design criteria and the miscellaneous considerations listed below. Each of the following criterion is addressed in detail in this guide.

1. Bearing fatigue life (rolling contact fatigue)
2. Bearing static capacity
3. Adequate case depth and core hardness
4. Adequate lubrication (surface failure)
5. Friction torque
6. Miscellaneous
 - A. External bolting
 - B. Cages or separators
 - C. Integral seals

The relationships used in the method for determining the basic dynamic capacity in oscillation, life adjustment factors, and equivalent load are supplied in Section 4.

4. Calculation of Bearing Fatigue Life

4.1. Lundberg-Palmgren Theory

In general, the method for demonstrating adequate life for rolling bearings is to use mathematical models to estimate the minimum life. For rolling bearings this typically is done using a model proposed by Lundberg and Palmgren [8, 9]. The Lundberg-Palmgren theory forms the basis of ISO Standard 281 [1], ANSI/ABMA Standard 9 and Standard 11 [3, 4], and the stress-life method [7]. Discussion of the Lundberg-Palmgren model as it can be applied using the ANSI/ABMA standards [3, 4], and the determination of coefficients and material factors are summarized in Rumbarger and Poplawski [10].

The Lundberg-Palmgren theory and its derivative methods determine rolling contact fatigue life according to the statistical probability of the survival of a bearing; the probability of failure is the complement of probability of survival. Bearing life L is expressed in terms of millions of revolutions of the rotating raceway. The usual industry practice is to specify bearing life according to a 10% probability of failure, or L_{10} . L_{10} also can be expressed in hours of operation. The Lundberg-Palmgren model refers to the visible evidence of rolling contact fatigue of the raceway surfaces. Although rolling elements (i.e., balls, rollers) also could fail, the Lundberg-Palmgren theory does not accommodate that phenomenon. Derivative methods, based on Lundberg-Palmgren (e.g., Barnsby et al. [7]), include the probability of rolling element failure.

Actual operational failure (ceasing to rotate satisfactorily) of a yaw or pitch bearing occur after the accumulation of damage. In some applications, this increased time or life margin can equal or exceed the L_{10} life. A periodic oil or grease analysis can be helpful in determining the approach of a field failure.

The life model is based on a consideration of the stressed volume of material in a ball-raceway or a roller-raceway contact and the number of stress repetitions. This method has been extended and used in life rating of oscillating rolling bearings; see Harris and Kotzalas [5].

4.2. Applied Loading

Ball and roller bearings used for wind turbine yaw and pitch bearings, as shown in Figure 1, are thrust-type bearings. The principal load is an eccentrically applied thrust which results in an axial load and an overturning moment load. The distribution of the applied load to the individual balls or rollers is predominantly one of thrust, where the number of loaded elements is greater than one-half. Ideally, the radial load is distributed over half (or fewer) of the rolling elements. Applied simultaneously with thrust and overturning moment loads, radial load affects the thrust type of load distribution but does not significantly alter it.

4.3. Basic Dynamic Axial Load Rating

Basic dynamic axial load rating, or thrust basic dynamic capacity, is defined as the constant centric thrust load that a rolling bearing theoretically could endure for a rating life (L_{10}) of 1 million revolutions. The load rating standards [1, 3, 4] provide the following formula for a basic dynamic axial load rating for thrust ball bearings.

$$C_a = f_{cm} (i \cos \alpha)^{0.7} Z^{2/3} D^{1.8} \tan \alpha \quad (1)$$

The material factor f_{cm} based on CVD 52100 steel material is given in Table 1. Compared with the values given in the standards [1, 3], these values have been extrapolated to accommodate large-diameter bearings (i.e., γ values to 0.002, $\gamma = D \cos \alpha / d_m$). Because standard values of factor f_{cm} are based on a raceway groove conformity of 0.52 ($f = \text{raceway groove curvature radius}/D$), and because yaw and pitch ball bearings frequently have a 0.53 conformity, Table 1 presents values of f_{cm} adjusted for the latter ball-raceway conformity. To obtain f_{cm} for 0.52 conformity, the f_{cm} values of Table 1 must be multiplied by 1.172.

Table 1. Values of f_{cm} for Yaw and Pitch Ball Bearings ($f_i = f_o = 0.53$)

γ	$f_{cm} (\alpha = 45^\circ)$	$f_{cm} (\alpha = 60^\circ)$
0.002	28.84	26.84
0.004	35.50	33.05
0.006	40.09	37.32
0.008	43.69	40.67
0.010	46.70	43.48
0.020	57.35	53.35
0.030	64.56	60.12
0.040	70.21	65.33
0.050	74.48	69.44
0.060	78.42	72.99
0.070	85.53	75.87
0.080	84.19	78.31
0.090	86.52	80.53
0.100	88.40	82.30

The equations for the basic dynamic axial load rating, C_a , depend on the ball diameter. According to the load rating standards [1, 3], the capacity is reduced for balls larger than 25.4 mm (1 in.), as indicated in equation 2.

$$C_a = 3.647 f_{cm} (i \cos \alpha)^{0.7} Z^{2/3} D^{1.4} \tan \alpha \quad (2)$$

For thrust roller bearings, the load rating standards provide the following equation for basic dynamic axial load rating.

$$C_a = f_{cm} (l_e \cos \alpha)^{7/9} Z^{3/4} D^{29/27} \tan \alpha \quad (3)$$

The f_{cm} values for thrust roller bearings are given in Table 2. Compared with the values provided in the standards [1, 4], these values have been extrapolated to accommodate large-diameter bearings (i.e., γ values to 0.002, $\gamma = D \cos \alpha / d_m$).

Table 2. Values of f_{cm} for $\alpha = 45^\circ$ Yaw and Pitch Roller Bearings

γ	f_{cm}
0.001	66.36
0.002	77.41
0.004	90.30
0.006	98.81
0.008	105.32
0.010	110.65
0.020	128.91
0.030	140.76
0.040	149.60
0.050	156.61
0.060	162.32
0.070	167.07
0.080	171.03
0.090	174.35
0.100	177.12

4.4. Dynamic Equivalent Axial Load

A typical load application is shown in Figure 3. Radial (F_r), thrust (F_a), and overturning moment (M) loads are shown as applied to the center of the bearing coordinate system. The radial and moment loads sometimes are given as orthogonal vectors. The root-mean-square (rms) values of the two orthogonal components then are used to determine the magnitude of F_r and M . The dynamic equivalent axial load (P_{ea}) is defined as a constant centric (uniformly distributed) axial load; under whose influence a rolling bearing would have the same life as it would attain under the actual load conditions.

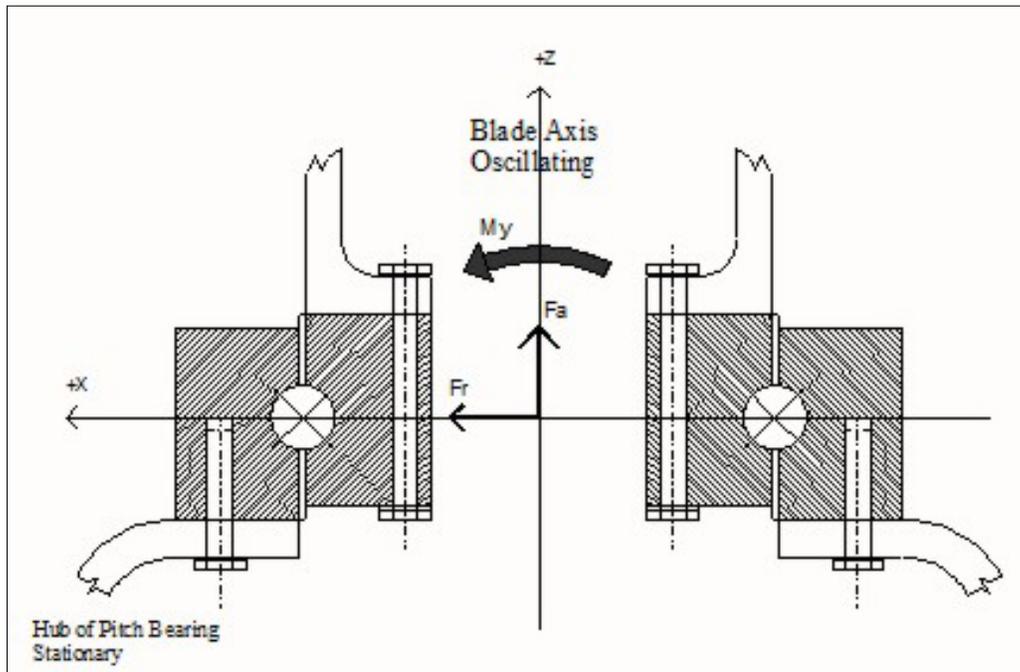


Figure 3. Application of loads for a single-bearing system

The equivalent axial load (P_{ea}) also is defined in terms of the individual ball or roller loads (Q_j) [5], where the subscript j refers to the angular location of the rolling element about the bearing axis. The following is for ball bearings.

$$P_{ea} = \left(\frac{1}{Z} \sum_{j=1}^{j=Z} Q_j^3 \right)^{1/3} Z \sin \alpha . \quad (4)$$

In equation 4, Z equals the total number of balls in the bearing (whether loaded or not loaded). The following equation is for cross-roller bearings.

$$P_{ea} = \left(\frac{1}{Z} \sum_{j=1}^{j=Z} Q_j^4 \right)^{1/4} Z \sin \alpha . \quad (5)$$

In equation 5, Z equals the number of rollers oriented to carry thrust load in one direction (whether loaded or not loaded). The dynamic equivalent thrust load also can be estimated from the applied loads as follows.

$$P_{ea} = 0.75F_r + F_a + \frac{2M}{d_m} \quad (6)$$

4.5. Bearing L_{10} Life

4.5.1. Basic L_{10} Life Equations

The life of an individual bearing is defined as the number of revolutions one of the bearing rings (or raceways) makes in relation to the other ring (or raceway) before the first evidence of fatigue develops in the material of one of the rings (or raceways) or rolling elements. For thrust ball bearings, the L_{10} life, expressed in millions of revolutions, is given by the following.

$$L_{10} = \left(\frac{C_a}{P_{ea}} \right)^3 \quad (7)$$

For thrust roller bearings the following is used.

$$L_{10} = \left(\frac{C_a}{P_{ea}} \right)^{10/3} \quad (8)$$

L_{10} life also can be determined in hours of operation, as follows, where speed N is in rpm.

$$L_{10}(hr) = \frac{L_{10} \cdot 10^6}{N \cdot 60} \quad (9)$$

The prorated bearing life for a bearing with a number of individual L_{10} life values at specified percentages of operating time is as follows.

$$L_{10} = \left(\sum_{k=1}^{k=n} \frac{t_k}{L_{10k}} \right)^{-1} \quad \sum_{k=1}^{k=n} t_k = 1 \quad (10)$$

Where t_k is the decimal fraction of time the bearing operated under the condition yielding life L_{10k} and n is the number of different operating conditions.

4.5.2. ANSI/ABMA Standard Modified Rating Life

4.5.2.1. Life Modification Factor for Reliability

According to ANSI/ABMA [3, 4], the modified rating life (L_{nm}) is the life modified for a different “reliability” (reliability = probability of survival = 1 – probability of failure) for special bearing materials and for conditions associated with lubrication. Zaretsky [6] applies additional factors such that the following is the case.

$$L_{nm} = a_1 a_2 a_3 \dots a_k L_{10} \quad (11)$$

Where:

a_1 = life modification factor for reliability;

a_2 = life modification factor for bearing steel or material;

a_3 = life modification factor for lubrication; and

a_k = life modification factors for other conditions.

From International Standard ISO 281, International Standard ISO 76, and American National Standard ANSI/ABMA 9-1990 [1, 3, 4], the life modification factor for reliability is shown in Table 3.

Table 3. Life Modification Factor a_1 for Reliability

Reliability (%)	L_{nm}	a_1
90.00	L10m	1.00
95.00	L5m	0.64
96.00	L4m	0.55
97.00	L3m	0.47
98.00	L2m	0.37
99.00	L1m	0.25
99.20	L0.8m	0.22
99.40	L0.6m	0.19
99.60	L0.4m	0.16
99.80	L0.2m	0.12
99.90	L0.1m	0.093
99.92	L0.08m	0.087
99.94	L0.06m	0.080
99.95	L0.05m	0.077

The a_1 factors in Table 3 can be used to estimate the increased reliability of a bearing design. As an example, consider a yaw bearing that has a modified rating life, L_{nm} equals 30 years, and the design specification calls for an L_{10} life of 20 years. The ratio of 2:3 would indicate a reliability of approximately 95%, as shown in Table 3.

4.5.2.2. Life Modification Factor for Material

According to the load rating standards, bearing rating life is based on CVD 52100 steel bearing rings and rolling elements, through-hardened to a minimum 58 HRC for both rings and rolling elements. As indicated in section 5 (below), yaw and pitch bearings tend to be manufactured using surface-hardening steel—either carburized or induction-hardened. One manufacturer of large-diameter four-point contact ball bearings and cross roller bearings adheres to a specification of 56 HRC minimum for the raceway hardness.

The a_2 life modification factor can be used to account for material hardness of less than HRC 58. Harris and Kotzalas [5] provide the following equation for the effect of hardness on bearing basic load rating.

$$C' = C \left(\frac{HRC}{58} \right)^{3.6} \quad (12)$$

Using the load-life relationships of equations 7 and 8, the life modification factor a_2 can be obtained from the following.

$$a_2 = \left(\frac{HRC}{58} \right)^{3.6p} \quad (13)$$

Where exponent $p = 3$ for ball bearings and $10/3$ for roller bearings. Using equation 13, Table 4 gives values for a_2 as a function of hardness.

Table 4. Life Modification Factor a_2 for Material Hardness

HRC	a_2 Ball Bearing	a_2 Roller Bearing
58	1.0	1.0
56	0.68	0.66
54	0.46	0.42

4.5.2.3. Life Modification Factor for Lubrication

The a_3 life modification factor is used to account for lubrication conditions. A common and distinguishing feature of yaw and pitch bearing applications is that both involve a rather slow oscillatory motion. For rolling bearings, the lubricant films that form between the raceway and rolling element surfaces mainly are the result of bearing internal speeds, operating temperatures, and lubricant viscous properties. If lubricant films are sufficient to completely separate the rolling contact surfaces, then bearing fatigue endurance is enhanced greatly. Conversely, when the generated lubricant films are so thin as to allow substantial metal-to-metal contact of the surfaces, bearing life is significantly less than that calculated using equations 7 and 8 unmodified.

To determine the effectiveness of lubrication, the generated lubricant film thickness is compared against the composite rms “roughness” of the rolling contact surfaces. A measure of lubrication effectiveness is the ratio Λ , defined by equation 14.

$$\Lambda = \frac{h_0}{(s_m^2 + s_{RE}^2)^{1/2}} \quad (14)$$

Where h_0 is the minimum lubricant film thickness occurring in the rolling element–raceway contacts, s_m is the raceway rms surface roughness, and s_{RE} is the ball or roller rms surface roughness. In a yaw or pitch bearing application, this is $\Lambda \ll 0.5$. This typically is the result of raceway surface finishes of 0.4 μm and slow speeds, notwithstanding grease lubricants having relatively high viscous properties.

The ability of a grease to function as a good boundary lubricant and to maintain the thickest oil film possible in the contact region is very important. The main function of the grease is to coat the rolling surfaces and to prevent corrosion damage or micropitting of the contact areas. The selected grease should have good boundary lubrication characteristics, such as antiwear and extreme-pressure additives. In small oscillatory motions, fretting corrosion or “false brinelling” of the rolling element contacts is the failure mechanism to prevent. The grease also should be able to protect the rolling surfaces from entrapped water. Only lubricants that have demonstrated good performance in yaw and pitch bearing applications should be used. To minimize the tendency to displace the lubricant from the contact region, the yaw or pitch bearing periodically should be rotated through a large-amplitude oscillation. The angle of rotation should be greater than three times the critical amplitude (θ_{crit}). This should be done at least daily and before any planned equipment shutdown. Based on the foregoing information, an a_3 life modification factor no greater than 0.1 is recommended.

4.5.2.4. Life Modification Factor for Flexible Supporting Structure

The bearing life rating standards and methods [1, 3, 4] are based on an assumption of rigid, flat, and circular bearing rings. Yaw and pitch bearings are constructed of large-diameter, thin, cross-section rings that are bolted to the supporting structures directly (*see* Figure 3). The bearing inner and outer raceways conform to the structural shapes of their supporting structures. Hard points in a supporting structure [6] result in a peak ball or roller load that is much greater than the element load expected under ideal mounting conditions. A wind turbine gearbox and platform mounting to the supporting tower generally is somewhat flexible, including some hard-point effects.

One method used to determine the influence of the supporting structures on bearing life is a complete finite element analysis (FEA) of the system. The two supporting structures are connected by spring elements that represent a group of the rolling elements. The resulting graph of the actual element loads to the ideal element loads indicates the effect of the structural deformations under load. Equations 4 and 5 are used to determine the dynamic equivalent thrust load.

Actual experience (FEA analysis) with the effects of various mounting structures on bearing life is limited. Table 5 shows some estimates of a_4 , the life modification factor for a flexible support.

Table 5. Life Modification Factor a_4 for Flexible Support Structure

Application	Condition	a_4
Pitch bearing	Hub deformations	0.85
Yaw bearing	Tubular tower	0.85
Yaw bearing	Four-leg tower	0.25

4.5.2.5. Life Calculation for Oscillating Bearings

Experience has shown that the fatigue life of rolling element bearings subjected to cyclic oscillation of the races under constant load cannot be described adequately by the same life formulas used to describe the fatigue life of continuously rotating bearings. The rating life (equations 7 and 8) for continuously rotating bearings must be modified for application to oscillating bearings.

For an oscillating bearing, the critical amplitude of oscillation (θ_{crit}) is defined as the angle of rotation of the inner raceway relative to the outer raceway, for which the raceway portion stressed by one rolling element (ball or roller) touches—but does not overlap—the raceway stressed by adjacent rollers.

$$\theta_{crit} = \frac{720^\circ}{Z(1 \mp \gamma)} \quad (15)$$

Where the upper sign refers to the outer raceway and the lower sign to the inner raceway.

The amplitude of oscillation is defined in Figure 4. Angle θ is one-half the total arc traced during one cycle of oscillation. Angle ϕ , used by Harris and Kotzalas [5], is one-quarter of the total arc traced during one cycle.

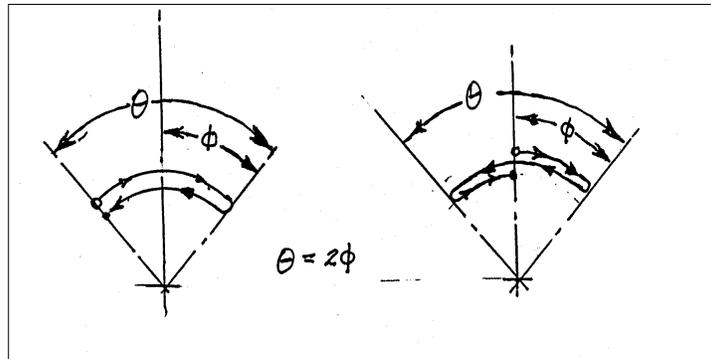


Figure 4. Amplitude angle for one cycle of oscillation

When $\theta > \theta_{crit}$, the contact stresses of the individual rolling elements overlap. The total stressed volume for each raceway then is proportional to the product of the arc length of the amplitude of oscillation, the element contact width b , and the depth to the critical shear stress amplitude. The total stressed volume and number of stress repetitions per cycle are identical to a bearing in continuous rotation when $\theta = 180^\circ$.

When $\theta < \theta_{crit}$, the individual element stresses do not overlap. Each rolling element has its own discrete stressed volume. These individual stressed volumes then must be combined statistically to calculate the overall capacity and life.

A third zone of operation is oscillation at very small amplitudes, where the stressed area (or footprint) between the element and the race only is partially uncovered and retraced. The amplitude θ_{dith} is defined as the width of the contact footprint divided by the radius of the rolling path. Repetition of oscillation cycles at small amplitudes is called “dither.”

$$\theta_{dith} = \frac{720^\circ b}{\pi \cdot d_m (1 \mp \gamma)} \quad (16)$$

Where the upper sign refers to the inner raceway contact and the lower sign refers to the outer raceway contact.

When $\theta < \theta_{crit}/2$, it is possible for fretting corrosion to occur. It therefore is advisable to rotate the bearings as often as possible to redistribute grease to the rolling element contacts. The angle of the rotation should be greater than θ_{crit} . The lubricant should be a grease that has proven capability for pitch and yaw bearings, and its base oil should have adequate viscosity and good antiwear additives.

When $\theta \leq \theta_{dith}$, fretting corrosion most likely will occur, and it is best to avoid operation under these very small oscillations. It is advisable to rotate the bearings as often as possible to redistribute grease to the rolling element contacts. The angle of the rotation should be greater than θ_{crit} . The lubricant should be a grease that has proven capability for pitch and yaw bearings, and its base oil should have adequate viscosity and good antiwear additives. Furthermore, it could be necessary to coat the raceways with thin dense chrome (TDC) or another hard coating, and to use solid-film lubricants.

4.5.2.6. False Brinelling and Fretting Corrosion

Fretting occurs between contacting surfaces that are pressed together and subjected to cyclic, relative motion of extremely small amplitude. It can occur in pitch and yaw bearings under specific conditions where the bearings are not rotating and are subjected to structure-borne vibrations caused by wind loads. Under these conditions, lubricant squeezes from between the rolling elements and raceways, and the motion of the surfaces is too small to replenish the lubricant. Natural oxide films that normally protect surfaces are disrupted, permitting metal-to-metal contact and causing adhesion of surface asperities.

Fretting begins with an incubation period during which the wear mechanism is mild adhesion and the wear debris is the black iron oxide magnetite (Fe_3O_4). Damage during incubation is termed false brinelling. If wear debris accumulates in amounts sufficient to dam lubricant from reaching the contact, then the wear mechanism becomes severe adhesion that breaks through the natural oxide layer and forms strong welds with parent material. Wear rate increases dramatically and damage escalates to fretting corrosion. Relative motion breaks welded asperities and generates the iron oxide hematite ($\alpha-Fe_2O_3$), a fine powder that is the reddish-brown color of cocoa. The wear debris is hard and abrasive and, in fact, has the same composition as jeweler’s rouge.

Fretting damages bearings by forming ruts along lines of contact. False brinelling ruts are shallow and the damage usually is benign. Fretting corrosion forms relatively deep ruts that can cause bearing failure. During operation, bearings with severe fretting corrosion can generate a sharp hammering noise, or the bearing can jam and cease to rotate. Furthermore, fretting corrosion can cause macropitting or initiate fatigue cracks which, if in high-stress areas, can propagate to failure.

The basic dynamic axial load rating for complete rotation must be modified to accommodate bearing oscillation when $\theta \geq \theta_{crit}$.

$$C_{a,osc} = C_a \left(\frac{180^\circ}{\theta} \right)^{1/p} \quad (17)$$

Where exponent p equals 3 for ball bearings and p equals 4 for roller bearings.

For ball bearings when $\theta < \theta_{crit}$, the following equation is used.

$$C_{a,osc} = C_a \left(\frac{180^\circ}{\theta} \right)^{3/10} Z^{0.033} \quad (18)$$

For roller bearings when $\theta < \theta_{crit}$, the following equation is used.

$$C_{a,osc} = C_a \left(\frac{180^\circ}{\theta} \right)^{2/9} Z^{0.028} \quad (19)$$

For an oscillating bearing, the speed is expressed in oscillations per minute (opm). The fatigue life of an oscillating rolling bearing is then calculated as follows.

For thrust ball bearings:

$$L_{10} = \left(\frac{C_{a,osc}}{P_{ea}} \right)^3 \quad (20)$$

where L_{10} is expressed in millions of oscillations.

For thrust roller bearings fatigue life is calculated as follows.

$$L_{10} = \left(\frac{C_{a,osc}}{P_{ea}} \right)^{10/3} \quad (21)$$

Equation 11 is used to calculate the modified rating life. Using equation 9, the life also can be expressed in hours, where N is the constant speed of oscillation (in opm).

The yaw and pitch bearings in a wind turbine application generally do not operate under constant load and speed; rather, the operating conditions vary. These variable conditions can be divided into a number of discrete sets of conditions, each associated with dynamic equivalent load (P_{eak}), speed of oscillation (N_k), oscillation amplitude (θ_k), and duration of operation (t_k). In this case, the dynamic equivalent axial load over the duty cycle can be calculated as follows.

$$P_{ea} = \left(\frac{\sum_{k=1}^{k=n} P_{eak}^p N_k t_k \theta_k^x}{\sum_{k=1}^{k=n} N_k t_k \theta_k^x} \right)^{1/p} \quad (22)$$

Where:

exponent $p = 3$ for ball bearings and $10/3$ for roller bearings;

exponent $x = 9/10$ for ball bearings and $8/9$ for roller bearings when $\theta > \theta_{crit}$; and

exponent $x = 1$ for both ball and roller bearings when $\theta \leq \theta_{crit}$.

4.5.3. ISO Standard Life

Equation 23 gives bearing fatigue life in millions of revolutions according to ISO Standard 281 [1].

$$L_{ISOm} = a_1 a_{ISO} \left(\frac{C_a}{P_{ea}} \right)^p \quad (23)$$

Where exponent p equals 3 for ball bearings and $10/3$ for roller bearings. Similar to equation 11, a_1 equals life modification factor for reliability; values of a_1 are obtained from Table 3.

However, a_{ISO} is an integrated life modification factor, a function of a fatigue limit load P_u (obtainable for each bearing from the manufacturers catalog) and the dynamic equivalent axial load P_{ea} (see [1], [5]). As shown in Harris and Kotzalas [5], a_{ISO} can be calculated using the following formula.

$$a_{ISO} = 0.1 \left[1 - \left(x_1 - \frac{x_2}{\kappa^{e_1}} \right)^{e_2} \left(\frac{\eta P_u}{P_{ea}} \right)^{e_3} \right]^{e_4} \quad (24)$$

In equation 24, κ is a measure of the adequacy of lubrication. It is related to Λ such that κ equals $\Lambda^{1.12}$. As indicated in section 4.5.2.3, in a yaw or pitch bearing application $\Lambda \ll 0.5$; and Λ equals 0.1 is a satisfactory assumption.

Therefore, κ can be assumed to be 0.076 for a yaw or pitch bearing. Table 6 provides values of constants x_1 and x_2 , and exponents $e_1 - e_4$ for thrust ball and roller bearings with marginal lubrication.

Table 6. Constants and Exponents for Thrust Bearings (Using Equation 24)

Bearing Type	x1	x2	e1	e2	e3	e4
Thrust ball	2.5671	2.2649	0.054381	0.83	1/3	-9.3
Thrust roller	1.5859	1.3993	0.054381	1.00	0.4	-9.185

In equation 24, η defines the level of particulate contamination in the grease lubricant in the bearing. As determined from ISO Standard 281 [1] and Harris and Kotzalas [5], the following equation can be used to calculate η for a grease-lubricated yaw or pitch bearing.

$$\eta = 0.173c_1\kappa^{0.68}d_m^{0.55}\left(1 - \frac{c_2}{d_m^{1/3}}\right)\eta < 1 \quad (25)$$

The constants c_1 and c_2 are defined in terms of the level of particulate contamination in the bearing grease lubricant; they are given in Table 7.

Table 7. Constants c_1 and c_2 Versus Contamination Level

Contamination Level	c_1	c_2
High cleanliness	0.0864	0.6796
Normal cleanliness	0.0432	1.141
Slight-to-typical contamination (dm < 500 mm)	0.0177	1.887
Slight-to-typical contamination (dm ≥ 500 mm)	0.0177	1.677
Severe contamination	0.0115	2.662
Very severe contamination	0.00617	4.06

ISO Standard 281 [1] applies specifically to bearings fabricated from good-quality CVD 52100 steel, through-hardened to a minimum of 58 HRC. As indicated in Section 2, yaw and pitch bearings are generally fabricated using surface-hardened steel rings. Therefore, the results of a life evaluation using ISO Standard 281 do not strictly apply; they might tend to be optimistic with regard to fatigue endurance. The surface-hardened rings tend to have a hardness less than 58 HRC, however, therefore the basic dynamic axial load rating C_a must be reduced using equation 12. Hence, for a yaw or pitch bearing, equation 23 becomes the following.

$$L_{ISOm} = a_1 a_{ISO} \left(\frac{C_a (HRC/58)^{3.6}}{P_{ea}} \right)^p \quad (26)$$

Considering the load and speed variation over a duty cycle, equation 22 can be used to determine the dynamic equivalent axial load (P_{ea}).

4.5.4. Stress-Life Method

The stress-life method for estimating bearing fatigue life makes use of the following formula.

$$L_{SLm} = a_1 a_{SL} \left(\frac{C_a}{P_{ea}} \right)^p \quad (27)$$

In equation 27, similar to equation 11, a_1 is the life modification factor for reliability (values of a_1 are obtained from Table 3). As noted in section 3 (above), a_{SL} is a factor that integrates—in the form of a subsurface and surface distribution of Von Mises stresses—all identified conditions that influence rolling contact fatigue. These conditions include the following.

- Hertz stresses
- Contact surface shear stresses due to sliding friction
- Residual stresses due to heat treatment and material processing
- Ring hoop stresses due to press fitting and rotation
- Lubricant contamination

Harris and Kotzalas [5] and Barnsby et al. [7] describe the stress-life method and calculation of the individual fatigue-influencing stresses in significant detail. To calculate a_{SL} for the bearing application, it is necessary to calculate a_{SLmj} for each rolling element–raceway contact and determine the life for each contact. For a yaw or pitch bearing, however, the dynamic equivalent axial load (P_{ea}) is assumed to apply equally to each contact. Therefore, the equivalent load on each contact ($Q_{mj} = P_{ea} / Z \sin \alpha$; a_{SLmj}) can be determined for this value of Q_{mj} and the Hertz stresses and friction due to bearing internal speeds and lubrication associated with Q_{mj} .

Furthermore, the stress-life method is applicable to bearings fabricated from component materials other than CVD 52100 steel. These materials are identified by a Von Mises fatigue limit stress ($\Phi_{VM,lim}$). Table 8, taken from Harris and Kotzalas [5], provides values of $\Phi_{VM,lim}$ for various steels that might be used to fabricate yaw and pitch bearing components. It is noted that, when carburizing or induction-hardened steels are used, a compressive residual stress also must be included in the determination of a_{SL} . A value of -170 MPa can be considered.

Table 8. Mises Fatigue Limit Stress ($\Phi_{VM,lim}$) for Various Steels

Steel	$\Phi_{VM,lim}$ (MPa)
CVD 52100, minimum 58 HRC	684
SAE 4320/8620 case-hardening, minimum 58 HRC	590
VIMVAR M50, minimum 58 HRC	717
VIMVAR M50NiL, minimum 58 HRC	579
Induction-Hardened	450

For every contact, a distribution of at least 100 values of Φ_{VM} both on and below the contact surface is included in the calculation of a_{SL} . A computer program is required to calculate a_{SL} . A program called ASMELife can be purchased from ASME International [7]. This program contains all the required lubrication and contamination data necessary to perform calculations. In the absence of specified input on component surface roughness and bearing internal dimensions, the program estimates the data.

If the bearing steel is not hardened to 58 HRC minimum, then equation 28 can be used to calculate life.

$$L_{SLm} = a_1 a_{SL} \left(\frac{C_a (HRC/58)^{3.6}}{P_{ea}} \right)^p \quad (28)$$

The considerations for calculating the life of oscillating application bearings, as explained above, also pertain to the calculations of the stress-life method.

5. Bearing Static Capacity

5.1. Permanent Deformation

A static load is a load acting on a nonrotating (non-oscillating) bearing. Permanent deformations appear in elements and raceways under a static load of moderate magnitude and increase gradually with increasing load. Experience shows that a total permanent deformation of 0.0001 of the rolling element diameter, at the center of the most heavily loaded element–raceway contact, can be tolerated in most bearing applications without impairment of the subsequent bearing operation. The basic static load rating of a rolling bearing is based on this consideration.

Tests indicate that a ball or roller load of the magnitude that yields a permanent deformation of magnitude $0.0001D$ corresponds to the calculated maximum Hertz stress shown in Table 9.

Table 9. Maximum Hertz Stress (Permanent Deformation = 0.0001D)

Bearing Type	Maximum Hertz Stress (MPa)
Ball	4,200
Roller	4,000

5.2. Maximum Ball or Roller Load

For a yaw or pitch bearing, the maximum ball or roller load (Q_{\max}) can be calculated as a function of the applied radial, thrust, and moment loading, as follows.

$$Q_{\max} = \left(\frac{2F_r}{Z \cos \alpha} + \frac{F_a}{Z \sin \alpha} + \frac{4M}{d_m Z \sin \alpha} \right) \quad (29)$$

5.3. Maximum Contact Stress

The maximum stress acting normal to the rolling element–raceway contact—Hertz stress—is the compressive stress at the center of the contact. The average Hertz contact stress is the element load divided by the contact area. For a ball bearing, the contact is an ellipse having semi-major axis a and semi-minor axis b ; the area of the ellipse is πab . The maximum Hertz stress is given by the following equation.

$$S_{\max} = \frac{1.5Q_{\max}}{\pi ab} \quad (30)$$

For a cylindrical roller-raceway, the contact approximates a rectangle of length l_e and half-width b ; the contact area is $2bl_e$. The maximum compressive contact stress is determined as follows.

$$S_{\max} = \frac{2Q_{\max}}{\pi b l_e} \quad (31)$$

5.4. Contact Dimensions

5.4.1. Point Contact

The condition in which no load has been applied and two surfaces contact each other at a single point is defined as “point contact.” When load is applied between the surfaces, flattening occurs near the contact point (shown in Figure 5); the area of the contact is elliptical in form. In Figure 6, considering a ball bearing, “Body a” represents the ball, and “Body b” represents either the inner or outer bearing raceway.

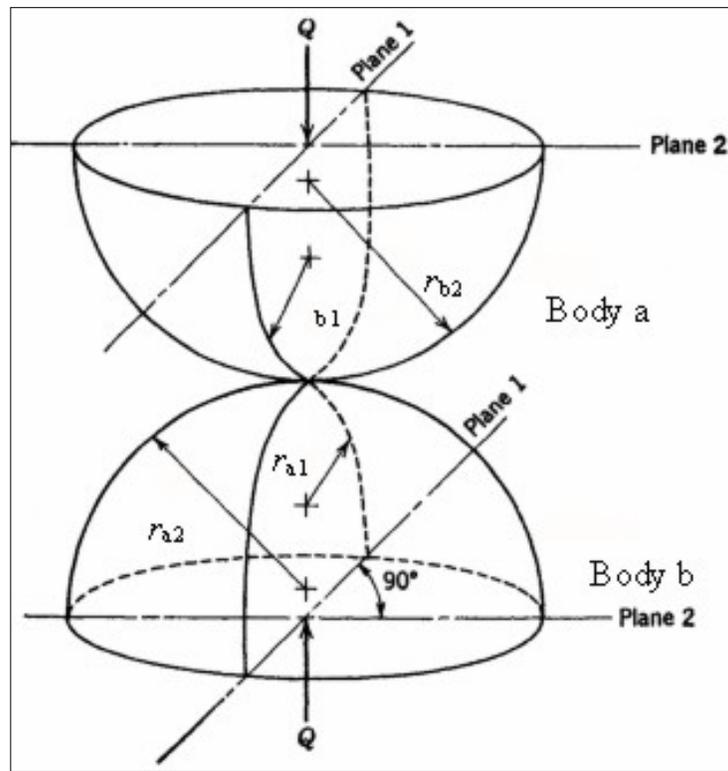


Figure 5. Schematic of two solid elastic curved bodies (Body a, Body b) compressed by load Q

Each body is represented by the principal radii of curvature in orthogonal planes 1 and 2 (i.e., r_{a1} , r_{a2} , r_{b1} , and r_{b2}). Curvature is defined as the reciprocal of radius (i.e., curvature $\rho = 1/r$). Radius always is positive, but curvature can be positive or negative. If the center of curvature lies within the solid body, curvature is positive; if it lies outside of the solid body, curvature is negative. Hence, for a ball, curvature is positive; for a raceway groove, curvature is negative.

With regard to the bodies in contact, two parameters (which are functions of contact body curvatures) are required to define the dimensions of the contact. These are $\Sigma\rho$, curvature sum:

$$\Sigma\rho = \rho_{a1} + \rho_{a2} + \rho_{b1} + \rho_{b2} \quad (32)$$

and curvature difference $F(\rho)$:

$$F(\rho) = \frac{\rho_{a1} - \rho_{a2} + \rho_{b1} - \rho_{b2}}{\sum \rho} \quad (33)$$

for a ball bearing,

$$\sum \rho_m = \frac{4}{D} - \frac{1}{f_m D} \pm \frac{2}{D} \left(\frac{\gamma}{1 \mp \gamma} \right) \quad m = \text{inner, outer} \quad (34)$$

and

$$F(\rho)_m = \frac{\frac{1}{f_m} \pm \left(\frac{2\gamma}{1 \mp \gamma} \right)}{4 - \frac{1}{f_m} \pm \left(\frac{2\gamma}{1 \mp \gamma} \right)} \quad m = \text{inner, outer.} \quad (35)$$

In equations 34 and 35, the upper signs refer to the inner raceway contact and the lower signs refer to the outer contact. For ball bearing contacts, the contact ellipse dimensions are given by the following.

$$a = 0.0236a^* \left(\frac{Q}{\sum \rho} \right)^{1/3} \quad (36)$$

$$b = 0.0236b^* \left(\frac{Q}{\sum \rho} \right)^{1/3} \quad (37)$$

In equations 36 and 37, a^* and b^* are functions of $F(\rho)$; they can be obtained by interpolation in Table 10 (from Harris and Kotzalas [5] for steel bodies). The principal radii for a four-point contact ball bearing are described in Figure 6. The equations for the radii of curvature for a four-point contact ball bearing are given in Table 11.

5.4.2. Line Contact

The condition in which no load has been applied and two surfaces just contact each other on a single line is called “line contact.” When load is applied between the surfaces flattening occurs near the contact line, and the area of the contact is rectangular in form.

The contact semi-width (b) of the rectangle of contact can be calculated directly as a function of the summation of curvature ($\sum \rho$). The equation for the contact semi-width b is given below. The curvatures for a cylindrical cross-roller bearing are provided in Table 12.

$$b = 0.00335 \left(\frac{Q}{l_e \sum \rho} \right)^{1/2} \quad (38)$$

Table 10. Values a^* and b^* Versus $F(\rho)$

$F(\rho)$	a^*	b^*
0.0	1.0	1.0
0.1075	1.0760	0.9318
0.3204	1.2623	0.8114
0.4795	1.4556	0.7278
0.5916	1.6440	0.6687
0.6716	1.8258	0.6245
0.7332	2.011	0.5881
0.7948	2.265	0.5480
0.83495	2.494	0.5186
0.87366	2.800	0.4863
0.90999	3.233	0.4499
0.93657	3.738	0.4166
0.95738	4.395	0.3830
0.97290	5.267	0.3490
0.983797	6.448	0.3150
0.990902	8.062	0.2814
0.995112	10.222	0.2497
0.997300	12.789	0.2232
0.9981847	14.839	0.2072
0.9989156	17.974	0.18822
0.9994785	23.55	0.16442
0.9998527	37.38	0.13050
1.0	∞	0.0

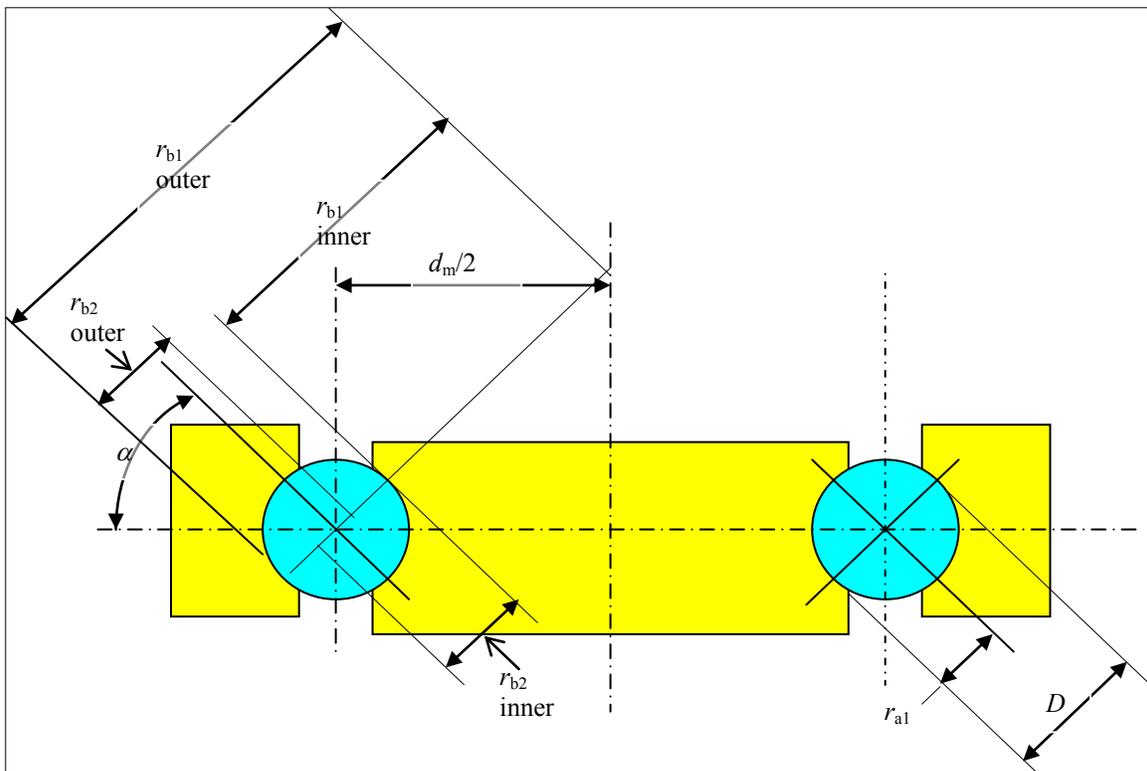


Figure 6. Radii of curvature for a four-point contact ball bearing

Table 11. Curvatures for a Four-Point Contact Ball Bearing

Curvature	Inner Raceway	Outer Raceway
ρ_{a1}	$\frac{2}{D}$	$\frac{2}{D}$
ρ_{a2}	$\frac{2}{D}$	$\frac{2}{D}$
ρ_{b1}	$\frac{2}{D\left(\frac{1-\gamma}{\gamma}\right)}$	$-\frac{2}{D\left(\frac{1+\gamma}{\gamma}\right)}$
ρ_{b2}	$-\frac{1}{f_i D}$	$-\frac{1}{f_o D}$

Table 12. Curvatures for a Cylindrical Cross Roller Bearing

Curvature	Inner Raceway	Outer Raceway
ρ_a	$\frac{2}{D}$	$\frac{2}{D}$
ρ_b	$\frac{2}{D\left(\frac{1-\gamma}{\gamma}\right)}$	$-\frac{2}{D\left(\frac{1+\gamma}{\gamma}\right)}$
$\Sigma\rho$	$\rho_a + \rho_b$	$\rho_a + \rho_b$

5.5. Static Load Factor

The static load factor (SF) is the ratio of the allowable ball or roller load to the actual ball or roller load. The static load factor should be greater than 1 to assure some margin of safety with regard to the static capacity. The SF is linear because it is based on load; SF equals 1.5 means that the ball or roller load could be increased by a factor of 1.5 before it equals the static capacity or maximum allowable non-Brinell load.

The static capacity is expressed in terms of the maximum Hertz contact stress S_{\max} (equation 30, equation 31). The SF can be calculated from the maximum Hertz contact stress for four-point contact ball bearings as follows.

$$SF = \left(\frac{4200}{S_{\max}} \right)^3 \quad (39)$$

The SF can be calculated for cylindrical cross roller bearings as follows.

$$SF = \left(\frac{4000}{S_{\max}} \right)^2 \quad (40)$$

It sometimes is necessary for the ball or roller load to be calculated when the maximum Hertz contact stress (S_{\max}) is known. This usually is the case when applying the SF to the load. For four-point contact ball bearings the following applies.

$$Q = \left[\frac{(0.0236)^2 \pi a * b * S_{\max}}{1.5} \right]^3 \frac{1}{(\sum \rho)^2} \quad (41)$$

For cylindrical cross roller bearings the following applies.

$$Q = \frac{l_e (0.00335 \cdot \pi \cdot S_{\max})^2}{4 \sum \rho} \quad (42)$$

5.6. Effect of Surface Hardness on Static Capacity

Static capacity is based on a component hardness of 58 HRC minimum. The static capacity decreases as hardness is reduced. For ball bearings, Harris and Kotzalas [5] provide this hardness reduction factor (f_s) in terms of Vickers hardness (HV), as follows.

$$f_s = 1.5 \left(\frac{HV}{800} \right)^2 \quad f_s \leq 1 \quad (43)$$

For roller bearings the following applies.

$$f_s = 2 \left(\frac{HV}{800} \right)^2 \quad f_s \leq 1. \quad (44)$$

Decreases in static capacity for reduced hardness also are shown in Table 13 as a function of the HRC surface hardness.

Table 13. Hardness Reduction Factor f_s Versus Hardness

HRC	HV	Ball Bearing	Roller Bearing
58	660	1	1
56	620	0.901	1
55	600	0.844	1
50	500	0.586	0.781

6. Estimation of the Case-Core Interface Depth

6.1. Design Stresses

Yaw and pitch bearings (as illustrated in Figure 1) are case-hardened. This heat-treatment process is intended to produce a raceway hardness of 58 HRC. Through the “case,” the hardness decreases gradually to HRC 50, and then decreases rapidly to the core hardness of the ring material. The depth below the surface to HRC 50 hardness is called the effective case depth (h_e). The application of a concentrated (Hertz contact) ball or roller load results in significant subsurface shear stresses which reach down into the core material (illustrated in Figure 7).

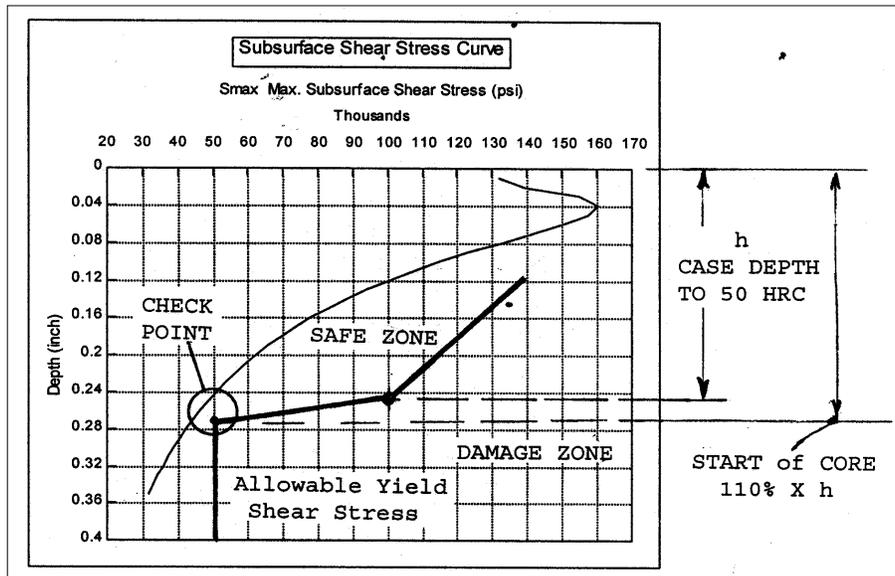


Figure 7. Typical subsurface shear stress curve

(Note that the data are in inches and psi; to convert inches to millimeters (mm), multiply by 25.4; to convert psi to MPa, divide by 145.)

The core deteriorates if the subsurface shear stress at the depth where the core hardness starts exceeds the yield stress in shear or the limit shear stress in fatigue of the core material. Such deterioration eliminates support for the hardened case material. Two types of failure then can occur. The core can be crushed by physically pressing the hardened case material into the core, or the lack of support for the core can cause the case to crack and disintegrate. The depth at which the core hardness starts can be estimated as 110% of the effective case depth (h_c) (see Figure 7).

The shear stress considered is the maximum shear stress (τ_{45}) which acts on a 45-degree plane and is one-half the difference between the maximum and minimum principal stresses acting at the point. The normal stresses acting at the center of a ball or roller contact are principal stresses. In determining the ability of the component to withstand the applied loading, Sague and Rumbarger [11] employ the maximum shear stress (τ_{45}) and the allowable yield and fatigue stresses in shear. Zwirlein et al. [12] use the triaxial combined stress and the tensile yield stress as the allowable stress. The latter is the Distortion Energy Hypothesis (DEH) approach. The maximum shear stress approach is used here.

6.2. Allowable Shear Stresses for Yield and Fatigue

The allowable yield shear and the allowable fatigue shear stresses are given in Table 14 and shown in Figure 8. These allowable stress values cover a range of core hardness from 20–40 HRC (226–370 HB). It is noted that the yield stress in shear equals 0.425 times the ultimate strength in tension, and the fatigue strength in shear equals 0.6 times the yield stress in shear.

Table 14. Allowable Subsurface Shear Stresses

Rockwell C Hardness (HRC)	Brinell Hardness (HB)	Yield Strength in Shear (MPa)	Fatigue Strength in Shear (MPa)
40	371	530.3	318.2
35	327	454.5	272.7
30	286	398.6	239.2
25	253	351.7	211.0
20	226	304.8	182.9

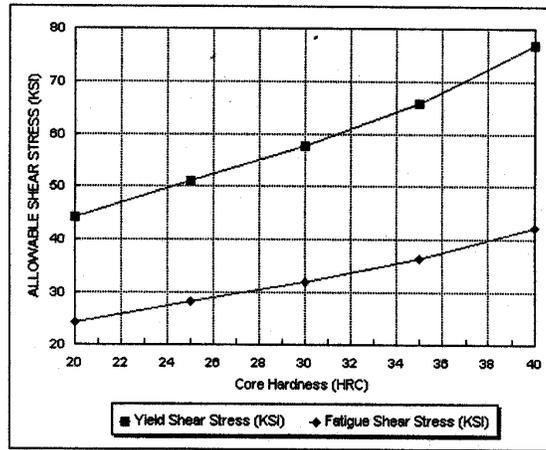


Figure 8. Allowable subsurface shear stresses
(Note that the data are in ksi; to convert ksi to MPa, divide by 0.145.)

6.3. Calculation of Subsurface Shear Stress

The subsurface shear stress can be calculated using the parameter ζ , a function of the ratio b/a of the contact surface and the dimensionless depth z/b below the surface. This relationship is illustrated in Figure 9 and further defined in Table 15. The desired value of ζ is found by interpolation for the desired depth, z/b , where z is the depth to the start of the core hardness or 110% of the case depth, as shown in Figure 7. Using either Figure 9 or Table 15 to determine ζ , the subsurface shear stress τ at depth z/b is then given by the following equation.

$$\tau = \frac{\zeta \cdot b \sum \rho}{1.8754 \cdot 10^{-5}} \quad (45)$$

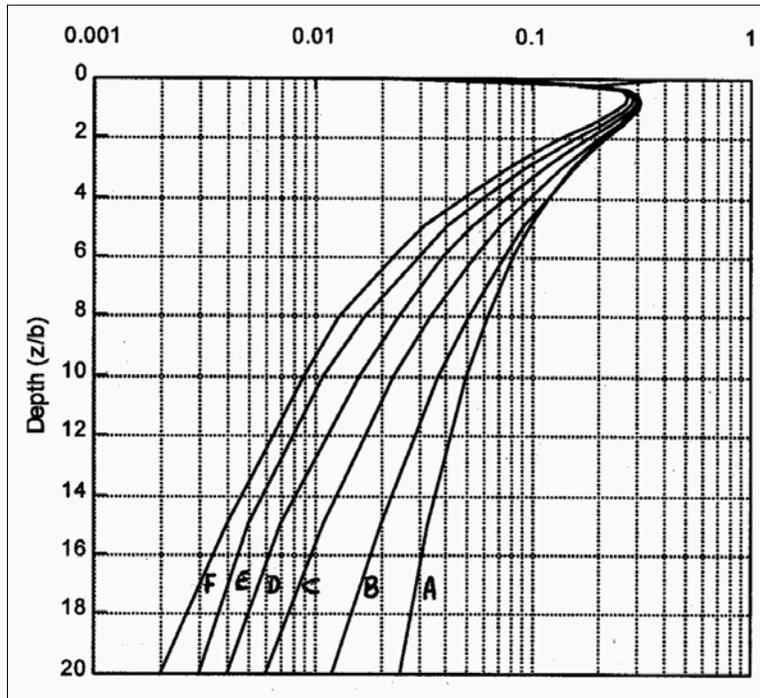


Figure 9. Subsurface shear stress parameter versus z/b and b/a
 (Curve A: $b/a = 0$; B: $b/a = 0.1$; C: $b/a = 0.2$; D: $b/a = 0.3$; E: $b/a = 0.4$; F: $b/a = 0.5$)

Table 15. Subsurface Shear Stress Parameter Versus z/b and b/a

z/b	$b/a = 0$	$b/a = 0.1$	$b/a = 0.2$
0.0	0.000	0.022	0.040
0.2	0.161	0.180	0.190
0.4	0.251	0.268	0.273
0.6	0.291	0.305	0.306
0.8	0.300	0.313	0.311
1.0	0.293	0.304	0.299
1.5	0.252	0.260	0.250
2.0	0.211	0.217	0.203
2.5	0.179	0.183	0.166
3.0	0.154	0.156	0.137
4.0	0.119	0.118	0.097
5.0	0.097	0.092	0.071
6.0	0.082	0.075	0.054
8.0	0.062	0.051	0.034
10.0	0.050	0.037	0.023
15.0	0.033	0.020	0.011
20.0	0.025	0.012	0.006

In Figure 9 and Table 15, b/a equals 0 represents the line contact.

6.4. Evaluation of the Case-Core Interface

In addition to achieving satisfactory fatigue endurance and sufficient static capacity, case-core interface design criteria must be satisfied. Failure to satisfy these criteria affects the bearing fatigue life and static capacity.

The yield shear stress associated with the greatest rolling element–raceway contact load (Q_{\max}) under the ultimate or worst-case loading condition should be less than the allowable yield shear stress at the depth where core hardness starts (i.e., 110% of the case depth). Failure to satisfy this condition requires a reduction in the static load factor (*see* Section 5). The allowable Hertz contact stress for the static capacity (*see* Section 5) must be reduced to a value that has a subsurface shear stress equal to the subsurface shear stress at the core boundary.

The subsurface shear stress associated with load Q_{\max} for the dynamic equivalent axial load should be equal to or less than the fatigue shear stress. A reduction in the L_{10} fatigue life is indicated if the subsurface shear stress exceeds the allowable fatigue shear stress at the core boundary. Considering equation 11, an additional life modification factor (a_5) is required to accommodate this condition, such that the following is true.

$$a_5 = \frac{\tau_{shear,allowable}}{\tau_{shear,actual}} \leq 1 \tag{46}$$

7. Effective Lubrication

A meaningful oil film thickness cannot be generated in a slowly and intermittently moving (oscillating) grease-lubricated yaw or pitch bearing. Therefore a clean grease with good boundary lubrication additives (especially for oscillating conditions) should be selected on the basis of experience for use in wind turbine yaw and pitch bearings. A seal system (integral or external) also is essential for achieving satisfactory operation.

A fretting-corrosion type of raceway and rolling element surface failure commonly is encountered in yaw and pitch bearings. The fretting corrosion appears as elliptical or rectangular footprints at ball or roller spacing in the bearing. The markings are tiny corrosion pits caused by the lubricant being forced out of the contact area (by a small load increase) and then not being able to re-enter the contact zone. The unprotected surface then is subject to corrosion pitting. Most grease rated for oscillation use can coat the rolling contact surfaces and maintain corrosion protection. In extreme cases, coating the raceways is an option. A TDC coating increases the bearing cost significantly but provides increased protection.

One manufacturer of wind turbine yaw and pitch bearings suggests using the Hertz contact stress limits given in Table 16 as means to limit fretting-corrosion types of failures.

Table 16. Hertz Stress Limits to Minimize Fretting Corrosion Failures in Yaw and Pitch Bearings

Operating Condition	Maximum Suggested Hertz Stress(MPa)
Ultimate	3,200
Maximum operating	2,800
Mean operating	2,400

8. Bearing Friction Torque

A practical running friction torque estimate for yaw and pitch bearings is important to the equipment designer. The torque estimate suggested is conservative and results in values that can be used for drive system or actuator design. The running torque estimate provided below is for yaw and pitch bearings of the four-point contact ball bearing and cross roller bearing configurations (*see* Figure 1). These bearings include plastic spacers between the rolling elements and integral seals. Starting friction might exceed the running friction by a factor of as much as 1.5. The following is the equation for running torque.

$$T = \mu \frac{d_m}{2} \left(\frac{4.4M}{d_m} + 2.2F_r + F_a \right) \quad (47)$$

For use with equation 47, Table 17 gives values of μ , the bearing coefficient of friction.

Table 17. Bearing Friction Coefficient μ Versus Bearing Type

Bearing Type	μ
Ball bearing with cage	0.003
Ball bearing with spacers	0.004
Cross cylindrical roller bearing	0.004

9. Miscellaneous Design Considerations

9.1. General

A number of important non-rolling items can adversely affect yaw and pitch bearing service life if they are not properly satisfied. A partial list includes:

- External bolting;
- Cages or separators; and
- Integral seals.

9.2. External Bolting

The proper operation of yaw and pitch bearings requires that the external mounting bolts (Figure 2, Figure 3) have sufficient preload and strength to maintain the rings in contact with their mounting surfaces at all times. The maximum rolling element load (Q_{\max}) is used, with proper consideration of the number of elements and the number of bolts, to determine the maximum bolt load. See References, entries 14 through 18, for a brief bibliography of technical papers covering a bolted joint.

9.3. Cages or Separators

Most yaw and pitch bearings are supplied with plastic spacers between each rolling element. A four-point contact ball bearing has cylindrical spacers with hemispherical indented ends. The cross roller bearing has plastic saddle-type spacers, which conform to the two adjacent rollers with axes of rotation 90 degrees apart. The ball spacers for larger balls (50-mm diameter and larger) often have a steel-plate reinforcement cast into the plastic.

Use of segmented cages is very rare for yaw and pitch bearings. The opening that the cage requires between the raceways significantly reduces the available load-carrying ball path. The bearing manufacturer should be consulted about spacer design. The spacer material also must be compatible with the selected lubricant.

9.4. Integral Seals

Wind turbine yaw and pitch bearings are usually supplied with integral rubbing lip seals. The bearing manufacturer should be consulted about materials, design, and placement of the integral seals. The seal material must be compatible with the selected lubricant.

Nomenclature

Symbol	Description	Units
a	semimajor axis of contact ellipse	mm
a^*	dimensionless semimajor axis of contact ellipse	mm
a_1	life modification factor for reliability	
a_2	life modification factor for materials	
a_3	life modification factor for lubrication	
a_4	life modification factor for flexible support structure	
a_{ISO}	integrated ISO life factor	
a_{SL}	integrated stress-life factor	
b	semiminor axis of contact ellipse	mm
b^*	dimensionless semiminor axis of contact ellipse	mm
c_1, c_2	constants to calculate contamination factor	
C_a	basic dynamic axial load rating	N
$C_{a,osc}$	basic dynamic axial load rating for oscillation	N
D	diameter of rolling element	mm
d_m	pitch diameter of bearing	mm
e	eccentricity	mm
e_1, e_2, e_3, e_4	exponents to calculate a_{ISO}	
f_{cm}	bearing geometry-material factor in load rating equation	
f_i	inner raceway groove radius/ D	
f_o	outer raceway groove radius/ D	
F_a	applied axial load	N
F_{ay}, F_{ax}	components of F_a along the y and x axes	N
F_r	applied radial load	N
f_s	hardness reduction factor	
$F(\rho)$	curvature difference	
i	number of rows of rolling elements	
h_e	effective case depth to 50 HRC	mm
h_0	minimum lubricant film thickness	micrometers
HB	Brinell hardness	
HRC	Rockwell C scale hardness	
HV	Vickers hardness number	
l_e	effective roller length	mm
L	bearing life for rotation	rev, hr
L_{ISOm}	ISO Standard modified rating life	rev, hr
L_{nm}	ANSI/ABMA modified rating life	rev, hr
L_{SLm}	stress-life modified rating life	rev, hr
L_{10}	bearing life for 90% survival (revolutions)	
M	applied moment load	kNm
N	bearing net rotational or oscillating speed	rpm, opm
P_{ca}	dynamic equivalent axial load	N
P_n	dynamic equivalent load-for-load speed condition n	N
P_u	fatigue limit load	N
Q_j	individual ball or roller load	N

Q_{\max}	maximum ball or roller load	N
r_{a1}, r_{a2}	principal radii of curvature for Body a	mm
r_{b1}, r_{b2}	principal radii of curvature for Body b	mm
s_m	raceway rms surface “roughness”	micrometers
s_{RE}	ball or roller rms surface “roughness”	micrometers
SF	static load factor	
S_{\max}	maximum Hertz contact stress on roller or ball with highest load	MPa
t_k	time fraction at operating condition k	
T	running friction torque	N-m
x_1, x_2	constants to calculate a_{ISO}	
X	dynamic radial load factor	
Y	dynamic axial load factor	
z	depth of maximum shear stress (length)	mm
Z	number of rolling elements in a row	
z/b	dimensionless depth parameter	
α	contact angle	°
γ	$D \cos \alpha / d_m$	
ζ	parameter for calculating subsurface shear stress	
η	lubricant contamination factor	
θ	amplitude of oscillation (one-half total arc in one cycle)	°
θ_{crit}	critical amplitude of oscillation	°
θ_{dith}	repetition of oscillation cycles at small amplitudes	°
κ	lubrication effectiveness factor	
Λ	lubrication film parameter	
μ	coefficient of friction	
ρ	curvature	mm ⁻¹
$\Phi_{VM, \text{lim}}$	Von Mises fatigue limit stress	MPa
$\Sigma \rho$	curvature sum	mm ⁻¹
τ	subsurface shear stress	MPa
τ_0	maximum orthogonal shear stress	MPa
τ_{45}	maximum shear stress	MPa
ϕ	angular location of individual ball or roller	°

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Appendix A. Wind Turbine Yaw Bearing

A.1. Operating Conditions and Bearing Dimensions

A wind turbine yaw bearing mounted in a tubular (cylindrical) tower is required to endure for 20 years, considering 50% operating time while subjected to loading according to the duty cycle in Table A-1.

Table A-1. Yaw Bearing Load Duty Cycle

Operating Condition	Time of Operation (%)	Wind Speed (m/sec)	Radial Load (kN)	Thrust Load (kN)	Moment Load (kNm)
1	32	6.250	30.60	226.0	89.27
2	16	8.944	52.96	226.0	178.4
3	15	11.61	57.18	226.0	233.7
4	20	13.42	52.51	226.0	251.3
5	15	17.89	40.94	226.0	273.2
6	2	33.36	36.05	226.0	299.9
Limit load	—		147.9	226.0	2430

The outer raceway of the bearing is stationary with respect to applied loading and rotates at 0.285 rpm. A cylindrical cross roller bearing, as illustrated in Figure A-1 and having the dimensions and material properties provided in Table A-2, has been selected for the application. The rollers are oriented at 45° contact angle, as shown in Figure A-1. The lower surface of the bearing inner ring is mounted to the tubular tower. The inner raceway is stationary. Alternate rollers carry the thrust load in a given direction. Hence, half the rollers are oriented to carry the applied thrust load.

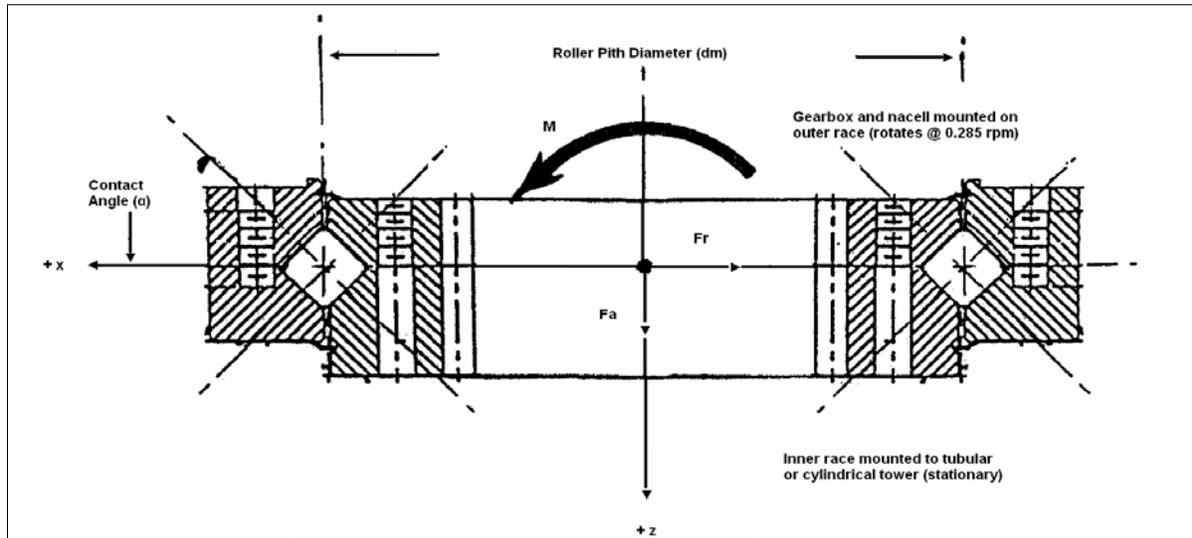


Figure A-1. Yaw bearing coordinates and loads

Table A-2. Yaw Bearing Dimensions and Material Properties

Parameter	Dimensions
Pitch diameter, d_m	1454 mm*
Roller diameter, D	44.45 mm*
Roller effective length, l_e	38.10 mm*
Number of rollers per row, Z	40
Contact angle, α	°
Raceway hardness	56 HRC minimum
Core hardness	250 HB minimum
Case depth	4.572 mm

*Bearing length dimensions were converted from inches to millimeters.

A.2. Dynamic Equivalent Loads

Dynamic equivalent loads for each operating condition of Table A-1 are calculated according to equation 6.

$$P_{ea} = 0.75F_r + F_a + \frac{2M}{d_m}$$

Also, the eccentricity of applied thrust load F_a can be calculated from the following.

$$e = \frac{2M}{d_m F_a} \quad (\text{A-1})$$

Using equation 6 and equation A-1, the dynamic equivalent axial P_{ea} for each operating condition of Table A-1 can be calculated. These loads are given in Table A-3.

Table A-3. Yaw Bearing Dynamic Equivalent Axial Loading for the Duty Cycle

Operating Condition	Time of Operation (%)	Dynamic Equivalent Axial Load, P_{ea} (N)	Eccentricity of Thrust Loading, e
1	32	371,700	0.543
2	16	511,100	1.085
3	15	590,300	1.422
4	20	611,100	1.529
5	15	632,500	1.662
6	2	665,600	1.825
Limit load	—	3,679,000	14.79

Using equation 8 and equation 10 show that the dynamic equivalent axial load over the duty cycle is as follows.

$$P_{ea} = \left(\sum_{k=1}^{k=6} P_{eak}^{10/3} t_k \right)^{3/10} \quad (\text{A-2})$$

$$P_{ea} = [(3.717)^{10/3} \cdot 0.32 + (5.111)^{10/3} \cdot 0.16 + (5.903)^{10/3} \cdot 0.15 + (6.111)^{10/3} \cdot 0.20 \\ + (6.325)^{10/3} \cdot 0.15 + (6.656)^{10/3} \cdot 0.02]^{0.3} \cdot 10^5 = 5.437 \cdot 10^5 N$$

Hence, the overall operating condition for evaluating rolling contact fatigue life is as follows.

$$P_{ea} = 543,700 N \text{ at } 0.285 \text{ rpm.}$$

A.3. Basic Dynamic Axial Load Rating

Using equation 3 yields the following.

$$C_a = f_{cm} (l_e \cos \alpha)^{7/9} Z^{3/4} D^{29/27} \tan \alpha$$

For the following bearing

$$\gamma = \frac{D \cos \alpha}{d_m} = \frac{44.45 \cos 45^\circ}{1454} = 0.02162$$

from the interpolation in Table 2 for $\gamma = 0.02162$, $f_{cm} = 130.8$. Therefore, the following results.

$$C_a = 130.8 (38.10 \cos 45^\circ)^{7/9} (40)^{3/4} (44.45)^{29/27} \tan 45^\circ = 1.587 \cdot 10^6 N$$

A.4. Basic L_{10} Life

The basic rating life for the bearing is determined using equation 8.

$$L_{10} = \left(\frac{C_a}{P_{ea}} \right)^{10.3} = \left(\frac{1.587 \cdot 10^6}{5.437 \cdot 10^5} \right)^{10.3} = 35.54 \cdot 10^6 \text{ rev}$$

The L_{10} fatigue life is converted to hours of operation using equation 9.

$$L_{10} (hr) = \frac{L_{10} (mr) \cdot 10^6}{N \cdot 60} = \frac{35.54 \cdot 10^6}{0.285 \cdot 60} = 2.078 \cdot 10^6 \text{ hr}$$

A.5. Modified L_{10} Life According to ANSI/ABMA Standard 11

The modified rating life is calculated using equation 11. For L_{10} life calculation, the life modification factor for reliability from Table 3 remains a_1 equals 1. The steel is 56 HRC, which is not as hard as the minimum requirement of 58 HRC standard, a life reduction factor a_2 equals 0.66 can be determined from Table 4.

As indicated in Section 4.5.2.3, as a result of the very slow speed of assumed rotation, the bearing will operate with boundary lubrication only, and a life modification factor for lubrication a_3 not greater than 0.1 can be considered. Owing to the flexible support structure for the bearing associated with the tubular tower mounting, a life modification factor a_4 equals 0.85 from Table 5 can be employed.

The following considers all of these effects.

$$L_{nm} = a_1 a_2 a_3 \dots a_k L_{10} = 1 \cdot 0.66 \cdot 0.1 \cdot 0.85 \cdot 2.078 \cdot 10^6 = 1.166 \cdot 10^5 \text{ hr}$$

The required fatigue life for the yaw bearing application is 50% operation over 20 years or 87,600 hours. Here, L_{10m} equals 116,600 hours, and bearing life appears satisfactory. It might be more appropriate to consider L_{1m} in lieu of L_{10m} . Table 3 shows that for 99% reliability, a_1 is equal to 0.25. This yields: $L_{1m} = 28,000$ hr. This life appears marginally satisfactory.

A.6. Modified L_{10} Life According to ISO Standard 281

ISO Standard L_{ISOm} is calculated using equation 23. In equation 23, the integrated life modification factor a_{ISO} is determined using equation 24 and equation 25. Moreover, for the slow rotation, grease-lubricated bearing, as stated in Section 4.5.3, a lubrication effectiveness factor κ equals 0.076 can be assumed. Additionally, from Table 6 the contamination constants, c_1 equaling 0.0432 and c_2 equaling 1.141 for a grease lubricant of normal cleanliness can be assumed. Hence, from equation 25 the following can be determined.

$$\eta = 0.173 c_1 \kappa^{0.68} d_m^{0.55} \left(1 - \frac{c_2}{d_m^{1/3}} \right) = 0.173 \cdot 0.0432 \cdot (0.076)^{0.68} (1454)^{0.55} \left(1 - \frac{1.141}{(1454)^{1/3}} \right) = 0.06394$$

From equation 24, using Table 6 to determine constants and exponents for a thrust roller bearing.

$$a_{ISO} = 0.1 \left[1 - \left(x_1 - \frac{x_2}{\kappa^{e_1}} \right)^{e_2} \left(\frac{\eta P_u}{P_{ea}} \right)^{e_3} \right]^{e_4}$$

To calculate a_{ISO} , it also is necessary to calculate the fatigue limit load P_u . As indicated in Section 4.5.3, P_u is found in bearing manufacturers' catalogs for most bearings. In this example, however, P_u must be calculated. ISO Standard 281 [1] gives the following formula to approximate P_u for thrust roller bearings.

$$P_u = \frac{C_{0a}}{8.2} \left(\frac{100}{D} \right)^{0.3} \quad (\text{A-3})$$

Where C_{0a} is the basic static axial capacity of the bearing. In turn, C_{0a} can be calculated according the method of ISO Standard 76 [2] as follows.

$$C_{0a} = 220(1 - \gamma) Z l_e D \sin \alpha \quad (\text{A-4})$$

Which produces the following.

$$C_{0a} = 220(1 - 0.02162) \cdot 40 \cdot 38.10 \cdot 44.45 \sin 45^\circ = 1.031 \cdot 10^7 N$$

$$P_u = \frac{1.031 \cdot 10^7}{8.2} \left(\frac{100}{44.45} \right)^{0.3} = 1.604 \cdot 10^6 N$$

$$a_{ISO} = 0.1 \left[1 - \left(1.5859 - \frac{1.3993}{(0.076)^{0.054381}} \right)^1 \left(\frac{0.06394 \cdot 1.604 \cdot 10^6}{543700} \right)^{0.4} \right]^{-9.185} \approx 0.1$$

Equation 23 is $L_{ISOm} = a_1 a_{ISO} \left(\frac{C_a}{P_{ea}} \right)^p$.

The raceway hardness is 56 HRC, therefore C_a must be reduced to account for this material deficiency. This reduction can be accomplished using equation 12.

$$C'_a = C_a \left(\frac{HRC}{58} \right)^{3.6} = 1.587 \cdot 10^6 \left(\frac{56}{58} \right)^{3.6} = 1.399 \cdot 10^6 N$$

$$L_{ISOm} = 1 \cdot 0.1 \cdot \left(\frac{1.399 \cdot 10^6}{543700} \right)^{10/3} = 2.335 \cdot 10^6 rev$$

Alternatively, the following can be used.

$$L_{ISO10} = \frac{2.335 \cdot 10^6}{0.285 \cdot 60} = 1.365 \cdot 10^5 hr$$

This life does not include consideration of the flexible support of the bearing on the tubular tower, but L_{ISO10} is comparable to L_{10m} equaling 116,600 hours, determined using the ANSI/ABMA Standard 11 method. The correct procedure for determining the effect of the flexible support is accomplished using finite element methods with a bearing computer program such as GENROL to calculate the roller-raceway load at each roller location. Thereafter, the dynamic equivalent axial load can be calculated with greater accuracy. Conversely, such increased accuracy might not be warranted, considering the boundary lubrication situation and the overall a_{ISO} having the minimum value of 0.1.

A.7. Modified L_{10} Life According to Stress-Life Method

To estimate L_{SL10} , the ASMELife computer program of Barnsby et al. [7] was used with the dynamic equivalent axial loads of the duty cycle as shown in Table A-3. In addition to the bearing dimensional and material property data in Table A-2, the following input was used with ASMELife.

- Grease with ISO VG460 mineral oil for a slow-rotation bearing
- Normal cleanliness of the grease with regard to particulate contamination
- Operating temperature assumed to be 50°C
- Bearing rings manufactured of SAE 8620 steel having compressive residual stress in the case of -170 MPa
- Bearing basic dynamic axial load rating reduced to 1.587×10^6 N to account for 56 HRC hardness of raceways
- Surface roughness of raceways and rollers of 0.4 microns

Using the inputs noted above, ASMELife estimated overall: $L_{SL10} = 130,700$ hr. A minimum lubricant effective parameter of κ equaling 0.055 also was calculated, however, and the following cautionary statement was output.

Lambda < 0.5 indicates LUBRICATION EFFECTIVENESS IS EXTREMELY MARGINAL even if the lubrication supply to the roller-raceway contacts appears sufficient. The asperities of the roller and raceway surfaces are in substantial contact; fatigue endurance can be substantially less than predicted.

Thus, all of the modified life prediction methods indicate satisfactory L_{10} life in the same approximate range; however, higher reliability life (e.g., L_1) might be marginal.

A.8. Maximum Roller Load

The yaw bearing static capacity is based on the maximum Hertz contact stress at the center of the most heavily loaded roller, Q_{max} . According to ISO Standard 76 [2], the maximum Hertz contact stress should not exceed 4,000 MPa for a roller bearing. Also, according to Table 13, there is no reduction in the basic static capacity for the roller-raceway owing to the 56 HRC minimum raceway hardness. From Table 13, f_s equals 1.0.

Maximum roller load, Q_{max} , is calculated using equation 29 for the dynamic equivalent load and the limit load situations.

$$Q_{max} = \left(\frac{2F_r}{Z \cos \alpha} + \frac{F_a}{Z \sin \alpha} + \frac{4M}{d_m Z \sin \alpha} \right)$$

Table A-4. Q_{max} for Equivalent and Limit Loads

Load	F_r (kN)	F_a (kN)	M (kNm)	Q_{max} (N)
Dynamic equivalent	46.93	226.0	217.0	32,414
Limit	147.90	226.0	243.0	254,800

A.9. Contact Semi-Width, Hertz Stress, and SF

The semi-width (b) of the contact area rectangle is determined using equation 38. In turn, equation 38 requires the calculation of curvature sum $\Sigma\rho$. From Table 12, for the inner raceway contact the following is the case.

$$\sum \rho = \frac{2}{D} - \frac{2}{D \left(\frac{1-\gamma}{\gamma} \right)} = \frac{2}{D(1-\gamma)} = \frac{2}{44.45(1-0.02162)} = 0.04599 \text{ mm}^{-1}$$

$$b_e = 0.00335 \left(\frac{Q}{l_e \sum \rho} \right)^{1/2} = 0.00335 \left(\frac{32414}{38.1 \cdot 0.04599} \right)^{1/2} = 0.4556 \text{ mm}$$

$$b_{\text{lim}} = 0.00335 \left(\frac{254800}{38.1 \cdot 0.04599} \right)^{1/2} = 1.277 \text{ mm}$$

Hertz contact stress is calculated using equation 31, and static load factor SF is calculated using equation 40; for example:

$$S_{\text{max}} = \frac{2Q_{\text{max}}}{\pi b l_e} \text{ and } SF = \left(\frac{4000}{\sigma_{\text{max}}} \right)^2$$

The results are given in Table A-5.

Table A-5. Q_{max} for Equivalent and Limit Loads

Load	b (mm)	S_{max} (MPa)	SF (NAmm)
Dynamic equivalent	0.4556	1189	11.32
Limit	1.277	3334	1.439

Static load factor equal to 1.439 shows that the yaw bearing has a satisfactory static capacity under the limit load conditions.

A.10. Case-Core Interface

From Table A-2 the case depth is determined to be 4.572 mm. As stated in Section 5.7.3, the core zone (i.e., HRC = 50) commences at 110% of the case depth or 5.029 mm. From Table A-5 for the limit load application, contact area semi-width equals 1.277 mm, such that dimensionless depth z/b equals 5.029/1.277 which equals 3.938. For the rectangular contact area b/a equals 0.

Therefore, from Table 15 by interpolation at b/a equaling 0, subsurface shear stress parameter ζ equals 0.121. The shear stress at this depth is then determined using equation 45.

$$\tau = \frac{\zeta \cdot b \sum \rho}{1.8754 \cdot 10^{-5}} = \frac{0.121 \cdot 1.277 \cdot 0.04599}{1.8754 \cdot 10^{-5}} = 378.9 \text{ MPa}$$

For the bearing, a minimum core hardness of 250 HB has been specified. From Table 14 at this condition, the yield strength in shear is 346.5 MPa, and the fatigue strength in shear is 207.9 MPa. Note that for the core, τ equals 378.9 MPa; this is greater than the 346.5 MPa allowable. Therefore core crushing or cracking of the core material under the limit load is

possible. Either higher core hardness or a greater case depth needs to be specified to achieve an allowable shear yield stress that is greater than 378.9 MPa.

For the applied dynamic equivalent axial load, calculated subsurface shear stress τ is determined using equation 45, however, from Table A-5, b equals 0.4556 mm and ζ equals 0.049. In this case, $\tau = 54.8$ MPa. Because 54.8 MPa is less than the fatigue strength in shear of 207.9 MPa, the case-core interface is satisfactory from a fatigue standpoint.

A.11. Evaluation for Potential Fretting Corrosion

According to Table 16, the calculated maximum operating Hertz stress does not exceed the suggested limit to avoid a fretting-corrosion type of failure. Maximum Hertz stress under the limit load condition is marginally greater than the recommended limit for fretting corrosion avoidance. These results are summarized in Table A-6.

Table A-6. Comparison of Calculated Hertz Stresses with Recommended Limit Stresses to Avoid Fretting in Bearing Raceways

Load	Hertz Stress (MPa)	Recommended Hertz Stress Limit to Avoid Fretting (MPa)
Dynamic equivalent	1189	2800 (maximum operating)
Limit	3334	3200 (ultimate)

A.12. Lubricant and Lubrication

A lubricant selection has not been included with this pitch bearing rating example. The grease selected should be recommended for oscillating motion conditions, and also should have excellent antiwear and extreme pressure additive characteristics. Frequent (daily) movement of the yaw bearing through a large angle (90 degrees) is recommended to redistribute grease to the rollers and roller path surfaces.

A.13. Bearing Friction Torque

The estimated rolling friction torque is calculated using equation 47 using a value of friction coefficient μ equaling 0.004. Therefore the following is the case.

$$T = 0.002d_m \left(\frac{4.4M}{d_m} + 2.2F_r + F_a \right)$$

Table A-7 provides calculated values of bearing friction torque for the average bearing load condition and the limit load condition.

Table A-7. Bearing Friction Torque for Average and Limit Loads

Load	F_r (kN)	F_a (kN)	M (kNm)	T (kNm)
Average	44.21	226.0	1.894	2.607
Limit	147.9	226.0	2430	23.26

Starting friction torque or a change in friction torque with a reversal of yaw bearing motion could be 1.5 times the running torque. Estimated running torque values are given in Table A-7.

The estimated running friction torque is conservatively high for sizing drives and actuators. The actual running friction torque might be less than the estimated maxima. The torque estimates include integral seal drags and spacer drag.

A.14. Design Summary

The yaw bearing design in Figure A-1 and Table A-2, operating under the loads and constant speed duty in Table A-1, has been evaluated in accordance with NREL Guideline DG03. The yaw bearing does not satisfy the criterion to avoid core crushing. Either the core hardness or the case depth must be increased. A minimum case depth of 5.08 mm with a 250 HB minimum core hardness (for example) would satisfy the case-core design criteria.

The yaw bearing design meets or exceeds all of the following design criteria.

- L_{10} fatigue life in rotation
- Static capacity
- Satisfactory lubrication
- Non-rolling items

Based on increased reliability (e.g., L_1), fatigue endurance could be marginal.

Appendix B. Wind Turbine Pitch Bearing

B.1. Operating Conditions and Bearing Dimensions

A wind turbine pitch bearing is required to function for 20 years, considering 50% operating time and loading according to the duty cycle of Table B-1.

Table B-1. Bearing Duty Cycle Speeds and Loads

Operating Condition	Wind Speed (m/sec)	Operating Time (%)	Speed (opm*) N_k	Oscillation Amplitude (°)	Radial Load (kN)	Axial Load (kN)	Moment Load (kNm)
1	31	37.75	13.5	0.552	13.4	32.5	217.1
2	45	15.96	19.9	0.792	25.0	64.7	366.1
3	58	13.51	26.2	1.264	26.1	84.0	372.2
4	67	17.67	27.7	1.308	23.9	88.4	325.3
5	89	13.36	28.4	1.122	19.6	94.0	228.6
6	112	1.75	28.5	1.076	17.8	88.4	187.7
Limit load					46.9	145.0	937.1

*Oscillations per minute

The loads were derived by the wind turbine designer from detailed load studies of rainbow cycles versus load. (NREL DG01 covers the determination of wind turbine loads.) The loads are applied to the bearing inner raceway, which oscillates relative to the outer raceway. The pitch bearing coordinate system and load application are shown in Figure B-1.

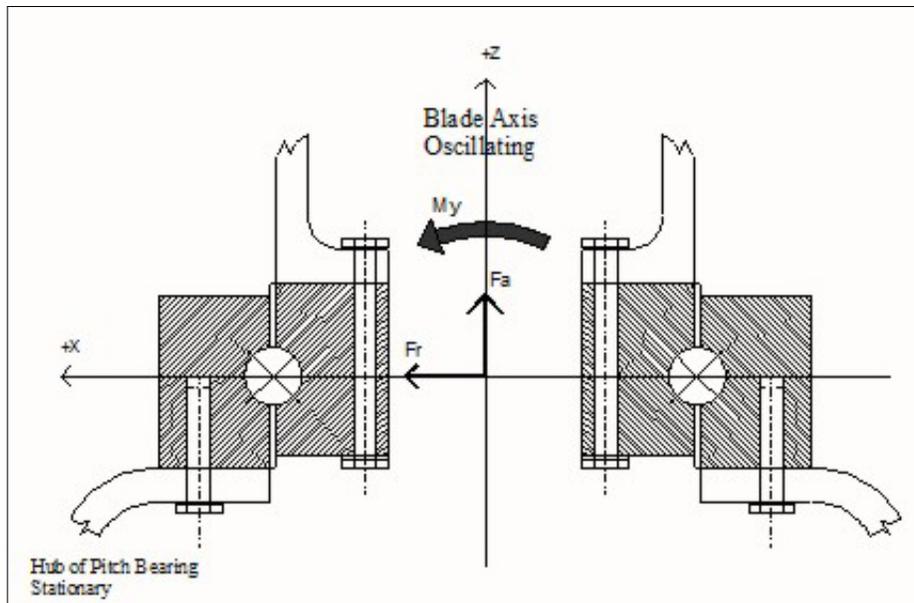


Figure B-1. Pitch bearing coordinates and loads

For this application a four-point contact ball bearing was selected, as illustrated in Figure B-2, which has the dimensions and properties given in Table B-2.

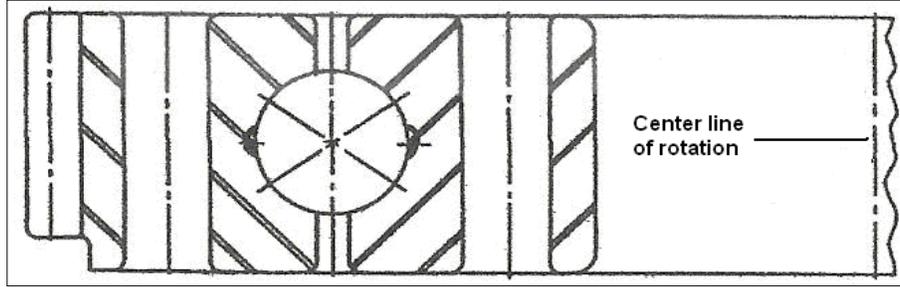


Figure B-2. Four-point contact ball bearing

Table B-2. Four-Point Ball Bearing Dimensions and Material Properties

Parameter	Dimension/Value
Pitch diameter, d_m	940 mm
Ball diameter, D	45 mm
Contact angle, α	45°
Number of balls per row, Z	52
Number of rows, i	1
Inner raceway groove radius/ D , f_i	0.52
Outer raceway groove radius/ D , f_o	0.52
Outer raceway critical amplitude of oscillation, $\theta_{crit,o}$	14.33°
Inner raceway critical amplitude of oscillation, $\theta_{crit,i}$	13.39°
Raceway surface hardness	58 HRC minimum
Core hardness	250 HB

B.2. Bearing Dynamic Equivalent Loads and Speeds

For each operating condition k ($k = 1 \dots 6$ and limit load) of Table B-1, a dynamic equivalent axial load can be calculated using equation 6. Table B-3 shows the calculated values P_{eak} .

$$P_{eak} = 0.75F_{rk} + F_{ak} + \frac{2M_k}{d_m}$$

Table B-3. Bearing Dynamic Equivalent Axial Loads over the Duty Cycle

Operating Condition	Operating Time Fraction (t_k)	Oscillation Amplitude θ_k (°)	Speed (osm) N_k	P_{eak} (kN)
1	0.3775	0.552	13.5	504.5
2	0.1596	0.792	19.9	862.4
3	0.1351	1.264	26.2	895.5
4	0.1767	1.308	27.7	798.5
5	0.1336	1.122	28.4	595.1
6	0.0175	1.076	28.5	501.1
Limit load				2174

For each operating condition of the duty cycle of Table B-1, $\theta_k < \theta_{crit}$. Therefore, the duty cycle loading specified in Table B-1 can be reduced to one dynamic equivalent axial load using equation 22 for exponent x equaling 1, as follows.

$$P_{ea} = \left(\frac{\sum_{k=1}^{k=n} P_{eak}^p N_k t_k \theta_k}{\sum_{k=1}^{k=n} N_k t_k \theta_k} \right)^{1/p}$$

Substituting the values of Table B-3 into this equation yields the following.

$$P_{ea} = \left[\frac{(504.5)^3 \cdot 13.5 \cdot 0.3775 \cdot 0.552 + \dots + (506.7)^3 \cdot 28.5 \cdot 0.0175 \cdot 1.076}{13.5 \cdot 0.3775 \cdot 0.552 + \dots + 28.5 \cdot 0.0175 \cdot 1.076} \right]^{1/3} = 765.3 \text{ kN}$$

The average speed of oscillation during the duty cycle is given by the following.

$$N_{ave} = \sum_{k=1}^{k=6} N_k t_k = 13.5 \cdot 0.3775 + 19.9 \cdot 0.1596 + 26.2 \cdot 0.1351 + 27.7 \cdot 0.1767 + 28.4 \cdot 0.1336 + 28.5 \cdot 0.0175$$

$$N_{ave} = 21.0 \text{ opm}$$

The equivalent oscillation amplitude can be determined as follows.

$$\theta_e = \frac{\sum_{k=1}^{k=n} N_k t_k \theta_k}{\sum_{k=1}^{k=n} N_k t_k} = \frac{13.5 \cdot .3775 \cdot 9.552 + \dots + 28.5 \cdot 0.0175 \cdot 1.076}{13.5 \cdot .3775 + \dots + 28.5 \cdot .0175} = 1.000^\circ$$

B.3. Bearing Basic Dynamic Axial Load Rating for Rotation

For a rotating ball bearing having balls greater than 25.4 mm (1 in) in diameter, as shown in Section 4.3, the basic dynamic axial load rating C_a is given by equation 2.

$$C_a = 3.647 f_{cm} (i \cos \alpha)^{0.7} Z^{2/3} D^{1.4} \tan \alpha$$

The geometry-material constant f_{cm} is determined from Table 1 as a function of contact angle α and parameter γ equaling $D \cos \alpha / d_m$.

$$\gamma = \frac{D \cos \alpha}{d_m} = \frac{45 \cos 45^\circ}{940} = 0.03385$$

By interpolating in Table 1, f_{cm} equals 66.74. Therefore the following is the case.

$$C_a = 3.647 \cdot 66.74 (1 \cdot \cos 45^\circ)^{0.7} (52)^{2/3} (45)^{1.4} \tan 45^\circ = 548860 \text{ N} = 548.86 \text{ kN}$$

Table 1 refers to bearings having 0.53 conformity (raceway groove radius/ D), and the bearing raceways have 0.52 conformity, as noted in Section 4.3, therefore C_a as determined above must be multiplied by 1.172, producing: $C_a = 643.3 \text{ kN}$.

B.4. Basic Dynamic Axial Load Rating for the Oscillating Bearing

The critical amplitude of oscillation for the four-point contact ball bearing inner race is 13.39° . Each of the amplitudes of oscillation in Table B1 is less than the critical amplitude of oscillation. For this condition, the basic dynamic axial load rating calculated for the rotating bearing must be modified according to equation 18 as follows.

$$C_{a,osc} = C_a \left(\frac{180^\circ}{\theta} \right)^{3/10} Z^{0.033} = 643.3 \left(\frac{180^\circ}{1.000} \right)^{0.3} (52)^{0.033} = 3480kN$$

B.5. Basic L_{10} Life

The basic rating life for the bearing is determined using equation 7, modified to employ $C_{a,osc}$.

$$L_{10} = \left(\frac{C_{a,osc}}{P_{ea}} \right)^3 = \left(\frac{3480}{765.3} \right)^3 = 94.02 \cdot 10^6 \text{ osc}$$

The L_{10} fatigue life is converted to hours of operation using equation 9.

$$L_{10}(hr) = \frac{L_{10}(mo) \cdot 10^6}{N_{ave} \cdot 60} = \frac{94.02 \cdot 10^6}{21 \cdot 60} = 74620hr$$

B.6. Modified L_{10} Life According to ANSI/ABMA Standard 11

The modified rating life is calculated using equation 11. For L_{10} life calculation, the life modification factor for reliability from Table 3 remains a_1 equaling 1. The steel is 58 HRC minimum according to the standard requirement, therefore life modification factor for material a_2 equaling 1 can be used.

As indicated in Section 4.5.2.3, considering oscillatory motion that is not conducive to the formation of lubricant films that can separate balls from raceways, the bearing will operate with boundary lubrication only, and a life modification factor for lubrication (a_3) that is not greater than 0.1 can be considered.

Owing to the flexible support structure for the bearing associated with the mounting, from Table 5 a life modification factor a_4 equal to 0.85 can be employed. Considering all of the following effects.

$$L_{nm} = a_1 a_2 a_3 \dots a_k L_{10} = 1 \cdot 1 \cdot 0.1 \cdot 0.85 \cdot 74620 = 6344hr$$

Required fatigue life for the pitch bearing application is 50% operation over 20 years or 87,600 hours. Because L_{10m} equals 6,344 hours, bearing life does not appear satisfactory. Therefore a bearing with increased basic dynamic axial load rating could be required. To satisfy the latter requirement, a double-row bearing might be considered. This would be an eight-point contact ball bearing (illustrated in Figure B-3), having the dimensions given in Table B-4.

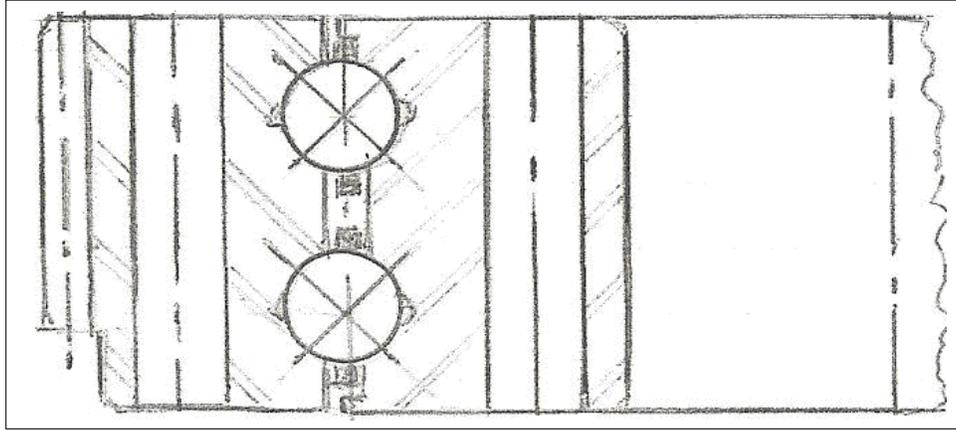


Figure B-3. Eight-point contact ball bearing

Table B-4. Eight-Point Ball Bearing Dimensions and Material Properties

Parameter	Dimension/Value
Pitch diameter, d_m	940 mm
Ball diameter, D	35 mm
Contact angle, α	45°
Number of balls per row, Z	65
Number of rows, i	2
Inner raceway groove radius/ D , f_i	0.52
Outer raceway groove radius/ D , f_o	0.52
Outer raceway critical amplitude of oscillation, $\theta_{crit,o}$	11.38°
Inner raceway critical amplitude of oscillation, $\theta_{crit,i}$	10.79°
Raceway surface hardness	58 HRC minimum
Core hardness	250 HB

For each operating condition of the duty cycle of Table B-1, as for the four-point contact ball bearing, $\theta_k < \theta_{crit}$. Therefore the duty cycle dynamic equivalent load remains the same (i.e., 765.3 kN). The basic dynamic axial load rating, however, is increased as a result of the greater number of bearing rows. The internal geometry also is altered as a result of the reduced ball diameter. Therefore the following is the case.

$$\gamma = \frac{D \cos \alpha}{d_m} = \frac{35 \cos 45^\circ}{940} = 0.02633$$

By interpolating in Table 1, f_{cm} equals 61.91. Therefore the following is the case.

$$C_a = 3.647 \cdot 61.91 \cdot (2 \cdot \cos 45^\circ)^{0.7} (65)^{2/3} (35)^{1.4} \tan 45^\circ = 675100 N = 675.1 kN$$

Table 1 refers to bearings having 0.53 conformity (raceway groove radius/ D), and the bearing raceways have 0.52 conformity, as stated in Section 4.3, therefore C_a as determined above must be multiplied by 1.172, giving: $C_a = 791.2$ kN.

The critical amplitude of oscillation for the eight-point contact ball bearing inner raceway is 10.79E. Each of the amplitudes of oscillation in Table B-1 is less than the critical amplitude of

oscillation. For this condition the basic dynamic axial load rating calculated for the rotating bearing must be modified according to equation 18.

$$C_{a,osc} = C_a \left(\frac{180^\circ}{\theta} \right)^{3/10} Z^{0.033} = 791.2 \left(\frac{180^\circ}{1.000} \right)^{0.3} (65)^{0.033} = 4312 kN$$

The basic rating life for the bearing is determined using equation 7, modified to employ $C_{a,osc}$.

$$L_{10} = \left(\frac{C_{a,osc}}{P_{ea}} \right)^3 = \left(\frac{4312}{765.3} \right)^3 = 178.9 \cdot 10^6 \text{ osc}$$

The L_{10} fatigue life is converted to hours of operation using equation 9.

$$L_{10}(hr) = \frac{L_{10}(mo) \cdot 10^6}{N_{ave} \cdot 60} = \frac{178.9 \cdot 10^6}{21 \cdot 60} = 141,980 hr$$

$$L_{nm} = a_1 a_2 a_3 \dots a_k L_{10} = 1 \cdot 1 \cdot 0.1 \cdot 0.85 \cdot 141,980 = 12,070 hr$$

As stated above, required fatigue life for the pitch bearing application is 50% operation over 20 years or 87,600 hours. Because L_{10m} equals 12,070 hours, the life for this bearing is not satisfactory. Therefore, a bearing with higher load capacity is required. Table 3 shows that, for 99% reliability, a_1 equals 0.25. This yields: $L_{1m} = 24,270$ hr. This life seems marginal.

B.7. Modified L10 Life According to ISO Standard 281

ISO Standard L_{ISOm} is calculated using equation 23. In equation 23, the integrated life modification factor a_{ISO} is determined using equation 24 and equation 25. Moreover, for the relatively slow oscillation, grease-lubricated bearing, as stated in Section 4.5.3, a lubrication effectiveness factor κ equaling 0.076 can be assumed. Additionally, from Table 7 the contamination constants c_1 equaling 0.0432 and c_2 equaling 1.141 for a grease lubricant of normal cleanliness can be assumed. Hence, from equation 25 the following is determined.

$$\eta = 0.173 c_1 \kappa^{0.68} d_m^{0.55} \left(1 - \frac{c_2}{d_m^{1/3}} \right) = 0.173 \cdot 0.0432 \cdot (0.076)^{0.68} (940)^{0.55} \left(1 - \frac{1.141}{(940)^{1/3}} \right) = 0.04942$$

From equation 24, using Table 6 to determine constants and exponents for a thrust roller bearing produces the following.

$$a_{ISO} = 0.1 \left[1 - \left(x_1 - \frac{x_2}{\kappa^{e_1}} \right)^{e_2} \left(\frac{\eta P_u}{P_{ea}} \right)^{e_3} \right]^{e_4}$$

To calculate a_{ISO} , it also is necessary to calculate the fatigue limit load P_u . As indicated in Section 4.5.3, P_u is found in bearing manufacturers' catalogs for most bearings. In this example,

however, P_u must be calculated. ISO Standard 281 [1] gives the following formula to approximate P_u for thrust ball bearings.

$$P_u = \frac{C_{0a}}{22} \left(\frac{100}{D} \right)^{0.5} \quad (\text{B-1})$$

Where C_{0a} is the basic static axial capacity of the bearing. In turn, C_{0a} can be calculated according the method of ISO Standard 76 [2] as follows.

$$C_{0a} = f_s Z D^2 \sin \alpha \quad (\text{B-2})$$

Where the f_s is a function of γ , determined from ISO Standard 76 [2] to be 59.39 for a thrust ball bearing. In equation B-2, Z is the number of balls carrying load in one direction; for the eight-point ball bearing, Z equals 130. Therefore the following is the case.

$$C_{0a} = \varphi_s Z D^2 \sin \alpha = 59.39 \cdot 130 \cdot (35)^2 \sin 45^\circ = 6.688 \cdot 10^6 \text{ N}$$

$$P_u = \frac{6.688 \cdot 10^6}{22} \left(\frac{100}{35} \right)^{0.5} = 5.138 \cdot 10^5 \text{ N}$$

$$a_{ISO} = 0.1 \left[1 - \left(2.5671 - \frac{2.2649}{(0.076)^{0.054381}} \right)^{0.83} \left(\frac{0.04942 \cdot 5.138 \cdot 10^5}{765300} \right)^{1/3} \right]^{-9.3} \approx 0.1$$

Because a_{ISO} equals 0.1, L_{ISOm} essentially is the same as L_{10m} , according to ANSI/ABMA Standard 9 [3] (i.e., 12,070 hours). A life reduction for load distribution among the bearing balls can be considered due to the flexibility of the bearing-mounting structure.

B.8. Modified L_{10} Life According to Stress-Life Method

Due to the inability of the oscillating bearing to form lubricant films of sufficient thickness to separate the balls from the raceways during bearing operation, and the resulting surface asperity-asperity contact experienced during boundary lubrication, a_{SL} also will not be greater than 0.1. Therefore, L_{SLm} is essentially the same as L_{10m} according to ANSI/ABMA Standard 9 [3], (12,070 hours).

B.9. Maximum Ball Load

The pitch bearing static capacity is based on the maximum Hertz contact stress at the center of the most heavily loaded ball, Q_{max} . According to Table 9, maximum Hertz stress should not exceed 4,200 MPa. Q_{max} is calculated using equation 29. The maximum ball load for the two-row, eight-point contact ball bearing is 55%—and not 50% of the maximum ball load for a single-row ball thrust bearing. A 55%/45% thrust load sharing of the two rows is considered the best possible load distribution ratio because of tolerances and variation of internal dimensions between the bearing rows. (The insufficient L_{10} life of the four-point contact ball bearing means that there is no need to consider that bearing further in this analysis.)

Maximum roller load (Q_{\max}) is calculated using a modified equation 29 for the dynamic equivalent load and the limit load situations, as follows.

$$Q_{\max} = 0.55 \left(\frac{2F_r}{Z \cos \alpha} + \frac{F_a}{Z \sin \alpha} + \frac{4M}{d_m Z \sin \alpha} \right)$$

Table B-5 provides the peak dynamic, dynamic equivalent and limit loads, and the calculated value of Q_{\max} for each load condition.

Table B-5. Q_{\max} for Equivalent and Limit Loads

Load	F_r (kN)	F_a (kN)	M (kNm)	Q_{\max} (kN)
Maximum dynamic load (3)	26.1	84.0	372.2	20.58
Dynamic equivalent	22.7	82.1	314.6	17.55
Limit	46.9	145.0	937.1	50.58

Note that, for the dynamic equivalent load, each component was calculated individually (e.g., F_{re} , F_{ae} , M_e). This is not completely accurate, however the result is more conservative than using a simple mean load component.

B.10. Contact Ellipse Dimensions

The maximum ball load for the limit load case and the maximum ball load under fatigue loading can be used to determine the contact ellipse dimensions and maximum Hertz stress for the eight-point contact ball bearing. For the ball bearing, contact stresses are greater at the inner raceway than the outer raceway, therefore only inner raceway stresses are estimated here. To determine the contact ellipse dimensions, $\Sigma\rho$ and $F(\rho)$ first are determined. Equation 34 produces the following.

$$\Sigma\rho_i = \frac{4}{D} - \frac{1}{f_i D} + \frac{2}{D} \left(\frac{\gamma}{1-\gamma} \right)$$

$$\Sigma\rho_i = \frac{4}{35} - \frac{1}{0.52 \cdot 35} + \frac{2}{35} \left(\frac{0.02633}{1-0.02633} \right) = 0.06089 \text{ mm}^{-1}$$

From equation 35 the following is determined.

$$F(\rho)_i = \frac{\frac{1}{f_i} + \left(\frac{2\gamma}{1-\gamma} \right)}{4 - \frac{1}{f_i} + \left(\frac{2\gamma}{1-\gamma} \right)} = \frac{\frac{1}{0.52} + \left(\frac{2 \cdot 0.02633}{1-0.02633} \right)}{4 - \frac{1}{0.52} + \left(\frac{2 \cdot 0.02633}{1-0.02633} \right)} = 0.9278$$

The dimensionless quantities a^* and b^* are functions $F(\rho)$. The values for a^* and b^* are calculated by interpolation from the dimensionless parameters of Table 10. Therefore a^*_i equals 3.571 and b^*_i equals 0.4276. For the limit load condition, dimensions a_i and b_i are calculated using equation 36 and equation 37.

$$a_i = 0.0236 a_i^* \left(\frac{Q}{\Sigma \rho_i} \right)^{1/3} = 0.0236 \cdot 3.571 \cdot \left(\frac{50580}{0.06089} \right)^{1/3} = 7.922 \text{ mm}$$

$$b_i = 0.0236 b_i^* \left(\frac{Q}{\Sigma \rho_i} \right)^{1/3} = 0.0236 \cdot 0.4276 \cdot \left(\frac{50580}{0.06089} \right)^{1/3} = 0.9486 \text{ mm}$$

B.11. Hertz Contact Stress and Static Load Factor

For the limit load condition, the maximum Hertz stress (S_{\max}) is calculated using equation 30.

$$S_{\max} = \frac{1.5 Q_{\max}}{\pi \cdot a_i b_i} = \frac{1.5 \cdot 50580}{\pi \cdot 7.922 \cdot 0.9486} = 3214 \text{ MPa}$$

The static load factor is calculated using equation 40.

$$SF = \left(\frac{4200}{S_{\max}} \right)^3 = \left(\frac{4200}{3214} \right)^3 = 2.232$$

The eight-point contact ball bearing design exceeds the requirement that the static factor (SF) under limit load meets or exceeds 1.5; hence, the design is satisfactory from a maximum static load standpoint.

B.12. Case–Core Interface

The case depth was not specified for the eight-point contact ball bearing; therefore, it is necessary to define an adequate value for that parameter based on the case avoiding crushing during application of the limit load. In Table B-2, bearing ring core hardness was specified at 250 HB. From Table 14, the allowable shear stress with regard to core crushing for this hardness is 346.5 MPa. According to equation 45, the following is the case.

$$\tau = \frac{\zeta \cdot b \Sigma \rho}{1.8754 \cdot 10^{-5}} = 346.5 \text{ MPa}$$

Here, $\Sigma \rho_i$ equals 0.06089 and b_i equals 0.9482, therefore the parameter ζ_i equals 0.1125. The dimensional depth at which ζ_i equals 0.1125 occurs can be determined from Table 15 via a double interpolation. The first interpolation is based on the inverse ellipticity ratio for the contact area. For the inner raceway-ball contact curvatures of the eight-point contact ball bearing, the following is the case.

$$\frac{b_i}{a_i} = \frac{b_i^*}{a_i^*} = \frac{0.4276}{3.571} = 0.1197$$

Hence, for ζ_i equaling 0.1155 and b_i/a_i equaling 0.1197, z/b equals 4.054. Therefore, the case depth is: 4.054 x 0.9482 mm = 3.844mm. The core depth is 110% of the case depth, thus core depth is 4.228 mm.

It next is necessary to check the adequacy of this case depth with regard to fatigue subsurface shear stress. From Table 14, the allowable fatigue shear stress is 207.9 MPa. Considering the peak dynamic load condition of Table B-1, as shown in Table B-5, Q_{\max} equals 20.58 kN. From equation 37 the following can be determined.

$$b_i = 0.0236b_i^* \left(\frac{Q}{\Sigma\rho_i} \right)^{1/3} = 0.0236 \cdot 0.4276 \cdot \left(\frac{20580}{0.06089} \right)^{1/3} = 0.7029 \text{ mm}$$

Core depth is 4.228 mm, therefore the non-dimensional core depth is as follows.

$$\frac{z}{b} = \frac{4.228}{0.7026} = 6.017 \text{ mm}$$

Interpolating in Table 15 for b_i/a_i equaling 0.1197 yields: $\zeta = 0.07067$. Using equation 45 produces the following.

$$\tau = \frac{\zeta \cdot b \sum \rho}{1.8754 \cdot 10^{-5}} = \frac{0.07067 \cdot 0.7029 \cdot 0.06089}{1.8754 \cdot 10^{-5}} = 161.2 \text{ MPa}$$

This fatigue shear stress is less than the 207.9 MPa allowable, giving a safety factor of 1.290 for the eight-point contact ball bearing.

B.13. Hertz Stress Limits for Avoidance of Fretting Corrosion

Table 16 indicates maximum Hertz stress limits as a means to limit or prevent fretting corrosion type of failure at the ball-groove contacts. (As noted in Section 6, these limits are suggested by a manufacturer of large bearings and have not been verified or accepted as standards.) Equation 30 is used to calculate maximum Hertz stress for a ball bearing as follows.

$$S_{\max} = \frac{1.5Q_{\max}}{\pi a_i b_i}$$

Values of Q_{\max} are given in Table B-4. Contact ellipse dimensions a_i and b_i are calculated using equation 36 and equation 37 for the peak dynamic and limit load conditions. These values and ensuing maximum Hertz stresses are given in Table B-6, where they also are compared against the suggested Hertz stress limits.

Table B-6. Comparison of Calculated Hertz Stresses with Recommended Limit Stresses to Avoid Fretting in Bearing Raceways

Condition	a_i (mm)	b_i (mm)	Q_{\max} (kN)	S_{\max} (MPa)	S_{limit}	$S_{\text{limit}} / S_{\max}$
Peak operating load	5.870	0.7029	20.58	2,383	2,800*	1.175
Limit load	7.919	0.9482	50.58	3,216	3,200**	0.995

* Suggested maximum operating Hertz stress from Table 16.

** Suggested ultimate Hertz stress from Table 16.

B.14. Dither Angle

The dither angle (θ_{dith}) is the zone of operation, at very small amplitude of oscillation, where the elliptical contact area between ball and raceway only is partially uncovered and retraced. The amplitude is defined as the width of the contact divided by the raceway radius. Equation 16 is used to calculate θ_{dith} for the ball-inner raceway contact of the eight-point contact ball bearing under the limit load condition.

$$\theta_{dith} = \frac{720^\circ b_i}{\pi \cdot d_m (1 - \gamma)} = \frac{720 \cdot 0.9482}{\pi \cdot 940 \cdot (1 - 0.02633)} = 0.2374^\circ$$

B.15. Lubrication

A lubricant selection has not been included in this pitch bearing example. The grease selected should be recommended for oscillating conditions, and also should have excellent antiwear and extreme pressure additives. Frequent (daily) movement of the blade through a large angle (three times the critical angle θ_{crit} or more) is recommended to redistribute grease to the ball groove surfaces. This minimum rotation ($3\theta_{crit}$) is calculated using equation 15 as follows.

$$3\theta_{crit} = 3 \cdot \frac{720^\circ}{Z(1 - \gamma)} = \frac{3 \cdot 720}{65 \cdot (1 - 0.02633)} = 34.13^\circ$$

B.16. Friction Torque

Bearing friction torque is estimated using equation 47 as follows.

$$T = \mu \frac{d_m}{2} \left(\frac{4.4M}{d_m} + 2.2F_r + F_a \right)$$

The friction coefficient μ is provided by Table 17 for the different types of pitch ball bearings (i.e., depending on mechanical means used to separate the balls).

Starting friction torque or a change in friction torque with a reversal of blade motion can be 1.5 times the running torque. Estimated running torque values for a bearing having a cage and a bearing having ball separators are given in Table B-7. The estimated running friction torque is conservative and high for the purpose of sizing drives and actuators. The friction torque estimates include the effect of drag caused by integral seals.

Table B-7. Bearing Friction Torque for Operating and Limit Loads

Operating Condition	Radial Load (kN)	Axial Load (kN)	Moment (kNm)	$T_{running}$ Cage ($\mu = 0.003$ kNm)	$T_{running}$ Spacers ($\mu = 0.004$ kNm)
1	13.4	32.5	217.1	1.52	2.03
2	25.0	64.7	366.1	2.59	3.45
3	26.1	84.0	372.2	2.66	3.54
4	23.9	88.4	325.3	2.35	3.13
5	19.6	94.0	228.6	1.70	2.27
6	17.8	88.4	187.7	1.42	1.89
Limit load	46.9	145.0	937.1	6.53	8.71

B.17. Miscellaneous

This numerical example does not include an evaluation of the mounting bolts. Other items such as cages, plastic separators, and integral rubbing seals are not subject to numerical analysis. These non-rolling items should be covered in the purchase specifications.

B.18. Summary

Four-point and eight-point contact ball bearings for a pitch bearing application were evaluated with regard to design adequacy in supporting the duty cycle conditions of Table B-1. The four-point contact ball bearing was found to be inadequate to meet the fatigue endurance requirement of 20-year life considering 50% operational time. The eight-point contact ball bearing met the life requirement and further analysis was conducted for that bearing design.

Further, it was found that the eight-point contact ball bearing design meets or exceeds all of the necessary remaining design criteria:

- Static capacity;
- Adequate case depth and core hardness;
- Satisfactory lubrication; and
- Non-rolling items.

The eight-point contact ball bearing design is more costly to manufacture than the four-point contact ball bearing design. Causes for the greater cost of the eight-point contact ball bearing are (1) the second row of balls, (2) a second cage or separators, and (3) the need to assemble and disassemble the bearing during final fit-up. The intent is to match the ball diameters, raceway groove radii, and diametral clearance or preload of the two ball rows.

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