

Issuance Date: Oct. 28,1963

## CIH EFFECTS TEST GROUP

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Report to the Test Director

## SECONDARY MISSILES GENERATED BY NUCLEAR-PRODUCED BLAST WAVES

By
I. Gerald Bowen

Mary E. Franklin
E. Royce Fletcher

Ray W. Albright

| Approved by: | CLAYTON S. WHITE |
| ---: | :--- |
|  | Director |
|  | Program 33 |

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#### Abstract

The generation of secondary missiles by blast waves was investigated in Operation Plumbbob for three nuclear detonations with estimated yields of 11,38 , and 44.5 kt . A trapping technique was used to determine the impact velocities for 17,524 missiles (stones, glass fragments, spheres, and military debris or steel fragments) which occurred in open areas, houses, and an underground shelter with an open entryway. The equivalent ideal-wave peak overpressures computed from measured blast data for the open-area stations varied from 3.8 to 21 psi . Two houses and an underground shelter were located where the overpressures were 3.8 and 65 psi , respectively. The effect of hill-and-dale terrain on the production of missiles was investigated on one of the shots. Precursor effects were noted on two of the shots at stations near Ground Zero (GZ).

Missile velocities measured at all stations except the underground shelter were compared with those computed by use of a model based on an ideal blast wave. An analytical procedure was presented by which translational velocities of man can be estimated using the measured velocities of spheres and stones.

Total distances of displacement were measured for 145 stones that weighed up to 20 kg and for 1528 fragments from a concrete-block wall.


## ACKNOWLEDGMENTS


#### Abstract

Recognition should be given to V. R. Clare and R. V. Taborelli who, as assistant project officers, shared responsibility for the completion of the field work at the Nevada Test Site (NTS). Mr. Clare's participation in the project was made possible by the interest and generosity of Chemical Warfare Laboratories, Edgewood, Md.

Weapons-test operations involve the coordinated efforts of many organizations. Some of those contributing directly to the execution of the secondary-missile project were

Civil Effects Test Group (now Civil Effects Test Operations), Division of Biology and Medicine, Atomic Energy Commission


Armed Forces Special Weapons Project (now Defense Atomic Support Agency), Department of Defense

Federal Civil Defense Administration (now Office of Civil and Defense Mobilization)
Ballistic Research Laboratories
Sandia Corporation

## Kirtland Air Force Base

## Indian Springs Air Force Base

## Holmes \& Narver, Inc., Architectural Engineers

Reynolds Electrical and Engineering Co., Inc.
Acknowledgment is made of the efforts of the following personnel of the Lovelace Foundation who assisted not only in the field work, but also in the tedious and exacting procedures involved in the analysis of the data and preparation of the test report:

Members of the Physics Department who were involved in the field work, data analysis, and related activities are J. J. Anderson, J. M. Craig, A. W. Dennis, K. A. Doherty, J. Kleinfeld, J. D. McCurdy, M. A. Osoff, R. F. D. Perrett, and W. R. Roeder.

Members of the Department of Medical Illustration who were responsible for photographic work at the Test Site and for the preparation of illustrative material for the report are R. A. Smith, G. S. Bevil, Holly M. Ferguson, E. M. Johnson, and R. A. MacMahon.

The final manuscript was prepared by Isabell D. Benton, K. Maureen Gilmore, Ruth P. Lloyd, Vicki J. Newsom, and Helen T. Vatoseow.

Mention should be made of Robert L. Corsbie, who was Director of the Civil Effects Test Group at the time of the field operations, and of Dr. C. S. White, Director of Research of the Lovelace Foundation. The secondary-missile studies were initiated through their leadership.

Operations at NTS were financed jointly by the U. S. Atomic Energy Commission, the Department of Defense, and the Federal Civil Defense Administration. Postoperation analytical work and related studies were supported by the USAEC.

## LIST OF SYMBOLS

| Symbol | Definition | Unit of measurement |
| :---: | :---: | :---: |
| A | Impact area | Sq in. |
| a,b,c | Regression coefficients |  |
| Abs | Absorber |  |
| Al | Aluminum sphere (fraction following type of sphere indicates diameter) | In. |
| $\alpha$ | Acceleration coefficient | Sq ft/lb |
| $\bar{\alpha}$ | Acceleration coefficient for missiles of average mass $M$ | Sq ft/lb |
| CB | Croquet ball |  |
| $\mathrm{c}_{0}$ | Speed of sound in undisturbed air | $\mathrm{Ft} / \mathrm{sec}$ |
| d | Distance traveled by missile | Ft |
| d_ | Minimum distance | Ft |
| $\mathrm{d}_{+}$ | Maximum distance | Ft |
| $\overline{\mathrm{d}}$ | Average distance | Ft |
| $\mathrm{d}_{\mathrm{x}}$ | Distance traveled by missile parallel to direction of propagation of blast wave (downwind) | Ft |
| $\mathrm{d}_{\times 50}$ | Geometric mean of $\mathrm{d}_{\mathrm{x}}$ | Ft |
| $\mathrm{d}_{\mathrm{y}}$ | Distance traveled by missile perpendicular to direction of propagation of blast wave (crosswind) | Ft |
| $\overline{\mathrm{D}}_{\mathrm{s}}$ <br> $\Delta \mathrm{V} \%$ | Average spatial density of missiles in trap $\left(\overline{\mathrm{V}}-\mathrm{V}_{\mathrm{p} 50}\right) / \mathrm{V}_{\mathrm{P} 50}$ | Missiles/sq ft |
| $\mathrm{E}_{\mathrm{gm}}$ | Geometric standard error of estimate in $\text { mass }=\operatorname{antilog} \mathrm{E}_{\mathrm{lm}}$ |  |
| $\mathrm{E}_{\mathrm{gv}}$ | ```Geometric standard error of estimate in velocity = antilog E Elv``` |  |
| $\mathrm{E}_{\mathrm{lm}}$ | Standard error of estimate of log mass | Log units |
| $\mathrm{E}_{\mathrm{lv}}$ | Standard error of estimate of log velocity | Log units |
| $\left(\mathrm{E}_{1 \mathrm{lv}}\right) \%$ | Standard error of estimate of log velocity | \% of velocity units |
| FPG | Plate glass, flat upon arrival at trap |  |
| FWG | Window glass, flat upon arrival at trap |  |
| G | Galileo |  |
| Gl | Glass sphere, large (average mass $=$ 72.6 mg ) |  |
| Gr | Gravel |  |
| Gs | Glass sphere, small (average mass = 36.0 mg ) |  |


| Symbol | Definition | Unit of measurement |
| :---: | :---: | :---: |
| Gx | Glass sphere, extra large (average mass = 242.4 mg ) |  |
| GZ | Ground Zero, the point on the surface vertically below the center of the burst |  |
| $\mathrm{h}_{1}$ | Height above ground at which spheres were placed | In. |
| $\overline{\mathrm{h}}_{2}$ | Average impact height above ground | In. |
| $\mathrm{I}_{\mathrm{p}}$ | Overpressure impulse | Psi-sec |
| k | Constant, added to depth of penetration for velocity calibration | In. |
| kt | Kiloton (kt), energy of nuclear (or atomic) explosion which is equivalent to that produced by the explosion of 1 kt (1000 tons) of TNT |  |
| m | Mass of missile | Mg , unless otherwise specified |
| M_ | Minimum m | Mg , unless otherwise specified |
| $\mathrm{M}_{+}$ | Maximum m | Mg , unless otherwise specified |
| $\overline{\mathrm{M}}$ | Mean or average mass | Mg , unless otherwise specified |
| $\mathrm{M}_{50}$ | Geometric mean mass | Mg , unless otherwise specified |
| MD | Military debris |  |
| n | Number of missiles |  |
| NS | Natural stones |  |
| Ny | Nylon sphere (fraction following type of sphere indicates diameter) | In. |
| P | Priscilla |  |
| p | Overpressure or pressure in excess of $\mathrm{p}_{0}$ | Psi |
| $\mathrm{p}_{0}$ | Pressure of undisturbed air or ambient pressure | Psi |
| $\mathrm{p}_{\text {s }}$ | Maximum overpressure or shock overpressure | Psi |
| PG | Plate glass |  |
| q | Dynamic pressure | Psi |
| R | Range, distance of station from GZ | Ft |
| S | Smoky |  |
| S | Depth of penetration of missile in absorber | In. |
| S. | Minimum s |  |
| $\mathrm{S}_{+}$ | Maximum s |  |
| $\mathrm{S}_{\text {dy }}$ | Standard deviation of $\mathrm{d}_{\mathrm{y}}$ |  |
| $\mathrm{S}_{\mathrm{gm}}$ | Standard geometric deviation of mass $=$ antilog $\mathrm{S}_{\mathrm{lm}}$ |  |
| $\mathrm{S}_{\mathrm{gv}}$ | Standard geometric deviation of velocity $=$ antilog $\mathbf{S}_{1 v}$ |  |
| $\mathrm{S}_{1 \mathrm{ld}_{\mathrm{x}}}$ | Standard deviation of $\log d_{x}$ | Log units |
| $\mathrm{S}_{\mathrm{lm}}$ | Standard deviation of log mass | Log units |
| $\mathrm{S}_{1 \mathrm{lv}}$ | Standard deviation of log velocity | Log units |
| $\mathrm{S}_{\mathrm{m}}$ | Standard deviation of $m$ |  |
| $\mathrm{S}_{\mathrm{v}}$ | Standard deviation of $v$ |  |
| St | Steel sphere (fraction following type of sphere indicates diameter) | In. |


|  | Symbol | Definition | Unit of measurement |
| :---: | :---: | :---: | :---: |
|  | t | Time after arrival of blast wave | Sec |
|  | $\mathrm{t}_{\mathrm{p}}^{+}$ | Duration of positive pressure phase of blast wave | Sec |
| 3 | v | Velocity | Ft/sec |
|  | $V_{-}$ | Minimum v | $\mathrm{Ft} / \mathrm{sec}$ |
|  | $\mathrm{V}_{+}$ | Maximum v | $\mathrm{Ft} / \mathrm{sec}$ |
|  | $\overline{\mathrm{V}}$ | Mean or average velocity | $\mathrm{Ft} / \mathrm{sec}$ |
|  | $\mathrm{V}_{50}$ | Geometric mean velocity | $\mathrm{Ft} / \mathrm{sec}$ |
| 3 | $\mathrm{V}_{\mathrm{p} 50}$ | Predicted velocity for missiles of mass $\mathbf{M}_{50}$ (if $\mathrm{M}_{50}$ not listed, $\overline{\mathrm{M}}$ ) | $\mathrm{Ft} / \mathrm{sec}$ |
|  |  | $\mathrm{V}_{\mathrm{p} 50}$ for gravel |  |
|  | $\left(\mathrm{V}_{\mathrm{p} 50}\right)_{\mathrm{R}}$ | $\mathrm{V}_{\mathrm{p} 50}$ assuming reflected pressure |  |
|  | WG | Window glass |  |
|  | WGH | Window glass inside concrete house |  |
|  | Roman | erals designate type of absorber identified in | e 2.1, page 30 |

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## Chapter 1

## INTRODUCTION

### 1.1 BACKGROUND

Experience with large-scale explosions, e.g., those of Hiroshima, Nagasaki, ${ }^{1}$ and Texas City, ${ }^{2,3}$ has demonstrated that missiles resulting from blast effects are a significant cause of biological damage. These casualty-producing missiles were mostly fragments of glass from broken window panes, but they could have been any object not securely anchored which could be translated by the high winds accompanying a blast wave. Indeed, in many instances people themselves became missiles by virtue of their involuntary translation by the blast winds.

A systematic study ${ }^{4}$ of the translational velocities of window-glass fragments and stones was made during the 1955 weapons tests in Nevada (Operation Teapot). During the following Nevada test series in 1957 (Operation Plumbbob), translational effects were investigated by five separate projects: (1) Project 4.1, which used window-glass fragments as missiles and swine as targets; ${ }^{5}$ (2) Project 33.1 , which used dogs in shelters as translational objects; ${ }^{6}$ (3) Project 33.3, which used anthropomorphic dummies in open areas as translational objects; ${ }^{7}$ (4) Project 33.4, which used window-glass fragments, gravel, and concrete blocks as missiles and dogs as targets; ${ }^{8}$ and (5) Project 33.2 whose studies are reported herein.

In addition to the field investigations noted above, a few laboratory type studies have been made which are pertinent to the evaluation of translational effects of blast waves. One study ${ }^{9}$ was aimed at establishing the penetrating potential of glass-fragment missiles into the abdominal cavity of dogs as a function of fragment mass and velocity at impact. Another study was concerned with the biological effects of direct impact of experimental subjects ${ }^{10}$ (mice, rats, guinea pigs, and rabbits) with a smooth hard surface, a situation similar to that which could occur as a result of translation by blast winds. A third study involved the use of a shock tube to accelerate goats and dummies; ${ }^{11}$ these goats and dummies were then allowed to decelerate by tumbling over a flat grassy surface. It was concluded that the principal source of damage to the goats was the decelerative tumbling.

Two other studies of an analytical nature should be mentioned since they were motivated by the voluminous field data contained in this report. The first study resulted in a mathematical model ${ }^{12}$ that allowed numerical computations of the velocity, displacement, and acceleration histories of arbitrary objects when exposed to classical blast waves such as those resulting from nuclear detonations. Before such a model could be used, it was necessary to determine certain aerodynamic parameters of the translated objects. Thus drop-test experiments ${ }^{13}$ were performed to permit the determination of acceleration coefficients for the experimental objects that were used in the present study (glass fragments, stones, etc.) as well as for mice, rats, guinea pigs, and rabbits. These efforts made it possible to present predicted velocities in this report for comparison with the ones determined experimentally.

### 1.2 CATEGORIES OF BIOLOGICAL EFFECTS OF BLAST

For purposes of orientation, the categories into which the biological effects of blast are usually divided are mentioned here briefly. ${ }^{14-17}$ These effects can be thought of as being of four distinct types: (1) primary; (2) secondary; (3) tertiary; and (4) miscellaneous.

The primary effects are those due to variations in environmental pressure caused by explosive events. As a general rule critical pathology is most marked in the air-containing or gans (the lungs, gastrointestinal tract, ear, and paranasal sinuses) and at those locations where there is the greatest variation in tissue density. ${ }^{6,18-23}$

Secondary blast effects are those due to missiles that are energized by the blast overpressures and winds or by ground shock and gravity.

Missiles may consist of fragments of window glass, stones, pieces of building debris, or any object other than man which is set in motion by the blast wave. Injury may result from penetration of the surface wall or organs of the body or from nonpenetrating impact of the missile.

If the biologic target is translated by the blast wave, ground shock, or gravity, the effect is called tertiary. Injury can occur during the accelerative phase of displacement; however, significant damage is more likely to occur during decelerative tumbling or upon impact with a stationary object.

The fourth category of blast damage consists of miscellaneous effects such as those due to blast-induced dust and fires as well as to gases, dust, or debris that have been heated aerodynamically or by direct thermal radiation.

### 1.3 OBJECTIVES

The purpose of the field tests reported herein was to produce information on blast-produced missiles which would be of value in assessing the secondary type blast injury described in the previous section. It will be apparent later that the results are also applicable to some extent to the evaluation of biological effects in the tertiary category.

Specifically, it was planned to determine individual translational velocities for various types of small objects (window-glass fragments, stones, spheres, etc.) by means of a trapping technique that was used first for this purpose during Operation Teapot. ${ }^{4}$ The technique used, described in Chap. 2, permitted the evaluation of velocities and masses for large samples of missiles that occurred near the location of the trap.

It was planned to obtain velocities, masses, and spatial distributions (where applicable) for the following types of missiles in the environments noted:

1. Window- and plate-glass fragments inside houses and in open areas where the windows were mounted without "benefit" of a house.
2. Natural (or native) stones in flat and hill-and-dale terrain.
3. Gravel that had been marked for identification and placed at various distances in front of traps in open areas.
4. Small metallic, nylon, and wooden spheres placed in front of traps in flat and hill-anddale terrain and in a shelter with an open entryway.
5. "Military" debris (fragments of steel) placed in front of traps in flat and hill-and-dale terrain.

Since the size of objects that could be accommodated by a missile trap is limited, other studies were planned in which only the total displacement was to be determined. This included the displacement of large stones (up to about 20 kg ) and of concrete blocks from a wall exposed to a blast wave.

The final and perhaps most significant objective was to compare missile velocities that were empirically determined with velocities that were computed* through use of the analytical work mentioned in the last paragraph of Sec. 1.1. From these comparisons it was hoped that

* The blast parameters used in these computations were determined from overpressure measurements made at each missile station by Ballistic Research Laboratories, Aberdeen, Md.
some degree of confidence could be established in the computational methods used. These methods could then be used to predict secondary-missile hazards for range - yield combinations different from those used in the test series.


## REFERENCES

1. Samuel Glasstone (Ed.), "The Effects of Nuclear Weapons," Superintendent of Documents, U. S. Government Printing Office, Washington, D. C., June 1957.
2. George Armistead, Jr., The Ship Explosions at Texas City, Texas, on April 16 and 17, 1947, and Their Results, Engineering Consultants Report to John G. Simonds and Company, Inc., Oil Insurance Underwriters, New York City; Washington, D. C., June 1, 1947.
3. Virginia Blocker and T. G. Blocker, The Texas City Disaster-A Survey of 3000 Casualties, Am. J. Surg., 78: 756-771(1949).
4. I. G. Bowen, A. F. Strehler, and M. B. Wetherbe, Distribution and Density of Missiles from Nuclear Explosions, Operation Teapot Report, WT-1168, December 1956.
5. G. M. McDonnel, H. A. Claypool, W. A. Moncrief, and J. D. Goldstein, Effects of Nuclear Weapons on a Large Biological Specimen (Swine), Project 4.1, Operation Plumbbob Report, ITR-1428, Nov. 5, 1957.
6. D. R. Richmond, R. V. Taborelli, I. G. Bowen, T. L. Chiffelle, F. G. Hirsch, B. B. Longwell, J. G. Riley, C. S. White, F. Sherping, V. C. Goldizen, J. D. Ward, M. B. Wetherbe, V. R. Clare, M. L. Kuhn, and R. T. Sanchez, Blast Biology - A Study of the Primary and Tertiary Effects of Blast in Open Underground Protective Shelters, Operation Plumbbob Report, WT-1467, June 30, 1959.
7. R. V. Taborelli, I. G. Bowen, and E. R. Fletcher, Tertiary Effects of Blast -Displacement, Operation Plumbbob Report, WT-1469, May 22, 1959.
8. V. C. Goldizen, D. R. Richmond, and T. L. Chiffelle, Missile Studies with a Biological Target, Operation Plumbbob Report, WT-1470, Jan. 23, 1961.
9. I. G. Bowen, D. R. Richmond, M. G. Wetherbe, and C. S. White, Biological Effects of Blast from Bombs: Glass Fragments as Penetrating Missiles and Some of the Biological Implications of Glass Fragmented by Atomic Explosions, USAEC Report AECU-3350, Lovelace Foundation for Medical Education and Research, June 18, 1956.
10. D. R. Richmond, I. G. Bowen, and C. S. White, Tertiary Blast Effects. 1. Effects of Impact on Mice, Rats, Guinea Pigs, and Rabbits, Aerospace Med., 32: 789-805(September 1961).
11. R. S. Anderson, F. W. Stemler, and E. B. Rogers, Air Blast Studies with Animals, Part II, Report CRDLR-3049, Army Chemical Research and Development Laboratories, April 1961.
12. I. G. Bowen, R. W. Albright, E. R. Fletcher, and C. S. White, A Model Designed to Predict the Motion of Objects Translated by Classical Blast Waves, USAEC Report CEX-58.9, June 29, 1961.
13. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determinations of Aero-dynamic-drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.
14. C. S. White, I. G. Bowen, D. R. Richmond, and R. L. Corsbie, Comparative Nuclear Effects of Biomedical Interest, USAEC Report CEX-58.8, Jan. 12, 1961.
15. C. S. White, Biological Blast Effects, Hearings Before the Special Subcommittee on Radiation of the Joint Committee of Atomic Energy, Congress of the United States, Eighty-Sixth Congress, First Session on Biological and Environmental Effects of Nuclear War, Part 1, pp. 311-372, U. S. Government Printing Office, Washington, D. C., June 22-26, 1959.
16. C. S. White, Blast Biology-A Summary, a Contribution to the Holifield Subcommittee Hearings, May 1, 1958. Published in Report of Hearings Before a Subcommittee on Government Operations, House of Representatives, Part I-Atomic Shelter Tests, pp. 80-93, U. S. Government Printing Office, Washington, D. C., 1958.
17. C. S. White, Biological Blast Effects, USAEC Report TID-5564, Lovelace Foundation for Medical Education and Research, September 1959.
18. C. S. White, T. L. Chiffelle, D. R. Richmond, W. H. Lockyear, I. G. Bowen, V. C. Goldizen, H. W. Merideth, D. E. Kilgore, B. B. Longwell, J. T. Parker, F. Sherping, and M. E. Cribb, Biological Effects on Pressure Phenomena Occurring Inside Protective Shelters Following A Nuclear Detonation, Operation Teapot Report, WT-1179, Oct. 28, 1957.
19. D. R. Richmond, M. V. Wetherbe, R. V. Taborelli, T. L. Chiffelle, and C. S. White, The Biologic Response to Overpressure. I. Effects on Dogs of Five- to Ten-second Duration Overpressures Having Various Times of Pressure Rise, J. Aviation Med., 28: 447-460 (1957).
20. D. R. Richmond, R. V. Taborelli, F. Sherping, M. B. Wetherbe, R. T. Sanchez, V. C. Goldizen, and C. S. White, Shock Tube Studies on the Effects of Sharp-rising, Longduration Overpressures on Biological Systems, USAEC Report TID-6056, Lovelace Foundation for Medical Education and Research, Dec. 21, 1960.
21. C. S. White and D. R. Richmond, Blast Biology, USAEC Report TID-5764, Lovelace Foundation for Medical Education and Research, Sept. 18, 1959.
22. D. R. Richmond, V. R. Clare, V, C. Goldizen, D. E. Pratt, R. T. Sanchez, and C. S. White, Biological Effects of Overpressure. II. A Shock Tube Utilized to Produce Sharp-rising Overpressures of 400 Milliseconds Duration and Its Employment in Biomedical Experiments, Aerospace Med., 32: 997-1008(1961).
23. D. R. Richmond, V. C. Goldizen, V. R. Clare, D. E. Pratt, F. Sherping, R. T. Sanchez, C. C. Fischer, and C. S. White, The Biologic Response to Overpressure. III. Mortality in Small Animals Exposed in a Shock Tube to Sharp-rising Overpressures of 3-4 Msec Duration, Aerospace Med., 33: 1-27(1962).

## Chapter 2

## MISSILE-ABSORBING TECHNIQUES <br> AND METHODS OF ANALYSIS

### 2.1 INTRODUCTION

It was possible to obtain impact velocities for large numbers of secondary missiles (objects translated by the blast wave) by techniques that required quite simple instrumentation. The field operation consisted of placing a suitable absorbing material downwind from the source of secondary missiles. Following the detonation the absorbing material was taken to the laboratory where each missile was extracted, and the depth of penetration and missile mass were measured. Impact velocity could then be determined by use of a calibration equation applicable to the type of absorber used and the type of missile caught.

This chapter will be concerned first with a description of the missile absorbers used and the methods of placing them in the field. Next will follow an account of the laboratory and analytical procedures used to arrive at calibration equations for each absorber and missile type combination. Finally, some of the statistical methods used to organize the large quantities of missile data obtained from the field tests will be reviewed.

### 2.2 MISSILE ABSORBERS

The missile-absorbing technique used in blast studies is characterized by the translated object's being accelerated by weak pressures applied over long distances in air and then being decelerated by stronger pressures over shorter distances in the absorber. This arrangement of pressure strengths is necessary so that the absorber will not be crushed by the dynamic pressure accelerating the missile as well as by the usually greater static pressure (or overpressure), especially if the latter is reflected at the surface of the absorber. Thus an absorber should be strong enough to withstand the pressures accompanying the blast wave yet weak enough to be penetrated by the missiles generated by the same wave. It should be noted that the blast wave does not decay appreciably between the time the missiles are generated and the time the wave reaches the absorbing material.

Mechanical properties other than compressive yield strength, described above, need to be considered in the choice of an absorber. It is important, for instance, that the shear strength be low so that each deformation be localized, i.e., the depth of penetration for each missile should not be influenced by the penetration of other missiles in the vicinity. Furthermore, it was found that the more nonresilient the material, the more reliably it could be calibrated. It is apparent that a material that would even partially return to its original shape after impact would be of little value in the measurement of impact velocities. In addition, obviously the material should be structurally uniform so that a velocity calibration obtained from using a sample of the absorber would apply to other material of the same type used in the field operation.

Another important consideration in the choice of an absorbing material is its resistance to heat. Even a temporary change in the mechanical properties of the absorber due to heating,
especially in the outer layer which is exposed to thermal radiation in most instances and to hot blast winds, could change the depth to which a missile would penetrate. Since the outer layer is most susceptible to thermal effects,* the errors introduced in the evaluation of missile velocities would be most significant for the objects with small depths of penetration.

The materials that were found to be suitable (with reservations) for the present study are listed in Table 2.1. Absorber types I, II, III, and IV are expanded polystyrenes. $\dagger$ Types V and VI are balsa wood, selected on the basis of density.

TABLE 2.1-ABSORBERS USED TO TRAP MISSILES*

| Type | Description | Density, <br> lb/cu ft | Compressive <br> yield strength, <br> psi | Shear strength, <br> psi | Maximum temp. <br> for continuous <br> use, ${ }^{\circ} \mathrm{F}$ |
| :--- | :--- | :---: | :---: | :---: | :---: |
| I | Special order | 1.54 |  |  |  |
| II | Styrofoam 22 | 1.6 to 2.0 | 16 to 32 | 27 to 36 | 175 |
| III | Q-103.15 | 2.8 to 3.2 | 50 to 80 | 53 to 62 | 175 |
| IV | Q-103.21 | 4.3 to 4.7 | 120 to 140 | 80 to 95 | 175 |
| V | Balsa wood | 7.85 |  |  |  |
| VI | Balsa wood | 10.78 |  |  |  |

* All absorbers except balsa wood were manufactured by Dow Chemical Co., Midland, Mich., using expanded polystyrene. Specifications for types II, III, and IV absorbers were supplied by the manufacturer. BaIsa wood was used end-grain only.

Type I absorber was prepared on a special order, $\ddagger$ but types II, III, and IV are stock items. These materials were tested in a shock tube for compressive yield strength under dynamic conditions. Samples that were 2 in . thick and 1 ft square were cemented to the closed end of the tube. For types II, III, and IV, it was found that the compressive yield strengths determined in the dynamic tests were approximately the same as those specified by the manufacturer for static loading (see the fourth column of Table 2.1). Thus these data served as a guide in the selection of the type of absorber to be used at various field installations.

The mechanical properties of the expanded polystyrene were found to be reasonably good. The principal difficulty encountered with the balsa wood was its nonuniformity. Homogeneity was improved by dividing pieces of wood into two groups according to density and making calibrations for each group separately. It was found that when the wood was used end-grain the deformations were localized to the areas of impact.

Since all types of absorbing material used were susceptible to modification by heat, it was necessary to provide thermal protection without appreciably changing the missile-catching properties of the absorber. In shot Priscilla this consisted in placing two 0.0007 -in. -thick layers of aluminum foil over the exposed side of the absorber. This proved to be insufficient protection in some instances; therefore additional protection was arranged for some of the installations in later shots (see Sec. 2.3 and Fig. 2.2).

### 2.3 CONSTRUCTION OF TRAP HOUSING AND ANCHORS AND WINDOW MOUNTS

Construction details for the trap housing that was used at most installations are illustrated in Fig. 2.1. The housing was designed to hold absorbing material 36 in . wide, 12 in . high, and 11 in. deep. Types II, III, and IV absorbing material were placed in the housing in 1-and 2-in.

[^1]2.4 CALIBRATION OF MISSILE ABSORBERS

### 2.4.1 Experimental Procedure

The air gun and the velocity-measuring device used in the calibration of absorbers are described in Ref. 1. Three sizes of gun barrels were used; the gun barrels were about 8 ft long and were 1,2 , and 3 in . in inside diameter. Sabots were made of various types of expanded polystyrene (see Table 2.1). The sabots consisted of cylindrical plugs with diameters somewhat smaller than, and lengths at least as large as, the diameter of the gun barrel to be used.

For the larger sized gun barrels it was not feasible to use a choke to stop the sabot, as described in Ref. 1. Instead, the following procedure was used: A hole about one-half as long as the sabot was drilled in one of its flat surfaces. The missile to be shot was placed at the bottom of the hole. $\ddagger$ The rim of the cup-like sabot, which contacted the target first, served to

[^2]decelerate the sabot before the impact of the missile. The advantage of this procedure was that the light beams that controlled the electronic timer were interrupted by a sabot of regular shape instead of by a missile of irregular shape followed by a sabot.

To serve as a check on the velocity determinations that were made with the gun, as well as to produce additional calibration data, free-fall experiments were performed in an elevator shaft where the usable free-fall distance was about 48 ft (with corresponding impact velocities up to approximately $55 \mathrm{ft} / \mathrm{sec}$ ). In these experiments the absorbing material was placed at the bottom of the shaft, and the missiles ( $7 / 16$ - to $15 / 16$-in.-diameter steel spheres) were dropped from a measured distance. Results obtained from another study ${ }^{2}$ were used to evaluate impact velocity. The penetration data obtained in this way were found to be comparable to those resulting from the air-gun experiments.

### 2.4.2 Glass Fragments with Random Orientations

Experimentation with the calibration of type II absorber with glass fragments showed that the depth of penetration was almost independent of impact orientation of the fragment provided the angle made by the flat side of the missile and the absorber was greater than about $15^{\circ}$. It was also found that the thickness of the glass from which the fragment was made was not significant in determining its depth of penetration. Two significant parameters, however, were missile mass and impact velocity. It was empirically determined that, for fragments of a given mass, the calibration data would fit an equation of the form

$$
\begin{equation*}
\log v=A+B \log s \tag{2.1}
\end{equation*}
$$

where $A$ and $B$ are constants if the missile masses are constant, $v$ is the impact velocity, and $s$ is the depth of penetration.

Further investigation showed that A and B could be represented within wide ranges of mass by

$$
\begin{equation*}
A=a+c \log m \quad B=b+d \log m \tag{2.2}
\end{equation*}
$$

where $a, c, b$, and $d$ are constants and m is missile mass. Thus, when Eqs. 2.1 and 2.2 are combined, the resulting calibration equation is

$$
\begin{equation*}
\log v=a+c \log m+(b+d \log m) \log s \tag{2.3}
\end{equation*}
$$

The experimental data for randomly oriented glass fragments in type II absorber consisted of values of impact velocity, mass, and depth of penetration for 258 shots. As an aid to the analysis of the data, the missiles were grouped according to mass; the range of masses within each group was $\pm 2.5$ per cent of the average. The average masses of seven groups of fragments thus formed were from 0.0274 to 11.406 g .

It was necessary to determine two fits with Eq. 2.3: one for missiles of small mass and the other for missiles of large mass. The resulting equations, along with appropriate plots, are presented in Fig. 2.4.

An enlarged version of the chart in Fig. 2.4 was used to evaluate velocities* for glass fragments that were caught in the field operation by the type II absorber. The velocity vs. mass analysis for each sample of missiles caught (described and illustrated later in the report) demonstrated that log velocity was an approximately linear function of log mass. Thus for analytical purposes it was decided to group the field data into constant log-mass and logvelocity intervals. The log intervals used (based on common logarithms, $\log _{10}$ ) were 0.1 for mass and 0.05 for velocity. These intervals, labeled a through v for velocity and A through $Z$ and AA through KK for mass, are plotted in Fig. 2.4. The appropriate group identifier was determined for each missile by means of simultaneous mass and depth-of-penetration entries on the chart.

[^3]A few of the absorbers that were placed behind windows (especially those containing plate glass) received impressions that indicated that fragments had struck with a flat surface forward and that no appreciable change in orientation occurred during deceleration. In most cases the larger fragments impacting in this manner did not remain in the absorber but fell to the ground. However, the missile could be described even though it was not retrieved, since the thickness and density of the glass were known and the area of the fragment could be estimated from the impression in the absorber.

Calibration experiments were designed for the flat type impact with the assumption that the missile could be described by two parameters: (1) mass per unit impact area or area density ( $\mathrm{m} / \mathrm{A}$ ) and (2) impact area (A). Average values of $\mathrm{m} / \mathrm{A}$ corresponding to double-strength window glass and plate glass used in the field tests were 4.957 and $9.498 \mathrm{~g} / \mathrm{sq}$ in. It was not feasible to shoot actual plates from the air gun; therefore plates were simulated by cementing 0.064 - to 0.130 -in. -thick Plexiglass disks to the end of balsa cylinders, and the total mass was adjusted to achieve the desired values of area density. These missiles, which were made to fit three sizes of gun barrels, had impact areas of $0.7466,3.032$, and 6.998 sq in. Three missiles were made with each of the above areas, but with different area densities, making a total of nine test objects.

Each of the nine test missiles was shot 10 times into type II absorber at velocities ranging from about 59 to $220 \mathrm{ft} / \mathrm{sec}$; the depth of penetration was from 0.026 to 1.96 in . Data for each missile were fitted by the least-squares method to the following form, area density and impact area being constant:

$$
\begin{equation*}
\log v=C+0.5 \log (s+k) \tag{2.4}
\end{equation*}
$$

where $v$ is the impact velocity, $C$ and $k$ are the regression coefficients, and $s$ is the depth of penetration.

Further analysis showed that k was a function of area alone and could be represented by

$$
\begin{equation*}
\log k=-0.7099+0.3502 \log A \tag{2.5}
\end{equation*}
$$

where $A$ is in square inches and $k$, to be added to $s$, is in inches.
By use of data for missiles of the same area density, $C$ in Eq. 2.4 could be represented by

$$
\begin{equation*}
\mathrm{C}=\mathrm{c}_{1}+\mathrm{c}_{2} \log \mathrm{~A} \tag{2.6}
\end{equation*}
$$

where $c_{1}$ and $c_{2}$ are regression coefficients but can be defined in terms of area density as

$$
\begin{equation*}
c_{1}=d_{1}+d_{2}\left(\frac{m}{\mathrm{~A}}\right)+\mathrm{d}_{3}\left(\frac{\mathrm{~m}}{\mathrm{~A}}\right)^{2} \tag{2.7}
\end{equation*}
$$

and

$$
\begin{equation*}
c_{2}=e_{1}+e_{2}\left(\frac{m}{A}\right)+e_{3}\left(\frac{m}{A}\right)^{2} \tag{2.8}
\end{equation*}
$$

When Eqs. 2.4, 2.6, 2.7, and 2.8 are combined and values for the regression coefficients are substituted, the following calibration equation results:
$\log v=2.3472+0.00045\left(\frac{m}{A}\right)-0.002244\left(\frac{m}{A}\right)^{2}$

$$
\begin{equation*}
+\left[-0.01756+0.00009\left(\frac{m}{A}\right)-0.000439\left(\frac{m}{A}\right)^{2}\right] \log A+0.5 \log (s+k) \tag{2.9}
\end{equation*}
$$

where $k$ is defined in Eq. 2.5 as a function of $A$, and the units are: for $v$, feet per second; $m$, grams; A, square inches; and $s$, inches. When only the data for test missiles with area densities that corresponded to double-strength window glass and plate glass were used, the standard error of estimate in log velocity was found to be 0.0122 log unit, or about 3 per cent.

An enlarged version of the nomogram in Fig. 2.5 was used to solve Eq. 2.9 for the purpose of evaluating velocities for the appropriate missiles caught in the field operation. Equation 2.5, which defines $k$ in terms of impact area, was solved by a simple graph (which is not shown). Use of the nomogram is illustrated in the lower left portion of Fig. 2.5; the illustration involves one step where values of $A$ and $m / A$ are entered and another where ( $s+k$ ) is entered and velocity is read.

### 2.4.4 Gravel and Natural Stones

Calibration data for gravel and natural stones were not significantly different from each other and were therefore combined for analysis. The experimental and analytical procedures followed were essentially the same as those described in Sec. 2.4.2 for glass fragments with random orientations. In some instances it was necessary to divide the data into two or more parts, according to missile mass, and to apply a regression equation of the form of Eq. 2.3 to each part separately. Calibration data for the balsa absorbers showed much more variability than did those for the more structurally uniform plastic absorbers. Detailed information in regard to the resulting calibration equations as well as their limits of applicability will be presented in Sec. 2.4.6 and Table 2.2.

### 2.4.5 Spheres and Military Debris

With the exception of the soda-glass spheres, the mass for each type and diameter of sphere could be considered constant. Thus, for spheres of constant mass, the following simpler type of calibration equation was used:

$$
\begin{equation*}
\log v=a+b \log (s+k) \tag{2.10}
\end{equation*}
$$

where $v$ is impact velocity, $a$ and $b$ are regression coefficients, $s$ is depth of penetration, and k is a correction term added to the total depth of penetration to yield the depth of a cylindrical deformation of the same diameter and volume as the one observed but with a flat bottom instead of the rounded one made by a sphere.

The correction $k$, defined above, was used only in instances where its application would reduce the standard error of estimate in log velocity. In some cases depths of penetration less thán the sphere radius were of interest. For these shallow deformations the actual depth was used to compute an equivalent depth --the equivalent depth is defined as the depth of a flatbottom cylindrical hole with the same diameter as the sphere and same volume as the actual deformation.

Soda-glass-sphere data for penetration in the plastic absorbers were analyzed in a manner similar to that used for glass fragments (Sec. 2.4.2). However, for the type $V$ balsa absorber, the calibration equation used was similar to Eq. 2.10 with $\mathrm{k}=0$; the results are applicable to spheres with masses within specified limits.

Detailed information regarding the individual calibration equations is given in Table 2.2.
The military debris used in this study consisted mostly of steel fragments that were produced by the deformation of small steel-encased charges of high explosives.

The depths of penetration for steel fragments of constant mass and velocity were averaged for a number of randomly oriented impacts. It was found that steel spheres of the same mass
and impact velocity would penetrate to a depth not significantly different from that for the average value for the fragments. Thus the steel-sphere penetration data were used to estimate the impact velocities of military debris, using the calibration for the sphere whose mass was nearest the steel fragment of interest. The steel spheres varied from $1 / 8$ to $15 / 16 \mathrm{in}$. in diameter and from 0.1308 to 54.95 g in mass (see Table 2.2).

### 2.4.6 Summary of Calibration Results

The results obtained from the calibration procedures discussed in the previous sections are listed in Table 2.2. The equations presented in tabular form are those which were used to determine impact velocity for missiles trapped in the various absorbers employed in the field operation. Other quantities are specified which make it possible to assess the limits of applicability of the calibration equations.

The numbers listed under $a, b, c$, and d are regression coefficients for the general calibration equation stated at the top of the table. The values given under $k$ are corrections to be added to the depth of penetration, $s$. In some instances a different form of regression equation was used, in which case the appropriate equation is presented as a footnote to the table.

Maximum and minimum values of the following parameters used in the calibration procedures are designated by the subscripts + and -, respectively: M for missile mass, grams; $s$ for depth of penetration, inches; and $V$ for impact velocity, feet per second.

The numbers listed under n in the table designate the number of missile penetrations used to determine the calibration equations. $\mathrm{E}_{\mathrm{lv}}$ is the standard error of estimate in log-velocity units and ( $\mathrm{E}_{\mathrm{lv}}$ ) \% is the same quantity expressed in per-cent-of-velocity units.

### 2.5 THRESHOLD VELOCITIES

Threshold velocity, as used in this report, is the lowest velocity of impact that can be evaluated for a given missile-absorber combination. The importance of the concept in the design of secondary-missile experiments was implied in Sec. 2.2. The use of threshold velocities in the interpretation of field data will be discussed in the latter part of this section.

With the exception of glass fragments that impacted flat, the criterion for computing threshold velocity was that the depth of penetration be just sufficient for the missile to be retained in the absorber. In the case of spheres, the "sufficient" depth was assumed to be equal to the radius of the sphere. For stones the critical, or threshold, depth was taken to be the radius of a sphere with the same mass and density as that of the stone. A similar assumption was made for randomly oriented glass fragments,* except that both the radius and diameter of the "equivalent" sphere were used. This resulted in a band of threshold velocities, as illustrated in Fig. 2.6; the upper limit is for a penetration depth of one diameter of the equivalent sphere, and lower limit, one radius. The reason for the greater uncertainty of the threshold velocities for glass fragments is that retention is more dependent on orientation of impact for plate-like missiles than for objects that are usually more spherical, such as stones.

Since it was not necessary to recover the impacting glass fragment if its broad surface had the same orientation as the surface of the absorber (see Sec. 2.4.3), the requirement for velocity determination was simply that the impression made in the absorber be detectable. Figure 2.7 is a plot of threshold velocity as a function of missile mass for window and plate glass with flat orientations at impact. The data in this figure were computed on the assumption that a $0.05-\mathrm{in}$. deformation is detectable and measurable.

Threshold velocities for natural stone and gravel are shown in Fig. 2.8 as a function of missile mass for absorber types II, III, IV, V, and VI. A density of $2.72 \mathrm{~g} / \mathrm{cm}^{3}$ was used for both natural stones and gravel to make the necessary computations.

Figure 2.9 displays threshold velocities for $1 / 8$-in. -diameter nylon spheres in absorber types II, III, and IV, and seven $1 / 8$ - to $15 / 16$-in. -diameter aluminum spheres impacting in ab-

[^4]| $\log v=a+c \log m+(b+d \log m) \log (a+k)$ |  |  |  |  | Where $v$ is in feet per recond; $m, g$ grans; and $A$, inches; and $a, b, c$, and $d$ are constants |  |  |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Minstlile | Abeorber type | 4 | b | c | d | k | M- | M. | *- | $9+$ | $v$. | $v$. | n | $E_{1 v^{(1)}}{ }^{(1)}$ | ${ }_{\left.\left(E_{1},\right)^{4}\right)^{(1)}}$ |
| $W_{\text {O }} ;{ }_{\text {PG }}$ | 1 | 2.1790 | 0.5 | -0.01955 | 0 | 0 | 0.0164 | 4.990 | 0.385 | 7.230 | 123 | 344 | 96 | 0.0456 | 10.5 |
|  | $11(2)$ | $\left\{\begin{array}{l}\text { 2.2464 } \\ 2.2124\end{array}\right.$ | ${ }^{0.5651}$ | -0. 1525 | 0. 1608 | 0 | ${ }_{1}^{0.0274}$ | 0.3814 11.406 | 0.065 0.445 | 1.913 4.450 | 120 69.0 | $\left.\begin{array}{l}302 \\ 262\end{array}\right\}$ | 259 | 0.0485 | 11.2 |
|  |  | [2.2124 | 0.5399 | -0.2704 | 0.1409 |  |  | 11.406 |  |  |  |  |  |  |  |
| FWG; FPG | 4 | soe footnote 3 |  |  |  |  |  |  | 0.026 | 1.96 | 59.0 | 220 | 90 | 0.0122 | 2.8 |
| ns; Gr | 1 | See footnote 4 |  |  |  |  |  |  |  |  |  |  |  |  |  |
|  |  | $\int^{2.2865}$ | 0.6126 | -0.1477 | 0.1763 | 0 | 0.0353 | 0.145 | 0.143 | 0.939 | 105 | ${ }^{297}$ |  |  |  |
|  | II | $\left\{\begin{array}{l}2.1758 \\ 2.2310\end{array}\right.$ | ${ }_{0}^{0.6128} 0$ | ${ }_{-0.1399}^{-0.2800}$ | 0.1763 0.1789 | $\bigcirc$ | 0.145 0.402 | 0.402 0.669 | 0.146 0.392 | ${ }_{1}^{1.367}$ | 100 95.3 | 234 231 | 240 | 0.0387 | 8.9 |
|  |  | 2.2125 | 0.5517 | -0.2414 | $-0.1607$ | 0 | 0.660 | 2.960 | 0.322 | 2.156 | 80.7 | ${ }^{205}$ ) |  |  |  |
|  |  | [2.5050 | 0.5429 | -0.1648 | 0.0233 | 0 | 0.011 | ${ }^{0.0433}$ | 0.048 | ${ }^{1.093}$ | 151 | 795 | 47 | 0.0548 | ${ }^{12.6}{ }^{1.8}$ |
|  | Hf | soet footsoles ${ }^{\text {a }}$ |  |  |  |  | ${ }^{0.0433}$ | 2.972 | 0.118 | ${ }^{1.288}$ | 116 | 346 | 190 | 0.0312 | $7.2)$ |
|  | v | 2.4908 | 0.5790 | ${ }^{-0.2273}$ | 0.0259 | 0 | 0.0311 | 3.291 | 0.080 | ${ }^{7.765}$ | 145 | 1015 | 243 | 0.0348 | 8.0 |
|  | v | ${ }^{2.7598}$ | 0.5284 | $\xrightarrow{-0.2306}$ | $\bigcirc$ | 0 | 6.251 0.030 | 10.010 0.733 1.70 | 0 | ${ }^{4.950}$ | ${ }_{151}^{108}$ | ${ }^{8699}$ | ${ }_{65}^{81}$ | - 0.0901 | ${ }_{24.2}^{20.9}$ |
|  | vI | $\left\{\begin{array}{l} 2.8830 \\ 2.8134 \end{array}\right.$ | $\begin{aligned} & 0.8142 \\ & 0.6037 \end{aligned}$ | $\underbrace{-0.1442}_{-0.2334}$ | $\begin{aligned} & 0.1089 \\ & 0.0535 \end{aligned}$ | $\begin{aligned} & 0 \\ & 0 \end{aligned}$ | $\begin{aligned} & 0.030 \\ & 0.233 \end{aligned}$ | $\begin{aligned} & 0.233 \\ & 7.950 \end{aligned}$ | $\begin{aligned} & 0.042 \\ & 0.080 \end{aligned}$ | $\begin{aligned} & 1.884 \\ & 3.100 \end{aligned}$ | 151 93.4 | $\begin{aligned} & 1015 \\ & 1015 \end{aligned}$ | ${ }_{85}^{65}$ | $\begin{aligned} & 0.2039 \\ & 0.0962 \end{aligned}$ | 24.2 22.3 |
| Hy $1 / 4$ | II | 2.5389 | 0.1950 | 0 | 0 | $-0.021$ | 0.01973 0.01973 |  | 0.070 | ${ }^{0.636}$ | 188 | 309 | 10 | 0. 0206 | 4.8 |
|  | ${ }^{\text {II }}$ | 2.7475 | 0.4057 | 0 | 0 |  |  |  | 0.280 | 0.631 | 328 | ${ }^{458}$ | 10 | ${ }^{0.0096}$ | 2.2 |
|  | rv | 2.8478 | 0.3330 | - | - | -0.021 | 0.01973 |  | 0.102 | 0.512 | 262 | 534 | 9 | 0.0107 | 2.5 |
| A $1 / \%$ | ${ }^{\text {II }}$ | 2.4119 | 0.3794 | 0 | 0 | 0 | 0.04734 |  | 0.159 | 0.802 | 128 | 233 | ${ }^{9}$ | ${ }^{0.0083}$ | 1.9 |
|  | III | 2.5628 | 0.4272 | 0 | 0 | 0 | 0.04734 |  | 0.139 | 0.868 | 150 | 342 | 14 | 0.0123 | ${ }^{2.8}$ |
|  | vv | 2.6897 | ${ }^{0.3934}$ | ${ }^{\circ}$ | 0 | ${ }^{-0.021}$ |  |  | ${ }^{0.133}$ | 0.435 0.539 | 201 | 338 | 10 | 0.0073 | 1.7 |
|  | $v$ | 3.2298 | 0.5945 | 0 | 0 | , | 0.04734 |  | 0.036 | 0.539 | 149 | 807 | 17 | 0.0526 | 12. |
| $\mathrm{Al} 1 / 6$ | II | 2.4453 | 0.4112 | 0 | 0 | 0 | 0.1537 |  | 0.242 | 2.098 | 123 | 311 | 10 | 0.0155 | 3.6 |
|  | iur | 2.4732 | 0.4350 | 0 | 0 | $\bigcirc$ | ${ }^{0.1537}$ |  | 0.143 | 0.948 | 126 | ${ }^{234}$ | 10 | ${ }^{0.0105}$ | 2.4 |
|  | ıv | 2.6008 | 0.4254 | 0 | 0 | -0.031 | 0.15370.1537 |  | 0.130 | ${ }^{0.500}$ | 149 | ${ }^{236}$ | 10 | ${ }^{0.0091}$ | 2.1 |
|  | v | 2,3873 | 0.5717 | 0 | 0 | 0 |  |  | 0.072 | 0.992 | 152 | ${ }^{\text {782 }}$ | 12 | 0.0525 | 12.1 |
| A $11 / 2$ | 1 | 2.2689 | 0.4326 | ${ }^{0}$ | 0 | $\bigcirc$ | 0.3767 |  | 0.282 | 1.963 | 102 | 248 | 10 | ${ }^{0.0120}$ | ${ }^{2.8}$ |
|  | III | 2,3962 | 0.4888 | 0 | 0 | 0 | 0. 0.3757 |  |  | 1.321 | 123 | 293 |  | ${ }^{0.0056}$ | 1.3 |
|  | sv | 2.5408 | 0.4472 | 0 | 0 | $-0.042$ |  |  | 0.192 | 0.760 | 151 | 299 | 9 | 0.0092 | 2.2 |
|  | v | 2.3098 | 0.5814 | 0 | 0 | 0 | 0.3767 |  | 0.084 | 1.993 | 145 | 907 | 11 | 0.0299 | 6.8 |
| A1\% | 1 | 2.1583 | 0.5019 | 0 | 0 | 0 | 1.2868 |  | 0.303 | 1.741 | ${ }^{88.9}$ | 192 | 10 | 0.0026 | 0.6 |
|  | in | 2.3082 | 0.5342 | 0 | 0 | 0 |  |  | 0.215 | 1.299 | 86.5 | 227 | 10 | 0.0092 | 2.1 |
|  | rv | 2.4623 | 0.4951 | 0 | 0 | -0.062 | $\begin{aligned} & 1.2682 \\ & 1.2662 \end{aligned}$ |  | 0.353 | 0.993 | 153 | 276 | 10 | 0.0092 | 2.2 |
| Al $1 / 2$ | 1 | 2.0707 | 0.5115 | 0 | 0 | - | 2.94412.944 |  | 0.416 | 1.574 | 15.6 | 150 | 10 | 0.0090 | 2.1 |
|  | i11 | 2.2515 | 0.6286 | 0 | 0 | - |  |  | 0.237 | 1.452 | 70.7 | 224 | 10 | 0.0077 | 1,8 |
| A $1 / 4$ | II | 1.9305 | ${ }^{0.8109}$ | 0 | 0 | 0 | 10.17210.172 |  | 0.393 | ${ }_{1}^{1.735}$ | 47.8 | 117 | 10 | ${ }^{0.0234}$ | 6.1 |
|  | ${ }^{11}$ | 2.1347 | 0.8587 | 0 | 0 | 0 |  |  | 0.307 | 1.195 | 62.3 | 153 | 10 | 0.0038 | 0.9 |
| A 1 \% $/ 11$ | 11 | 1.8827 | 0.6125 | 0 | 0 | 0 | 19.828 |  | 0.677 | 1.900 | 59.8 | 110 | 10 | 0.0125 | 2.9 |
|  | 11 | 2,1116 | 0.6995 | 0 | 0 | 0 | 19.828 |  | 0.285 | 1.283 | 46.8 | 138 | 10 | 0.0066 | 1.5 |
| St $1 / 1$ | 11 | 2.2178 | 0.349 | 0 | - | -0.021 | 0.1308 |  | 0.091 | 0.852 | 82.7 | 157 | 9 | 0.0153 | 3.2 |
|  | II | 2.3687 | 0.4530 | 0 | 0 | 0 | 0.13080.13080.1208 |  | 0.134 | 0.554 | 95.8 | 181 | 10 | . 0.0112 | ${ }^{2.6}$ |
|  | v | 2.4838 | 0.4271 | 0 | 0 | $-0.021$ | 0.1306 |  | 0.094 | 0.382 | 96.7 | 190 | 10 | ${ }^{0.0071}$ | 1.7 |
|  | v | 2.8971 | 0.5195 | 0 | 0 | 0 | 0.1308 |  | 0.042 | 0.995 | 90.6 | 505 | 17 | 0.0514 | 10.7 |
| St $1 / 4$ | 11 | 2.0408 | 0.4317 | 0 | 0 | -0.042 | 1.0494 |  | 0.172 | 1.281 | 48.1 | 121 | 10 | 9.0029 | 0.8 |
|  | mi | 2.2075 | 0.5219 | 0 | 0 | , |  |  | 0.252 | 0.839 | 78.3 | 146 | 10 | 0.0072 | 1.6 |
|  | v | 2,3592 | 0.4703 | 0 | 0 | $\rightarrow 0.042$ | 1.04341.0434 |  | 0.183 | 0.622 | 91.6 | 179 | 10 | 0.0079 | 2.0 |
|  | $v$ | 2.5231 | 0.5036 | 0 | 0 | 0 |  |  | 0.138 | 2.257 | 114 | 439 | ${ }^{21}$ | 0.0562 | 11.8 |
| B1\% | rv | 686 | 0.5226 | 0 | 0 | $\bigcirc$ | 3.5211 |  | 0.387 | 4.580 | 109 | 412 | 11 | 0.0060 | 1.4 |
| s. $1 / 14$ | $\mathrm{H}^{(6)}$ | 1,9471 | 0.4771 | 0 | 0 | (3) | 5.5971 |  | 0.175 | 0.427 | 29.6 | 59.3 | 24 | 0.0663 | 1.4 |
|  | $\mathrm{mi}^{(8)}$ | 2.0610 | 0.5991 | 0 | 0 | 0 | 5.59715.5971 |  | 0.104 | 1.515 | 29.8 | 145 | 32 | ${ }^{0.0061}$ | 1.4 |
|  | iv | 2.2206 | 0.5027 | 0 | 0 | 0 |  |  | 0.419 | 4.658 | 108 | 360 | 12 | 0.0039 | 0.9 |
| st $1 / 2$ | $110^{(6)}$ | 1.8639 | 0.3896 | 0 | 0 | ${ }^{(9)}$ | 9.353 |  | 0.180 | 0.522 | 29.7 | 53.5 | 24 | 0.0037 | 0.9 |
|  | 11 | 2.0414 | 0.5188 | 0 | 0 | (9) | ${ }_{8}^{8.353}$ |  | 0.192 | 1.655 | 35.7 | 139 | 10 | ${ }^{0.0032}$ | 0.7 |
|  | rv | 2.1784 | 0.5351 | 0 | 0 | 0 | 8.353 <br> 8.353 |  | 0.494 | 4.232 | 101 | 324 | 12 | 0.0071 | 1.6 |
|  | $v$ | 2.4118 | 0.5095 | 0 | 0 | 0 |  |  | 0.178 | 2.975 | 94.3 | 44 | ${ }^{21}$ | 0.059a | 13.6 |
| 85 \% | 1 | 1.8577 | 0.5367 | 0 | 0 | (1) | ${ }^{11.874}$ |  | 0.194 | 0.5732 | 29.7 | 53.8 | 24 | 0.0059 | 1.4 |
|  | III | 2.0192 | 0.5154 | 0 | 0 | (10) |  |  | 0.245 | 1.447 | 39.2 | 137 | 10 | 0.0037 | 0.7 |
|  | v | 2.1344 | 0.5518 | 0 | 0 | 0 | 11.87411.874 |  | 0.516 | 4.412 | 93.7 | 306 | 12 | 0.0067 | 1.5 |
| 8t $1 /$ | s | 2.086 | 0.5510 | 0 | 0 | 0 | .110 |  | 0.767 | 5.201 | 103 | 298 | 15 | 0.0160 | 3.7 |
| 9t1/4 | sy | 2.0256 | 0,5447 | 0 | 0 | 0 | 54.9554.95 |  | 0.971 | 4.311 | 99.4 | 248 | 12 | 0.0125 | 2.9 |
|  | v | 2.2848 | 0.4842 | 0 | 0 | 0 |  |  | 0.142 | 3.375 | 70.4 | 326 | 8 | 0.0355 | 8.8 |
| Ge. al. ox | 11 | 2.1878 | 0.5235 | -0.1861 | 0.0970 | (11) | $0.030 \quad 0.300$ |  | 0.181 | 2.048 | 118 | 294 | ${ }^{4}$ | 0.0145 | 3.4 |
|  | in | 2.3171 | 0.5835 | -0.1842 | 0.0950 | 0 | 0.0978 | 0.249 | 0.110 | 1.455 | 114 | 381 | 60 | 0.0185 | 4.3 |
|  | Iv | 2.4540 | 0.5398 | -0,1834 | 0.720 | 0 | 0.0378 | 0.248 | 0.051 | 1.141 | 139 | 429 | 58 | 0.0109 | 2.5 |
| Ge | v | 2.9987 | 0.6019 | 0 | 0 | $\bigcirc$ | 0.0416 | 0.0439 | 0.060 | 0.577 | 159 | 765 | 10 | 0.042 | 10.2 |
| ${ }^{\text {cl }}$ | $v$ | 2.9280 | 0.5191 | 0 | 0 | 0 | 0.0691 | 0.0727 | 0.040 | 1.012 | 153 | 807 | 10 | 0.0385 | 8.4 |
| cs | $\mathrm{T}^{(6)}$ | 1.8915 | 0.4448 | 0 | 0 | 0 | 358.0 | 357.5 | 0.124 | 0.427 | 30.0 | 48 | 16 | 0.0361 | 8.3 |
|  <br> ${ }^{(2)}$ Tigure 2.4 is a plot of theece equations. <br> ${ }^{(3)} \log \mathrm{v}=2.3472+0.00045(\mathrm{~m} / \mathrm{A})-0.002244(\mathrm{~m} / \mathrm{A})^{2}+1-0.01756+0.00009(\mathrm{~m} / \mathrm{A})-0.000439(\mathrm{~m} / \mathrm{A})^{2} 1 \log \mathrm{~A}+0.5 \log (\mathrm{~s}+\mathrm{k})$, where $\log \mathrm{k}=-0.7089+$ $0.3502 \log A ; A$, mquare inctes: $m$, gramin; and $4.958 \leq(\mathrm{m} / \mathrm{A}) \leq 9.802$ (eee Fig. 2.5 ). <br> ${ }^{(4)}$ I. G. Bowen, A. F. Strehler, and M, B. Wetberbe, Diotribution and Denelty of Misalles from Nuclean Eaplosions, Operation Tempot Report, wT -I1A8, |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| Decamber 1958, p. 21. <br> ${ }^{(5)} \log y=\log [1000 /(1.8984 \log m+4.4704)]+\left[0.550 .5+0.0388 \log m+0.00882(\log m)^{2}\right.$ (6)Celibrition data obthined by drop method. <br>  <br> ( ${ }^{(1)}$ Combined data from air-gun and drop method. <br> (9) For $\mathrm{E}=0.250, \mathrm{k}=-0.083$; for $=\times 0.250,(\mathrm{~s}+\mathrm{k})=5.333 \mathrm{~s}^{2}(0.750-\mathrm{s})$. <br>  <br> (1110.02513 (m) ${ }^{-1 / 2}$. |  |  |  |  |  |  |  |  |  |  |  |  |  |  |  |

sorber types II, III, IV, and V. Note that the nylon spheres, because of their lesser density, require considerably higher velocities to penetrate the depth of one radius than do aluminum spheres of the same size.

Threshold velocities shown in Fig. 2.10 are for steel spheres with diameters from $1 / 8$ to $9 / 16$ in. for absorber types II and III and from $1 / 8$ to $15 / 16 \mathrm{in}$. for absorber types IV and V. It is interesting to note that, for the more dense absorbers (types IV and V), the threshold velocities are about the same for the small as for the large spheres. For the two less-dense absorbers (types II and III), however, threshold velocities decrease with sphere diameter up to about $1 / 2$ in. The data for the $9 / 16$-in.-diameter sphere suggest that larger spheres would have higher threshold velocities.

TABLE 2.3- THRESHOLD VELOCITIES FOR SPHERES IN TYPES II, III, IV, AND V ABSORBERS

| Spheres | Velocity, ft/sec |  |  |  | Mass, mg |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  | Type II | Type III | Type IV | Type V |  |
| Ny $1 / 8$ | 186 | 181 | 202 |  | 19.7 |
| Al $1 / 8$ | 90 | 112 | 134 | 206 | 47.3 |
| Al $3 / 16$ | 83.5 | 105 | 123 | 199 | 154 |
| Al $1 / 4$ | 75.5 | 90.1 | 114 | 193 | 377 |
| Al $3 / 8$ | 62.3 | 83.1 | 104 |  | 1,266 |
| Al $1 / 2$ | 58.1 | 74.7 |  |  | 2,944 |
| A1 $3 / 4$ | 46.1 | 70.5 |  |  | 10,172 |
| Al $15 / 16$ | 48.0 | 76.2 |  |  | 19,828 |
| St $1 / 8$ | 54.3 | 66.6 | 78.3 | 118 | 131 |
| St $1 / 4$ | 37.3 | 54.5 | 77.9 | 117 | 1,043 |
| St $3 / 8$ |  |  | 77.4 |  | 3,532 |
| St $7 / 16$ | 35.3 | 46.3 | 77.4 |  | 5,597 |
| St $1 / 2$ | 36.4 | 43.5 | 71.8 | 127 | 8,353 |
| St $9 / 16$ | 41.0 | 44.1 | 67.7 |  | 11,870 |
| St $3 / 4$ |  |  | 71.0 |  | 28,161 |
| $\mathrm{St}^{15} / 16$ |  |  | 70.2 | 134 | 54,950 |
| GS | 117 | 113 | 150 |  | 40.0 |
| Gs |  |  |  | 182 | 42.7 |
| Gl | 102 | 103 | 139 |  | 72.6 |
| Gl |  |  |  | 222 | 70.9 |
| Gx | 80 | 89 | 125 |  | 242.4 |

Threshold velocities for soda-glass beads with an average density of $2.55 \mathrm{~g} / \mathrm{cm}^{3}$ are plotted in Fig. 2.11. Consistent with the calibration equations (see Table 2.2), threshold velocities are shown as functions of sphere mass for the plastic absorbers (types II, III, and IV) and for two sphere-mass values for the balsa absorber (type V) (see points labeled 'Small Spheres" and "Large Spheres" in Fig. 2.11).

For the convenience of the reader, the sphere threshold velocities that are presented graphically in Figs. 2.9 to 2.11 are listed in Table 2.3. The nomenclature used in the first column to describe the spheres is given in the List of Symbols, pages 7 to 9 .

Although the assumptions made in computing threshold velocities were somewhat arbitrary, the results showed a reasonable agreement with the field data. Very few missile velocities were evaluated which were below the computed threshold; however, this does not mean that every missile that struck the trap with above-threshold velocities was retained in the absorber. (Also, some of the missiles that were not firmly imbedded in the absorber were dislodged during transport of the traps from the field to the laboratory.) Actually, a definite threshold velocity cannot be established for any missile. A more realistic concept is that of a band, or range, of threshold velocities as a function of missile mass, such as is portrayed for glass fragments in Fig. 2.6.

In spite of the limitations noted above, the computed threshold velocities proved to be quite useful in the interpretation of the field data. For example, if the mean of measured velocities was near the threshold, it could be assumed that the sample was truncated at the lower end, and therefore the computed mean was too high. Other discrepancies may result when the actual missile velocities are lower than the threshold value or values. This situation could result in a few missiles being caught because of their shapes and orientations at impact, e.g., a sliver of glass impacting on a sharp point. This again would result in the mean of the measured velocities being too high since the calibration equations were obtained for missiles of random shapes and orientations at impact. Also to be considered is the circumstance where the velocities measured for a sample of uniform missiles are above the threshold yalue but the expected velocity (based on blast-wave parameters) is below the threshold. This, along with collaborating evidence, would lead one to suspect that the absorber had been softened by thermal radiation before the time of impact or that the missile itself was hot.

### 2.6 STATISTICAL ANALYSIS OF FIELD DATA

In the computation of statistical parameters describing the velocities and masses of nonspherical missiles from a given sample (trap or group of traps), it was assumed that the distributions were log normal. A graphical verification is presented in Sec. 6.2.6 of the normalcy of distributions of log mass and log velocity by making use of data for 2523 glass fragments that were trapped in two houses.

Another type of test was developed (see the Appendix) by establishing the following theoretical relation between the ordinary mean of a log-normal distribution and its geometric mean and standard geometric deviation:

$$
\begin{equation*}
\frac{\bar{x}}{x_{50}}=\exp \left[\frac{\left(\ln _{e} S_{g x}\right)^{2}}{2}\right] \tag{2.11}
\end{equation*}
$$

where $\quad \bar{x}=(\Sigma x) / n \quad$ (ordinary mean of variable $x$ )
$\mathrm{x}_{50}=$ antilog $[(\Sigma \log \mathrm{x}) / \mathrm{n}] \quad$ (geometric mean)
$S_{g x}=$ antilog $\sqrt{\left[\Sigma\left(\log x-\log x_{50}\right)^{2}\right] /(n-1)} \quad$ (standard geometric deviation)
$n=$ number of $x$ values in the sample
The relation between $\bar{x} / x_{50}$ and $S_{g x}$, expressed by Eq. 2.11, is plotted as a solid line in Fig. 2.12. Note that, as the dispersion of the distribution (indicated by $\mathrm{S}_{\mathrm{gx}}$ ) increases, the magnitude of the mean also increases relative to the geometric mean.*

The points plotted on the chart in Fig. 2.12 represent velocity and mass parameters that were obtained from 111 missile samples (presented in detail later in the report). Note that the missile-velocity points (in the lower-left portion of the chart) are uniformly scattered about the theoretical line, indicating general agreement with the log-normal assumption. The missile-mass points, however, have a slight tendency to fall more to the right than to the left of the theoretical line. This means that, in general, the samples contained too few small missiles to satisfy the log-normal assumption. The scarcity of missiles of low masses could have been due to one or more of the following:

1. Some of the smaller missiles, because of their size, may have been overlooked in the absorber at the time the missiles were extracted.
2. Limitations in the calibration procedure prohibited use of missiles that were extremely small. $\dagger$

* The geometric mean and the median are identical for a log-normal distribution.
$\dagger$ Masses of the missiles used for the calibrations are listed in Table 2.2. Actually, the calibration equations were used to evaluate velocities for missiles somewhat smaller than those used in the calibrations; e.g., the smallest missiles used to calibrate type II absorber for glass fragments weighed 0.0274 g , but velocities were evaluated for fragments as small as 0.010 g .

3. The gravel used had been screened to remove both the small and the large stones, and this screening had resulted in truncated samples.

It is appropriate to discuss briefly the significance of the statistical parameters that were defined in Eq. 2.11. Consider, for example, the distribution of missiles according to mass, where $\overline{\mathrm{M}}$ is the mean, $\mathrm{M}_{50}$ is the geometric mean, and $\mathrm{S}_{\mathrm{gm}}$ is the standard geometric deviation. It can be shown that 84.13 per cent of the missiles from a given $\log$-normal sample have velocities less than $\mathrm{M}_{50} \times \mathrm{S}_{\mathrm{gm}}$ and that 15.87 per cent have velocities less than $\mathrm{M}_{50} / \mathrm{S}_{\mathrm{gm}}$. Thus 68.26 per cent of the missiles have masses greater than $M_{50} / S_{g m}$ and less than $M_{50} \times S_{g m}$. In some instances it is of interest to know the total mass of a sample of $n$ missiles where only the geometric mean and the geometric standard deviation are known. An estimate of the total mass can be obtained by using Eq. 2.11 to obtain the mean mass and then multiplying this quantity by $n$.

In general, the impact velocities measured for missiles of a given sample were not independent of their masses. It was found that the following relation satisfactorily expressed the dependence of impact velocity on missile mass:

$$
\begin{equation*}
\log v=a+b \log m \tag{2.12}
\end{equation*}
$$

where $v$ is impact velocity, $m$ is missile mass, and $a$ and $b$ are regression coefficients.
Note that the log-normal distributions discussed above are recognized in Eq. 2.12 by the use of $\log v$ and $\log m$ as variables instead of $v$ and $m$. The coefficients a and $b$ were determined by the least-squares method for each missile sample with the substitution $y=\log v$ and $\mathrm{x}=\log \mathrm{m}$. The geometric standard error of estimate, $\mathrm{E}_{\mathrm{gv}}$, was also determined for each sample, considering $\log v$ to be the dependent variable. The significance of $\mathrm{E}_{\mathrm{gv}}$ is the same as that of $\mathrm{S}_{\mathrm{gv}}$, except that the reference for $\mathrm{E}_{\mathrm{gv}}$ is the "geometric mean" velocity as a function of mass found from Eq. 2.12 instead of simply the geometric mean of the sample. Thus, if the regression velocity is given by antilog $(a+b \log m)$, then 84.13 per cent of the missiles from a $\log$-normal distribution would have velocities less than [antilog $(a+b \log m)] \mathrm{E}_{\mathrm{gv}}, 15.87$ per cent would have velocities less than [antilog $(a+b \log m)] / E_{g v}$, and 68.26 per cent would have velocities between the two limits. In general, $\mathrm{E}_{\mathrm{gv}}$ for a given missile sample is less than $\mathrm{S}_{\mathrm{gv}}$. However, if missile velocities are independent of their masses, then $\mathrm{E}_{\mathrm{gv}}$ has approximately the same value as $\mathrm{S}_{\mathrm{gv}}$, and Eq. 2.12 expresses the geometric mean velocity ( $\mathrm{V}_{50}$ ) for all values of mass.

The equation used to compute $\mathrm{E}_{\mathrm{gv}}$ is

$$
E_{g v}=\operatorname{antilog} \sqrt{\left[\sum_{i=1}^{n}\left(a+b \log m_{i}-\log v_{i}\right)^{2}\right] /(n-2)}
$$

where $m_{i}$ and $v_{i}$ are paired values of mass and velocity and $a$ and $b$ are regression coefficients.

## REFERENCES

1. I. G. Bowen, A. F. Strehler, and M. B. Wetherbe, Distribution and Density of Missiles from Nuclear Explosions, Operation Teapot Report, WT-1168, Dec. 14, 1956.
2. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determinations of Aerody-namic-drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.


SECTION A


Fig. 2.1-Construction details of trap housing.

9


Fig. 2.2-Photograph showing trap anchors, aluminum foil for thermal protection, and added thermal shield 1 ft in front of the trap.
 plate glass was 0.25 in . thick.


Fig. 2.4-Plot of calibration equations for glass missiles in type II absorber


Fig. 2.5-Nomogram for determining velocities of missiles striking type II absorber flat. Diagram in lower-left corner indicates steps necessary.


Fig. 2.6 - Threshold velocities for glass fragments. Roman numerals designate type of absorber.
出


Fig. 2.7 - Threshold velocities for window and plate glass striking type II absorber flat (assuming $\mathrm{s}=0.05 \mathrm{in}$.) .


Fig. 2.8-Threshold velocities for natural stones and gravel. Roman numerals designate type of absorber.


Fig. 2.9-Threshold velocities for nylon and aluminum spheres. Roman numerals designate type of absorber.
*


Fig. 2.10-Threshold velocities for steel spheres. Roman numerals designate type of absorber.


Fig. 2.11-Threshold velocities for soda-glass spheres. Roman numerals designate type of absorber.


Fig. 2.12-Relation between the ratio of the mean to the geometric mean and the standard geometric deviation ( $\bar{x} / \mathrm{x}_{50} \mathrm{vs} . \mathrm{S}_{\mathrm{gx}}$ ) for $\log -$ normal distributions. Data points are for missile samples which are presented later in the report. Points marked by trap numbers are those with large deviations from the theoretical line.

### 3.1 GENERAL

One of the more important objectives (see Sec. 1.3) of the secondary-missile study was to compare the velocities measured for various secondary missiles with those which could be computed (or predicted) by use of appropriate values of the blast-wave and missile parameters. Two auxiliary studies had to be carried out before this objective could be reached. They are reported elsewhere. ${ }^{1,2}$ The first of these involved the solution of a mathematical model designed to simulate the salient phenomena of missile production by ideal or classical blast waves. The second was concerned with the measurement of appropriate aerodynamic parameters for irregular objects such as those used in the field operation. Through use of the blastwave data measured by the Ballistics Research Laboratories, ${ }^{3}$ the computations were made specific for field situations.

This chapter describes briefly the work previously reported and discusses its application to the present study.

### 3.2 PREDICTION OF MISSILE VELOCITIES

For the sake of simplicity, it was assumed that the only force acting on the missile was due to the difference in the missile and wind velocities. The field experience indicated that objects being translated by blast winds tend to be lofted; thus the effects of surface (or ground) friction are minimized. The lofting effect, however, would be dependent on the strength and nature of the blast winds as well as on the physical characteristics of the displaced object.

The blast wave was assumed to be the ideal, or classical, type, unaffected by precursor or hill-and-dale effects. Winds and dynamic pressures associated with the ideal wave of given shock strength and duration were evaluated by use of the relations derived from numerical studies made by H. L. Brode of Rand Corporation.

No allowance was made in the secondary-missile model for the decay of the blast wave during the time (or distance) required for the missile to reach maximum velocity. This simplification would be justified at large ranges from GZ where both distance of missile travel and the decay rate of blast wave are small. At the smaller ranges, however, the blast wave experiences more significant attenuation over the distance required to accelerate a missile to maximum velocity. This effect could not be evaluated from the field experience since the blast waves at the shorter ranges were significantly modified by precursor effects.

The analytical procedure used in the missile model identified a missile by one parameter the acceleration coefficient $(\alpha)$, defined as the product of the area presented to the wind and the drag coefficient divided by the mass ( $\alpha=A C_{D} / \mathrm{m}$ ) and assumed to be constant for a given mis sile. Two objects of vastly different shapes, sizes, and weights could have the same accelera-
tion coefficient and thus experience similar velocity vs. time histories when exposed to any particular blast wave. Use was made of this concept to investigate the displacement velocities for man by trapping objects smaller than man but possessing approximately equivalent acceleration coefficients, namely, $7 / 16^{-}, 1 / 2^{-}$, and $9 / 16$-in.-diameter steel spheres (see Ref. 1).

### 3.3 DETERMINATION OF THE IDEAL BLAST WAVE FROM THE FIELD DATA

Overpressure and dynamic pressure were measured as functions of time at most of the missile stations by Ballistic Research Laboratories (BRL) mechanical type gauges. ${ }^{3}$ Since the velocity-prediction model was solved for the ideal blast wave, it was desirable to determine the equivalent ideal wave for each of the measured blast waves. This was done in the case of the overpressure pulse by finding the ideal wave with the same impulse and duration as those measured by the gauges. The overpressures of the ideal wave as a function of time were then evaluated* and plotted for comparison on the graph showing the measured values of overpressure as a function of time.

Dynamic pressure as a function of time was determined for the ideal wave by making use of the maximum overpressure of the ideal wave and the measured duration of the positive over pressure. The relation between the ratio of durations of the positive dynamic pressure and the positive overpressure as a function of maximum overpressure is set forth in Sec. 2.3.4 of Ref. 1.

Section 2.3.2 of Ref. 1 describes the expression used for dynamic pressure vs. time for blast waves specified by maximum overpressure and duration.

### 3.4 ACCELERATION COEFFICIENTS FOR SMALL NONSPHERICAL MISSILES

Acceleration coefficients, defined in Sec. 3.2, could be determined for spheres of known presented area and mass by use of a drag coefficient of $0.47 . \dagger$ Acceleration coefficients for irregular objects such as stones and glass fragments were not so readily determined. Experiments were performed in which the test objects were dropped a known distance (about 48 ft ) in a measured time. Acceleration coefficients could then be determined by comparing the measured drop times with the time required for the object to fall the same distance without air drag. ${ }^{2}$ It should be pointed out that in these experiments the velocities encountered were relatively low and the compressibility effects of the air were small.

### 3.5 GLASS-FRAGMENT STUDIES

The drop-test studies reported in Ref. 2 indicated that orientation of the missile with respect to the wind was not important in determining acceleration coefficients for double-strength window fragments with masses less than 0.220 g and for plate-glass fragments with masses less than 0.860 g . As the fragment masses increased from these lower limits, their orientation became more important; e.g., $2-\mathrm{g}$ window-glass fragments have acceleration coefficients for the edgewise orientation which are about 40 per cent lower than those obtained when the maximum areas are presented to the wind. The scatter in the velocity data obtained for a typical window -glass sample was too large to be explained by the orientation effect (see Fig. 6.19).

Velocities predicted for glass fragments on the basis of a free-field blast wave ignored any possible modification of the wave by the window installations in open areas or by the structure containing the window in the case of the house installations. In some instances, particularly for the houses, the modification noted (as signified by missile velocities) was great enough to suggest that velocities also be computed for a blast wave with a duration the same as that for the free-field wave and with a maximum overpressure equal to the reflected overpressure assuming normal incidence of the free-field blast wave. Although this procedure

[^5]cannot be rigorously defended by theory, its usefulness as an empirical guide in the prediction of missile velocities is apparent, provided, of course, that it conforms with the experimental evidence available.

### 3.6 NATURAL-STONE, GRAVEL, MILITARY-DEBRIS, AND SPHERE STUDIES

The point of origin and the distance of travel of the natural (or native) stones that were caught in the traps were unknown. Predicted velocities were computed by making the assumption that the displacement of the missile before striking the trap was that distance required by each missile to reach maximum velocity. Thus natural stones displaced distances other than the optimum would have velocities lower than the predicted values.

At the missile stations in open areas on shots Priscilla and Galileo, screened gravel, which had been dipped in paint for identification, was placed in front of traps at two or three distances. The greatest distance used at each station* was estimated to be that which would be necessary for a typical stone (about 0.1 g ) to attain 98 per cent of its maximum velocity. The shorter distances were about 39 and about 15 per cent of the greatest distance. This procedure allowed a comparison of predicted and measured velocities for various known distances of travel.

Military debris was marked with paint and placed in the same manner as the gravel (see Chaps. 4 and 5). Spheres of various sizes, some marked with paint or dye, were also placed at the distances used for gravel. The sphere samples were placed at ground level and at various distances above the ground on appropriately designed supports.

## REFERENCES

1. I. G. Bowen, R. W. Albright, E. R. Fletcher, and C. S. White, A Model Designed to Predict the Motion of Objects Translated by Classical Blast Waves, USAEC Report CEX-58.9, June 29, 1961.
2. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determinations of Aero-dynamic-drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.
3. J. J. Meszaros and J. H. Keefer, Blast Measurements for CETG Projects, Operation Plumbbob Report, ITR-1501, June 6, 1958. (Classified).
[^6]
## Chapter 4

## SHOT PRISCILLA, EXPERIMENTAL PROCEDURE AND RESULTS

### 4.1 PROSPECTUS

Before the detonation of shot Priscilla (estimated yield, 38 kt ) in Frenchman Flat, plans were made to investigate the production of secondary missiles at 19 locations (see area map, Fig. 4.1). Eleven of the stations were in open areas at ranges of 6120 to 2030 ft , seven were in closed shelters at ranges of 1360 to 860 ft , and one was in a shelter with open entryway at a range of 900 ft .

The number appearing in the designators for the stations in the open areas indicates the expected value of maximum overpressure; e.g., at $10 \mathrm{P}, 10 \mathrm{psi}$ was the anticipated maximum overpressure. The letter " $P$ " in the designators represents shot Priscilla, and "PP" represents the trap installations associated with a study ${ }^{1}$ of biological damage caused by glass fragments using swine as targets.*

At stations $4 \mathrm{P}, 5 \mathrm{P}, 6 \mathrm{P}$, and 8 P , experiments were designed to study the translation of (1) fragments from windows mounted in open areas; (2) marked gravel and military debris; (3) marked spheres of various types; (4) natural stones; and (5) large stones, blocks, and bricks marked for identification. Similar experiments were conducted at stations $10 \mathrm{P}, 15 \mathrm{P}$, and 20P except that the glass-fragment studies were omitted. Velocities were obtained for all missile types except the large marked stones, blocks, and bricks; the total distance of translation was measured for these missiles.

The experiment inside the open shelter, OPS, was concerned with the translational velocities of "human-equivalent" spheres. $\dagger$ Incidental to this experiment, velocity data were obtained for a number of small stones of unknown source.

The experiments inside the closed shelters were designed to measure the velocity of particles that might spall from the walls of the shelter owing to earth shock. Postshot examinations showed no evidence of significant spalling. $\ddagger$

The material in this chapter is presented by station, starting with the one most remote from GZ. The only exception to this procedure was made for the large-stone study; the displacement data for this study (obtained at seven stations) are discussed in Sec. 4.15. Most of the results, because of their voluminous nature, are presented graphically along with pertinent statistical parameters. For purposes of comparison, predicted or computed missile velocities are shown on the data graphs. Two summary tables - one for the blast-wave parameters (Table 4.5) and the other for statistical parameters (Table 4.6)- describe missile data.
*Glass-fragment data were also collected at stations $4 \mathrm{P}, 5 \mathrm{P}, 6 \mathrm{P}$, and 8 P for the swine study (Project 4.1) and for Project 33.4 which conducted a similar study ${ }^{2}$ but used dogs as targets.
$\dagger$ Spheres of such a size and weight that they acquire approximately the same velocity as would a human being under the same circumstances.
$\ddagger$ The closed shelters were tested by Projects 3.1 and 3.2. Details relevant to the performance of these shelters may be found in Refs. 3 and 4.

### 4.2.1 Experimental Plan

The experimental plan* for station 4 P is illustrated in Fig. 4.2. Three of the five window installations provided for the exposure of animals to glass-fragment missiles. Dogs were used for the study ${ }^{2}$ made by Project 33.4 and pigs were used for the study ${ }^{1}$ made by Project 4.1.

Military debris, mostly steel fragments resulting from explosions, was painted for identification and was placed in front of installations 4 P 4 and 4 P 5 at $4.5,10.9$, and $28 \mathrm{ft}, \dagger$ a different color being used at each location. About 275 pieces of debris varying in mass from 1 to 2220 g were used at each distance.

Gravel was also painted for identification and placed in front of installations 4P4, 4P5, and 4P6 at the same distances as the military debris (see Fig. 4.2). (Note that an area in front of installations 4P6 and 4P7 was stabilized with asphalt to provide a more ideal surface over which gravel and spheres were to be translated.)

Painted spheres were placed in front of installation 4P7 at the same three distances used for military debris and gravel. The smaller spheres were packaged in tissue-paper containers, some placed on the asphalt surface and others suspended above the ground by wire frames (see Fig. 4.12). The heights above ground level, in inches, at which the spheres were placed are recorded in Table 4.6 for the spheres that were caught in traps. The larger steel spheres ( $1 / 2$ and $9 / 16$ in. in diameter) were hung on wire frames and held in aluminum-foil containers that were constructed and mounted in such a way that the blast winds would rip them open and release the spheres.

The sphere samples, described in the following paragraphs, for this station were also used at stations $5 \mathrm{P}, 6 \mathrm{P}, 8 \mathrm{P}, 10 \mathrm{P}, 15 \mathrm{P}$, and 20 P . The distances of placement from the traps varied from station to station, but the samples exposed consisted of the same amounts.

At the shortest distance 10 steel spheres $7 / 16 \mathrm{in}$. in diameter were placed on the asphalt surface and 10 steel spheres $9 / 16 \mathrm{in}$. in diameter were hung from the wire frame.

At the intermediate distance, 10 steel spheres $1 / 2 \mathrm{in}$. in diameter were suspended from the wire frame, but none were placed at ground level.

For each of the three distances, 2110 small spheres were placed at ground level and 1055 were suspended from the wire frame. All samples contained the spheres listed below in the indicated proportions:

| $1 / 8$-in.-diameter nylon ( $\mathrm{Ny} 1 / 8$ ) | 5.2\% |
| :---: | :---: |
| 1/8-in.-diameter aluminum (Al $1 / 8$ ) | 10.4\% |
| $3 / 16$-in.-diameter aluminum ( $\mathrm{Al}^{3} / 16$ ) | 5.2\% |
| $1 / 4$-in.-diameter aluminum ( $\mathrm{Al}^{1 / 4}$ ) | 0.7\% |
| $3 / 8$-in.-diameter aluminum ( $\mathrm{Al}^{3} / 8$ ) | 0.1\% |
| $1 / 8$-in.-diameter steel (St $1 / 8$ ) | 10.4\% |
| 1/4-in.-diameter steel (St $1 / 4$ ) | 4\% |
| 36.0 mg (av.) soda glass (Gs) | 53.5 |
| 72.6 mg (av.) soda glass (G1) | 13.1 |

A summary of the results at station 4 P for window glass, plate glass, natural stones, gravel, and spheres appears in Table 4.6.

Displacement data obtained for the large stones, building blocks, and bricks are presented in Sec. 4.15 and Table 4.4.

### 4.2.2 Blast Parameters

A method was discussed in Sec. 3.3 for obtaining the peak overpressure of an ideal blast wave whose overpressure impulse and duration are the same as those measured in the field. This procedure was found to be necessary in order to arrive at predicted velocities for various missiles by use of a mathematical model ${ }^{5}$ based on the ideal blast wave. The computed as

[^7]well as the measured blast parameters obtained for the various stations are summarized in Table 4.5. Unfortunately, gauge failure prohibited the measurement of overpressure vs. time at stations $4 P$ and $5 P$. Therefore values of overpressure were determined for these stations by means of a regression equation based on the computed overpressures at five stations where records were obtained. (Refer to footnote ** in Table 4.5.) A similar procedure was followed to estimate the duration of the blast wave. Thus the overpressure and duration used to make velocity predictions at station 4P were 4.54 psi and 1.027 sec , respectively. These values were used to compute the dynamic pressure vs. time curve for an ideal wave which is shown in Fig. 4.3 as a dashed line. Illustrated as a solid line in the same chart is the dynamic pressure ( q ) measured by the BRL gauge. The measured q record appears erratic and indicates pressures generally lower than those computed for the ideal wave.

### 4.2.3 Window-glass Installation 4P1

Installation 4P1 consisted of two traps: 4P1b stacked above 4P1a. This installation was placed 7.8 ft behind a window of $1 / 8$-in.-thick double-strength glass. Figure 4.4 is a postshot view of the two traps. Note that the aluminum foil used for thermal protection was ruptured by the glass fragments and torn in some places by blast winds.

The velocity and mass of individual fragments are plotted in Fig. 4.5 for trap 4P1a and in Fig. 4.6 for trap 4P1b. The numbers appearing with some of the points indicate the number of missiles in the velocity and mass intervals represented by those points. The points without numbers represent only one missile. (For a summary of results see Table 4.6.)

Note that for both traps most of the missiles had velocities that were greater than those predicted on the basis of the incident maximum overpressure (lower line of predicted velocities). The prediction line appearing in the upper part of each chart was made for the assumption that the blast wave had a maximum overpressure equal to the reflected (normal) value for the incident wave, * i.e., 10.34 psi instead of the incident maximum overpressure of 4.54 psi .

The slopes of the regression equations describing the data in Figs. 4.5 and 4.6 are -0.0924 and -0.0838 , respectively, whereas the average slopes of the prediction lines are much closer to zero. A partial explanation of this discrepancy is that small fragments require higher impact velocities in order to be retained by the absorber (type II) than do large fragments. This is illustrated by the threshold-velocity chart, Fig. 2.6.

### 4.2.4 Window-glass Trap 4P2b (Above Dog Trap 4P2A)

Installation 4 P 2 was located 12.8 ft behind a window. It consisted of a single missile trap, 4 P 2 b , placed above a dog trap, ${ }^{2} 4 \mathrm{P} 2 \mathrm{~A}$, which was 31.5 in . high. Figure 4.7 is a postshot view of the installation taken after the dog had been removed.

The glass in each outside window installation extended from ground level to a height of 64 in. (see Fig. 2.3). The upper edge of the absorber in the trap at this location was 55 in. above ground level and the lower edge was 33 in . above ground level. Thus the upper edge of the absorber was only 9 in . lower than the top of the window. Unless a lofting effect compensated for the effect of gravity, the spatial density of missiles would be expected to decrease with increasing height above the ground. A comparison of the total missiles caught in trap 4P2b with those caught by other traps at this station is difficult because similar window installations were placed at different distances from the traps. For installation 4P1, placed nearer the window ( 7.8 ft compared with 12.8 ft for installation 4P2), the ground-level trap caught 68 missiles and the one placed 15 in . above the ground caught 58 missiles. Although the number of missiles ( 68 and 58) caught by installation 4P1 traps was greater than that (48) caught by trap 4P2b, their average masses were smaller. It is interesting to note that the trap placed highest above the ground ( $4 \mathrm{P} 2 \mathrm{~b}, 31.5 \mathrm{in}$. above the ground) caught the largest total mass $\dagger$ of glass, 148.8 g com-
*This concept is discussed in more detail in Chap. 3.
$\dagger$ The total mass of missiles caught can be obtained by multiplying the average mass, $\overline{\mathrm{M}}$, by the number of missiles, $n$ (both obtained from Table 4.6).
pared with 144.0 g for trap 4P1a and 64.7 g for trap 4 P 1 b . Figure 4.8 and Table 4.6 present data for missiles recovered from trap 4P2b.

### 4.2.5 Plate-glass Trap 4P3b (Above Dog Trap 4P3A)

### 4.2.6 Military-debris and Gravel Installations 4P4 and 4P5

The placement of military debris and gravel at installations 4P4 and 4P5 was described in the second paragraph of Sec. 4.2.1 and illustrated in Fig. 4.2. Figure 4.10 is a preshot view of installation 4P4 (similar to installation 4P5); piles of gravel and debris are shown.

The postshot condition of both installations is illustrated in Fig. 4.11. The slightly dark areas on the surface of the absorber are thermal effects.

No military debris was caught in any of the four traps. A total of 17 pieces of gravel was recovered: 0 from trap 4 P 4 b , 9 from trap 4 P 5 a , and 4 from trap 4 P 5 b . All gravel caught originated from the 10.9 - and $28.0-\mathrm{ft}$ distances (none from 4.5 ft ). Two to six natural stones (total 14) were caught in each of the four traps. Because the sample sizes were small, the data for both natural stones and gravel were combined with similar data obtained at other traps at station 4P. Analysis of the gravel data is discussed in the next section and that for the natural stones is discussed in Sec. 4.2.10 (see also Table 4.6).

### 4.2.7 Gravel and Sphere Installations 4P6 and 4P7

(a) General. One-third cubic foot of painted gravel was placed at each of three distances (4.5, 10.9, and 28 ft ) in front of installation 4P6 (see Fig. 4.2). Spheres were placed at the same distances in front of installation 4P7. (For description of spheres, see Sec. 4.2.1.) Figure 4.12 is a preshot view of the asphalt area; both the gravel and the spheres are shown. Note that the protective covers for the traps were in place when the photograph was taken.
(b) Traps $4 P 6 a$ and $4 P 6 b$. At installation $4 \mathrm{P} 6,8$ pieces of gravel were recovered from the lower trap and 10 from the upper trap. Only one gravel missile was caught which originated from the pile at the 4.5 -ft distance. For purposes of analysis, the data for these missiles were combined with those obtained from installations 4P4 and 4P5. Velocity vs. mass is plotted in Fig. 4.13 for 14 gravel missiles whose translation distance was 10.9 ft . Similar data are shown in Fig. 4.14 for 20 missiles that traveled 28.0 ft before impact. Both plots indicate that the individual velocities were generally higher than those predicted. Other missiles undoubtedly impacted with the absorber but were not caught because of insufficient velocity or disadvantageous orientation at impact (see Sec. 2.5).

Two natural-stone missiles were caught in trap 4P6b. The data for these missiles were combined for analysis with those for natural stones caught in other station 4 P traps (see Sec. 4.2.10 and Table 4.6).
(c) Traps $4 P 7 a$ and $4 P 7 b$. Results obtained for 15 spheres caught by these traps are presented in Table 4.6. The largest sample obtained consisted of 11 small glass spheres whose average velocity was $135 \mathrm{ft} / \mathrm{sec}-39.2$ per cent higher than the predicted velocity of $97 \mathrm{ft} / \mathrm{sec}$. Deviations from the predicted velocity for the smaller samples were as much as 76.4 per cent higher. These discrepancies probably reflect the inaccuracies inherent in the trapping technique when the depths of penetration are small; i.e., impact velocities were near the threshold for retention of the missile in the trap.

Data for two natural stones caught at this installation, combined with others at this station, are presented in Sec. 4.2.10 and Table 4.6.

### 4.2.8 Window-glass Installation 4P8

This installation was similar to installation 4P1 (Sec. 4.2.3) except that the window was placed 17.8 ft from installation 4 P 8 traps (compared with a $7.8-\mathrm{ft}$ separation for installation 4P1). Figure 4.15 is a postshot view of installation 4P8; fragments of glass imbedded in the absorber are shown.

Velocity vs. mass is plotted in Fig. 4.16 for 41 missiles recovered from trap 4P8a, and a similar analysis is portrayed in Fig. 4.17 for 54 missiles from trap 4P8b (upper trap). Only a small difference is observed between the data obtained at this installation (see Table 4.6) and those obtained from installation 4P1 where the window was considerably nearer the trap. In the instance of the greater translational distance, 25 per cent fewer missiles were caught and their masses were somewhat smaller, but the fragment velocities measured under the two conditions were not significantly different.

### 4.2.9 Window-glass Trap 4P9b (Above Pig Trap 4P9A)

At this installation a pig ${ }^{1}$ was exposed in a box somewhat smaller than that used for dogs (see Fig. 4.18). The missile trap, 4P9b, placed above the pig installation, was 27 in . above ground level.

Data obtained for 62 fragments are plotted in Fig. 4.19. There appears to be little difference between these data and those obtained from other window installations at this station (see Table 4.6) even though the translational distances and the trap heights were different. As in the previous cases, a large portion of the fragments had velocities that were higher than those predicted on the basis of the incident peak overpressure but lower than those predicted for the "reflected" condition (see Sec. 3.5).

### 4.2.10 Natural-stone Data from Station 4P Traps

Velocity and mass data obtained for 18 natural stones caught in six traps* are plotted in Fig. 4.20, and the results are given in Table 4.6. Similar to the gravel trapped at station $4 P$, the velocities tend to be higher than predicted-particularly for the missiles of low mass (see Sec. 2.5).

### 4.3 STATION 4PP (PIG STUDY), 6120-FT RANGE

This station consisted of a double-trap installation inside an enclosure containing 70 pigs. The primary aim of the pig study ${ }^{1}$ (Project 4.1 ) was to determine damaging effects of glassfragment missiles. The 80 -ft-long 13 -ft-wide enclosure was orientated so that a long side faced GZ. The pen was made of 5 - by 5 -in. -mesh hog wire, except for the side toward the approaching blast wave; this side consisted of a 4.2 -ft-high wall of double-strength glass. $\dagger$ Panes of glass 32 in . wide and 20 in . high were mounted in a 2 - by 4 -in.-lumber framework. The trap installation was placed 8.8 ft behind the central section of the glass wall. The pigs were restrained, preshot, in smaller pens made of electric fences. These enclosures were located at the same average distance from the glass wall as the traps. Thus shielding of the traps by the pigs was prevented.

Analyses for 81 missiles caught in the lower trap, 4 PPa , and 68 caught in the upper trap, 4 PPb , are presented graphically in Figs. 4.21 and 4.22 , respectively, and are also given in Table 4.6 with station 4 P window-glass data. A few more missiles were caught in these traps than at window-glass installations at station 4P; however, their masses and velocities were about the same. [Note that stations 4 P and 4 PP had the same range from GZ although they were at different locations (see Fig. 4.1).]
*Note that none of the six traps listed in Fig. 4.20 were behind windows.
$\dagger$ To prevent the pigs from escaping after the arrival of the blast wave, hog wire was also placed 18 in . in front of the glass wall.
4.4 STATION 5P, 5320-FT RANGE

### 4.4.1 Experimental Plan and Blast Parameters

The experimental plan for this station, illustrated in Fig. 4.23, was almost identical to that described in Sec. 4.2.1 for station 4P. A notable difference was that at station 5 P the gravel and sphere installations were placed on opposite edges of an area that was stabilized with concrete; this area was used by another project studying the displacement of anthropomorphic dummies ${ }^{6}$ (see Fig. 4.23). The window installations were the same type as those at station $4 P$, but the gravel, military-debris, and sphere studies differed in that the placement distances were somewhat greater at station 5 P .

Failure of the ground-baffle gauge prohibited the measurement of overpressure vs. time at this station. The methods used to estimate the peak overpressure and duration of the positivepressure phase of the blast wave are discussed in Sec. 4.2.2. For station 5 P the estimated values used to compute predicted missile velocities were 5.51 psi and 0.964 sec , respectively.

The dynamic pressure (q) measured as a function of time is plotted in Fig. 4.24 and, for comparison, the $q$ values associated with an ideal blast wave (dashed line) are also shown. This "ideal" curve represents the $q$ values actually used in the mathematical model ${ }^{5}$ to arrive at predicted values of missile velocity. Lack of consistency in the measured values of dynamic pressure is demonstrated by a comparison of this q record (Fig. 4.24) with the one obtained for station 6P (Fig. 4.44). Even though station 6P was 550 ft nearer GZ , the measured dynamic pressures were generally lower than at station 5 P .

Station 5P summary of results for window glass, plate glass, natural stones, gravel, and spheres is given in Table 4.6, and displacement data for large stones, building blocks, and bricks are given in Table 4.4 (see also Sec. 4.15).

### 4.4.2 Window-glass Installation 5P1

Installation 5P1 was located 7.8 ft behind a window of double-strength glass. (See Fig. 4.25 for postshot view of this installation.) Velocity and mass data for 48 fragments recovered from the lower trap are plotted in Fig. 4.26, and similar data are plotted in Fig. 4.27 for 32 missiles from the upper trap. It is of interest to note that in each chart the geometric mean velocity is approximately equal to the average of the predicted velocities (see also Table 4.6).

### 4.4.3 Window-glass Trap 5P2b (Above Dog Trap 5P2A)

Trap 5P2b, which was anchored above a dog trap ( 31.5 in . high), was located 12.5 ft behind a standard window (see Fig. 4.23). A relatively large number (88) of fragments was recovered; however, the data for trap 5P2b in Fig. 4.28 demonstrate that the velocities measured were lower in relation to the predicted values than was evident at installation 5P1 (see Table 4.6) where the missiles were caught at lower heights above ground level.

### 4.4.4 Plate-glass Trap 5P3b (Above Dog Trap 5P3A)

Trap 5P3b, which was situated above a dog trap, was located 12.8 ft behind a plate-glass window. Figure 4.29 is an enlarged postshot view of the absorber surface. Although this photograph presents evidence that several large fragments struck the trap, the geometric mean mass for the nine missiles that were recovered was only 877 mg . Velocities measured for these fragments (Fig. 4.30) were generally a little higher than predicted (see also Table 4.6).

### 4.4.5 Military-debris and Gravel Installations 5P4, 5P5, and 5P6

Gravel mixed with military debris was placed at three distances in front of installations 5P4 and 5P5. These materials were placed directly on the surface of the dry lake bed (Frenchman Flat). Installation 5P6, however, was located behind the large concreted area, and two of the three piles of gravel for this installation were on the concrete (see Figs. 4.23 and 4.31).

Figure 4.32 is a postshot view of installation 5P4. Note that both the upper and lower traps were slightly damaged by thermal radiation.

TABLE 4.1 - PARTIAL RESULTS FOR SPHERES RECOVERED FROM STATION 5P7
(Samples of less than five not included; complete data in Table 4.6.)

| Spheres* | Distance translated, ft | No. of missiles | Height above ground, in. |  | Velocity, ft/sec |  |  | \% Deviation $\dagger$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Placed | In trap, av. | Threshold | Measured, av. | Predicted | 100\% (M-P)/P | 100\% (P-T)/T |
| G11 (71.8) | 12.5 | 9 | 0 | 16.8 | 102 | 124 | 102 | 21.6 | 0 |
| $\text { St } 1 / 8$ | 12.5 | 6 | 0 | 17.0 | 54.3 | 82 | 67 | 22.4 | 23.4 |
| Gs (36.8) | 12.5 | 18 | 0 | 17.6 | 118 | 138 | 111 | 24.3 | -5.9 |
| AI $3 / 16$ | 12.5 | 5 | 0 | 22.0 | 83.5 | 126 | 82 | 53.7 | -1.8 |
| A1 $3 / 16$ | 32.0 | 6 | 0 | 8.7 | 83.5 | 116 | 105 | 10.2 | 25.7 |
| $\text { Gs }(41.6)$ | 32.0 | 7 | 29 | 11.1 | 117 | 138 | 126 | 9.5 | 7.7 |
| $\overline{G S}(37.0)$ | 32.0 | 33 | 0 | 12.9 | 118 | 141 | 128 | 10.2 | 8.5 |
| G1] (70.8) | 32.0 | 9 | 0 | 14.4 | 102 | 133 | 117 | 13.7 | 14.7 |

*Numbers after Al (aluminum) and St (steel) are diameters (in in.). Gs and Gl represent two sizes of soda-glass spheres whose average masses (in mg) are indicated in parentheses.
$\dagger 100 \%$ ( $\mathrm{M}-\mathrm{P}$ )/P is the per cent deviation of average measured velocity from predicted velocity; $100 \%(\mathrm{P}-\mathrm{T}) / \mathrm{T}$ is the per cent deviation of predicted velocity from threshold velocity

No military debris was trapped at station 5 P. The gravel missiles that were caught were distributed in the following way:

It is apparent from the above tabulation that the upper traps (with "b" suffix) at each installation generally caught more missiles than the corresponding lower ones and that more gravel was trapped from the greater than from the lesser distances of placement. The gravel translated only a short distance before impact probably lacked sufficient velocity to cause the necessary penetration for trapping the missile. The most interesting thing to be noted, however, is that ten times as many gravel missiles were caught at station 5 P 6 , which was behind the concrete area, as were caught at both stations 5P4 and 5P5, even though only twice as much gravel was placed before station 5P6 as in front of stations 5P4 and 5P5 (see Fig. 4.23).

Figures 4.33 and 4.34 represent analyses of the combined gravel data from these traps for translational distances of 12.5 and 32.0 ft , respectively. The gravel translated 32.0 ft was somewhat heavier and had slightly higher velocities than that translated 12.5 ft . Both sets of data are in good agreement with the predicted results (see Table 4.6).

### 4.4.6 Sphere Installation 5P7

Installation 5P7 was located on the right side (looking toward GZ) of the concrete area opposite installation 5P6 (see Fig. 4.23). Spheres were placed in front of installation 5P7 in the same manner as described in Sec. 4.2.1 and illustrated in Fig. 4.12.

It is interesting to note that the thermal radiation incident on this installation apparently increased with height above ground (see Fig. 4.35). This could have been caused by several different effects. However, the most plausible reason that the lower trap received less heating is that it was partially protected by a layer of dust generated close to the concrete surface in front of the installation by action of the thermal pulse itself. The formation of such a dust layer was documented by the motion pictures that were made by Project 33.3 to study the translation of anthropomorphic dummies due to blast winds. ${ }^{6}$ The differential-heating effect observed at this installation was present but to a lesser degree at installation 5P6, which was also behind the concrete slab. At installations 5P4 and 5P5, where there was no stabilization of the native soil, there was no noticeable difference in the thermal effects on the upper and lower traps (see Fig. 4.32).

Data for 120 spheres caught at installation 5P7 are summarized in Table 4.6. For comparison, some of these data were organized in a different fashion and are presented in Table 4.1. It is noteworthy that, of the spheres caught, those originating from 12.5 ft had higher average striking heights than those translated 32.0 ft . An explanation of this is that the missiles translated 12.5 ft had insufficient velocities to penetrate the lower trap (compare in Table 4.1 the threshold velocities with those predicted) but could penetrate the upper one whose absorber surface had been softened temporarily by heating. Thus the velocities determined for the spheres translated 12.5 ft were too high - from 21.6 to 53.7 per cent greater than predicted (see Table 4.1). The fact that the spheres translated 32.0 ft had average velocities only 9.5 to 13.7 per cent greater than predicted may be explained by (1) their average height at impact being lower (i.e., more of them struck the lower trap, which was relatively undamaged by thermal) and (2)
they struck the traps after the spheres translated 12.5 ft , allowing the absorber more time to cool by action of the blast winds, thereby restoring its natural resistance to missile penetration. Attention is called to the last column in Table 4.1 which gives the per cent deviation of the predicted from the threshold velocities. For the spheres translated 12.5 ft , three of the four samples had predicted velocities that were the same as or less than the threshold. The predicted velocities for the spheres translated 32.0 ft , on the other hand, were 7.7 to 25.7 per cent higher than for threshold values.

Data for two natural stones caught in trap 5P7b are recorded in the summary table (Table 4.6).

### 4.4.7 Window-glass Installation 5P8

Installation 5P8 was located 17.8 ft behind a window (see Fig. 4.36 for postshot view of this installation). The amount of protection afforded the trap installation from thermal radiation is apparent by comparing this photograph with the one depicting the sphere traps after the detonation (Fig. 4.35). A factor that enhanced the thermal protection by windows was the color coding of the glass (see Fig. 4.9).

The glass-fragment data obtained from the 5P8 traps (Figs. 4.37 and 4.38 and Table 4.6) are not significantly different from the data obtained from similar $5 P$ installations, even though the distance between trap and window was considerably greater in the present instance.

### 4.4.8 Window-glass Trap 5P9b (Above Pig Trap 5P9A)

Trap 5P9b was placed above a pig trap in a manner similar to that illustrated in Fig. 4.18 for trap 4P9b. The distance from the traps to the window was also the same as for installation 4P9 (12.8 ft).

A comparison of the data obtained at the two installations (Figs. 4.19 and 4.39 and Table 4.6) indicates that the one nearest to GZ (5P9b) collected 16 per cent fewer fragments whose geometric mean mass was 28 per cent smaller but whose mean velocity was 8 per cent higher. In both instances the geometric mean velocities were approximately the same as the predicted ones.

### 4.5 STATION 5PP (PIG STUDY), ${ }^{1}$ 5320-FT RANGE

The experiment at this station was the same as that at station 4PP (Sec. 4.3) except that the distance from the glass wall to the traps was 11.7 ft instead of 8.8 ft and the total length of the wall was 120 ft instead of 80 ft .

The data from these two traps (Figs. 4.40 and 4.41) are fairly representative of those obtained from the window installations at station 5 P, the principal difference being that more fragments were caught at station 5PP and their masses were slightly lower.

Data for three natural stones caught at this station are presented in Table 4.6.

### 4.6 STATION 6P, 4770-FT RANGE

### 4.6.1 Experimental Plan and Blast Parameters

The experimental design for this station was essentially the same as at station 4 P (Sec. 4.2.1) except in the placement of the dogs ${ }^{2}$ by Project 33.4. Instead of locating a dog behind the plate-glass window, one was housed at a separate installation (6P8A) and marked gravel was placed at three distances in front of the installation. This installation is shown in the layout chart (Fig. 4.42) on the right side of the stabilized area.

Figure 4.43 contains a plot of overpressure vs. time measured at this station. Shown on the same chart, as a dashed line, is overpressure vs. time computed for an ideal blast wave whose overpressure impulse and duration are the same as those measured (see Sec. 3.3). Except for small deviations, the measured overpressure curve is in good agreement with the curve for the ideal wave.

The measured dynamic pressures, which are plotted in Fig. 4.44 as a function of time, are somewhat erratic and, in general, are lower than those computed for an ideal wave.

The data for large stones, building blocks, and bricks displaced at station 6 P are presented in Sec. 4.15. Station 6P results for window glass, plate glass, natural stones, gravel, military debris, and spheres are summarized in Table 4.6.

### 4.6.5 Military-debris and Gravel Installations 6P4 and 6P5

The military debris that was placed $5.5,14.0$, and 36.0 ft in front of these installations was similar to that described in Sec. 4.2.1. In addition to the military debris, $1 / 6 \mathrm{cu} \mathrm{ft}$ of marked gravel was placed at each of the three distances (see layout chart, Fig. 4.42).

Both lower and upper traps at these installations were slightly damaged by thermal radiation (see Fig. 4.55 for postshot view of installation 6 P 4 ).

TABLE 4.2 -PARTIAL RESULTS FOR SPHERES RECOVERED FROM STATION 6P6
(Samples of less than five not included; complete data in Table 4.6.)

| Spheres* | Distance translated, ft | $\begin{aligned} & \text { No. of } \\ & \text { missiles } \end{aligned}$ | Height above ground, in. |  | Velocity, ft/sec |  |  | \% Deviation $\dagger$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Placed | In trap, av. | Threshold | Measured, av. | Predicted | 100\% (M-P)/P | 100\% (P-T)/T |
| G1 (72.6) | 5.5 | 5 | 13 | 7.1 | 102 | 147 | 97 | 51.5 | -4.9 |
| Gs (39.0) | 5.5 | 6 | 13 | 9.0 | 117 | 157 | 106 | 48.1 | -9.4 |
| GS (38.4) | 5.5 | 7 | 0 | 10.8 | 117 | 153 | 107 | 43.0 | -8.5 |
| St ${ }^{-1 / 8}$ | 5.5 | 10 | 0 | 18.4 | 54.3 | 91.6 | 62 | 47.7 | 14.2 |
| Gs (39.4) | 14.0 | 18 | 18 | 13.3 | 117 | 158 | 132 | 19.7 | 12.8 |
| $\mathrm{Al}^{-1 / 8}$ | 14.0 | 11 | 0 | 14.8 | 90.0 | 129 | 127 | 1.6 | 41.1 |
| St $1 / 8$ | 14.0 | 7 | 0 | 15.5 | 54.3 | 87.3 | 80 | 9.1 | 47.3 |
| GI (72.9) | 14.0 | 21 | 0 | 17.6 | 102 | 136 | 122 | 11.5 | 19.6 |
| GS (47.0) | 14.0 | 36 | 0 | 18.8 | 113 | 149 | 130 | 14.6 | 15.0 |
| $\mathrm{Al}^{3} / 16$ | 14.0 | 10 | 0 | 22.1 | 83.5 | 128 | 108 | 18.5 | 29.3 |
| St $1 / 8$ | 36.0 | 7 | 0 | 12.6 | 54.3 | 91.6 | 93 | -1.5 | 71.3 |
| $\mathrm{Al}^{1 / 8}$ | 36.0 | 10 | 0 | 12.7 | 90.0 | 137 | 146 | -6.2 | 62.2 |
| $\mathrm{Al}^{3} / 16$ | 36.0 | 6 | 0 | 12.7 | 83.5 | 122 | 125 | -2.4 | 49.7 |
| GS (47.1) | 36.0 | 10 | 30 | 12.8 | 113 | 147 | 149 | -1.3 | 31.9 |
| G1] (71.3) | 36.0 | 12 | 0 | 14.3 | 102 | 135 | 142 | -4.9 | 39.2 |
| GS (37.2) | 36.0 | 43 | 0 | 14.8 | 118 | 156 | 155 | 0.6 | 31.3 |

*Numbers after Al (aluminum) and St (steel) are diameters (in in.). Gs and G1 represent two sizes of soda-glass spheres whose average masses (in mg) are indicated in parentheses.
$\dagger 100 \%(\mathrm{M}-\mathrm{P}) / \mathrm{P}$ is the per cent deviation of average measured velocity from predicted velocity; $100 \%(\mathrm{P}-\mathrm{T}) / \mathrm{T}$ is the per cent deviation of predicted velocity from threshold velocity.

The 55 missiles that were caught by the traps at installations 6 P 4 and 6 P 5 were distributed in the following way:

|  |  |  | Gravel |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: |
| Trap | Natural <br> stones | Military <br> debris | At 5.5 ft | At 14.0 ft | At 36.0 ft |
| 6P4a | 0 | 0 | 0 | 0 | 0 |
| 6P4b | 14 | 0 | 1 | 4 | 0 |
| 6P5a | 4 | 0 | 0 | 1 | 0 |
| 6P5b | 25 | 1 | 0 | 3 | 2 |

The upper traps caught more missiles than the lower ones, a result similar to that at the corresponding 5 P installations. Note that trap 6 P 4 a caught no missiles.

The only missile type caught in sufficient numbers to merit plotting was natural stones at 6P4b (Fig. 4.56) and 6P5b (Fig. 4.57). In both cases the measured velocities were about the same as, or lower than, those predicted.* The data for these natural stones, combined with others at station 6P, are also discussed in Sec. 4.6.9 (see Table 4.6 for results).

The data for gravel caught from the two larger distances were combined with similar data from installation 6P7 and are presented in Sec. 4.6.6.

The one piece of military debris caught in trap 6P5b had a mass of 5.53 g and a measured impact velocity of $74 \mathrm{ft} / \mathrm{sec}$, which is 32 per cent higher than the predicted velocity of $56 \mathrm{ft} / \mathrm{sec}$ (see Table 4.6).

### 4.6.6 Sphere Installation 6P6 and Gravel Installation 6P7

(a) General. These installations, along with trap 6P8A, were located behind an area that was stabilized with asphalt. Figure 4.58 is a preshot view of the area looking away from GZ. The BRL pressure instrumentation can be seen on the right side of the photograph. The packets held by wire frames, as well as those on the surface below the frames, contained an assortment of spheres (see Sec. 4.2 .1 for description). Marked gravel was located at three distances in front of the other two installations, 6P7 and 6P8A. Note that the protective covers had not been removed from the missile traps and that the dog trap was empty at the time the photograph was taken.

Figures 4.59 and 4.60 are postshot views of these installations. Although the lower trap at the sphere installation (Fig. 4.59) appears to have been less affected by thermal radiation than the upper one, the damage incurred was noticeably greater than that for the lower trap at the corresponding sphere installation at station 5P (see Fig. 4.35). The photograph of the gravel traps (Fig. 4.60) shows no apparent difference in the thermal radiation incident on the lower and upper traps.
(b) Data from Installation 6P6. A total of 251 spheres was caught at installation 6P6 (134 by the lower trap and 117 by the upper trap). Complete data for these spheres are listed in Table 4.6; however, for purposes of discussion, certain data were abstracted and presented in Table 4.2 in a form similar to that used for installation 5P spheres (see Table 4.1 and Sec. 4.4.6). Inspection of Table 4.2 reveals that the spheres originating 5.5 ft from the traps had average measured velocities 43.0 to 51.5 per cent higher than those predicted. The velocities for the spheres translated 14.0 ft were 1.6 to 19.7 per cent higher than predicted and those for the spheres translated 36.0 ft ranged from 6.2 per cent lower to 0.6 per cent higher than predicted. These observations are in general agreement with the hypothesis presented in Sec. 4.4.6 for installation 5P spheres; viz., the spheres placed at the smaller distances arrived at the trap while the absorber was still soft due to thermal action, whereas those arriving later from the
*Refer to Sec. 3.6 for a discussion of measured velocities of natural stones in relation to those predicted.
greater distances found the absorber restored to its natural hardness due to the cooling action of the blast winds. It is of interest to note that this effect was absent at the 7G4 sphere installation (shot Galileo, Chap. 6) where the maximum overpressure was $8.38 \mathrm{psi}^{*}$ but where the traps were given extra thermal protection (see Fig. 6.84 and Table 6.2).

Since both traps at installation 6P6 were appreciably affected by thermal radiation, the spheres from the small distance, whose velocities are presumed to have been relatively small, penetrated both traps. Thus the average striking heights (see column 5 in Table 4.2) for the spheres translated 5.5 ft were small in comparison with those for corresponding spheres at station 5P where the lower trap was relatively free of thermal damage.

Information in the last column of Table 4.2 indicates that three of the four types of spheres that were translated 5.5 ft had predicted velocities lower than the threshold. This would indicate that such sphere types were caught only because the absorber had been modified through thermal action.

Data for 31 natural stones obtained from trap 6P6a and for 58 from trap 6P6b are plotted, in Figs. 4.61 and 4.62, respectively. Both samples indicate that the smaller stones had high velocities and the larger ones had low velocities relative to the predicted ones. Table 4.6 gives the results of the analysis of natural stones at installation 6P6 as well as their data combined with data for all natural stones at station 6 P (see also Sec. 4.6.9).
(c) Data from Installation 6P7. The lower trap at this installation (6P7a) caught two pieces of gravel originating from 36.0 ft and one natural stone. The upper trap ( 6 P 7 b ) caught 7 natural stones, 1 gravel missile from 5.5 ft , 12 from 14.0 ft , and 5 from 36.0 ft . Data for the natural-stone missiles were combined for analysis with similar data obtained from other station 6P traps (see Sec. 4.6.9 and Table 4.6). The data for the one gravel missile translated 5.5 ft were combined with similar data for one from trap 6P4b but were not plotted. Data obtained from traps 6P4b, 6P5a, 6P5b, and 6P7b for 20 gravel missiles translated 14.0 ft are plotted in Fig. 4.63. Data for nine gravel missiles translated 36.0 ft from traps 6P5b, 6P7a, and 6P7b are plotted in Fig. 4.64. Velocities for the gravel translated 14.0 ft are in good agreement with those predicted, the predicted velocity line being between the regression line and the upper standard-error-of-estimate line (see Fig. 4.63). Velocities for the larger gravel missiles translated 36.0 ft are considerably lower than those predicted (see Fig. 4.64). Table 4.6 gives a summary of the results of the analysis of the combined data for the gravel at each distance.

### 4.6.7 Window-glass Installation 6P9

This installation was located 22.8 ft behind a standard window. Slight scorching of the wood in the upper trap is indicated by the postshot photograph (Fig. 4.65), although the absorber was found to be free from thermal damage. Similar scorching did not occur at other glass installations (see Fig. 4.50) at this station where the windows were placed nearer the traps.

Glass-fragment missile data obtained from traps 6P9a and 6P9b are plotted in Figs. 4.66 and 4.67, respectively. In both instances only a few missiles had measured velocities exceeding those predicted.

Data for 39 natural stones caught in the upper trap are displayed in Fig. 4.68 and are given in Table 4.6. Data for five natural stones caught in the lower trap as well as the 39 from the upper trap were combined with similar data obtained at other station 6 P traps (see Sec. 4.6.9 and Table 4.6). The line of predicted maximum velocity satisfactorily explains the higher velocities measured.

### 4.6.8 Window-glass Trap 6P10b (Above Pig Trap 6P10A)

Installation 6P10 was located 12.8 ft behind a standard window (see Fig. 4.69 for postshot view of this installation). There is evidence in this photograph that several large fragments of glass struck the absorber but were not caught. Data for 32 fragments that were retained by the


$\qquad$


absorber are presented graphically in Fig. 4.70. Only two missiles had velocities above those predicted.

Data for 10 natural stones caught by this trap are plotted in Fig. 4.71. All velocities measured were somewhat lower than the predicted maximum velocities. The natural-stone data are discussed, combined with others at station 6P, in Sec. 4.6.9 (see Table 4.6 also).

### 4.7 STATION 6PP (PIG STUDY), ${ }^{1}$ 4770-FT RANGE

The experiment at this station was similar to the ones at stations 4PP, 5PP, and $6.7 \mathrm{PP} . \ddagger$ This station had a 160 -ft-long glass wall; the traps were located near its center and 16.0 ft downwind. Since this station was at the same range as station 6 P , the same blast parameters were used to compute predicted missile velocities (see Table 4.5).

Results obtained at station 6PP are displayed graphically in Figs. 4.72 and 4.73 for the lower and upper traps, respectively. Data for the lower trap are quite similar to those obtained at installation 6P9 (see Figs. 4.66, 4.67, and 4.72), which was at the same range but was 22.8 ft from a standard window. However, data from the upper trap ( 6 PPb ) indicates that a greater number of fragments were caught and that their masses were smaller and their velocities higher. It is of interest to note that the total mass§ of the 390 fragments from the upper trap was 310 g , which is only 5 g greater than the total mass of 170 fragments from the lower trap.

One natural stone was caught in trap 6PPa and eight were caught in trap 6 PPb . Data for these missiles were combined with similar data from the station 6 P traps, which were also at 4770 -ft range. Results obtained from the combined data were discussed in Sec. 4.6 .9 and are given in Table 4.6 with station 6P data.

### 4.8 STATION 6.7PP (PIG STUDY), ${ }^{1}$ 4470- FT RANGE

The experimental plan for this station was the same as for station 6PP (discussed in Sec. 4.7) except that station 6.7 PP was 300 ft nearer GZ and the traps for station 6 PP were 18.0 ft behind the glass wall. Since blast-wave measurements for station 6.7PP were not available, values of peak overpressure and duration of the positive pressure were obtained from regression equations derived from measurements made at other Priscilla stations. $\pi$ These quantities, 6.99 psi for peak overpressure and 0.891 sec for duration, were used to compute predicted missile velocities.

The postshot photograph of station 6.7PP (Fig. 4.74) provides evidence of some scorching of the exposed wood surfaces of the trap housings. The absorber, however, was found to be undamaged by heating effects. This, in contrast to the observation of thermal damage to station
8 6 P absorbers not behind windows, serves to illustrate the thermal protection provided by ordi-

[^8]nary double-strength window glass. It should be pointed out that the glass used in the pig studies (stations 4PP, 5PP, 6PP, and 6.7PP) was unpainted, whereas that in the standard windows (stations $4 \mathrm{P}, 5 \mathrm{P}, 6 \mathrm{P}$, and 8 P ) was painted for the purpose of color coding.

The glass-fragment data obtained by station 6.7PP traps (Figs. 4.75 and 4.76 ) are related to the predicted velocities in a manner similar to that observed for station 6PP data. However, fewer total missiles were recovered at station 6.7PP than at 6PP. This discrepancy is evidently attributable to the fact that more diligence was exercised in one instance than in the other in recovery from the absorber of small fragments that were difficult to find.

Four natural stones having an average velocity of $140 \mathrm{ft} / \mathrm{sec}$ were recovered from trap 6.7PPa. Additional data for these missiles are listed in Table 4.6.

### 4.9 STATION 8P, 3930-FT RANGE

### 4.9.1 Experimental Plan and Blast Parameters

The chart in Fig. 4.77 illustrates the experimental plan for station 8P. The principal difference between the plan for this station and the one for station 6 P is that the gravel, military debris, and spheres were placed at greater distances from the traps at station 8 P since a somewhat stronger blast wave was expected at this station (see Sec. 3.6). Another notable difference is that a more rugged absorber (type III) was used in all station 8 P traps except those behind windows for which the windows themselves provided adequate protection against thermal radiation.

Figure 4.78 is an interesting preshot photograph of the 8 P station taken at a height of about 15 ft above ground level. Installation 8 P 1 is in the background and 8 P 10 is in the foreground. Note the sandbags placed on the lee side of the installations prepared for the exposure of animals. Displacement results obtained for the large stones and building blocks, to be seen in a line in the foreground in Fig. 4.78, are reported in Sec. 4.15.

Overpressure measured as a function of time at this station is shown graphically in Fig. 4.79. The dashed curve on this chart depicts the overpressure vs. time relation for an ideal blast wave whose impulse and duration are the same as those measured* ( $2.574 \mathrm{psi}-\mathrm{sec}$ and 0.823 sec , respectively). The maximum overpressure of the ideal blast wave that was used in the prediction of missile velocities was found by computation to be 8.60 psi . This value is somewhat lower than the gauge maximum of 9.20 psi shown on the chart as a spike. However, the overall agreement between the measured and computed curves is good.

Dynamic pressure vs. time measured at this station is shown in Fig. 4.80. The dashed line represents the dynamic pressure computed for the ideal blast wave whose parameters were discussed in the preceding paragraph. Although there are large fluctuations in the measured curve, the average values are in reasonable agreement with the computed ones up to about 0.055 sec . After that time the measured curve is consistently lower than the computed one.

### 4.9.2 Window-glass Installation 8P1

Installation 8 P 1 was located 7.8 ft behind a standard window (see Fig. 4.81 for preshot view of this installation). The dark appearance of the absorber was due to dust discoloration rather than thermal effects.

Data for 103 fragments caught in the lower trap (8P1a) and 100 from the upper trap are plotted in Figs. 4.82 and 4.83 , respectively. With a few exceptions the predicted-velocity lines form upper limits of the measured missile velocities.

Data for six natural stones caught by trap 8P1a are presented in Sec.4.9.9 in combination with similar data from other station 8 P traps.

### 4.9.3 Window-glass Trap 8P2b (Above Dog Trap 8P2A)

Trap 8P2b, which was placed above a dog trap, was 31.5 in . high. The installation was 12.8 ft behind a standard window.

[^9]Data for 497 fragments recovered from this trap (plotted in Fig. 4.84) indicate that most of the velocities measured were less than the predicted values. A relatively large number of missiles were recovered from this trap. At least part of the increase may be accounted for by the abundance of small fragments recovered.

Velocity and mass data for 25 natural stones from trap 8P2b are presented graphically in Fig. 4.85. The velocities measured were considerably lower than those predicted for stones that had traveled the optimum distance to maximize velocity.

### 4.9.4 Plate-glass Installation 8P3

At this location a standard plate-glass installation was placed 12.8 ft from the traps. The postshot photograph (Fig. 4.86) is remarkable in that it shows large depressed areas in the absorber caused by fragments of plate glass striking flat. In this photograph the absorber, which was originally white, appears gray due to the fine dust deposited by action of the blast wave.

Data for 25 fragments caught in the lower trap are shown in Fig. 4.87. The measured velocities were significantly lower than those predicted, especially for the larger missiles. This may have been caused by the fact that the orientations of the larger fragments were not truly random, as assumed in the calibration procedures for all fragments except those striking flat. An inspection of the lower trap in Fig. 4.86 indicates that the larger fragments appear to have struck almost flat, whereas none of them were judged to have struck in this orientation when the absorber was examined in the laboratory.

The upper trap ( 8 P 3 b ) caught 33 fragments whose orientations at impact were not flat (Fig. 4.88) and 7 whose orientations were flat (Fig. 4.89). The 33 fragments with non-flat orientations show velocity vs. mass relations similar to those noted for the lower trap. The "flat" fragments, however, were much larger and had measured velocities only slightly lower than those predicted. In agreement with theory, the larger of the flat fragments had somewhat higher average velocities than the smaller ones.

### 4.9.5 Military-debris and Gravel Installations 8P4 and 8P5

The chart in Fig. 4.77 illustrates the method of placement of military debris and gravel, color-coded for each of three distances, in front of installations 8P4 and 8P5. The postshot photograph (Fig. 4.90) indicates that the surface of the absorber at installation 8 P 5 was somewhat damaged by thermal radiation (note beaded appearance). The condition of installation 8P4 traps was about the same as that of 8P5 traps (Fig. 4.90).

No military debris was recovered from any of the four traps. The distribution, by trap and by displacement, of 214 gravel and natural-stone missiles caught is as follows:

|  | Natural <br> Trap | Gravel |  |  |  |
| :--- | :---: | :---: | :---: | :---: | :---: |
|  | stones | At 6.5 ft | At 16.8 ft | At 43.0 ft | Total <br> gravel |
| 8P4a | 0 | 5 | 23 | 3 | 31 |
| 8P4b | 6 | 5 | 103 | 3 | 111 |
| 8P5a | 2 | 8 | 8 | 2 | 18 |
| 8P5b | 0 | 20 | 10 | 16 | 46 |
|  | Total | 8 | 38 | 144 | 24 |

Data for the eight natural stones were combined for purposes of analysis with naturalstone data obtained from other traps at station 8 P . The results are presented in Table 4.6. A similar procedure was followed for the gravel missiles where the sample size was less than eight.

The results obtained for the larger samples of gravel are plotted by trap and by displacement distance in Figs. 4.91 and 4.97. It is noteworthy that the upper traps caught more missiles than the lower ones and also that the two largest samples originated from the $16.8-\mathrm{ft}$ distance.

TABLE 4.3 -PARTIAL RESULTS FOR SPHERES RECOVERED FROM STATION 8 P6
(Samples of less than five not included; complete data in Table 4.6.)

| Spheres* | Distance translated, ft | No. of missiles | Height above ground, in. |  | Velocity, ft/sec |  |  | \% Deviation $\dagger$ |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  |  |  | Placed | In trap, av. | Threshold | Measured, av. | Predicted | 100\% (M-P)/P | 100\% (P-T)/T |
| Gs (47.6) | 16.8 | 7 | 0 | 14.0 | 109 | 173 | 181 | -4.4 | 66.1 |
| GS (37.5) . | 16.8 | 5 | 18.5 | 17.4 | 113 | 191 | 187 | 2.1 | 65.5 |
| St $1 / 8$ | 43.0 | 14 | 0 | 10.9 | 66.6 | 118 | 133 | -11.3 | 99.7 |
| G1 (71.9) | 43.0 | 9 | 0 | 19.5 | 104 | 191 | 196 | -2.6 | 88.5 |
| Gs (37.9) | 43.0 | 28 | 0 | 19.8 | 113 | 204 | 212 | -3.8 | 87.6 |

*St $1 / 8$ : Steel spheres, $1 / 8 \mathrm{in}$. in diameter.
Gs: Small soda-glass spheres, average mass (mg) in parentheses.
G1: Large soda-glass spheres, average mass ( mg ) in parentheses.
$\dagger 100 \%(\mathrm{M}-\mathrm{P}) / \mathrm{P}$ is the per cent deviation of average measured velocity from predicted velocity; $100 \%(\mathrm{P}-\mathrm{T}) / \mathrm{T}$ is the per cent deviation of predicted velocity from threshold velocity.
$+$

Although there was little difference in the velocities of missiles caught in the upper and lower traps, provided the displacement distance was the same, there is to be noted an increase in velocity with increased distance of displacement. The regression lines describing the measured median velocities are in good agreement with the predicted velocities for the gravel displaced 16.8 and 43.0 ft (Figs. 4.91, 4.92, 4.95 to 4.97 ); however, the measured velocities for

|  | Gravel |  |  | Natural <br> Trap |
| :--- | :---: | :---: | :---: | :---: |
|  | At 6.5 ft | At 16.8 ft | At 43.0 ft | stones |
| 8P7a | 0 | 7 | 2 | 2 |
| 8P7b | 0 | 60 | 14 | 4 |

No gravel was caught from the 6.5 -ft distance, although some was caught from this distance at installation 8P5 (see Figs. 4.93 and 4.94). Graphical data for the three largest samples listed above are presented in Figs. 4.100 to 4.102 . In each instance the predicted velocities were only slightly higher than the median represented by the regression line.

Data for the three smaller samples listed above were combined for purposes of analysis with similar data obtained at station 8 P and are presented in Table 4.6.

### 4.9.7 Window-glass Installation 8P9

This installation, which was located 22.8 ft behind a standard window, was similar to installation 6P9 (described in Sec. 4.6.7).

The following data are useful in comparing the results from the two installations:

| Trap | No. of <br> missiles | Geometric <br> mean mass, <br> mg | Geometric <br> mean velocity, <br> ft/sec |
| :---: | :---: | :---: | :---: |
| 6P9a | 178 | 419 | 123 |
| 6P9b | 161 | 541 | 132 |
| 8P9a | 180 | 318 | 154 |
| 8P9b | 129 | 403 | 161 |

There is no apparent reason why only 129 fragments were recovered from trap 8P9b in comparison to 178,161 , and 180 from the other traps. It should be pointed out that the number of missiles recovered-especially small ones-is dependent to some extent on the diligence of search by the technician extracting the fragments from the absorber. Other significant patterns, however, may be noted in the above tabulation of results. The data for the higher overpressure ( 8 P ) indicate smaller missiles and higher velocities than the data for the lower overpressure. In contrast, the upper (b) traps in comparison to the lower (a) ones at the same station yielded both larger missiles and higher velocities.

Glass-fragment missile data for traps 8P9a and 8P9b are plotted in Figs. 4.103 and 4.104, respectively. Both sets of data show that the predicted maximum velocity defines an upper limit for the measured velocities.

### 4.9.8 Window-glass Trap 8P10b (Above Pig Trap 8P10A)

This installation, which was located 12.8 ft behind a standard window, was similar to the installation at station 6P (described in Sec. 4.6.8). (Figure 4.69 is a postshot view of this installation.) Results obtained are shown graphically in Fig. 4.105 for trap 8P10b and in Fig. 4.70 for trap 6P10b. The following summarizes the data obtained at the two installations:

| Trap | No. of <br> missiles | Geometric <br> mean mass, <br> mg | Geometric <br> mean velocity, <br> $\mathrm{ft} / \mathrm{sec}$ |
| :---: | :---: | :---: | :---: |
| 6P10b <br> 8P10b | 32 | 1010 | 110 |

The reason for the large difference in geometric mean mass of the missiles caught in the two traps is made apparent by examination of the plotted data in Figs. 4.70 and 4.105. At least as many large missiles were caught in trap 8 P 10 b as in trap 6P10b, but many more smaller ones were recovered from trap 8 P 10 b . The difference in geometric mean velocity between the two traps is undoubtedly significant and indicates that higher missile velocities are produced at higher overpressures.

Data for 20 natural stones caught in trap 8 P 10 b are plotted in Fig. 4.106. The fact that the measured velocities are considerably lower than those predicted is not significant-especially considering the small sample caught - since the points of origin of the natural stones are not known.

### 4.9.9 Combined Analysis for Natural Stones and Gravel at Station 8P

In previous sections missile data have been presented for each trap. In this section all data for natural stones caught in various traps at station 8 P have been combined, as well as the data for gravel missiles translated equal distances. The results of these analysis are recorded in Table 4.6; however, for purposes of discussion, the following data were extracted:

|  | Natural stones | Gravel |  |  |
| :---: | :---: | :---: | :---: | :---: |
|  |  | At 6.5 ft | At 16.8 ft | At 43.0 ft |
| Number | 85 | 38 | 211 | 41 |
| Geometric mean mass, mg | 80.9 | 175 | 178 | 232 |
| Geometric mean velocity, $\mathrm{ft} / \mathrm{sec}$ | 181 | 180 | 183 | 193 |
| Predicted geometric mean velocity, $\mathrm{ft} / \mathrm{sec}$ | 246 | 155 | 197 | 214 |
| Deviation of measured from predicted velocity, \% | -26 | 16 | -7.1 | -9.8 |

The predicted velocity of $246 \mathrm{ft} / \mathrm{sec}$ for natural stones with a mass of 80.9 mg was computed for the displacement which would maximize velocity for stones of this size. Thus it is not surprising that the geometric mean of measured velocities is 26 per cent lower than the predicted velocity, since the source of the stones is unknown. The probable reason that velocities measured for the gravel displaced 6.5 ft were higher than predicted is discussed in Sec. 4.4.6.

The velocities measured for the gravel placed at 16.8 and 43.0 ft are in reasonable agreement with theory.

### 4.10 STATION 10P, 2730-FT RANGE

### 4.10.1 Experimental Plan and Blast Parameters

The experimental plan for station 10P, depicted in Fig. 4.107, is similar to those previously discussed except that window and plate glass were not used at station 10P. Note that two of the installations, 10P2 and 10P3, contained only one trap. All installations except 10P1 were reinforced by sandbags placed on the lee side of the traps. Types III, IV, and V absorbers (see Chap. 2) were used at this station.

Overpressure vs. time measured at this station is plotted in Fig. 4.108. The deviations of the measured from the ideal overpressures are quite significant, the measured curve being characterized by a long rise time and an irregular, but relatively flat, peak. The dynamic pressure record obtained at this station, Fig. 4.109, shows even greater deviations from the ideal than the overpressure record. It is significant to the interpretation of the missile data obtained at this station that the dynamic pressure reached relatively high values, but was slow in development. Thus translational velocities attained after short displacements could be expected to be inordinately low compared to those later attained after greater displacements.

The displacement data for large stones, building blocks, and bricks at station 10P are presented in Sec. 4.15.

### 4.10.2 Military-debris and Gravel Installation 10P1

Figure 4.110 is a postshot view of installation 10 P 1 . No data were obtained from the upper trap, which contained type III absorber, because of excessive erosion. Gravel and naturalstone data from the lower trap are plotted in Figs. 4.111 to 4.113 . Velocities of the gravel measured after 19.1 ft of travel were low relative to those predicted on the basis of the ideal blast wave defined in Figs. 4.108 and 4.109 (see discussion in Sec. 4.10.1). The velocities for the gravel translated 49 ft (Fig. 4.112) were about the same as those predicted, whereas the velocities of the natural stones (Fig. 4.113) measured at various stages during the displacement cycle ranged up to $400 \mathrm{ft} / \mathrm{sec}$ higher than those predicted.

Thirteen pieces of military debris were caught in trap 10P1a. Nine of these originated from the $49-\mathrm{ft}$ distance and were combined for analysis with similar data obtained from installation 10P2 (see Sec. 4.10.3). Four of the 13 pieces caught were displaced only 7.4 ft . The data for these missiles are recorded in Table 4.6. It is sufficient to say here that their masses ranged from 12 to 271 g and their velocities from 110 to $203 \mathrm{ft} / \mathrm{sec}$.

### 4.10.3 Military-debris and Gravel Installation 10P2

Figure 4.114 is a postshot view of installation 10 P 2 . Note the damaged sandbags behind the installation and dry lake silt deposited in front of the trap.

Data for 31 pieces of gravel translated 19.1 ft before being caught are plotted in Fig. 4.115. The velocities shown in this figure are significantly higher than those in Fig. 4.111 for a similar type of experiment. A difference between the two situations, however, was that the trap absorber yielding the lower missile velocities (10P1a) was Styrofoam (type IV), whereas the other was balsa wood (type V). The balsa absorber was much less uniform than the Styrofoam and therefore yielded a less reliable velocity calibration. On the other hand, the balsa absorber was more resistant to the erosion effects due to the severe exposure conditions at this location. One circumstance that would tend to discredit the balsa data at installation 10P2 is that velocities of about the same magnitude were measured for the stones translated 19.1 ft (Fig. 4.115) as for those translated 49 ft before striking the trap (Fig. 4.116). In the Styrofoam trap, however, the gravel traveling the greater distance had higher velocities (see Figs. 4.111 and 4.112).

Data for 186 natural stones obtained from installation 10P2 are plotted in Fig. 4.117. The velocities are generally significantly higher than those to be expected from an ideal or classical blast wave whose overpressure impulse and duration are the same as those measured (3.329 $\mathrm{psi}-\mathrm{sec}$ and 0.737 sec , respectively). It is also noteworthy that the velocities of the larger stones were only slightly lower than those for the smaller stones.

Six pieces of military debris that had traveled 19.1 ft were caught in installation 10P2 (see Table 4.6). These missiles had masses that ranged from 14 to 144 g and velocities that ranged from 165 to $310 \mathrm{ft} / \mathrm{sec}$.

Data for three military-debris missiles displaced 49 ft were combined for analysis with similar data from trap 10P1a. Graphical data from both traps are shown in Fig. 4.118. Note that one missile penetrated through the balsa absorber to the plywood support to which the balsa was cemented. Data for this missile were not included in the analysis. The data for military debris from these two traps seem to be in agreement in contrast to the data for gravel, as noted above.

### 4.10.4 Gravel Installation 10P3 and Sphere Installation 10P4

The postshot condition of installation 10P3, depicted in Fig. 4.119, was similar to that of installation 10P2 (Fig. 4.114) except that installation 10P3 had accumulated a larger pile of native silt in front of it. Figure 4.120 shows installation 10P3 on the right and installation 10P4 on the left. Note that the dry lake bed, which had been smooth, suffered violent upheavels due to the shot. Although it is not evident from this photograph, the area in front of these installations had been paved with asphalt (see Fig. 4.107). The upper trap (10p4b) at the installation, shown on the left in Fig. 4.120, was found to be unusable for missile evaluation because of excessive erosion. This trap contained the same absorber (type IV) as the lower trap at installation 10P1, which did survive the traumatic environment produced by the explosion.

The only gravel caught in installation 10P3 which could be positively identified was that originating from the sample placed 49 ft from the trap. Velocities for the 78 gravel missiles caught in this installation (Fig. 4.121) are consistent with data for similar missiles obtained from installation 10P2 (Fig. 4.116). However, these velocities measured using the balsa absorber were significantly higher than those determined using type IV Styrofoam (Fig. 4.112).

Gravel missiles whose identification was doubtful were included with the natural-stone sample (Fig. 4.122). The velocities determined for this mixed sample were generally higher than those predicted.

Velocities for 66 gravel missiles (Fig. 4.123) and 96 natural stones (Fig. 4.124) caught in trap 10P4a are in general agreement with similar data from traps with balsa absorbers at this station.

Although 165 spheres were caught in trap 10 P 4 a , the point of origin generally could not be determined. The thin coat of paint that the spheres had been given for identification purposes was destroyed by action of erosion and thermal radiation. However, two $1 / 2$-in. -diameter steel spheres that were caught were identified since those spheres had been placed only at the 19.1-
ft distance from the trap. The impact velocities determined for these missiles were 197 and $198 \mathrm{ft} / \mathrm{sec}, 146$ per cent higher than the predicted velocity of $81 \mathrm{ft} / \mathrm{sec}$. Complete data for these two and the other 163 spheres that were caught are recorded in Table 4.6. The column listing the predicted velocities, $\mathrm{V}_{\mathrm{p} 50}$, contains two velocities for each type of missile if the translational distance is unknown - the lower value corresponding to a displacement of 7.4 ft and the higher value to a displacement of 49 ft . The column in Table 4.6 containing deviations of measured from predicted velocities, $\Delta \mathrm{V} \%$, lists two figures for most types of missiles for the same reason stated above. Measured velocities were higher than those predicted on the assumption of a $49-\mathrm{ft}$ displacement - and even for a $7.4-\mathrm{ft}$ displacement.

The velocities ranged from 0.5 to 143 per cent higher than predicted.

### 4.11 STATION 15P, 2280-FT RANGE

### 4.11.1 Experimental Plan and Blast Parameters

The experiment at this station (see Fig. 4.125) was similar to the one at station 10 P . All installations at station 15 P , however, contained single traps, and the marked missiles were placed at somewhat greater distances from the traps than at station 10P.

The overpressure measured at station 15P (Fig. 4.126) indicates similar anomalies as noted at station 10P (Fig. 4.108). At the nearer range, compared with the greater one, the overpressure duration decreased from 0.737 to 0.661 sec and the overpressure impulse increased from 3.329 to $3.829 \mathrm{psi}-\mathrm{sec}$. Even though the blast waves of these stations were definitely not of the ideal or classical type, the changes noted above are in the proper direction for such a wave.

The dynamic pressure measured at station 15P, recorded in Fig. 4.127, indicates even greater variability in pressure than the corresponding station 10P record (Fig. 4.109).

Section 4.15 includes the displacement data for large stones, building blocks, and bricks at station 15P.

### 4.11.2 Military-debris and Gravel Installations 15P1 and 15P2

Figure 4.128 is a preshot photograph of installations 15 P 1 (left) and 15 P 2 (right), looking toward GZ. In the installation 15P1 trap, the blast and thermal effects destroyed the type IV Styrofoam absorber; however, the trap housing remained intact but eroded along the leading edges.

The postshot view of installation 15P2 (Fig. 4.129) indicates that this installation stopped a considerable amount of soil and rocks. However, only 16 stones whose origin could be determined were caught in the trap. The data for these missiles, which were displaced 9.4 ft , are plotted in Fig. 4.130. These data indicate that relatively high velocities were attained in a short distance and that there was little dependence of velocity on missile mass.

Ten military-debris missiles were caught in installation 15P2 - two translated 9.4 ft ; three, 24.2 ft ; and five, 62.0 ft . Data for these missiles are plotted in Fig. 4.131, with individual coding for distance of translation. It is interesting that distance of translation made little difference in the measured impact velocities.

Data for 274 natural stones recovered from installation 15P2 are set forth in Fig. 4.132. Note that data for two missiles with velocities greater than $800 \mathrm{ft} / \mathrm{sec}$ are plotted above the upper edge of the chart. Data are given in the figure caption for two large stones that penetrated the entire thickness of the balsa absorber. The fact that the line indicating predicted velocities goes through the center of the data does not indicate agreement between measured and predicted velocities. The velocity predictions were made on the assumption of maximum velocity resulting from optimum distance of travel; therefore the line of predicted velocities should describe the higher missile velocities measured which presumedly resulted from optimum displacement.

### 4.11.3 Gravel Installation 15P3 and Sphere Installation 15P4

Figure 4.133 depicts the arrangement of traps and missiles, "planted" preshot, for installations 15P3 (right) and 15P4. Figure 4.134 is another preshot view of these installations (15P3 at the upper left) illustrating the placement of sandbags behind the traps.

Figure 4.135 is a postshot view of installation 15P3. Note that the balsa absorber was completely removed from the trap housing by action of the blast wave and that the housing itself was left partly filled with native soil.

The postshot view of installation 15P4 (Fig. 4.136) shows that the balsa stayed in place but was severely eroded. Note that surprisingly little material accumulated in front of the trap.

Data for 20 identifiable gravel missiles caught in installation 15P4 are plotted in Fig. 4.137. These missiles, after traveling 62.0 ft , had velocities remarkably near those predicted on the assumption of an ideal blast wave. Data for 232 natural-stone missiles caught in this trap (Fig. 4.138) also conform fairly well to the maximum velocities predicted.

Twenty-eight spheres, none of which could be identified by the color code, were caught in installation 15P4. Complete data for these missiles are recorded in Table 4.6. Because the distances of translation were unknown, predicted velocities were computed for the shortest and the greatest distance. These are recorded in the column marked $\mathrm{V}_{\mathrm{p} 50}$. The next column indicates that the average measured velocities varied from 26 per cent lower than predicted to 43.6 per cent higher.

### 4.12 STATION 20P, 2030-FT RANGE

### 4.12.1 Experimental Plan and Blast Parameters

The experimental plan for station 20P, similar to that for station 15 P , is illustrated by the layout chart in Fig. 4.139. All four traps at this station had balsa absorbers; however, only one of them (20P3) was found to be usable for evaluation of missiles.

The overpressure and dynamic pressure records (Figs. 4.140 and 4.141) obtained at this station are very similar in type to those already discussed for stations 10P and 15P (see Secs. 4.10.1 and 4.11.1).

The displacement data for large stones, building blocks, and bricks at station 20P are presented in Sec. 4.15.

### 4.12.2 Military-debris and Gravel Installations 20P1 and 20P2

Figure 4.142 is a preshot view of installations 20P1 (left) and 20P2. The balloon in the background is at the approximate location of GZ. Figures 4.143 and 4.144 are postshot views of installations 20P1 and 20P2, respectively. At installation 20P1 the balsa absorber was completely removed by the blast wave; at installation 20 P 2 it was only partly removed. That part which remained, however, yielded no usable missile data.

### 4.12.3 Gravel Installation 20P3 and Sphere Installation 20P4

Figure 4.145 is a preshot view of installations 20P3 (right) and 20P4. Note the gravel placed in front of installation 20P3 and the spheres in packets on wire supports in front of installation 20P4.

Figure 4.146 is a postshot photograph depicting a localized disruption of the dry lake bed on and near the area stabilized with asphalt (see Fig. 4.139). This upheaval was typical of others that were observed after the detonation at various spots in the regions close to GZ.

Figure 4.147 is a postshot view of installation 20P3 indicating the poor condition of the balsa absorber owing to the abrasive action of high-velocity silt and stones. Velocities were obtained for 88 stones recovered from this trap. Minimal velocities were evaluated for 11 additional stones that penetrated the entire thickness of the balsa and were found imbedded in the plywood support. Data for these 11 missiles are indicated as triangles on the plot in Fig. 4.148. From the data shown in this figure, it is evident that there were many missiles that had velocities considerably in excess of those to be expected from an ideal blast wave whose overpressure impulse is the same as that measured at this station ( $4.211 \mathrm{psi}-\mathrm{sec}$ ) (see Fig. 4.140). Although the samples of spheres were placed in front of installation 20 P 4 , one $1 / 8$-in.-diameter aluminum sphere, which had a velocity at impact of $357 \mathrm{ft} / \mathrm{sec}$, was retrieved from installation 20P3 (see Table 4.6).

Figure 4.149 is a postshot view of trap 20P4, which was judged to be unsuitable for the evaluation of missile velocities due to excessive erosion.

### 4.13 UNDERGROUND SHELTER WITH OPEN ENTRYWAY, OPS (UK 3.7)

### 4.13.1 Experimental Plan and Blast Parameters

An underground shelter constructed and tested during Operation Upshot-Knothole was made available to this project (33.2) for the study of translation effects due to winds associated with the blast wave. The shelter was located 900 ft from GZ (see station-location chart, Fig. 4.1). The plan view of the OPS shelter (Fig. 4.150) illustrates the construction of the structure as well as the experimental arrangement used in the present study. Note that the stairway is orientated toward GZ and that it connects to the shelter itself by means of an open, although somewhat tortuous, passageway.

The pressure instrumentation placed in the shelter entrance (see Figs. 4.150 and 4.151) failed to function. However, a pressure gauge placed at ground level near the shelter measured 65.4 psi maximum overpressure.

To make the experimental results more meaningful, test objects were chosen whose acceleration coefficients closely approximated those of man; ${ }^{5-7}$ i.e., had people occupied the shelter, their impact velocities would have been approximately the same as those measured for the test objects. The devices used were steel spheres with $7 / 16^{-}, 1 / 2^{-}$, and $9 / 16^{-i n}$. diameters similar to those used at the surface stations. In addition, three croquet balls were used whose masses had been increased with brass plugs so that an acceleration coefficient of $0.035 \mathrm{sq} \mathrm{ft} / \mathrm{lb}$ was obtained.

The placement positions of the spheres are indicated on the shelter drawing (Fig. 4.150); e.g., 20 steel spheres $1 / 2 \mathrm{in}$. in diameter were suspended 5.4 ft above the floor and 14.8 ft from the wall to which the type IV absorbing material was cemented. Figure 4.151 is a photograph taken near the missile-absorbing wall, looking toward the open entryway. The spheres were held in aluminum-foil bags so constructed and suspended that the spheres were readily released by action of the blast winds. The bags were taped to transverse wires of about the same strength as ordinary clothesline.

### 4.13.2 Sphere Data

Figure 4.152 is a postshot view of the absorbing wall. Aluminum foil similar to that used in the above-ground traps for thermal protection for the absorbers was partly blown away. Before the photograph shown in Fig. 4.152 was taken, the remaining foil had been removed and the impact points of the sphere were marked with a felt pen. The absorbing material (type IV) was found to be in good condition; no effects of thermal radiation, abrasion, or overpressure were shown.*

Impact points labeled 1 through 6 (Figs. 4.152 and 4.153 ) are for $1 / 2$-in.-diameter steel spheres that were placed 5.4 ft above the floor and 14.8 ft from the absorber. Three of these spheres struck the absorber at heights greater than the placement height, the average impact height being only 0.2 ft lower than that of placement. Thus lofting due to nonhorizontal winds is indicated. The average velocity of the $1 / 2$-in.-diameter spheres was $129 \mathrm{ft} / \mathrm{sec}$ (see Table 4.6). Velocities ranged from 99.1 to $159 \mathrm{ft} / \mathrm{sec}$, the higher values tending to be associated with missiles striking the upper-right portion of the absorber shown in Fig. 4.153.

Points labeled $b$ and $c$ in Figs. 4.152 and 4.153 mark the impact location of $9 / 16$-in.diameter steel spheres that, because of their impact location, probably originated from the group 9.8 ft from the absorber and 4.4 ft above the floor. Their average impact height was 4.6 ft above the floor. The average impact velocity ( $52.9 \mathrm{ft} / \mathrm{sec}$ ) was considerably lower than that for the $1 / 2$-in. spheres ( $129 \mathrm{ft} / \mathrm{sec}$ ), both because of a lower acceleration coefficient and a shorter distance of translation.

[^10]The croquet-ball imprint labeled "A" was probably made by the ball originally placed 9.8 ft away and 4.4 ft above the floor (see position marked with a large triangle and letter A in Fig. 4.153). It is to be noted that the points of impact of all spheres tended to be to the right of their original positions (see Fig. 4.152). Since croquet-ball A impacted to the left of its original position, one might speculate that it may have struck the right wall at a grazing angle before impacting with the absorber.

Of the 63 spheres used at this installation, only 9 struck the absorber with sufficient velocity to be captured or to make an impression sufficiently deep to allow identification of the missile and evaluation of velocity. A few impressions were noted which did not meet the above requirements. With one exception, the missiles that made sufficiently deep impressions did so in the upper-right quadrant of the absorbing wall (see Figs. 4.152 and 4.153 ). This would suggest that the blast or pressure wave did not fill the chamber uniformly but had a swirling motion, both horizontally and vertically, which allowed higher winds to develop on the upperright side than on the lower-left side (looking toward the absorbing wall). Another effect that might account for relatively few spheres striking the absorber is that the ground shock, which arrived before the blast wave, may have prematurely released some of the spheres from their aluminum-foil containers.

### 4.13.3 Molten-metal and Natural-stone Missiles

Sixty-nine missiles that were retrieved from the absorber were apparently formed from molten metal. They were almost spherical in shape, with masses that varied from 1 to 71 mg , similar to the beads that are commonly produced by welding operations. No attempt was made to estimate the impact velocity of these missiles since the holes they made in the Styrofoam indicated that they were hot at the time of impact; i.e., penetration was enhanced by melting the Styrofoam.

Data were obtained for 194 stone-like missiles whose origin was unknown. Many of these objects had the appearance of concrete chips. For want of a better title, they were called natural-stone missiles. The velocity vs. mass data, plotted in Fig. 4.154, indicate that their masses were small compared to those of the natural stones caught at the above-ground stations. Owing to calibration limitations, stones with masses less than 10 mg were omitted from the analysis. Measured velocities varied from 164 to $755 \mathrm{ft} / \mathrm{sec}$ (see Table 4.6), the smaller stones tending to have slightly higher velocities.

In order to better understand the production of the natural-stone missiles in this shelter, spatial-distribution charts were prepared which show as a function of location of impact the number of missiles per square foot (Fig. 4.155), the average masses (Fig. 4.156), and the average velocities (Fig. 4.157). The distribution chart in Fig. 4.155 indicates that most of the missiles impacted on the right side of the trap-a result similar to that obtained for the spheres evident in Figs. 4.152 and 4.153. The data in Fig. 4.156 indicate that the variation in the mass averages for various area segments was small ( 20.9 to 41.4 mg ). However, the velocity data plotted in Fig. 4.157 show a significant tendency for missiles striking in the upper right region (looking toward the absorber) to have higher velocities than those impacting in the remaining area. This result is consistent with the velocity data obtained for the $1 / 2$-in.-diameter spheres shown in Fig. 4.153.

### 4.14 UNDERGROUND SHELTERS WITH CLOSED ENTRYWAYS

The purpose of this study was to investigate a possible missile hazard within closed shelters due to spalling of concrete from the walls. In the seven shelters investigated, no missiles were caught, and there was no evidence of appreciable spalling. Pertinent blast parameters and details of shelter construction may be found in Refs. 3 and 4. The locations of these structures are indicated on the station-location chart (Fig. 4.1) at ranges from 860 to 1360 ft from GZ. Four of the shelters were of the arch type construction, ${ }^{3}$ and three were made with 8 -ftdiameter concrete conduits. ${ }^{4}$

A single trap containing type II absorber was placed, face up, near the center of each of the arch type shelters. As illustrated in Fig. 4.158, each trap was secured to the floor with chain and stud bolts. This anchor was not disturbed in any case by ground shock.

The trap arrangement was somewhat different in the three conduit type shelters. In order to increase the missile-collecting area, 16 strips of 2 - by 6 - by 36 -in. Styrofoam were cemented to the surface of the shelter (see Fig. 4.159).

### 4.15 LARGE-STONE, CONCRETE-BLOCK, AND BRICK DISPLACEMENT

### 4.15.1 General

This phase of the secondary-missile project involved measurement of the total displacement experienced by various test objects due to action of the blast wave. Additional studies would be required, making use of the experimental data reported here, in order to obtain estimates of the velocities attained by the displaced objects.

Twenty-five stones, two concrete blocks, and two ordinary bricks were placed near each of the seven above-ground missile stations already described. The placement positions are marked on the layout charts in Figs. 4.2, 4.23, 4.42, 4.77, 4.107, 4.125, and 4.139 for stations $4 \mathrm{P}, 5 \mathrm{P}, 6 \mathrm{P}, 8 \mathrm{P}, 10 \mathrm{P}, 15 \mathrm{P}$, and 20 P , respectively. The stones contained in each group, whose individual masses varied from about 150 g to 20 kg , were painted a distinctive color for later identification.

Figure 4.160 depicts a typical placement of large missiles at station $4 P$. The postshot photograph of the same installation (Fig. 4.161) shows that all displacements were relatively small but that the small stones traveled farther than the large ones. Note also that the concrete block or brick which initially presented the greater area to the wind (see Fig. 4.160) was displaced farther than its mate which presented a smaller area.

### 4.15.2 Large-stone Data

The relation between mass and distance displaced for the stones is shown graphically in Figs. 4.162 to 4.168 for each of the seven stations. After trying various types of plots, it was found that $\log$ mass vs. distance made the data as linear as any other and also had certain advantages; viz., zero distance could be plotted, and the points were separated into approximately equal mass intervals. In computing regression lines, either log mass or distance could be considered to be the dependent variable since scatter in the data was undoubtedly due to factors other than the measurement of mass or distance, e.g., variability in shape of the stones, nonhomogeneous blast wave, etc. It was decided to compute the regression lines by minimizing the square of the deviations in $\log$ mass since this procedure produced much more stable results (or regression lines) for the data from the precursor region (see Figs. 4.166 to 4.168 ) than that which minimized the square of the distance deviations.

Results of the statistical analyses described above are listed in the captions of the figures presenting the displacement data for the individual stations* (Figs. 4.162 to 4.168 ). The units of mass and distance used in the regression equations are the same as those used in plotting the data, viz., kilograms and feet. The geometric standard error of estimate in mass, $\mathrm{E}_{\mathrm{gm}}$, is a measure of the scatter of the mass points about the regression line. The quantity $M_{50}$ is the geometric mean mass of the stone sample. The average displacement of the stones at each station is indicated (in feet) by the quantity $\overline{\mathrm{d}}$.

The average displacements of stones for the three stations most distant from GZ varied from 2.29 ft at station 4 P (Fig. 4.162) to 1.15 ft at station 6 P (Fig. 4.164 ). It is probably not significant, in view of the variability of the data, that the stones at the most distant of these three stations had the highest average displacement. The stones at the station next closest to GZ (station 8P at $3930-\mathrm{ft}$ range) had a somewhat higher average displacement ( 7.50 ft ) (see Fig. 4.165).

Station 10P, at the $2730-\mathrm{ft}$ range, was 1200 ft closer to GZ than station 8 P . The stones at station 10P, which had an average displacement of 739 ft , almost spanned this separation in station locations. Only 16 of the 25 stones placed at station 10 P were recovered after the detonation. Some of these were smaller than they were originally due to splitting or chipping

[^11]TABLE 4.4 -MASSES AND DISTANCES DISPLACED FOR LARGE STONES, BLOCKS, AND BRICKS, SHOT PRISCILLA

|  | Station 4P |  | Station 5P |  | Station 6P |  | - Station 8P |  | Station 10P |  | Station 15P |  | Station 20P |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | m* | $\mathrm{d}^{*}$ | m* | $d^{*}$ | m* | $d^{*}$ | m* | $d^{*}$ | $\mathrm{m}^{*}$ | $d^{*}$ | m* | $\mathrm{d}^{*}$ | m* | $\mathrm{d}^{*}$ |
| Large stones | 0.287 | 6.5 | 0.235 | 0.8 | 0.206 | 5.1 | 0.227 | 15.2 | 0.308 | 249 | 0.620 | 1217 | 0.152 | 1617 |
|  | 0.332 | 5.5 | 0.277 | 0.1 | 0.256 | 0 | 0.254 | 19.2 | 0.665 | 1071 | 3.318 | 1237 | 0.384 | 872 |
|  | 0.391 . | 4.9 | 0.317 | 3.4 | 0.357 | 2.3 | 0.366 | 5.4 | 0.724 | 772 | 5.283 | 1814 | 0.384 | 1375 |
|  | 0.551 | 4.5 | 0.485 | 0.1 | 0.461 | 0 | 0.419 | 0 | 0.886 | 868 | 8.296 | 1039 | 0.394 | 910 |
|  | 0.641 | 4.1 | 0.528 | 2.9 | 0.514 | 3.6 | 0.528 | 14.5 | 0.998 | 943 | 12.975 | 1150 | 0.941 | 235 |
|  | 0.740 | 3.8 | 0.577 | 2.9 | 0.575 | 1.1 | 0.614 | 6.5 | 2.332 | 396 | 18.855 | 1745 | 1.027 | 1600 |
|  | 0.986 | 3.6 | 0.815 | 4.3 | 0.832 | 3.4 | 0.796 | 7.0 | 2.942 | 945 |  |  | 1.329 | 1190 |
|  | 1.015 | 1.8 | 0.895 | 2.3 | 0.972 | 0 | 0.971 | 24.0 | 3.278 | 713 |  |  | 1.858 | 899 |
|  | 1.159 | 3.4 | 1.053 | 0.6 | 1.250 | 0.5 | 1.061 | 13.0 | 3.787 | 1141 |  |  | 1.905 | 931 |
|  | 1.476 | 3.2 | 1.071 | 0.2 | 1.503 | 1.8 | 1.310 | 8.5 | 5.175 | 826 |  |  | 1.952 | 573 |
|  | 1.979 | 3.1 | 1.295 | 1.8 | 1.589 | 4.3 | 1.500 | 10.6 | 5.821 | 647 |  |  | 2.347 | 356 |
|  | 2.365 | 1.2 | 1.671 | 1.1 | 1.980 | 0.1 | 1.933 | 12.6 | 6.777 | 800 |  |  | 3.215 | 1258 |
|  | 2.862 | 1.5 | 2.261 | 0.7 | 2.120 | 1.0 | 2.189 | 6.9 | 9.734 | 608 |  |  | 3.249 | 881 |
|  | 3.233 | 0.9 | 2.643 | 2.0 | 2.579 | 0 | 2.711 | 6.7 | 12.653 | 333 |  |  | 3.252 | 1683 |
|  | 3.555 | 2.3 | 3.089 | 0.1 | 3.042 | 0.3 | 3.064 | 4.5 | 14.119 | 742 |  |  | 3.955 | 1190 |
|  | 4.702 | 0.7 | 3.324 | 2.9 | 3.300 | 1.0 | 3.325 | 13.0 | 18.844 | 772 |  |  | 3.969 | 1361 |
|  | 5.104 | 0.3 | 4.547 | 1.8 | 4.580 | 1.0 | 4.533 | 6.0 |  |  |  |  | 4.296 | 1525 |
|  | 6.069 | 1.1 | 5.980 | 0.6 | 5.407 | 0.3 | 5.732 | 2.7 |  |  |  |  | 4.564 | 871 |
|  | 7.083 | 0 | 6.837 | 0 | 6.100 | 0.3 | 7.028 | 2.5 |  |  |  |  | 5.895 | 1461 |
|  | 9.073 | 0.1 | 8.685 | 0 | 8.796 | 1.8 | 8.278 | 0.8 |  |  |  |  | 6.022 | 1004 |
|  | 10.606 | 0 | 9.215 | 1.3 | 9.310 | 0 | 9.243 | 1.8 |  |  |  |  | 6.089 | 1199 |
|  | 13.381 | 0.3 | 10.484 | 0.7 | 10.472 | 0 | 10.477 | 1.9 |  |  |  |  | 9.652 | 1356 |
|  | 14.980 | 2.0 | 12.419 | 0 | 10.816 | 0 | 12.532 | 2.4 |  |  |  |  | 10.377 | 845 |
|  | 16.946 | 0 | 16.195 | 0 | 16.509 | 0.8 | 16.027 | 0.3 |  |  |  |  | 10.766 | 1069 |
|  |  |  | 17.273 | 0 | 18.680 | 0.2 | 17.754 | 1.5 |  |  |  |  |  |  |
| Concrete |  |  |  |  |  |  |  |  |  |  |  |  |  |  |
| blocks | 15.491 | 2.5 | 14.014 | 1.0 | 14.293 | 12.8 | 15.239 | 40.5 |  |  |  |  |  |  |
|  | 15.695 | 0.3 | 14.090 | 4.8 | 15.670 | 0.3 | 15.367 | 3.6 |  |  |  |  |  |  |
| Bricks | 2.507 | 6.7 | 3.373 | $2: 9$ | 2.636 | 5.5 | 2.525 | 10.4 |  |  |  |  |  |  |
|  | 1.957 | 4.0 | 2.530 | 1.7 | 1.961 | 0 | 1.910 | 14.1 |  |  |  |  |  |  |

Note: Blank space indicates that object was not recovered after the shot.
*Mass (m), in kilograms; distance (d), in feet.
during translation. (Similar observations were made for stones placed at stations 15P and 20P.) The data presented in Fig. 4.166 for station 10P show that the distance translated had no significant dependence on stone mass.

Stone-translation data for stations 15P (2280-ft range) and 20P (2030-ft range) are plotted in Figs. 4.167 and 4.168 , respectively. Only six of the stones at station 15P were recovered, 900 ft from GŻ (see Sec. 4.13). The pressure instrumentation inside the shelter failed to function; however, the maximum overpressure measured at ground level near the shelter was 65.4 psi (see entry at bottom of Table 4.5). Missile traps were placed inside seven shelters with closed entryways ${ }^{3,4}$ at distances of 1360 to 860 ft from GZ.

### 4.16.2 Tabulated Results

A summary of all results obtained for shot Priscilla is given in Table 4.6. The data in each of three major divisions of Table 4.6 are listed by trap, or combination of traps at a particular station, in the order of decreasing range from GZ.

TABLE 4.5-BLAST PARAMETERS, SHOT PRISCILLA
(See List of Symbols.)
$\mathrm{p}_{0}=13.3 \mathrm{psi} \quad c_{0}=1120 \mathrm{ft} / \mathrm{sec}\left(17.0^{\circ} \mathrm{C}\right) \quad$ Estimated yield: $38 \mathrm{kt} * \quad$ Terrain, dry lake bed
(Frenchman Flat)

| Station | Range, ft | Blast <br> line | $\begin{aligned} & \left(I_{p}\right)_{m}, \dagger \\ & \text { psi-sec } \end{aligned}$ | $\begin{gathered} \left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{r}}, \ddagger \\ \mathrm{psi}-\mathrm{sec} \end{gathered}$ | $\begin{aligned} & \left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{m}, \dagger} \\ & \mathrm{sec} \end{aligned}$ | $\begin{gathered} \left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{r}}, \S \\ \mathrm{sec} \end{gathered}$ | $\underset{\mathrm{psi}}{\left(\mathrm{p}_{\mathrm{s}}\right)_{\mathrm{m}} \dagger}$ | $\begin{gathered} \left(\mathrm{p}_{s}\right)_{c}, \\ \mathrm{psi} \end{gathered}$ | $\begin{gathered} \left(\mathrm{p}_{\mathrm{s}}\right)_{\mathrm{r}}, * * \\ \mathrm{psi} \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 4 P | 6120 | 33.2 |  | 1.832 |  | 1.027 |  | (4.64) | 4.54 |
| 4 PP | 6120 | Main |  |  |  |  |  |  |  |
| 5 P | 5320 | 33.2 |  | 2.035 |  | 0.964 |  | (5.59) | 5.51 |
| 5 PP | 5320 | Main |  |  |  |  |  |  |  |
| 6 P | 4770 | 33.2 | 2.202 | 2.208 | 0.920 | 0.917 | 6.6 | 6.38 | 6.40 |
| 6 PP | 4770 | Main |  |  |  |  |  |  |  |
| 6.7PP | 4470 | Main |  |  |  | 0.891 |  |  | 6.99 |
| 8 P | 3930 | 33.2 | 2.574 | 2.553 | 0.823 | 0.841 | 9.2 | 8.60 | 8.34 |
| 10 P | 2730 | 33.2 | 3.329 | 3.354 | 0.737 | 0.713 | 9.3 | 13.0 | 13.7 |
| 15P | 2280 | 33.2 | 3.829 | 3.838 | 0.661 | 0.658 | 15.2 | 17.3 | 17.5 |
| 20P | 2030 | 33.2 | 4.211 | 4.187 | 0.610 | 0.624 | 15.2 | 21.4 | 20.6 |
| OPS | 900 |  |  |  |  |  | 65.4 |  | 62.4 |

*Estimation made by comparing the overpressure-impulse data measured for stations 6P and 8P with data for a surface burst as described in The Effects of Nuclear Weapons. $\dagger$ Determined from BRL mechanical-gauge records. (Gauges failed at stations 4P and 5P.) $\ddagger$ Overpressure impulse computed by regression equation derived from ( $\left.\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{m}}$ values

$$
\log \left(I_{P}\right)_{\mathrm{r}}=3.0982-0.7487 \log R
$$

§Overpressure duration computed by regression equation derived from $\left(t_{p}^{+}\right)_{m}$ values

$$
\log \left(t_{\mathrm{p}}^{+}\right)_{\mathrm{r}}=-1.6972+0.4512 \log R
$$

TPeak overpressure computed for a classical blast wave of impulse $\left(I_{p}\right)_{m}$ and of duration $\left(t_{p}^{+}\right)_{m}$. Measured values of impulse and duration were not obtained at 4 P and 5 P , therefore regression values, $\left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{r}}$ and $\left(t_{p}^{+}\right)_{r}$, were used.
**Peak overpressure computed by regression equation derived from $\left(p_{s}\right)_{c}$ values

$$
\log \left(p_{s}\right)_{r}=5.8300-1.3657 \log R
$$

A summary of the large-stone displacement data is presented at the bottom of Table 4.6. The regression coefficients e and fare explained in the table. It should be noted that the symbol $d$ is used here to designate the total distance of translation, whereas in other parts of the table it represents the distance traveled by the missile before striking the trap. The symbol $\bar{d}$ designates the average distance of translation. Minimum and maximum distances are represented by $d_{-}$and $d_{+}$, respectively.

### 4.16.3 Glass-fragment Missiles, Shot Priscilla

Impact velocities were evaluated for 3728 window-glass fragments caught in 32 traps placed at 6120- to 3930 -ft ranges. At the greater ranges, compared to the smaller ones, fewer missiles were caught, and their masses were larger and their velocities smaller.

The predicted velocities for the fragments caught in the lower overpressure region ( 4.5 to 5.5 psi ) were generally near the geometric mean of the measured velocities. This is in contrast to the predicted velocities applicable to the higher overpressure regions ( 6.4 to 8.6 psi ), which were generally near the highest values of the measured velocities. At stations 4 P and 5 P , windows were placed $7.8,12.8$, and 17.8 ft from the traps. At stations 6 P and 8 P the distances were $7.8,12.8$, and 22.8 ft from the traps. In no instance was there a significant difference in missile velocity due to distance of translation. Thus the velocities of window-glass fragments were found to be (1) less dependent on the blast-wave parameters than specified by

TABLE $4.6-$-SUMMARY OF RESULTS, SHOT PRISCILL
Regression Equation: $\log \mathrm{v}=\mathrm{a}+\mathrm{b} \log \mathrm{m}$


| Missile | Trap | Absorber type | d | n | $\mathrm{h}_{1}$ | $\bar{h}_{2}$ | $\mathrm{V}_{\mathrm{ps}}$ | $\Delta \mathrm{V}$ \% | $\overline{\mathrm{v}}$ | $\mathrm{S}_{\mathrm{v}}$ | v- | $\mathrm{v}_{+}$ | $\bar{M}$ | $\mathrm{S}_{\mathrm{m}}$ | $\mathrm{M}_{-}$ | $M_{+}$ | $\bar{\alpha}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Ny $1 / 8$ | 4 Pra | II | 28.0 | 1 | ${ }^{30}$ | 2.5 | 135 | 40.0 | ${ }^{189}$ |  |  |  | 19.73 |  |  |  | 0.9210 |
| $\mathrm{Al}_{1 / 8}$ | ${ }_{4 P 7 \mathrm{~b}}$ | II | 28.0 | 1 | 30 | 19.0 | ${ }^{99}$ | ${ }^{34.3}$ | 133 |  |  |  | 47.34 |  |  |  | 0.3839 |
| St $1 / 8$ | 4P7b | ${ }^{1}$ | $\begin{gathered} 4.5 \\ 28.0 \end{gathered}$ | 2 | - | 22.1 | $50^{(21)}$ | 76.4 | 88.2 | 4.7 | 85.0 | 91.5 | 130.8 |  |  |  | 0.1389 |
| Gs | $4 \mathrm{P7a}$ and b | п | $\begin{aligned} & 1,4.5^{(22)} \\ & 3,10.9 \end{aligned}$ | 11 | $22^{(23)}$ | 17.9 | $97{ }^{(21)}$ | 39.2 | 135 | 30.8 | 52.0 | 182 | 39.1 | 6.06 | 31.0 | 51.2 | 0.438 |
| ${ }_{\text {Al }}{ }^{\text {Ny }} 1 / 8$ | ${ }_{\text {5P7a and }}^{\text {5P7b }}$ | II | 32.0 | 1 | 0 | 24.2 | 166 | 29.5 | 215 |  |  |  | 19.73 |  |  |  | 0.9210 |
|  |  |  | 12.5 | 3 | 0 | 16.4 | 106 | 15.1 | 122 | 15.2 | 108 | 138 | 47.34 |  |  |  | 0.3839 |
|  |  |  | 32.0 | 3 | 0 | ${ }^{6.8}$ | 124 | 0 | 124 150 1 | ${ }_{22.6}^{6.2}$ | 117 | 129 | 47.34 <br> 4734 |  |  |  | 0.3839 08399 |
| $\mathrm{Al}^{3 / 16}$ |  |  | 32.0 | 4 | 29 | 18.8 | 124 | 21.0 | 150 | 22.6 | 134 | 182 | ${ }^{47.34}$ |  |  |  | 0.3839 |
|  | 5P7a and b | II | 4.9 12.5 | 5 | 14 0 | 4.1 22.0 | ${ }_{82}^{69}$ | 82.6 53.7 | 126 126 | 19.9 | 110 | 153 | 153.7 153.7 |  |  |  | 0.2660 0.2660 |
|  |  |  | 12.5 | 1 | 19 | 25.6 | 82 | 68.3 | 138 |  |  |  | 153.7 |  |  |  | ${ }_{0} .2660$ |
|  |  |  | 12.5 32.0 | 6 | ${ }_{0}$ | ${ }_{8.7}$ | ${ }^{82}$ | ${ }_{10.2}$ | 116 | 7.7 | 107 | 125 | 153.7 |  |  |  | ${ }_{0}^{0.2660}$ |
| St $1 / 8$ | 5P7a and b | ı | 12.5 | 6 | 0 | 17.0 | 67 | 22.4 | 82 | 8.7 | 67 | 92 | 130.8 |  |  |  | 0.1389 |
|  |  |  | 12.5 | 1 | 19 | 27.8 | 67 | 32.8 | 89 |  |  |  | 130.8 |  |  |  | 0.1389 |
|  |  |  | 32.0 | 4 |  | 8.0 | ${ }^{76}$ | 23.7 | 94 | 10.8 | 86 | 110 | 130.8 |  |  |  | 0.1389 |
| st $1 / 4$ | 5P7a | II | 4.9 | 1 | 14 | 4.4 | 37 | 43.2 | 53 |  |  |  | 1043 |  |  |  | 0.0697 |
| Gs | 5P7a and b | II | 4.9 | 1 | 0 | 9.9 | 85 | 55.3 | 132 |  |  |  | 39.0 |  |  |  | 0.438 |
|  |  |  | 12.5 | 18 |  | 17.6 | 111 | ${ }^{24.3}$ | 138 | 13.5 | 122 | 176 | 36.8 | ${ }^{3.63}$ | 31.0 | 42.8 | 0.446 |
|  |  |  | 12.5 | 2 | 19 | 18.8 | 103 | 50.5 | 155 | 26.9 | ${ }^{136}$ | 174 | 42.7 | 5.73 | 38.6 | 46.7 | 0.424 |
|  |  |  | 32.0 | 33 | 0 | 12.9 | 128 | 10.2 | 141 | 17.3 | 113 | 170 | 37.0 | 3.01 | 32.6 | 43.4 | 0.444 |
|  |  |  | 32.0 | 7 | 29 | 11.1 | 126 | 9.5 | 138 | 13.8 | 115 | 151 | 41.6 | 4.30 | 33.9 | 46.3 | 0.428 |
| G1 | 5P7a and b | II | 4.9 | 1 | 0 | 11.0 | 78 | 44.9 | 113 |  |  |  | 67.4 |  |  |  | 0.363 |
|  |  |  | 12.5 | 9 | 0 | 16.8 | 102 | 21.6 | 124 | 15.0 | 110 | 160 | 71.8 | 2.60 | 67.6 | 75.4 | 0.356 |
|  |  |  | 12.5 | 1 | 19 | 7.4 | 102 | 14.7 | 117 |  |  |  | 70.7 |  |  |  | 0.358 |
|  |  |  | 32.0 | 9 | 0 | 14.4 | 117 | 13.7 | 133 | 11.6 | 115 | 152 | 70.8 | 4.08 | 62.2 | 76.2 | 0.358 |
|  |  |  | 32.0 | 3 | 29 | 13.0 | 117 | 10.3 | 129 | 15.9 | 111 | 139 | 70.7 | 2.55 | 68.5 | 73.5 | 0.358 |
| Ny $1 / 8$ | ${ }_{6 P G a}$ and b | II | 14.0 | 2 | 0 | 21.0 | 173 | 28.9 | ${ }^{223}$ | 2.6 | 221 | 225 | 19.73 |  |  |  | 0.9210 |
|  |  |  | 14.0 36.0 | 3 | 18 30 | 24.4 9.6 | ${ }_{191}^{173}$ | ${ }_{22,5}^{22.0}$ | 211 234 | 18.8 15.5 | 198 218 | ${ }_{249}^{224}$ | 19.73 19.73 |  |  |  | 0.9210 0.9210 |
| A1 $1 / 3$ | ${ }_{6}$ PGa and b | п | 5.5 | 2 | 0 | 8.9 | 100 | 35.0 | 135 | 25.0 | 117 | 152 | 47.34 |  |  |  | 0.3839 |
|  |  |  | 5.5 | 2 | 13 | 5.5 | 100 | 54.0 | 154 | 12.3 | 146 | 163 | 47.34 |  |  |  | 0.3839 |
|  |  |  | 14.0 | 11 | 0 | 14.8 | 127 | 1.6 | $\begin{array}{r}129 \\ \hline 187\end{array}$ | 6.8 | 117 | 143 | 47.34 4734 |  |  |  | 0.3839 0.839 |
|  |  |  | 36.0 | 10 | 0 | 12.7 | 146 | - 6.2 | 137 | 6.4 | 132 | 151 | 47.34 |  |  |  | ${ }^{0.3839}$ |
|  |  |  | 36.0 | 1 | 30 | 9.4 | 146 | - 4.1 | 140 |  |  |  | 47.34 |  |  |  | 0.3839 |
| $\mathrm{Al}^{3} / 16$ | 6P6a and b | II | 14.0 | 10 | 0 | 22.1 | 108 | 18.5 | 128 | 7.4 | 114 | 136 | 153.7 |  |  |  | 0.2660 |
|  |  |  | 14.0 | 1 | 18 | 20.5 | 108 | 20.4 | ${ }^{130}$ |  |  |  | 153.7 |  |  |  | ${ }^{0.2660}$ |
|  |  |  | 36.0 | 6 | 0 | 12.7 | 125 | - 2.4 | 122 | 5.0 | 115 | 128 | 153.7 |  |  |  | ${ }^{0.2660}$ |
|  |  |  | 36.0 | 1 | 30 | 21.7 | 125 | -8.0 | 115 |  |  |  | 153.7 |  |  |  | 0.2660 |
| A1 $3 / 8$ | ${ }^{\text {6Pfa }}$ | І | * | 1 | * | 10.2 | 58; 90 | 27.9; -17.6 | 74.2 |  |  |  | 376.7 |  |  |  | 0.1292 |
| St $1 / 8$ | ${ }_{6 P 6 a}$ and b | I | 5.5 | 10 | 0 | 18.4 | 62 | 47.7 | 91.6 | 16.2 | 78 | 124 | 130.8 |  |  |  | 0.1389 |
|  |  |  | 5.5 | 4 | 13 | 11.2 | ${ }^{62}$ | 59.7 | 99.0 | 13.7 | 85 | 117 | 130.8 |  |  |  | 0.1389 |
|  |  |  | 14.0 | 7 | 0 | 15.5 | 80 | 9.1 | 87.3 | 6.3 | 80 | 97 | 130.8 |  |  |  | 0.1389 |
|  |  |  | 36.0 |  | 0 | ${ }_{18.6}^{12.6}$ | ${ }_{93}^{93}$ | - 1.5 | ${ }_{85}^{91.6}$ | ${ }^{8.8}$ | 79 | ${ }_{9}^{105}$ | 130.8 1308 |  |  |  | 0.7389 |
|  |  |  | 36.0 | 2 | 30 | 8.4 | 93 | - 7.8 | 85.7 | 11.2 | 78 | 94 | 130.8 |  |  | - | 0.1389 |
| St $1 / 4$ | ${ }_{6 P 6 a}$ and b | II | 5.5 | 1 | 13 |  | ${ }^{46}$ | 104.3 | 94 |  |  |  | 1043 |  |  |  | 0.0697 |
|  |  |  | 14.0 | 2 | 0 |  | 58 | 5.2 | 61 | 1.1 | 60 | ${ }^{6}$ | 1043 |  |  |  | 0.0697 |
| Gs | 6P6a and b | II | 5.5 | 7 | 0 | 10.8 | 107 | 43.0 | 153 | 8.9 | 143 | 168 | 38.4 | 3.71 | 31.6 | 42.6 | 0.443 |
|  |  |  | 5.5 | 6 | 13 | 9.0 | 106 | 48.1 | 157 | 15.2 | 141 | 178 | 39.0 | 3.68 | 33.5 | 43.0 | 0.438 |
|  |  |  | 14.0 | 36 | 0 | 18.8 | 130 | 14.6 | 149 | 9.0 | 130 | 178 | 47.0 | 3.14 | 40.9 | 52.9 | 0.406 |
|  |  |  | 14.0 | 18 | 18 | 13.3 | 132 | 19.7 | 158 | 15.4 | 134 | 192 | 39.4 | ${ }^{4.35}$ | ${ }^{33.8}$ | 50.5 | ${ }^{0.436}$ |
|  |  |  | ${ }^{36.0}$ | 43 |  | 14.8 | 155 | 0.6 | 156 | 10.6 | 134 | 188 | 37.2 | 3.25 | 32.3 | 44.9 | 0.444 |
|  |  |  | 36.0 | 10 | 30 | 12.8 | 149 | - 1.3 | 147 | 5.1 | 136 | 154 | 47.1 | 3.08 | 42.5 | 52.4 | 0.405 |
| GI | 6P6a and b | II | 5.5 | 3 | 0 | 4.8 | 97 | 36.1 | 132 | 3.9 | 129 | 136 | 71.6 | 1.49 | 69.9 | 72.8 | 0.355 |
|  |  |  | 5.5 |  | 13 | 7.1 | 97 | 51.5 | 147 | 9.7 | 134 | 157 | 72.6 | 2.34 | 70.0 | 75.1 | 0.354 |
|  |  |  | 14.0 | 21 | 0 | 17.6 | 122 | 11.5 | 136 | 9.5 | 115 | 157 | 72.9 | 2.84 | 65.7 | 77.8 | ${ }^{0.353}$ |
|  |  |  | 14.0 | 1 | 18 | 19.3 | 122 | 15.6 | 141 |  |  | , | 76.0 |  |  |  | 0.351 |
|  |  |  | 36.0 | 12 | 0 | 14.3 | 142 | - 4.9 | 135 | 10.7 | 119 | 152 | 71.3 | 2.75 | 65.1 | 75.2 | 0.355 |
|  |  |  | 36.0 | 4 | ${ }^{30}$ | 7.2 | 141 | - 7.1 | 131 | 9.7 | 123 | 145 | 73.5 | 1.50 | 71.5 | 75.1 | 0.352 |
| Ny $1 / 8$ | ${ }_{8 P 6 a}$ and b | III | 43.0 | 1 | $\bigcirc$ | 5.9 | 258 | 17.8 | 304 |  |  |  | 19.73 |  |  |  | 0.9210 |
|  |  |  | - | 2 | * | 12.7 | 189; 258 | 29.6;-5.0 | 245 | 5.9 | 240 | 249 | 19.73 |  |  |  | 0.9210 |
|  | 8P6a and $b$ | 피 | 16.8 | 1 | 18.5 | 22.8 | 177 | 14.1 | ${ }^{202}$ |  |  |  | 47.34 |  |  |  | ${ }^{0.3839}$ |
|  |  |  | 43.0 | 4 | 0 | 19.7 | 302 | - 38.4 | 186 | 8.2 | 178 | 197 | 47.34 |  |  |  | 0.3839 |
|  |  |  | 43.0 | ${ }_{1}^{3}$ | ${ }^{29.5}$ | 16.9 19.9 | 302 $138 ;$ | $-\quad \begin{gathered} 38.4 \\ 45.7 ;-33.4 \end{gathered}$ | 186 201 | 16.8 | 167 | 201 | 47.34 47.34 |  |  |  | 0.3839 0.3839 |
| $\mathrm{Al}^{3 / 16}$ | 8P6a and b | III |  |  |  |  | 118 | 26.3 | 149 |  |  |  | 153.7 |  |  |  | 0.2660 |
|  |  |  | 16.8 | 3 | 0 | 19.6 | 154 | 7.1 | 165 | 5.5 | 159 | 170 | 153.7 |  |  |  | 0.2660 |
|  |  |  | 16.8 | 1 | 18.5 | 12.9 | 154 | - 12.3 | 135 |  |  |  | 153.7 |  |  |  | ${ }^{0.2660}$ |
|  |  |  | 43.0 | 4 | 0 | 15.5 | 176 | - 3.4 | 170 | 18.0 | 145 | 185 | 153.7 |  |  |  | 0.2660 |
|  |  |  | 43.0 | 1 | 29.5 | 11.2 | 176 | - 22.7; | 136 |  |  |  | 153.7 |  |  |  | ${ }^{0.2660}$ |
| St $1 / 8$ | 8P6a and b | III | 6.5 | 4 | 0 | 10.1 | 88 | 15.9 | 102 | 7.2 | ${ }^{95}$ | 112 | 130.8 |  |  |  | 0.1389 |
|  |  |  | 16.8 16.8 | ${ }_{3}^{2}$ | $\stackrel{0}{18.5}$ | 23.2 6.4 | 115 115 | $\begin{array}{r}1.7 \\ \hline 16.5\end{array}$ | 113 134 | 6.9 13.7 | 109 126 | 118 150 | 130.8 <br> 130.8 |  |  |  | 0.1389 0.1389 |
|  |  |  | 43.0 | 14 | 0 | 10.9 | 133 | -11.3 | 118 | 11.0 | 99 | 137 | 130.8 |  |  |  | 0.1389 |
|  |  |  | ${ }^{43.0}$ | 4 | 29.5 | ${ }^{6.3}$ | 133 | -23.3 | 102 | ${ }^{3.5}$ | ${ }_{9}^{99}$ | 107 | 130.8 |  |  |  | ${ }^{0.1389}$ |
| St $1 / 4$ | 8P6a and b |  | 6.5 | ${ }_{1}^{4}$ | * | 16.7 10.4 | ${ }^{88 ;}{ }_{63} 133$ | $\underset{\substack{22.7 \%-18.8 \\ 28.6}}{\text { 2, }}$ | 108 81 | 13.0 | 95 | 123 | ${ }_{1043}^{130.8}$ |  |  |  | 0.1389 0.0697 |
|  |  | iII | 16.8 | 2 | 18.5 | 13.6 | 80 | 18.8 | 95 | 18.7 | 82 | 109 | 1043 |  |  |  | 0.0697 |
|  |  |  | 43.0 | 1 | 29.5 | 24.7 | 94 | 2.1 | 96 |  |  |  | 1043 |  |  |  | 0.0697 |
| St ${ }^{1 / 16}$ | 8P6a and b | ii | 6.5 | 2 | 0 | 17.5 | 47 | 57.4 | ${ }^{74}$ | 7.9 | 69 | 80 | 5597 |  |  |  | 0.0398 |
| Gs | ${ }_{8 P 6 \mathrm{a}}$ and b | ill | 16.8 | 7 | ${ }^{0}$ | 14.0 | 181 | - 4.4 | 173 | 20.7 | 139 | 198 | 47.6 | 2.6 | 43.8 | 50.5 | 0.405 |
|  |  |  | 16.8 43.0 | 28 | ${ }^{18.5}$ | 17.4 | ${ }_{212} 8$ |  | ${ }_{294}^{191}$ | ${ }_{192}^{22.3}$ | ${ }_{166}^{167}$ | ${ }_{236}^{216}$ | ${ }_{37}^{37.5}$ | 2.7 | ${ }^{355}$ | 42.1 | ${ }_{0}^{0.444}$ |
|  |  |  | ${ }^{43.0}$ | 28 6 | * | 19.8 21.8 | 212 1405 | ${ }_{-}^{3.8}{ }_{40.7}^{3 .} \mathbf{- 3 . 9}$ | 204 197 | 19.2 | 186 | ${ }_{223}^{236}$ | 37.9 46.4 | 2.8 3.1 | 33.9 42.2 | 45.5 51.0 | - |
| GI] | 8P6a and b | iI | 6.5 | 2 | 0 | 13.7 | 133 | 14.3 | 152 | 5.2 | 149 | 156 | 71.9 | 2.7 | 70.0 | 73.8 | 0.356 |
|  |  |  | ${ }_{16.8}^{16.8}$ | ${ }^{3}$ | 0 | 18.3 | 172 | ${ }^{3.5}$ | 178 | 12.4 | 170 | 192 | 72.5 | 2.1 | 70.2 | 74.0 | 0.354 |
|  |  |  | 16.8 | 1 | 18.5 | 9.8 | 171 | 1.2 | 173 |  |  |  | 74.8 |  |  |  | 0.351 |
|  |  |  | 43.0 | 9 | 0 | 19.5 | 196 | - 2.6 | 191 | 13.1 | 164 | 208 | 71.9 | 2.4 | 66.8 . | 74.6 | 0.356 |
|  |  |  | * ${ }^{43.0}$ | ${ }_{1}^{2}$ | ${ }^{29.5}$ | 16.4 25.9 | ${ }_{134}^{196} 195$ | ${ }_{-}^{-16.8} 12.6 ;-22.6$ | 163 151 | 1.4 | 153 | 173 | ${ }_{73.2}^{72.0}$ | 0.3 | 71.8 | 72.2 | ${ }^{0.355}$ |
| $\mathrm{Al}^{1 / 6}$ | 10P4a | $\checkmark$ | * | 32 | * | 7.3 | 214; 312 | 104; 39.7 | 436 | 74.4 | 316 | 617 | 47.34 |  |  |  | 0.3839 |
| ${ }_{\text {Al }}{ }^{3} / 16$ | 10P4a 10P4a | v | * | ${ }_{3}^{10}$ | * | 7.3 | ${ }_{1}^{1855} 275$ | $105 ; 37.8$ 130.-52.4 | 379 | 34.4 | ${ }_{3}^{330}$ | ${ }_{413}^{441}$ | 153.7 |  |  |  | 0.2660 |
| $-\mathrm{Al}^{1 / 2}$ | ${ }^{1084 a}$ |  |  |  | * |  | lis3:-246 |  | ${ }^{2315}$ | ${ }_{271}^{33.2}$ | -352 | ${ }_{262}^{413}$ | ${ }_{130.8}^{376.7}$ |  |  |  | ${ }_{0}^{0.1389}$ |
| St ${ }_{\text {St }}$ /1/8 | 10P4a 10P4a | $\stackrel{\text { v }}{ }$ | * | ${ }_{7}^{12}$ | * | 6.9 6.5 | 140; 210 100; 147 | 50.7: 0.5 $139 ; 62.6$ | 211 239 | ${ }_{86.1}^{27.1}$ | 182 180 | ${ }_{420}^{262}$ | ${ }_{1043}^{130.8}$ |  |  |  | 0.1389 0.0697 |
| St $1 / 2$ | 10P4a | v | 19.1 | 2 | 23 | 7.7 | ${ }_{81}$ | 143 | 198 | 1.1 | 197 | 199 | 8353 |  |  |  | 0.0348 |
| Gs | 10P4a | v | * | 73 | * | 7.7 | 220; 325 | 66.8; 12.9 | 367 | 59.3 | 264 | 554 | 42.7 | 5.6 | 30.4 | 51.7 | 0.424 |
| GI | 10P4a | v | * | 26 | * | 8.2 | 210; 305 | 88.1; 29.5 | 395 | 59.3 | 301 | 545 | 70.9 | 4.0 | 58.9 | 75.1 | 0.358 |
| $\mathrm{Al}_{1}^{1 / 8}$ | 15P4 | v | * | 3 | * | 5.5 | 285; 408 | 34.3; - 6.1 | ${ }^{383}$ | 25.4 | 353 | 400 | 47.34 |  |  |  | 0.3839 |
| A1 ${ }_{\text {Al }} / 1 / 16$ | 15P4 | v | * | ${ }_{5}^{4}$ | * | 8.4 | 250; 370 194; 285 | $43.6 ;-3.0$ $8.8 ;-26.0$ | ${ }_{211}^{359}$ | 36.1 17.0 | 325 187 | ${ }_{234}^{410}$ | 153.7 130.8 |  |  |  | 0.2660 0.1389 |
| St ${ }_{\text {St }} / 1 / 4$ | 15 P 4 <br> 15 P 4 <br> 1 | $\stackrel{\rightharpoonup}{v}$ | * | 1 | * | ${ }_{7.3}^{8.5}$ | 194; 203 | $8.8 ;-26.0$ $37.3 ;-3.9$ | ${ }^{2195}$ | 17.0 |  | 234 | 130.8 1043 |  |  |  | 0.1389 0.0697 |
|  | 15 l | v | : | 9 | * | 6.3 | 300; 430 | 14.0; -20.5 | 342 395 | ${ }_{5}^{52.0}$ | ${ }_{347}^{293}$ | 432 430 | ${ }^{35.4}$ | ${ }^{5.2}$ | ${ }^{27.0}$ | ${ }_{74.8}^{44.8}$ | 0.451 0.356 |
| G] | 15 P 4 | v | * | 6 | * | 4.0 | 278; 408 | 42.1; - 3.2 | 395 | 30.0 | 347 | 430 | 71.7 | 3.6 | 66.8 | 74.9 | 0.356 |
| A1 $1 / 8$ | 20 P 3 | v | * | 1 |  |  |  |  | 357 |  |  |  | 47.34 |  |  |  | 0.3839 |
| $\begin{aligned} & \mathrm{stt}^{1 / 2} \\ & \text { st } / 2 \mathrm{sin} \\ & \mathrm{CB} \end{aligned}$ | ops | IV | 14.8 | ${ }^{6}$ | ${ }^{64.7}$ | ${ }_{61.9}^{64}$ |  |  | ${ }^{129}$ | ${ }^{22.8}$ | ${ }_{59}^{99.1}$ | 159 | 8353 1887 |  |  |  | ${ }^{0.0348}$ |
|  | ${ }_{\text {OPS }}^{\text {OPS }}$ | $\underset{\text { IV }}{\text { IV }}$ | 9.8 9.8 | ${ }_{1}^{2}$ | 52.5 52.5 | 54.9 46.3 |  |  | 52.9 45.0 | 0.4 | 52.6 | 53.2 | 11874 356000 |  |  |  | ${ }_{0.035}^{0.0310}$ |

LaRGe-Stone data (log m $=\mathrm{e}+\mathrm{f} \cdot \mathrm{d} \mathrm{m}, \mathrm{kg} ; \mathrm{d}$, ft

| Missile | Station | n | e | $f$ | $\mathrm{E}_{\mathrm{gm}}$ | $M_{50}$ | $\mathrm{S}_{8 \mathrm{~m}}$ | M- | $\mathrm{M}_{+}$ | ¢ | d_ | $\mathrm{d}_{+}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Large stones | 4 P | 24 | 0.9616 | -0.2579 | 1.68 | 2.357 | 3.45 | 0.287 | 16.946 | 2.29 | 0 | 6.46 |
|  | ${ }_{5 P}$ | 25 | 0.5682 | -0.1956 | 3.27 | 2.139 | 3.72 | 0.235 | 17.273 | 1.22 | 0 | 4.25 |
|  | ${ }_{6 P}$ | 25 | 0.5475 | -0.1817 | 3.16 | 2.180 | 3.67 | 0.206 | 18.680 | 1.15 | 0 | 5.08 |
|  | ${ }_{8 P}$ | 25 | 0.7674 | -0.0573 | 2.78 | 2.178 | 3.70 | 0.227 | 17.754 | 7.50 | 0 | 24.0 |
|  | 10 P | 16 | 0.5872 | -0.0001 | 3.30 | 3.105 | 3.31 | 0.308 | 18.844 | 739. | 249 | 1141. |
|  | 15P | 6 | 0.1490 | 0.0004 | 2.93 | 5.296 | 3.04 | 0.620 | 18.855 | 1367. | 1039 | 1814. |
|  | 20 P | 24 | 0.3504 | 0.000009 | 3.96 | 2.293 | 3.06 | 0.152 | 10.766 | 1094. | 235 | 1683. |

*Source undetermined due to erosion and thermal damage of missile.
(1) Combination of NS from traps 4 P4a and $b ; 4 \mathrm{P} 5 \mathrm{a}$ and $\mathrm{b} ; 4 \mathrm{PGb}$; 4 P .


(4) Combination of Gr, $=12.5$, from traps 5P4b; 5P5a; 5 PFa and b
${ }^{(6)}$ ) cmine
(1) Combination of Gr, d=14.0, from traps 6 P 4 ; $6 \mathrm{P5a}$ and b ; $6 \mathrm{P7b}$.



(4) Combination of Gr, $\mathrm{d}=7.4$, from traps 10P1a and 10 P2.
(5) Combination of $\mathrm{Gr}, \mathrm{d}=7.4$, from traps 10 Pla and 10 PP .

(18) Combination of $\mathrm{MD}, \mathrm{d}=49.0$, from traps 10 P
(i8) Combination of Ns from traps 15 P 2 and 15 P .
$\left.{ }^{9} 9\right)^{9}$ Combination of $\mathrm{Gr}, \mathrm{d}=62.0$, from installations 15 P 2 and 15 P 4
(20) ${ }^{\text {MD }}$ from
(22) Weighed average of $\mathrm{V}_{\text {ppo }}$ 's at d 's.
(23)
within the limits investigated.

From the above observations it must be assumed that certain phenomena which are not accounted for in the model have a noticeable influence on the velocity attained by glass fragments under the conditions of the experiments reported. Part of this extraneous influence on missile velocity may be due to the mechanism of breakage of glass panes. If a pane supported along its edges is bent, a certain amount of potential and kinetic energy is stored in the pane before actual breakage occurs. Fragments near the center of the pane possessing the greater part of this energy would "pop out" at higher velocities than those near the perimeter. It should be pointed out that the energy thus temporarily stored in each pane is not necessarily at the leading edge of a classical blast wave. The defractive loading effect described above would be enhanced by the process of reflection but would be mitigated provided the blast wave arrived on the lee side of the pane before it shattered. Also, if shattering occurred before appreciable bending had taken place, as might be the case for a relatively strong blast wave, then the defractive effect would be minimal since the pressure difference between the front and rear of the pane would quickly vanish when the glass is broken.

The effects postulated in the preceding paragraph would tend to equalize fragment velocities produced by blast waves of different strengths and also for different distances of translation. The different distances of translation follow from the assumption that the velocities are imparted to a fragment by diffractive loading in a very short time during which the missile travels a short distance.

The dispersion of fragment velocities, which was noted in all the experimental data except for the fragments striking flatwise, is a reasonable result of the method of mounting the glass panes. Since the edges of the panes were restrained, fragments arising near the perimeter of the pane would be expected to have lower velocities (and more tumbling) than those arising near the center.

Six traps placed behind plate-glass installations caught a total of 88 fragments. Velocities evaluated for 12 large fragments striking the trap flat were much more uniform than the velocities for the fragments striking in random orientation. Velocities for the flat missiles were only slightly lower than those predicted.

### 4.16.4 Marked-gravel and Natural-stone Missiles, Shot Priscilla

Velocities were determined for 799 gravel missiles with masses between 10 mg and 1.3 g which had been color coded and placed at measured distances from the traps. For samples greater than five which were caught at nonprecursor stations (4P, 5P, 6P, and 8P), the geometric mean velocities were generally in good agreement with the predicted ones. The least satisfactory agreement was obtained for 14 gravel missiles caught at station 4 P after a displacement of 10.9 ft . In this instance the geometric mean of the measured velocities was 112 $\mathrm{ft} / \mathrm{sec}, 20$ per cent higher than the predicted value of $93 \mathrm{ft} / \mathrm{sec}$. This deviation may have been partly due to the lower-velocity missiles' having insufficient penetration for retention in the absorber.

The geometric means of measured velocities for gravel placed at stations 10P and 15P in the precursor region were as much as 39 per cent higher than the values predicted assuming an ideal blast wave with the same overpressure impulse as that measured.

Velocities were evaluated for a total of 1756 natural-stone missiles, including 194 stonelike objects caught in the OPS shelter with open entryway. Because predicted velocities were based on the assumption of optimum distance of travel for maximum velocity, the values tended to be higher than those measured.

### 4.16.5 Sphere Data, Shot Priscilla

Of a total of approximately 67,000 spheres placed in front of traps, impact velocities were obtained for 712. The predicted and measured velocities were generally in agreement. In instances where agreement was not good, the deviations were probably due to (1) inaccuracies in the trapping technique for small depths of penetration and (2) softening of the outer layer of absorbing material due to action of the thermal pulse.

### 4.16.6 Military-debris Data, Shot Priscilla

Velocities were estimated for 32 military-debris missiles whose masses ranged from 4.5 to 289 g . Only one piece of debris was caught at a nonprecursor station, 6P, where the maximum overpressure was 6.4 psi . Velocities for the military-debris missiles caught in the precursor region varied from 110 to $373 \mathrm{ft} / \mathrm{sec}$.

### 4.16.7 Missiles in Shelters

Missile studies were conducted in eight underground shelters that were located 860 to 1360 ft from GZ. Seven of the eight shelters had closed entryways. Missile traps were placed in these shelters in order to determine the velocity of any particles that might spall from the concrete walls. There was no evidence of appreciable spallation.

Missile-absorbing material was cemented to a wall of a shelter with open entryway in such a way that velocities could be determined for experimental spheres. The aerodynamic properties of the spheres used were such that their impact velocities would be approximately the same as for man. Velocities evaluated for nine such spheres ranged from 45 to $159 \mathrm{ft} / \mathrm{sec}$ for situations where the distances of translation were 9.8 and 14.8 ft . Velocities ( 165 to 755 $\mathrm{ft} / \mathrm{sec}$ ) were also obtained for 194 stone-like objects whose masses varied from 10 to 618 mg .

### 4.16.8 Displacement of Large Stones, Concrete Blocks, and Bricks

Twenty-five stones, two concrete blocks, and two ordinary bricks were placed near each of the seven above-ground stations 2030 to 6120 ft from GZ. The stones in each group of 25 had masses ranging from about 150 g to 20 kg . The purpose of the experiment was to obtain only the total displacement since the large sizes of the missiles prohibited measurement of velocity by the trapping technique. The greatest displacement experienced by any of the objects placed at the nonprecursor stations ( $4 \mathrm{P}, 5 \mathrm{P}, 6 \mathrm{P}$, and 8 P ) was 24 ft ; some of the experimental objects were not moved. Of the 46 stones recovered, which had been placed at the precursor stations ( $10 \mathrm{P}, 15 \mathrm{P}$, and 20 P ), the greatest total distance of displacement measured was 1814 ft and the least was 249 ft . Thus this experiment demonstrated the great difference in translational capability between the precursor and nonprecursor blast waves.

## REFERENCES

1. G. M. McDonnel, H. A. Claypool, W. A. Moncrief, and J. D. Goldstein, Effects of Nuclear Weapons on a Large Biological Specimen (Swine), Project 4.1, Operation Plumbbob Report, ITR-1428, Nov. 5, 1957.
2. V. C. Goldizen, D. R. Richmond, and T. L. Chiffelle, Missile Studies with a Biological Target, Operation Plumbbob Report, WT-1470, Jan. 23, 1961.
3. W. J. Flatau, R. A. Brechenridge, and C. K. Wiehle, Blast Loading and Response of Underground Concrete-Arch Protective Structures (U), Project 3.1, Operation Plumbbob Report, WT-1420. (Classified)
4. G. H. Albright, J. C. Ledoux, and R. A. Mitchell, Evaluation of Buried Conduits as Personnel Shelters, Project 3.2, Operation Plumbbob Report, WT-1421.
5. I. G. Bowen, R. W. Albright, E. R. Fletcher, and C. S. White, A Model Designed to Predict the Motion of Objects Translated by Classical Blast Waves, USAEC Report CEX-58.9, June 29, 1961.
6. R. V. Taborelli, I. G. Bowen, and E. R. Fletcher, Tertiary Effects of Blast-Displacement, Operation Plumbbob Report, WT-1469, May 22, 1959.
7. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determination of Aerodynamic Drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.


Fig. 4.1-Station locations for shot Priscilla in Frenchman Flat, NTS.


Fig. 4.2-Station 4P layout chart. Trap designators for shot Priscilla were identified by a number that indicated anticipated pressure, the letter P (for Priscilla), a number for the trap, and a small letter: "a" for a ground-level trap and " $b$ " for a trap stacked above another.


Fig. 4.3-Dynamic pressure vs. time for station 4P. (Gauge measuring overpressure vs. time failed at this station.)


Fig. 4.4-Traps 4P1a (bottom) and 4P1b (top), placed 7.8 ft behind window, postshot.


Fig. 4.5-Analysis of window-glass missiles from trap 4P1a: $d=7.8 \mathrm{ft} ; \mathrm{n}=68 ; \log \mathrm{v}=2.3099-0.0924 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.24 ; \mathrm{M}_{50}=1230$ $\mathrm{mg} ; \mathrm{V}_{50}=106 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.6-Analysis of window-glass missiles from trap $4 \mathrm{Plb}: \mathrm{d}=7.8 \mathrm{ft} ; \mathrm{n}=58 ; \log \mathrm{v}=2.2937-0.0838 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.23 ; \mathrm{M}_{50}=1150$ mg ; $\mathrm{V}_{50}=109 \mathrm{ft} / \mathrm{sec}$.
$\therefore \quad *$


Fig. 4.7-Trap 4P2b, placed above dog trap (4P2A). Distance to window installation was 12.8 ft . Bottom of missile trap was 31.5 in . above ground level.
Predicted Velocity
Predicted Velocity (Reflected)
Regression Line

4P2b Stondard Error of Estimate
(2) Geometric Means (Moss and Velocity)


Fig. 4.8-Analysis of window-glass missiles from trap 4P2b: $\mathrm{d}=12.8 \mathrm{ft} ; \mathbf{n}=48 ; \log \mathrm{v}=2.2928-0.0836 \log \mathrm{~m} ; \mathrm{Egv}_{\mathrm{g}}=1.23 ; \mathrm{M}_{50}=1470$ $\mathrm{mg} ; \mathrm{V}_{50}=107 \mathrm{ft} / \mathrm{sec}$.

[^12]

Fig. 4.9-Installation 4P3 looking toward GZ, preshot. Trap was above a dog trap, 31.5 in. above ground level and 12.8 ft from the plate-glass installation.


Fig. 4.10-Traps 4P4a and b looking toward GZ, preshot. Note piles of military debris mixed with marked gravel 4.5 and 10.9 ft in front of the traps. Piles on the right side of the picture were placed in front of traps 4P5a and b.


Fig. 4.11-Front surface of traps 4 P 4 a and b , postshot. Slight thermal damage to the absorber was noted.
?


Fig. 4.12-Installations 4P6 (right) and 4P7 (left), preshot. Note packets of spheres on wire supports as well as on the ground. Piles of marked gravel can be seen on the right in front of installation 4 P 6 traps.
-.- - Predicted Velocity
—......- Prédicted Velocity: (Reflected)*
-_-_- Regression Line
4P4b, 4P5a, 4P5b, 4P6a, 4P6b
Grovel ( $d=10.9$ )


Mass, mg

Fig. 4.13-Analysis of gravel missiles from station 4 P traps: $\mathrm{d}=10.9 \mathrm{ft} ; \mathrm{n}=14 ; \log \mathrm{v}=2.5462-0.2183 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=$ $186 \mathrm{mg} ; \mathrm{V}_{50}=112 \mathrm{ft} / \mathrm{sec}$.

4P4b, 4P5a, 4P5b, 4P6a, 4P6D Grovel ( $\delta=28.0$ )

## ———— Predicted Velocity <br> ____ Predicted Velocity (Roflected) ${ }^{*}$ <br> -....-.-. Regression Line Geometric Meons (Moss ond Velocity)

## 



Fig. 4.14-Analysis of gravel missiles from station 4 P traps: $\mathrm{d}=28.0 \mathrm{ft} ; \mathrm{n}=20 ; \log \mathrm{v}=2.6592-0.2785 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=$ $98 \mathrm{mg} ; \mathrm{V}_{50}=127 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.15 - Traps 4 P 8 a and b , placed 17.8 ft behind window, postshot.
$\dot{\theta}$
$\star$


Fig. 4.16-Analysis of window-glass missiles from trap 4P8a: $d=17.8 \mathrm{ft} ; \mathrm{n}=41 ; \log \mathrm{v}=2.3025-0.0864 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=$ $677 \mathrm{mg} ; \mathrm{V}_{50}=114 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.17-Analysis of window-glass missiles from trap 4P8b: $d=17.8 \mathrm{ft} ; \mathrm{n}=54 ; \log \mathrm{v}=2.1823-0.0526 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.34 ; \mathrm{M}_{50}=$ $1040 \mathrm{mg} ; \mathrm{V}_{50}=106 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.18-Trap 4P9b, postshot, placed 12.8 ft from mounted window and above a 27 -in.-high pig trap.



Mass, mg
Fig. 4.19—Analysis of window-glass missiles from trap 4 P9b: $d=12.8 \mathrm{ft} ; \mathrm{n}=62 ; \log \mathrm{v}=2.3052-0.0854 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.24 ; \mathrm{M}_{50}=$ $1240 \mathrm{mg} ; \mathrm{V}_{50}=110 \mathrm{ft} / \mathrm{sec}$.


- Soomentric Means (Mass and Velocity)

4P4a, 4P4b, 4P5a, 4P5b, 4P6b, 4P7b
Natural Stones

103


Fig. 4.20-Analysis of natural-stone missiles from station 4P traps: $n=18 ; \log v=2.5098-0.2019 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.07 ; \mathrm{M}_{50}=28.9 \mathrm{mg} ;$ $\mathrm{V}_{50}=164 \mathrm{ft} / \mathrm{sec}$.
Predicted Velocity
Predicted Velocity (Reflected)


Fig. 4.21-Analysis of window-glass missiles from trap 4PPa: $d=8.8 \mathrm{ft} ; \mathrm{n}=81 ; \log \mathrm{v}=2.3067-0.0779 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.29 ; \mathrm{M}_{50}=1070$ $\mathrm{mg} ; \mathrm{V}_{50}=118 \mathrm{ft} / \mathrm{sec}$.



Fig. 4.22-Analysis of window-glass missiles from trap 4PPb: $d=8.8 \mathrm{ft} ; \mathrm{n}=68 ; \log \mathrm{v}=2.3964-0.1250 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=960$ $\mathrm{mg} ; \mathrm{V}_{50}=106 \mathrm{ft} / \mathrm{sec}$.

*


Fig. 4.24 - Dynamic pressure vs. time for station 5P.


Fig. 4.25-Traps 5P1a and b, postshot.


Fig. 4.26-Analysis of window-glass missiles from trap $5 \mathrm{P} 1 \mathrm{a}: \mathrm{d}=7.8 \mathrm{ft} ; \mathrm{n}=48 ; \log \mathrm{v}=2.3506-0.0948 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=1301$ $\mathrm{mg} ; \mathrm{V}_{50}=114 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.27-Analysis of window-glass missiles from trap 5P1b: $d=7.8 \mathrm{ft} ; \mathrm{n}=32 ; \log \mathrm{v}=2.3596-0.1031 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=734$ $\mathrm{mg} ; \mathrm{V}_{50}=116 \mathrm{ft} / \mathrm{sec}$


Mass, mg
Fig. 4.28-Analysis of window-glass missiles from trap $5 \mathrm{P} 2 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=88 ; \log \mathrm{v}=2.2671-0.0829 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=1000$ $\mathrm{mg} ; \mathrm{V}_{50}=104 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.29-Trap 5P3b, postshot.


Mass, mg
Fig. 4.30-Analysis of plate-glass missiles from trap $5 \mathrm{P} 3 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=9 ; \log \mathrm{v}=2.0396+0.0140 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.22 ; \mathrm{M}_{50}=877$ $\mathrm{mg} ; \mathrm{V}_{50}=121 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.31 - Traps 5P6a and $b$, preshot. Note the piles of gravel (two on the concrete pad 32.0 and 12.5 ft from the traps and one on the ground 4.8 ft from the trap, between the pad and traps).
$*$


Fig. 4.32 - Military-debris and gravel installation 5P4, postshot.


Fig. 4.33-Analysis of gravel missiles from traps 5P4b, 5P5b, and 5P6a and b: $d=12.5 \mathrm{ft} ; \mathrm{n}=38 ; \log \mathrm{v}=2.5578-0.2123 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=$ $1.10 ; \mathrm{M}_{50}=180 \mathrm{mg} ; \mathrm{V}_{50}=120 \mathrm{ft} / \mathrm{sec}$.


Mass, mg

Fig. 4.34-Analysis of gravel missiles from station 5 P traps: $\mathrm{d}=32.0 \mathrm{ft} ; \mathrm{n}=89 ; \log \mathrm{v}=2.6168-0.2240 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=228 \mathrm{mg}$; $\mathrm{V}_{50}=123 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.35 - Traps 5P7a and b , postshot. Note that the thermal damage is greater in the upper trap.


Fig. 4.36 - Traps 5P8a and b, postshot.


Fig. 4.37-Analysis of window-glass missiles from trap $5 \mathrm{P} 8 \mathrm{a}: \mathrm{d}=17.8 \mathrm{ft} ; \mathrm{n}=40 ; \log \mathrm{v}=2.2142-0.0450 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.23 ; \mathrm{M}_{50}=1011$ $\mathrm{mg} ; \mathrm{V}_{50}=120 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.38-Analysis of window-glass missiles from trap 5 P8b: $d=17.8 \mathrm{ft} ; \mathrm{n}=43 ; \log \mathrm{v}=2.0389+0.0012 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.23 ; \mathrm{M}_{50}=$ $881 \mathrm{mg} ; \mathrm{V}_{50}=110 \mathrm{ft} / \mathrm{sec}$.


5P9b

121


Fig. 4.39-Analysis of window-glass missiles from trap $5 \mathrm{P} 9 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=52 ; \log \mathrm{v}=2.3511-0.0935 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.23 ; \mathrm{M}_{50}=$ $895 \mathrm{mg} ; \mathrm{V}_{50}=119 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.40-Analysis of window-glass missiles from trap $5 \mathrm{PPa}: \mathrm{d}=11.7 \mathrm{ft} ; \mathrm{n}=123 ; \log \mathrm{v}=2.2919-0.0772 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=$ $609 \mathrm{mg} ; \mathrm{V}_{50}=119 \mathrm{ft} / \mathrm{sec}$.


Regression Line
Standard Error of Estimate
Q Geometric Means (Mass and Velocity)


Fig. 4.41 -Analysis of window-glass missiles from trap $5 \mathrm{PPb}: \mathrm{d}=11.7 \mathrm{ft} ; \mathrm{n}=158 ; \log \mathrm{v}=2.2809-0.0781 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.22$; $\mathrm{M}_{50}=$ $588 \mathrm{mg} ; \mathrm{V}_{50}=116 \mathrm{ft} / \mathrm{sec}$.

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Fig. 4.43-Overpressure vs. time at station 6P.


Fig. 4.44-Dynamic pressure vs. time at station 6P.


Fig. 4.45-Analysis of window-glass missiles from trap 6P1a: $d=7.8 \mathrm{ft} ; \mathrm{n}=67 ; \log \mathrm{v}=2.1924-0.0294 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=$ $594 \mathrm{mg} ; \mathrm{V}_{50}=129 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.46-Analysis of window-glass missiles from trap $6 \mathrm{P} 1 \mathrm{~b}: \mathrm{d}=7.8 \mathrm{ft} ; \mathrm{n}=41 ; \log \mathrm{v}=2.2258-0.0447 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=$ $981 \mathrm{mg} ; \mathrm{V}_{50}=124 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.47-Analysis of natural-stone missiles from trap 6P1b: $n=10 ; \log v=2.5835-0.2181 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=89.8 \mathrm{mg} ; \mathrm{V}_{50}=$ $144 \mathrm{ft} / \mathrm{sec}$.
-.-.- Predicted Velocity
——————Predicted Velocity (Reflected)*
-_-ー------ Regression Line
Standord Error of Estimate
Q Geometric Means (Mass and Velocity)


Fig. 4.48-Analysis of window-glass missiles from trap $6 \mathrm{P} 2 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=105 ; \log \mathrm{v}=2.2484-0.0762 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18$; $\mathrm{M}_{50}=651$ $\mathrm{mg} ; \mathrm{V}_{50}=108 \mathrm{ft} / \mathrm{sec}$.


Mass, mg
Fig. 4.49-Analysis of natural-stone missiles from trap $6 \mathrm{P} 2 \mathrm{~b}: \mathrm{n}=20 ; \log \mathrm{v}=2.3675-0.1110 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=68.6 \mathrm{mg}$; $V_{50}=146 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.50-Traps 6P3a and b, postshot, 12.8 ft behind plate-glass installation. Note large indentations made in upper trap by pieces of glass which were flat upon arrival at trap.
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Fig. 4.51-Analysis of plate-glass missiles which arrived flat at trap $6 \mathrm{P} 3 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=5 ; \log \mathrm{v}=1.8697+0.0390 \log \mathrm{~m}(\mathrm{mg})$; $\mathrm{E}_{\mathrm{gV}}=1.01$ (thus standard-error-of-estimate lines are almost identical to regression line); $\mathrm{M}_{50}=223,000 \mathrm{mg} ; \mathrm{V}_{50}=120 \mathrm{ft} / \mathrm{sec}$.

$\otimes$ Geometric Means (Mass and Velocity)


Fig. 4.52-Analysis of plate-glass missiles from traps 6 P 3 a and $\mathrm{b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=8 ; \log \mathrm{v}=2.1563-0.0148 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19$; $\mathrm{M}_{50}=2347 \mathrm{mg} ; \mathrm{V}_{50}=128 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.53-Analysis of natural-stone missiles from trap $6 \mathrm{P} 3 \mathrm{a}: \mathrm{n}=19 ; \log \mathrm{v}=2.3752-0.1219 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.11 ; \mathrm{M}_{50}=59.8 \mathrm{mg}$; $\mathrm{V}_{50}=144 \mathrm{ft} / \mathrm{sec}$.
—————— Predicted Velocity
--------. Regression Line
\& Geometric Means (Mass and Velocity)
Natural Stones

Fig. 4.54-Analysis of natural-stone missiles from trap 6P3b: $n=49 ; \log v=2.4344-0.1336 \log m ; E_{g v}=1.12 ; M_{50}=47.7 \mathrm{mg} ; V_{50}=$ $162 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.55 - Military-debris and gravel installation 6P4, postshot.


Mass, mg

Fig. 4.56-Analysis of natural-stone missiles from trap $6 \mathrm{P} 4 \mathrm{~b}: \mathrm{n}=14 ; \log \mathrm{v}=2.5749-0.2138 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=123 \mathrm{mg} ; \mathrm{V}_{50}=$ $134 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.57-Analysis of natural-stone missiles from trap 6P5b: $n=25 ; \log v=2.5385-0.1744 \log m ; E_{g v}=1.13 ; M_{50}=67.4 \mathrm{mg} ; \mathrm{V}_{50}=$ $166 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.58-Traps 6P6a and $b, 6 \mathrm{P} 7 \mathrm{a}$ and b , and 6P8A (right to left), preshot. Spheres on the ground and on wire mounts are on the asphalt pad in front of traps 6P6a and $b$. Piles of gravel are set out in front of other traps. A BRL gauge is shown at the right side of pad.
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Fig. 4.59-Traps 6P6a and b, postshot.


Fig. 4.60-Gravel installation 6P7, behind asphalt area, postshot.


Fig. 4.61-Analysis of natural-stone missiles from trap 6P6a: $n=31 ; \log v=2.5957-0.2306 \log m ; E_{g v}=1.10 ; M_{50}=39.0 \mathrm{mg}$ $V_{50}=169 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.62-Analysis of natural-stone missiles from trap $6 \mathrm{P} 6 \mathrm{~b}: \mathrm{n}=58 ; \log \mathrm{v}=2.5897-0.2192 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=49.0 \mathrm{mg} ; \mathrm{V}_{50}=$ $166 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.63-Analysis of gravel missiles from station 6 P traps: $\mathrm{d}=14.0 \mathrm{ft} ; \mathrm{n}=20 ; \log \mathrm{v}=2.4110-0.1257 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}} \because 1.11 ; \mathrm{M}_{50}=$ $147 \mathrm{mg} ; \mathrm{V}_{50}$ ■ $138 \mathrm{ft} / \mathrm{sec}$.
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Fig. 4.64-Analysis of gravel missiles fron station 6 P traps: $\mathrm{d}=36.0 \mathrm{ft} ; \mathrm{n}=9 ; \log \mathrm{v}=2.6117-0.2311 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=$ $67.7 \mathrm{mg} ; \mathrm{V}_{50}=154 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.65 - Traps 6P9a and b, postshot, 22.8 ft behind window-glass mount.
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Fig. 4.66-Analysis of window-glass missiles from trap $6 \mathrm{P} 9 \mathrm{a}: \mathrm{d}=22.8 \mathrm{ft} ; \mathrm{n}=178 ; \log \mathrm{v}=2.2053-0.0446 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=$ $419 \mathrm{mg} ; \mathrm{V}_{50}=123 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.67-Analysis of window-glass missiles from trap $6 \mathrm{P} 9 \mathrm{~b}: \mathrm{d}=22.8 \mathrm{ft} ; \mathrm{n}=161 ; \log \mathrm{v}=2.2215-0.0368 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=$ $541 \mathrm{mg} ; \mathrm{V}_{50}=132 \mathrm{ft} / \mathrm{sec}$.
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Fig. 4.68-Analysis of natural-stone missiles from trap $6 \mathrm{P} 9 \mathrm{~b}: \mathrm{n}=39 ; \log \mathrm{v}=2.4510-0.1338 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.09 ; \mathrm{M}_{50}=62.1 \mathrm{mg} ; \mathrm{V}_{50}=$ $163 \mathrm{ft} / \mathrm{sec}$


Fig. 4.69 - Trap 6P10b, postshot, 12.8 ft behind window-glass mount and above a 27 -in.-high pig trap. Outline of pig is visible in lower trap.

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b

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| $\ldots$ | Predicted Velocity |
| :--- | :--- |
| $\ldots$ | $=-\quad$Predicted Velocity (Reflected) |
| Regression Line |  |

_-_-_---_ Standard Error of Estimote
* Geometric Meons (Mass and Velocity)


Mass, mg
Fig. 4.70—Analysis of window-glass missiles from trap 6P10b: $d=12.8 \mathrm{ft} ; \mathrm{n}: 32 ; \log \mathrm{v}=2.1360-0.0310 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=$ $1010 \mathrm{mg} ; \mathrm{V}_{50}=110 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.71-Analysis of natural-stone missiles from trap 6P10b: $n=10 ; \log v=2.4627-0.1615 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.11 ; \mathrm{M}_{50}=117 \mathrm{mg} ;$ $V_{50}=135 \mathrm{ft} / \mathrm{sec}$.


6PPo
Window Glass

| -. | Predicled Velocity |
| :---: | :---: |
|  | Predicted Velocity (Reflected)* |
|  | Regression Line |
|  | Standard Error of Estimate |
| $\otimes$ | Geometric Means (Mass and Velocity) |



Fig. 4.72-Analysis of window-glass missiles from trap 6PPa: $\mathrm{d}=16.0 \mathrm{ft} ; \mathrm{n}=170 ; \log \mathrm{v}=2.2534-0.0534 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.22 ; \mathrm{M}_{50}=$ $478 \mathrm{mg} ; \mathrm{V}_{50}=129 \mathrm{ft} / \mathrm{sec}$.

6PPb
Window Glass


Fig. 4.73-Analysis of window-glass missiles from trap $6 \mathrm{PPb}: \mathrm{d}=16.0 \mathrm{ft} ; \mathrm{n}=390 ; \log \mathrm{v}=2.3090-0.0678 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{\mathbf{5 0}}=$ $292 \mathrm{mg} ; \mathrm{V}_{50}=138 \mathrm{ft} / \mathrm{sec}$.
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Fig. 4.74-Traps 6.7PPa and b, postshot.


Fig. 4.75-Analysis of window-glass missiles from trap $6.7 \mathrm{PPa}: \mathrm{d}=18.0 \mathrm{ft} ; \mathrm{n}=112 ; \log \mathrm{v}=2.1445-0.0086 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16$; $M_{50}=430 \mathrm{mg} ; V_{50}=132 \mathrm{ft} / \mathrm{sec}$.
 Standard Error of Estimate Geometric Means (Mass and Velocity)
6.7PPb

Window Glass


Mass, mg
Fig. 4.76-Analysis of window-glass missiles from trap 6.7PPb: $d=18.0 \mathrm{ft} ; \mathrm{n}=194 ; \log \mathrm{v}=2.2698-0.0281 \log \mathrm{mg} ; \mathrm{E}_{\mathrm{gv}}=1.19$; $\mathrm{M}_{50}=344 \mathrm{mg} ; \mathrm{V}_{50}=142 \mathrm{ft} / \mathrm{sec}$.



Fig. 4.78-Station 8P, preshot.


Fig. 4.79—Overpressure vs. time at station 8 P.


Fig. 4.80 - Dynamic pressure vs. time at station 8P.


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Fig. 4.81-Traps 8P1a and b, postshot.
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Fig. 4.82-Analysis of window-glass missiles from trap 8P1a: $\mathrm{d}=7.8 \mathrm{ft} ; \mathrm{n}=103 ; \log \mathrm{v}=2.2463-0.0410 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=$ $245 \mathrm{mg} ; \mathrm{V}_{50}=141 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.83-Analysis of window-glass missiles from trap 8P1b: $d=7.8 \mathrm{ft} ; \mathrm{n}=100 ; \log \mathrm{v}=2.2032-0.0276 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19$; $\mathrm{M}_{50}=$ $415 \mathrm{mg} ; \mathrm{V}_{50}=135 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.84-Analysis of window-glass missiles from trap $8 \mathrm{P} 2 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=497 ; \log \mathrm{v}=2.2647-0.0443 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gy}}=1.18 ; \mathrm{M}_{50}=$ $229 \mathrm{mg} ; \mathrm{V}_{50}=145 \mathrm{ft} / \mathrm{sec}$.


Mass, mg
Fig. 4.85-Analysis of natural-stone missiles from trap $8 \mathrm{P} 2 \mathrm{~b}: \mathrm{n}=25 ; \log \mathrm{v}=2.4717-0.1466 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.11 ; \mathrm{M}_{50}=43.6 \mathrm{mg} ; \mathrm{V}_{50}=$ $170 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.86 - Traps 8 P 3 a and b , postshot. Shadow of plate-glass frame, 12.8 ft from trap, can be seen. Note large indentations made in the absorber by glass that arrived flat and the dust and dirt on sur-
face of white absorber.


Fig. 4.87-Analysis of plate-glass missiles from trap $8 \mathrm{P} 3 \mathrm{a}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=25 ; \log \mathrm{v}=2.3450-0.0954 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=$ $1778 \mathrm{mg} ; \mathrm{V}_{50}=108 \mathrm{ft} / \mathrm{sec}$.
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Fig. 4.88-Analysis of plate-glass missiles from trap $8 \mathrm{P} 3 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=33 ; \log \mathrm{v}=2.2992-0.0694 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=$ $723 \mathrm{mg} ; \mathrm{V}_{50}=126 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.89-Analysis of plate-glass missiles that arrived flat at trap $8 \mathrm{P} 3 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=7 ; \log \mathrm{v}=1.0918+0.0568 \log \mathrm{~m}(\mathrm{mg})$; $\mathrm{E}_{\mathrm{gV}}=1.06 ; \mathrm{M}_{50}=123,000 \mathrm{mg} ; \mathrm{V}_{50}=155 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.90 - Traps 8P5a and b, postshot. Gravel and military debris were set out in front of the traps at three locations. Surface of absorber was somewhat damaged by thermal energy from the bomb.

8P4a Stondard Error of Estimote
\& Geometric Means (Moss and Velocity)


Fig. 4.91-Analysis of gravel missiles from trap $8 \mathrm{P} 4 \mathrm{a}: \mathrm{d}=16.8 \mathrm{ft} ; \mathrm{n}=23 ; \log \mathrm{v}=2.5226-0.1159 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=146 \mathrm{mg}$; $V_{50}=187 \mathrm{ft} / \mathrm{sec}$.



Fig. 4.92-Analysis of gravel missiles from trap $8 P 4 b: d=16.8 \mathrm{ft} ; \mathrm{n}=103 ; \log \mathrm{v}=2.6573-0.1701 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.11$; $\mathrm{M}_{50}=157 \mathrm{mg}$; $\mathrm{V}_{50}=192 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.93-Analysis of gravel missiles from trap $8 \mathrm{P} 5 \mathrm{a}: \mathrm{d}=6.5 \mathrm{ft} ; \mathrm{n}=8 ; \log \mathrm{v}=2.5909-0.1486 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=183 \mathrm{mg} ; \mathrm{V}_{50}=$ $180 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.94-Analysis of gravel missiles from trap $8 \mathrm{P} 5 \mathrm{~b}: \mathrm{d}=6.5 \mathrm{ft} ; \mathrm{n}=20 ; \log \mathrm{v}=2.6573-0.1801 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{50}=165 \mathrm{mg}$; $\mathrm{V}_{50}=181 \mathrm{ft} / \mathrm{sec}$.


8P5o
Gravel ( $\mathrm{d}=16.8$ )


Mass, mg
Fig, 4.95 -Analysis of gravel missiles from trap $8 \mathrm{P} 5 \mathrm{a}: \mathrm{d}=16.8 \mathrm{ft} ; \mathrm{n}=8 ; \log \mathrm{v}=2.9724-0.3212 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.09 ; \mathrm{M}_{50}=152 \mathrm{mg} ;$ $\mathrm{V}_{50}=187 \mathrm{ft} / \mathrm{sec}$.
-.....- Predicted Velocity (Reflected)*
_-_-_-_-_ Regression Line
Standard Error of Estimol
Q Geometric Means (Mass and Velocity)


Fig. 4.96-Analysis of gravel missiles from trap $8 P 5 b: d=16.8 \mathrm{ft} ; \mathrm{n}=10 ; \log \mathrm{v}=2.6275-0.1571 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.11 ; \mathrm{M}_{50}=128 \mathrm{mg}$; $\mathrm{V}_{50}=198 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.97-Analysis of gravel missiles from trap $8 P 5 b: d=43.0 \mathrm{ft}, \mathrm{n}=16 ; \log \mathrm{v}=2.4748-0.0661 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{50}=178 \mathrm{mg}$; $\mathrm{V}_{50}=212 \mathrm{ft} / \mathrm{sec}$.
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Fig. 4.98-Sphere installation 8P6, postshot. (Behind asphalt area.)


8P6b
Natural Stones


Mass, mg
Fig. 4.99-Analysis of natural-stone missiles from trap $8 \mathrm{P} 6 \mathrm{~b}: \mathrm{n}=10 ; \log \mathrm{v}=2.6403-0.1456 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.07 ; \mathrm{M}_{50}=40.9 \mathrm{mg} ;$ $V_{50}=254 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.100-Analysis of gravel missiles from trap 8P7a: $d=16.8 \mathrm{ft} ; \mathrm{n}=7 ; \log \mathrm{v}=2.6117-0.1564 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=324 \mathrm{mg} ;$ $\mathrm{V}_{50}=166 \mathrm{ft} / \mathrm{sec}$.


Slondord Error of Estimose
Grovel ( $d=16.8$ )


Fig. 4.101-Analysis of gravel missiles from trap $8 \mathrm{P} 7 \mathrm{~b}: \mathrm{d}=16.8 \mathrm{ft} ; \mathrm{n}=60 ; \log \mathrm{v}=2.5813-0.1536 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{50}=239 \mathrm{mg} ;$ $V_{50}=164 \mathrm{ft} / \mathrm{sec}$.

## —.-.- Predicted Velocity <br> - Predicted Velocity (Raflected) <br>  <br> -------- Stondord Error of Extimate <br> Geometric Mpans (Mass and Velocity)



Mass, mg

Fig. 4.102-Analysis of gravel missiles from trap $8 \mathrm{P} 7 \mathrm{~b}: \mathrm{d}=43.0 \mathrm{ft} ; \mathrm{n}=14 ; \log \mathrm{v}=2.6552-0.1639 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.09 ; \mathrm{M}_{50}=363 \mathrm{mg} ;$ $V_{50}=172 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.103-Analysis of window-glass missiles from trap $8 \mathrm{P9a}: \mathrm{d}=22.8 \mathrm{ft} ; \mathrm{n}=180 ; \log \mathrm{v}=2.2339-0.0191 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=$ $318 \mathrm{mg} ; \mathrm{V}_{50}=154 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.104-Analysis of window-glass missiles from trap 8P9b: $d=22.8 \mathrm{ft} ; \mathrm{n}=129 ; \log \mathrm{v}=2.2433-0.0143 \mathrm{log} \mathrm{m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=$ $403 \mathrm{mg} ; \mathrm{V}_{50}=161 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.105-Analysis of window-glass missiles from trap $8 \mathrm{P} 10 \mathrm{~b}: \mathrm{d}=12.8 \mathrm{ft} ; \mathrm{n}=204 ; \log \mathrm{v}=2.3216-0.0470 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=$ $302 \mathrm{mg} ; \mathrm{V}_{50}=160 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.106-Analysis of natural-stone missiles from trap $8 P 10 b: n=20 ; \log v=2.4586-0.1099 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=79.1 \mathrm{mg} ;$ $V_{50}=178 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.107-Station 10P layout chart.
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Fig. 4.109—Dynamic pressure vs. time at station 10P.
$\square$


Fig. 4.110 - Traps 10Pla and b, postshot.

IOPIo


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Fig. 4.111-Analysis of gravel missiles from trap 10P1a: $d=19.1 \mathrm{ft} ; \mathrm{n}=18 ; \log \mathrm{v}=2.2904+0.0290 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.26 ; \mathrm{M}_{50}=305 \mathrm{mg}$; $\mathrm{V}_{50}=230 \mathrm{ft} / \mathrm{sec}$.



Fig. 4.112-Analysis of gravel missiles from trap 10P1a: $d=49.0 \mathrm{ft} ; \mathrm{n}=22 ; \log \mathrm{v}=2.3316+0.0539 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.22 ; \mathrm{M}_{50}=322 \mathrm{mg}$; $\mathrm{V}_{50}=293 \mathrm{ft} / \mathrm{sec}$.

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Fig. 4.113-Analysis of natural-stone missiles from trap 10Pla: $n=44 ; \log v=2.6137-0.0370 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.43 ; \mathrm{M}_{50}=1185 \mathrm{mg}$; $\mathrm{V}_{50}=316 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.114-Installation 10P2, postshot.

| —————— <br> $-\cdots \quad$ | Predicted Velocity <br> Predicted Velocity (Reflected)* <br> Regression Line <br> Standard Error of Estimate <br> Geormefric Means (Moss and Velocity) |
| :---: | :---: |
|  |  |
|  |  |
| . | $-100$ |
| 1 |  |
|  | *Not Applicable $\quad 50,000$ |

Mass, mg

Fig. 4.115-Analysis of gravel missiles from installation 10P2: $d=19.1 \mathrm{ft} ; \mathrm{n}=31 ; \log \mathrm{v}=2.7155-0.0512 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=$ $151 \mathrm{mg} ; \mathrm{V}_{50}=402 \mathrm{ft} / \mathrm{sec}$.
————— Predicted Velocity
——..-..- Predicted Velociry (Reflected) ${ }^{\text {* }}$
___-_-_-_ Regression Line Stondord Error of Estimote
Q Geomefric Means (Mass and Velocity)


Fig. 4.116-Analysis of gravel missiles from installation 10P2: $d=49.0 \mathrm{ft} ; \mathrm{n}=48 ; \log \mathrm{v}=2.6245-0.0068 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=$ $148 \mathrm{mg} ; \mathrm{V}_{50}=407 \mathrm{ft} / \mathrm{sec}$.
—.-. - Predicted velocity
——.-..- Predicted Velocity (Refiected) ${ }^{\text {* }}$
_ Regression Line
Standard Error of Estimote

- Geometric Means (Mass and Velocity)


Mass, mg
Fig. 4.117—Analysis of natural-stone missiles from installation $10 \mathrm{P} 2: \mathrm{n}=186 ; \log \mathrm{v}=2.6949-0.0312 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=$ $161 \mathrm{mg} ; \mathrm{V}_{50}=423 \mathrm{ft} / \mathrm{sec}$.
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-


Mass, g
Fig. 4.118-Analysis of military-debris missiles from trap 10 P 1 a and installation 10P2. Data for the missile that penetrated plywood are not included in analysis: $d=49.0 \mathrm{ft} ; \mathrm{n}=11 ; \log \mathrm{v}=2.9824-0.1345 \log \mathrm{~m}(\mathrm{mg}) ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=31.06 \mathrm{~g} ; \mathrm{V}_{50}=239 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.119 - Installation 10P3, postshot.


Fig. 4.120-Installation 10P3 and traps 10P4a and b, postshot.

_-_-_-_-_-_ Regression Line
Standard Euror of Estimate
Q Geometric Means (Mass and velocity)
IOP3
Grovel $\quad(d=49.0)$


Mass, mg
Fig. 4.121—Analysis of gravel missiles from installation 10P3: $d \div 49.0 \mathrm{ft} ; \mathrm{n}=78 ; \log \mathrm{v}=2.7511-0.0522 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=$ $195 \mathrm{mg} ; \mathrm{V}_{50}=428 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.122-Analysis of natural-stone missiles from installation $10 \mathrm{P} 3: \mathrm{n}=226 ; \log \mathrm{v}=2.6881-0.0217 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=158 \mathrm{mg} ;$ $\mathrm{V}_{50}=438 \mathrm{ft} / \mathrm{sec}$.
O
——.-. - Predicted velocity
-.....- Predicted Velocity (Reflected)*
-....-.-.- Regression Line
Stondard Error of Estimote
Q Geometric Means (Mass ond Velocity)


Mass, mg
Fig. 4.123-Analysis of gravel missiles from trap 10P4a: $d=49.0 \mathrm{ft} ; \mathrm{n}=66 ; \log \mathrm{v}=2.6338-0.0017 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=190$ $\mathrm{mg} ; \mathrm{V}_{50}=427 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.124-Analysis of natural-stone missiles from trap 10P4a: $n=96 ; \log \mathrm{v}=2.7112-0.0371 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.17 ; \mathrm{M}_{50}=188 \mathrm{mg}$; $\mathrm{V}_{50}=424 \mathrm{ft} / \mathrm{sec}$.

15 P STATION, RANGE 2280
MD Military Debris
(B) blue
Gr Gravel
(W) white
S Spheres
(Y) yellow


Fig. 4.125-Station 15P layout chart

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Fig. 4.127-Dynamic pressure vs. time at station 15P.


Fig. 4.128-Installations 15P1 (left) and 15P2, preshot. GZ is in background with balloon that carried nuclear device aloft. Instal-
lation 15P2 trap is anchored with sand bags.


Fig. 4.129 - Installation 15P2, postshot. Trap is almost buried by native soil.


Mass, mg
Fig. 4.130—Analysis of gravel missiles from installation 15P2: $d=9.4 \mathrm{ft} ; \mathrm{n}=16 ; \log \mathrm{v}=2.6707-0.0184 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=177$ $\mathrm{mg} ; \mathrm{V}_{50}=426 \mathrm{ft} / \mathrm{sec}$.

-     - 一 Predicled Velocity, $d=62.0$
——— Predicted Velocity, $d=24.2$ ------------.- Predicted Velocity, $\sigma=9.4$ - Regression Line
- Geometric Meons (Mass ond Velocity)
$d=62.0 \cdot, d=24.24, d=9.4{ }^{\circ}$ No


Mass, gm
Fig. 4.131-Analysis of military-debris missiles from installation 15P2. Note three predicted velocity lines: $\mathrm{d}=9.4$, 24.2 , and 62.0 $\mathrm{ft} ; \mathrm{n}=10 ; \log \mathrm{v}=2.5651-0.1196 \log \mathrm{~m}(\mathrm{mg}) ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=34.5 \mathrm{~g} ; \mathrm{V}_{50}=240 \mathrm{ft} / \mathrm{sec}$.
$\qquad$


Mass, mg
Fig. 4.132-Analysis of natural-stone missiles from installation 15P2: $\mathrm{n}=273 ; \log \mathrm{v}=2.6884-0.0229 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=208$ $\mathrm{mg} ; \mathrm{V}_{50}=432 \mathrm{ft} / \mathrm{sec}$. Two large stones that had masses of 9,307 and $19,125 \mathrm{mg}$ passed through the balsa absorber and penetrated the plywood back of the trap. The minimum velocities that they would have had were 458 and $389 \mathrm{ft} / \mathrm{sec}$, respectively. These data were not included in the above analysis.


Fig. 4.133-Stabilized area at station 15P. Gravel installation 15P3 appears on the right and sphere installation 15P4 on the left.


Fig. 4.134—Installations 15P3 and 15P4, preshot.


Fig. 4.135-Installation 15P3, postshot. Installation was destroyed.


Fig. 4.136-Installation 15 P 4 , postshot.


Fig. 4.137-Analysis of gravel missiles from installation 15P4: $d=62.0 \mathrm{ft} ; \mathrm{n}=20 ; \log \mathrm{v}=2.8394-0.0967 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=$ $223 \mathrm{mg} ; \mathrm{V}_{50}=409 \mathrm{ft} / \mathrm{sec}$.


15P4


Mass, mg
Fig. 4.138-Analysis of natural-stone missiles from installation 15P4: $\mathrm{n}=232 ; \log \mathrm{v}=2.7026-0.0361 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{50}=264$ $\mathrm{mg} ; \mathrm{V}_{50}=412 \mathrm{ft} / \mathrm{sec}$.

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20 P STATION, RANGE 2030


Fig. 4.139-Station 20P layout chart.

$\qquad$ - -


Fig. 4.141-Dynamic pressure vs. time at station 20P.


Fig. 4.142-Installations 20P1 and 20P2, preshot, looking toward GZ.
\$


Fig. 4.143-Destroyed installation 20P1, postshot.


Fig. 4.144-Destroyed installation 20P2, postshot.


Fig. 4.145-Station 20P, preshot, showing stabilized area and installations 20P3 and 20P4 with spheres and gravel set out.


Fig. 4.146-Station 20P, stabilized area, postshot.


Fig. 4.147-Installation 20P3, postshot.


Fig. 4.148-Analysis of natural-stone missiles from installation $20 \mathrm{P} 3: \mathrm{n}=88 ; \log \mathrm{v}=2.8412-0.0659 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=521 \mathrm{mg}$; $V_{50}=459 \mathrm{ft} / \mathrm{sec}$. The minimum velocities are plotted for the 11 missiles that penetrated through the absorber into the plywood back of the trap housing. The masses and velocities for those missiles are not included in the analysis.


Fig. 4.149-Installation 20P4, postshot (not usable due to erosion).


Fig. 4.150-Station OPS layout chart.


Fig. 4.151-Station OPS, preshot, showing spheres and three croquet balls in foil bags. Photograph was taken near the absorbing material looking toward the entrance.


Fig. 4.152-Station OPS, postshot, showing places marked on absorber where spheres and croquet ball impacted. Numbers and letters by impressions correspond to those in Fig. 4.153.


Fig. 4.153-Drawing for station OPS, showing placement of spheres before shot and places on trap surface where spheres impacted. Letters and numbers by recovered spheres correspond with those in Fig. 4.152.


Fig. 4.154-Analysis of natural-stone missiles from station OPS: $\mathrm{n}=194 ; \log \mathrm{v}=2.6493-0.0506 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.35 ; \mathrm{M}_{50}=28.8 \mathrm{mg} ;$ $\mathrm{V}_{50}=376 \mathrm{ft} / \mathrm{sec}$.


Fig. 4.155-Spatial distribution of natural-stone missiles recovered from station OPS. Numbers indicate missiles per square foot.


Fig. 4.156-Spatial distribution of the average masses (in mg ) of natural-stone missiles recovered from station OPS. The average mass of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 4.157 - Spatial distribution of the average velocities (in $\mathrm{ft} / \mathrm{sec}$ ) of natural-stone missiles recovered from station OPS. The average velocity of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 4.158-Typical trap installation in arch type shelters. See Ref. 3 for details of shelter construction.


Fig. 4.159 - Typical installation of missile absorber in conduit type shelters. See Ref. 4 for details of shelter construction.


Fig. 4.160-Station 4P and large stones, preshot.


Fig. 4.161 - Station 4P and large stones, postshot.


Fig. 4.162-Mass vs. distance for large stones displaced at station 4P: range $=6120 \mathrm{ft} ; \mathrm{n}=24 ; \log \mathrm{m}=0.9616-0.2579 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=1.68 ; \mathrm{M}_{50}$ * $2.357 \mathrm{~kg} ; \overline{\mathrm{d}}=2.29 \mathrm{ft}$.


Fig. 4.163-Mass vs. distance for large stones displaced at station 5 P: range $=5320 \mathrm{ft} ; \mathrm{n}=25 ; \log \mathrm{m}=0.5682-0.1956 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=3.27 ; \mathrm{M}_{50}=$ $2.139 \mathrm{~kg} ; \overline{\mathrm{d}}=1.22 \mathrm{ft}$.


Fig. 4.164-Mass vs. distance for large stones displaced at station 6 P : range $=4770 \mathrm{ft} ; \mathrm{n}=25 ; \log \mathrm{m}=0.5475-0.1817 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=3.16 ; \mathrm{M}_{50}=$ $2.180 \mathrm{~kg} ; \overline{\mathrm{d}}=1.15 \mathrm{ft}$.


Fig. 4.165- Mass vs. distance for large stones displaced at station $8 \mathrm{P}:$ range $=3930 \mathrm{ft} ; \mathrm{n}=25 ; \log \mathrm{m}=0.7674-0.0573 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=2.78 ; \mathrm{M}_{50}=$ $2.178 \mathrm{~kg} ; \overline{\mathrm{d}}=7.50 \mathrm{ft}$.


Fig. 4.166—Mass vs. distance for large stones displaced at station $10 \mathrm{P}:$ range $=2730 \mathrm{ft} ; \mathrm{n}=16 ; \log \mathrm{m}=0.5872-0.0001 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=3.3 \mathrm{C}$; $\mathrm{M}_{50}=3.105 \mathrm{~kg} ; \overline{\mathrm{d}}=739 \mathrm{ft}$.


Fig. 4.167 -Mass vs. distance for large stones displaced at station 15P: range $=2280 \mathrm{ft} ; \mathrm{n}=6 ; \log \mathrm{m}=0.1490+0.0004 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=2.93$; $\mathrm{M}_{50}=5.296 \mathrm{~kg} ; \overrightarrow{\mathrm{d}}=1367 \mathrm{ft}$.


Fig. 4.168-Mass vs. distance for large stones displaced at station 20 P: range $=2030 \mathrm{ft} ; \mathrm{n}=24 ; \log \mathrm{m}=0.3504+0.000009 \mathrm{~d} ; \mathrm{E}_{\mathrm{gm}}=3.06$; $\mathrm{M}_{50}=2.293 \mathrm{~kg} ; \mathrm{d}=1094 \mathrm{ft}$.


Fig. 4.169-Displacement of marked large stones for seven stations, shot Priscilla.

## Chapter 5

## SHOT SMOKY, EXPERIMENTAL PROCEDURE AND RESULTS

### 5.1 GENERAL

The primary purpose for participation in shot Smoky* was to determine the effect of hill-and-dale terrain upon the translation of native (or natural) stones, steel spheres, and military debris. All experiments were made in open areas at ranges of 2548 to 5680 ft , where the measured overpressures varied from about 13 to 5 psi . The yield estimated for this shot, on the basis of certain blast parameters, was 44.5 kt (see Table 5.1).

Locations of the nine stations used in this shot are shown in Fig. 5.1: two on flat terrain on the south blast line, three on hills and three in dales on the northeast line, and one in a dale on the north line. Two trap bases were installed at each station, one base for a single trap and the other for two traps.

A total of 405 steel spheres with diameters of $7 / 16,1 / 2$, and $9 / 16$ in. was placed at various distances in front of the traps. In addition, a total of about 3850 pieces of military debris whose masses varied from approximately 1 to 1000 g was set out. Figure 5.2 illustrates a typical placement of debris and spheres. The spheres were placed a short distance above ground level in a shallow trough supported by $1 / 8-\mathrm{in}$. steel rods.

Experience in shot Priscilla indicated that under certain conditions additional shielding was necessary to protect the absorbing material against thermal radiation. This was accomplished, as illustrated in Fig. 5.3, by mounting 0.0015 -in.-thick aluminum foil on a wooden frame about 1 ft in front of the face of the trap. The foil was ruptured and blown aside by the blast, and therefore it presented no obstruction to the missiles striking the traps.

The northeast, south, and north blast lines are discussed in Secs. 5.2, 5.3, and 5.4, respectively. In each section the material pertaining to the terrain of the blast line, along with a discussion of the effects of the terrain on the blast wave, is followed by a station-by-station presentation of the blast-wave and missile data.

A summary of the blast parameters for all stations used in this shot is presented in Table 5.1 (two extra stations are included where there were no missile studies). An explanation of the various parameters tabulated is included in the table. However, the reader is reminded that the computed value of peak overpressure, $\left(p_{s}\right)_{c}$, was obtained by finding the classical (or ideal) wave whose impulse and duration were equal to those values measured by the BRL gauges. ${ }^{1}$ The difference between computed and measured values of overpressure is a rough measure of the nonconformity of the measured wave to an ideal one. This point will be made much clearer upon examination of the overpressure vs. time curves to be presented later in the sections that describe each of the nine stations.

Only 2 of the approximately 3850 pieces of military debris placed in front of the Smoky traps were recovered. Five of 405 steel spheres were recovered. Data pertaining to these 7 objects are presented at the bottom of Table 5.2. Velocities and masses were determined for

[^13]2876 natural-stone missiles caught in the traps placed in this shot. Plots of these data, by trap, will be found in the sections that describe each station. In addition, statistical parameters* for all recovered missiles are summarized in Table 5.2. It should be noted in particular that Table 5.2 also contains the results of a statistical analysis of the data for natural stones combined from all three traps located at each of the nine stations. A mass vs. velocity plot was not made of the combined data at each station.

TABLE 5.1-BLAST PARAMETERS, SHOT SMOKY
(See List of Symbols.)

|  | $\mathrm{p}_{0}=12.4 \mathrm{psi}$ |  | $\mathrm{c}_{0}=1118 \mathrm{ft} / \mathrm{sec}\left(15.2^{\circ} \mathrm{C}\right)$ |  | Estimated yield: $44.5 \mathrm{kt}^{(1)}$ |  |  | Terrain: hill, dale, and flat |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| Station | Range, ft | Blast line | Terrain | $\begin{aligned} & \left(I_{p}\right)_{m}{ }^{(2)} \\ & \text { psi-sec } \end{aligned}$ | $\begin{aligned} & \left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{r}},^{(3)} \\ & \mathrm{pssi}-\mathrm{sec} \end{aligned}$ | $\begin{gathered} \left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{m}}, \\ \mathrm{sec} \end{gathered}$ | $\begin{gathered} \left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{r}}{ }^{(4)} \\ \mathrm{sec} \end{gathered}$ | $\underset{\text { psi }}{\left(p_{s}\right)_{m},}$ | $\begin{gathered} \left(\mathrm{p}_{\mathrm{s}}\right)_{\mathrm{c}},{ }^{(5)} \\ \mathrm{psi} \end{gathered}$ | $\begin{gathered} \left(p_{s}\right)_{r},{ }^{(6)} \\ \mathrm{psi} \end{gathered}$ |
| 9S | 5680 | S | Flat | 2.049 | 2.010 | 1.118 | 1.081 | 5.1 | 4.80 | 4.84 |
| 8 S | 4980 | NE | Dale | 2.280 | 2.257 | 0.984 | 1.014 | 5.0 | 6.20 | 5.92 |
|  | 4155 | S | Flat | 2.671 | 2.645 | 0.929 | 0.929 | 6.5 | 7.90 | 7.82 |
| 7 S | 4115 | NE | Hill | 2.480 | 2.668 | 0.932 | 0.925 | 7.4 | 7.25 | 7.93 |
|  | 3875 | S | Flat | 2.840 | 2.813 | 0.904 | 0.899 | 7.4 | 8.70 | 8.70 |
| 6 S | 3739 | NE | Dale | 3.014 | 2.902 | 0.750 | 0.883 | 7.7 | 11.5 | 9.19 |
| 5 S | 3722 | N | Dale | 2.883 | 2.914 | 0.972 | 0.881 | 6.3 | 8.18 | 9.25 |
| 4 S | 3406 | S | Flat | 3.113 | 3.150 | 0.868 | 0.844 | 6.9 | 10.1 | 10.6 |
| 3 S | 3218 | NE | Hill | 3.071 | 3.311 | 0.839 | 0.821 | 8.5 | 10.4 | 11.6 |
| 2 S | 2914 | NE | Dale | 4.024 | 3.612 | 0.793 | 0.783 | 11.5 | 15.0 | 13.5 |
| 1S | 2548 | NE | Hill | 3.962 | 4.064 | 0.728 | 0.734 | 13.1 | 16.4 | 16.5 |

${ }^{(1)}$ Estimation made by comparing overpressure impulse data measured for stations 8 S and 9 S with data for a surface burst described in The Effects of Nuclear Weapons.
${ }^{(2)}$ Determined from BRL mechanical-gauge records.
${ }^{(3)}$ Overpressure impulse computed by regression equation derived from $\left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{m}}$ values
$\log \left(\mathrm{L}_{\mathrm{p}}\right)_{\mathrm{r}}=3.5982-0.8776 \log \mathrm{R}$
${ }^{(4)}$ Overpressure duration computed by regression equation derived from ( $\left.\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{m}}^{\prime}$ values

$$
\log \left(t_{p}^{t}\right)_{\mathrm{r}}=-1.7792+0.4829 \log R
$$

${ }^{(5)}$ Peak overpressure computed for a classical blast wave of impulse ( $\mathrm{L}_{\mathrm{p}}$ ) m and of duration $\left(\mathrm{t}_{\mathrm{p}}^{+}\right) \mathrm{m}$.
${ }^{(6)}$ Peak overpressure computed by regression equation derived from $\left(p_{s}\right)_{c}$ values
$\log \left(p_{s}\right)_{r}=6.4370-1.5321 \log R$

### 5.2 NORTHEAST BLAST LINE

### 5.2.1 Terrain Effects

A profile of the entire northeast blast line is shown in Fig. 5.4. This chart indicates generally rising land from GZ to the first station (1S). The terrain remains fairly high and hilly until the last station ( 8 S ) is reached. This station is on much lower ground and is almost out of the line of sight to the point of detonation.

Selected blast parameters taken from Table 5.1 are also plotted in Fig. 5.4. Overpressure values, $p_{s}$, particularly computed ones, $\left(p_{s}\right)_{c}$, show a marked tendency to be low at the hill stations and high in the dales when compared with the average or regression values. There are no marked deviations of the duration values from the regression line except for station 6S. Here the low measured duration is reflected in a computed maximum overpressure that is particularly high. $\dagger$

Figures 5.5 and 5.6 (similar to Fig. 5.4) were made to a larger scale to show in more detail the positions of the traps and gauges in relation to the hills and dales. The same blast parameters shown in Fig. 5.4 are plotted on these charts.

[^14]Regression Equation: $\log v=a+b \log m$

${ }^{(1)}$ Combination NS from $1 \mathrm{~S} 1 \mathrm{a}, \mathrm{b}$; 1 S 2.
${ }^{(2)}$ Combination NS from 2S1; 2S2a,b.
${ }^{(3)}$ Combination NS from 3S1a, b; 3 S 2 .
${ }^{(4)}$ Combination NS from 4S1; 4S2a,b.
${ }^{(5)}$ Combination NS from 5S1a,b; $5 S 2$.
${ }^{(6)}$ Combination NS from 6S 1a,b; $6 S 2$.
(9) Combination NS from 7S1a;b; 7S2.
${ }^{(8)} \mathrm{C}$
${ }^{(9)}$ Combination NS from 9S1; 9S2a,b. (10) $\Delta V=\left(V-V_{p 50}\right) 100 \% / V_{p 50}$.
(11) Estimated.

### 5.2.2 Station $1 S$

Hill station 1 S , which was located at the 2548 - ft range on the northeast blast line, was. nearer GZ than any other station for shot Smoky. The terrain from GZ to 15 was generally: rising, the steepest incline being about 250 ft from the station (see Fig. 5.4). Figure 5.7 indicates the positions of the pressure gauges, military debris, and spheres with respect to the traps. Figure 5.8 is a preshot view of this installation which shows, in addition to the details on the layout chart (Fig. 5.7), a 10 -ft pressure-gauge installation and an experimental jeep used by other projects. Note that an extra thermal shield was used on only one trap installation.

Overpressure vs. time measurements at ground level are recorded in Fig. 5.9. The over-pressure-vs.-time curve, also shown, for the ideal wave was obtained by procedures outlined in Ref. 1 using the measured values of overpressure impulse ( $3.962 \mathrm{psi}-\mathrm{sec}$ ) and of duration ( 0.728 sec ). This curve has a peak overpressure of 16.4 psi , whereas the measured curve was 13.1 psi . The difference between the measured blast wave and the ideal, although significant, was not as great as will be seen for station 4 S , which was located at a greater range ( 3406 ft ) on the south blast line. Figure 5.10 is a record of the dynamic pressure obtained 3 ft above the ground. Also shown on this chart is the dynamic pressure associated with the ideal overpressure wave illustrated in Fig. 5.9. Note the latter portion of the measured record, which seems to indicate that the instrument zero was drifting.

Figures 5.11 and 5.12 are closeup photographs of the two trap installations after the detonation. Both photographs show erosion of the wood surfaces, especially the trap housing. Note the absence of the frame for the thermal shield in Fig. 5.11 and the large collection of native debris in front of installation 1S2 (Fig. 5.12). Balsa wood was used as the absorber in all three traps at this station. Thermal damage to the absorber was negligible -even for trap 1S2 which did not have extra thermal protection.

None of the steel spheres (85) or the military debris (about 500 pieces) set out was recovered. Velocity and mass data for natural stones caught by the three traps are plotted in Figs. 5.13 to 5.15 . At installation 1 S 1 the lower trap (1S1a) caught more missiles than the upper trap ( 238 vs .174 ), but the geometric mean velocity for the lower trap was slightly smaller ( $475 \mathrm{ft} / \mathrm{sec}$ vs. $498 \mathrm{ft} / \mathrm{sec}$ ). Only 93 stones were recovered from installation 1 S 2 , which perhaps indicates shielding by the debris (see Fig. 5.12). These 93 missiles had a geometric mean velocity of $486 \mathrm{ft} / \mathrm{sec}$, which indicated good agreement with the data from installation 151 traps. It is of interest to note that on these charts the predicted-velocity curves made on the basis of the ideal blast wave (see Fig. 5.9 and Chap. 3) show fair agreement with the data from all three traps. It should be remembered that the distance of travel for a natural stone is not known; therefore the velocity is predicted assuming the displacement necessary to obtain maximal velocity. All distances of displacement other than this optimum one would result in lower velocities. Thus, ideally, the predicted curve should lie near the top of the velocity distribution. Some scatter above the predicted curve can be explained by variations in the acceleration coefficient for individual missiles as well as by limitations in accuracy inherent in the missile-absorbing technique.

### 5.2.3 Station 2 S

Station 2S was placed in a dale just beyond station 1S (see Figs. 5.4 and 5.5). The experimental arrangement for this station is shown graphically in Fig. 5.16 and pictorially in Fig. 5.17. The two jeeps in Fig. 5.17 were part of another project. Pressure instrumentation had not been installed at the time the photograph was taken.

The measured overpressure vs. time record for station 2S (Fig. 5.18) indicates somewhat closer conformity to the ideal wave than was noted for station 1S. Dynamic pressure vs. time (recorded in Fig. 5.19) is considerably lower than for the hill station (1S) but is just as variable. The dynamic pressure measured between 0.5 and 0.6 sec is of the same order of magnitude as the overpressure (Fig. 5.18) for the same time interval. This undoubtedly indicates an erroneous response of the $q$ gauge.

The postshot photograph of installation 2S1 (Fig. 5.20) indicates that the Styrofoam absorber (type IV) with a thermal protector survived the burst with little damage. Similarly, the balsa absorber (type VI) placed at installation 2S2, but without the extra shield, was in good
condition (Fig. 5.21). Erosion of the trap housing noted in these photographs was considerably less than that for station 1S traps.

Approximately 550 pieces of military debris were placed in front of station 2 S traps. One piece originating from a distance of 18.7 ft was caught by trap 2 S 1 . The mass of the piece of debris was 11.58 g and the velocity with which it struck the absorber was estimated to be $121 \mathrm{ft} / \mathrm{sec}$ (see Table 5.2).

Natural-stone data obtained from traps 2S1, 2S2a, and 2S2b are presented graphically in Figs. 5.22, 5.23, and 5.24, respectively. It is noteworthy that velocities evaluated using the Styrofoam absorber (type IV. at trap 2S1) are somewhat lower than those obtained using the balsa absorber (type VI at traps 2S2a and 2S2b). The threshold velocities for the type IV absorber are lower than for type VI, which results in missiles of lower velocities being caught in type IV (trap 2S1). However, missiles of higher velocities were recorded in the balsa than in the Styrofoam. A recheck of the calibration for each of these absorbers failed to rectify this discrepancy. It has been found that Styrofoam is much more uniform in structure than balsa; therefore more credibility should be given to the data from trap $2 S 1$ than to the other two.

Predicted missile velocities were made from dynamic-pressure data for the ideal wave, which are represented in Fig. 5.19 by a dashed line. Up to about 0.45 sec , the measured curve, although oscillating, corresponds roughly to the ideal-wave curve. The effective dynamic pressure seems to be satisfactorily represented by the "ideal" curve; this is substantiated, in part, by the fact that the predicted-velocity line in Fig. 5.22 lies near the upper limit of the scatter of velocity points.

### 5.2.4 Station 3 S

Station 3S was placed on a hill at the $3218-\mathrm{ft}$ range, slightly higher in elevation than the hill location of station 1S (see Fig. 5.5). The plan for this station is shown diagrammatically in Fig. 5.25. Spheres were not studied, but approximately 550 pieces of military debris were set out. In the preshot photograph of this station (Fig. 5.26), it can be seen that the ground immediately in front of the traps had been leveled with a grader. A thermal shield was used for the installation on the right.

Overpressure and dynamic-pressure records obtained at this station are plotted in Figs. 5.27 and 5.28 , respectively. The measured overpressure curve deviates from the ideal curve in that more than 0.1 sec elapsed between arrival of the blast wave and maximum overpressure. The dynamic pressure developed to maximum at an even slower rate, the entire record being characterized by large fluctuations.

The postshot photographs of the installations (Figs. 5.29 and 5.30) indicate that all three traps were in good condition.

No military debris was caught at this station. Data for the 275 natural-stone missiles caught are plotted in Figs. 5.31 to 5.33 . That the predicted-velocity line is considerably lower than the higher velocity points for all three traps is consistent with the fact that the measured dynamic pressures were higher than the computed ones for an ideal wave. The lower velocity points for the Styrofoam trap (type IV at trap 3 Sib ) compared with the low points for the other traps reflect the difference in threshold velocities for the two types of absorbers (refer to Chap. 2). The upper velocities recorded by the two absorber types are nearer the same amount than those observed for station 2 S (see Sec. 5.2.3). Absence of small missiles in trap 3S1b data is probably attributable to the fact that they were overlooked when the missiles were extracted from the trap.

### 5.2.5 Station 6 S

The next station on the northeast blast line was 6 , located in a dale at the $3739-\mathrm{ft}$ range (see Figs. 5.4 and 5.6). The experimental plan depicted in Fig. 5.34 indicates that absorber type II, along with extra thermal protection, was used in all three traps. Figure 5.35 is a preshot photograph of this station. The sharp rise in the terrain (shown in the background) which appears to be in the direction of GZ was actually on the right of the blast line looking toward the location of the burst.

Figures 5.36 and 5.37 are records of overpressure and dynamic pressure, respectively, for station 6S. Note that the measured overpressure and dynamic-pressure curves in the
initial portions are lower than the ideal-wave curves. Moreover, the measured dynamic pressures were lower than the computed ones for the ideal wave for most of the duration of the wave.

Figures 5.38 and 5.39 are postshot photographs of the two installations at station 6 S . Note the plant stems that stuck to the face of the traps. The frame for the thermal absorber can be seen clearly in Fig. 5.38; however, it was destroyed at the other installation (Fig. 5.39).

The results obtained at this station for natural-stone missiles are presented graphically in Figs. 5.40 to 5.42 . The measured missile velocities are generally much lower than those predicted for the ideal wave. This agrees with the fact that the measured dynamic pressure was lower than that computed for the ideal wave (see Fig. 5.37). Note, however, that the velocities for two missiles caught by trap 6S1a (Fig. 5.40) are in agreement with the predicted values. One piece of military debris (about 550 pieces were set out) was caught by trap 6 S 2 . Data for this missile are recorded in Table 5.2.

### 5.2.6 Station 7S

Station 7 S , which was located at the $4115-\mathrm{ft}$ range, was the third of the three hill stations on the northeast blast line. This station was placed on a hill slightly higher and with slopes somewhat greater than the other two ( 1 S and 3 S ) (see Figs. 5.4 and 5.6).

The plan for this station is shown graphically in Fig. 5.43 and pictorially in Fig. 5.44. Type III absorber, along with thermal shields, was used in all three traps.

The overpressure data presented in Fig. 5.45 indicate that the principal deviation of the measured quantity from that computed for an ideal wave was the longer rise time. Owing to instrument failure, no dynamic pressure was obtained.

Closeup photographs taken after the detonation (Figs. 5.46 and 5.47 ) indicate that the installations were in good condition.

None of the military debris (about 550 pieces) placed at this station was recovered from the traps. Data for 247 natural stones that were trapped are plotted in Figs. 5.48 to 5.50 . Some of the velocity points are higher than the predicted-velocity lines, a result similar to that obtained from the other hill stations, but to a lesser degree. It is interesting to note that velocities obtained at this station correspond roughly with those obtained at station 6 S , a dale station 376 ft closer to GZ. Over twice as many missiles were caught at the dale station ( 6 S ) than were caught at station 7 S . This may be explained in part by the fact that an absorber with a lower density, and thus lower threshold velocities, was used at station 6S.

### 5.2.7 Station 8 S

The last station on the northeast blast line was 85 , located in a long flat dale at the $4980-\mathrm{ft}$ range (see Figs. 5.4 and 5.6). The arrangement at this station (see Fig. 5.51) varied somewhat from others on this line in that the pressure gauges were placed between the two trap installations. Seventy steel spheres, as well as the usual amount of military debris, were placed at this location. Figure 5.52 is a photographic view of installation 8 S2 looking away from GZ.

The data plotted in Fig. 5.53 show that even at this range the measured overpressure curve has a fairly long rise time and a flat top which endured for about 0.1 sec . The measured dynamic-pressure curve in Fig. 5.54 is quite erratic, particularly from 0.2 to 0.5 sec .

Postshot photographs (Figs. 5.55 and 5.56 ) show the two installations to be in good condition. A few plant stems can be seen partly imbedded in the absorber.

No spheres or military debris were caught. Data for the 223 natural-stone missiles that were trapped are plotted in Figs. 5.57 to 5.59 . The upper trap, 8 S 2 b , caught 162 missiles in contrast to only 35 for trap 8 S 2 a and 26 for trap 8 S 1 . Unlike the results obtained for the previous dale station ( $6 S$ ), the velocities predicted on the basis of the ideal wave are in fair agreement with the measured ones.

### 5.3 SOUTH BLAST LINE

The south blast line was flat desert terrain that gradually sloped away from GZ. Figure 5.60 contains a profile of this line, as well as overpressure and duration data for four sta-
tions -only two of which were used for missile studies (4S and 9S). It is interesting to note an actual increase in the measured peak overpressure from station 4 S at 3406 - ft range to the BRL station at 3875 ft . However, peak overpressures computed for the ideal wave from measured impulses and durations decrease monotonically with increasing range, forming a remarkably smooth curve.

### 5.3.1 Station 4S

Secondary-missile investigations at station 4 S were conducted in cooperation with another project ${ }^{2}$ that was designed to study, by means of motion pictures, the displacement of anthropomorphic dummies simulating $165-1 b$ men. Since efforts to obtain motion pictures failed because of dust obscuration, it was fortunate that some velocity data were obtained in the present study for spheres that also simulated men* - at least insofar as velocity of translation $\dagger$ is concerned.

Figure 5.61 portrays graphically the experimental design for both the missile and dummy projects. An asphalt road that was parallel to the blast line passed between the concrete stabilized area and the pressure instrumentation. Figure 5.62 is a preshot photograph of installation 4 S 2 . The flat terrain characteristic of this blast line can be seen in the background.

The blast data plotted in Fig. 5.63 illustrate a significant deviation of the measured from the ideal-wave overpressures. Since the terrain was flat, it can be assumed that the deviations noted were due to thermal effects. This assumption is strengthened by the fact that the measured dynamic pressure, Fig. 5.64, is significantly higher than the corresponding ideal-wave pressure. Note, however, the low level of measured dynamic pressure for the first 0.05 sec .

The balsa absorber in trap 4 S1 suffered little thermal damage (see Fig. 5.65), even without the extra thermal protection. The traps at installation 4S2, portrayed postshot in Fig. 5.66, also endured the thermal effects without serious damage. Absorber type III with a thermal shield, the remains of which can be seen in the photograph, was used at traps 4 S 2 a and b . Figure 5.66 shows four spheres on the right side of the lower trap and one in the upper trap. Velocities of $70,71,74$, and $83 \mathrm{ft} / \mathrm{sec}$ (from left to right) were computed for the $7 / 16-\mathrm{in}$. steel spheres in the lower trap. Average height of impact was 6.5 in . above the ground. The sphere ( $7 / 16$-in. steel) in the upper trap had a velocity of $79 \mathrm{ft} / \mathrm{sec}$ at an impact height of 20.4 in . These spheres had been placed 9 in . above the ground and 17.1 ft in front of the traps (see Figs. 5.61 and 5.62). The velocity predicted for the spheres (see Table 5.2 ) was $78 \mathrm{ft} / \mathrm{sec}$.

The average velocity at impact for the five spheres mentioned above was $75.2 \mathrm{ft} / \mathrm{sec}$. If one assumes that the average velocity during transit was between 37.6 and $75.2 \mathrm{ft} / \mathrm{sec}$, the time required to traverse 17.1 ft is found to be between 0.45 and 0.23 sec . Dynamic pressure measured during either of these periods ( 0 to 0.45 sec or 0 to 0.23 sec ) was considerably above that for the ideal wave (see Fig. 5.64), which was the basis for the predicted velocity of 78 ft / sec. From this one might speculate that dynamic pressures as high as those recorded in Fig. 5.64 did not exist at the location of the spheres. It should be noted (see Fig. 5.61) that the initial position of the spheres was only 9 in . above the ground and that the dynamic pressure was measured 3 ft above the surface at a distance of 190 ft from the spheres.

The results obtained for natural-stone missiles caught at this station are plotted in Figs. 5.67 to 5.69 . A significant number of missiles whose velocities exceeded the predicted values were caught in each trap. A difference, also noted at other stations, between the response of the absorbers (balsa at installation 4S1 and Styrofoam at traps 4S2a and b) to natural-stone missiles was that the Styrofoam absorber caught missiles that had lower velocities because of its lower threshold velocities. The balsa trap (4S1) caught a larger proportion of small missiles (note position of the geometric mean) whose velocities tended to be somewhat higher than those of the small missiles caught in the Styrofoam traps. The latter yielded a scattering of large missiles which impacted at high velocities, a reasonable result considering the nature of the measured dynamic-pressure curve displayed in Fig. 5.64.

* Total displacements measured after the shot were: standing dummy, 255.7 ft downwind and 43.7 ft to the right; and prone dummy, 160 ft downwind and 31.5 ft to the right.
$\dagger$ The acceleration coefficient of the spheres that were caught is slightly higher than the average value for a tumbling man. References 1 and 3 contain a more complete treatment of this subject.


### 5.3.2 Station 9S

Figure 5.70 is a layout chart for station 9 S . This station was located at the $5680-\mathrm{ft}$ range on the south blast line. The chart indicates the placement of 70 steel spheres and about 550 pieces of military debris. Installation 9S2 is shown in Fig. 5.71.

The overpressure vs. time data (Fig. 5.72) for this station display a closer correspondence

### 5.4 NORTH BLAST LINE, STATION 5S

The location of station 5 S , at the 3722 - ft range on the north blast line, is illustrated in Fig. 5.77. The station was located about 900 ft beyond the mountain peak at an elevation approximately 300 ft lower than that of the peak. This was the only station on shot Smoky that was not on a direct line of sight with the point of detonation of the bomb. Hence it was not necessary to use extra thermal protection for the absorbers. The only missiles studied were natural stones (see Fig. 5.78). Figure 5.79 is a view of the station looking up the mountain toward GZ.

Unlike the overpressure records for the other dale stations (see Figs. 5.18, 5.36, and 5.53), Fig. 5.80 illustrates that the initial rise was very sharp - the principal modification being its failure to peak in the manner characteristic of the ideal or classical wave. A dynamic-pressure record was not obtained.

Figures 5.81 and 5.82 show that the foil covering the absorber, except for a small patch on the right side of the lower trap in Fig. 5.81, was still in place after the shot.

Results obtained for the 119 natural-stone missiles caught at this station are presented graphically in Figs. 5:83 to 5.85 . Missile velocities were significantly lower than those which could be expected for an ideal wave.

### 5.5 SUMMARY, SHOT SMOKY

Three traps were placed at each of nine stations located on three blast lines. The station nearest to $\mathrm{GZ}(1 \mathrm{~S})$ had a range of 2548 ft and a measured overpressure of about 13 psi , and the most distant one (9S) had a range of 5680 ft and a measured overpressure of about 5 psi .

Hill-and-dale effects were studied at six stations on the northeast blast line and at one station on the north line. For natural-stone missiles, comparisons were made between measured velocities and the ones predicted on the basis of an ideal blast wave whose overpressure impulse and duration were the same as those measured. In general, the hill stations (1S, 3S, and 7 S ) produced missiles with velocities that were higher than those predicted, and the dale stations ( $2 \mathrm{~S}, 5 \mathrm{~S}, 6 \mathrm{~S}$, and 8 S ), lower than predicted. The effect was particularly noticeable at the dale station ( 5 S ) on the north line.

Two stations were placed on the south blast line where the terrain was flat. The blast wave incident at the $3406-\mathrm{ft}$ station (4S) was significantly modified by surface thermal effects which resulted in higher dynamic pressures and higher missile velocities than expected for an ideal wave. The blast wave that reached the second station on the south line ( 9 S at 5680 ft ) was
almost ideal in form, producing natural-stone velocities in good agreement with those predicted.

A total of 2876 natural-stone missiles was caught by the 27 traps used in this shot: 34 per cent was caught by the lower (a) traps at the installations where the traps were stacked, 41 per cent by the upper (b) traps, and only 25 per cent by the traps not stacked.

About 550 pieces of military debris were placed in front of the traps at each of eight stations. A total of 405 steel spheres ( $1 / 16-1 / 2-$, and $9 / 16$-in.-diameter steel) was placed at four stations. Only two pieces of military debris and five spheres were recovered.

Results of the missile studies for shot Smoky are summarized in Table 5.2.* Data resulting from the analysis of all natural stones caught at each station are listed. Some parameters are given here for the first time. The following symbols are used in this table:
$\overline{\boldsymbol{\alpha}} \quad$ Acceleration coefficient of the average mass of the missile sample
used to compute predicted values of missile velocity, $\mathrm{sq} \mathrm{ft} / \mathrm{lb}$
a,b Regression-equation coefficients
d Distance of travel of the missile before striking the trap, ft
$\bar{D}_{s} \quad$ Spatial density of missiles caught, number per sq ft
$\Delta \mathrm{V} \%$ Per cent of difference in average velocity from predicted velocity
$\mathrm{E}_{\mathrm{gv}} \quad$ Geometric standard error of estimate in velocity $=\operatorname{antilog} \mathrm{E}_{\mathrm{lv}}$
$h_{1} \quad$ Height above ground at which the missile was placed, in.
$\overline{\mathrm{h}}_{2} \quad$ Average height above ground at which the missiles struck, in.
MD Military debris
$\mathrm{n} \quad$ Number of missiles in sample
NS Natural stone
$S_{\mathrm{gv}} \quad$ Geometric standard deviation of velocity $=$ antilog $\mathrm{S}_{\mathrm{lv}}$
St Steel sphere
$V_{-} \quad$ Minimum velocity
$V_{+} \quad$ Maximum velocity
$\vec{V} \quad$ Average velocity
$V_{50} \quad$ Geometric mean velocity
$\mathrm{V}_{\mathrm{P} 50} \quad$ Predicted value of velocity for the geometric mean mass
All velocity parameters have units of feet per second. The last five columns of the table contain mass ( mg ) parameters corresponding to the quantities discussed for velocity.

## REFERENCES

1. I. G. Bowen, R. W. Albright, E. R. Fletcher, and C. S. White, A Model Designed to Predict the Motion of Objects Translated by Classical Blast Waves, USAEC Report CEX-58.9, June 29, 1961.
2. R. V. Taborelli, I. G. Bowen, and E. R. Fletcher, Tertiary Effects of Blast-Displacement, Operation Plumbbob Report, WT-1469, May 22, 1959.
3. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determinations of Aerodynamic-drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.

* Table 2.1 describes the absorber types.


Fig. 5.1-Station locations for shot Smoky in Area 2C, NTS.


Fig. 5.2-Typical placement of military debris and large steel spheres (on trough-like support).


Fig. 5.3-Typical trap installation showing use of extra thermal shield, which consisted of aluminum foil held in frame approximately 1 ft in front of traps.


Fig. 5.4-Profile of northeast blast line, shot Smoky. Range: 0 to 5500 ft . Note that the vertical and horizontal scales are the same. See Table 5.1 for explanation of blast-wave parameters.


Fig. 5.5-Profile of northeast blast line, shot Smoky. Range: 2500 to 3600 ft . See also Fig. 5.4.


Fig. 5.6-Profile of northeast blast line, shot Smoky. Range: 3600 to 5100 ft . See also Figs. 5.4 and 5.5 .


Fig. 5.7-Station 1S layout chart. The small letter suffix by the trap designators indicates level of the stacked traps: "a" for ground level and " $b$ " for one above another trap.


Fig. 5.8-Station 1S, preshot, at 2548 - ft range on the northeast blast line.


Fig. 5.9-Overpressure vs. time at station 1S.
$\qquad$


Fig. 5.10-Dynamic pressure vs. time at station 1S.


Fig. 5.11-Traps 1S1a and b, postshot. Note debris in front of traps and erosion of trap housing.

4


Fig. 5.12-Installation 1S2, postshot. Note debris in front of trap and erosion of trap housing.


Geometric Means (Mass and velocity
Natural Stones


Mass, mg
Fig. 5.13-Analysis of natural-stone missiles from trap 1S1a: $n=238 ; \log \mathrm{v}=2.8102-0.0664 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=102 \mathrm{mg} ; \mathrm{V}_{50}=$ $475 \mathrm{ft} / \mathrm{sec}$.


Fig. 5.14-Analysis of natural-stone missiles from trap 1S1b: $n=174 ; \log v=2.8191-0.0557 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=150 \mathrm{mg} ; \mathrm{V}_{50}=$ $498 \mathrm{ft} / \mathrm{sec}$.
-.-. - Predicted Velocity

-     -         - Predicted Velocity (Reflected)* Regression Line
--------- Standard Error of Estimote
( Geometric Means (Mass and Velocity)

Mass, mg
*Not Apolicoble
Fig. 5.15-Analysis of natural-stone missiles from trap $1 \mathrm{~S} 2: \mathrm{n}=93 ; \log \mathrm{v}=2.8227-0.0599 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=181 \mathrm{mg} ; \mathrm{V}_{50}=$ $486 \mathrm{ft} / \mathrm{sec}$.
$\qquad$

18.7
 Shield

$\begin{array}{ll}\text { STATION: } 2 \mathrm{~S} & \text { TERRAIN: DALE } \\ \text { RANGE: 2914 } & \text { BLAST LINE: NE }\end{array}$



## Roman numeral in parenthesis designates type of missile absorber

Fig. 5.16-Station 2S layout chart.


Fig. 5.17-Station 2 S , preshot, at 2914-ft range on the northeast blast line.

4 4.


Fig. 5.18-Overpressure vs. time at station 2 S.

()
d)


Fig. 5.20-Installation 2 S 1 , postshot. Note debris accumulated in front of trap.


Fig. 5.21-Traps 2 S2a and b, postshot.

*Not Applicable Fig. 5.22
-_.-. - Pedicted Velocily
-...... - Predicted Velocity (Reflected)*
---.-.-.- Stondard Error of Extimote
Stondard Error of Estimote
Geometric Moans (Moss and Velocity)
2S20


Fig. 5.23-Analysis of natural-stone missiles from trap $2 \mathrm{~S} 2 \mathrm{a}: \mathrm{n}=307 ; \log \mathrm{v}=2.8089-0.0658 \log \mathrm{~m} ; \mathrm{Egv}_{\mathrm{g}}=1.12 ; \mathrm{M}_{50}=61.3 \mathrm{mg} ; \mathrm{V}_{50}=$ $491 \mathrm{ft} / \mathrm{sec}$.
$Q$ a 0



Fig. 5.24-Analysis of natural-stone missiles from trap $2 \mathrm{~S} 2 \mathrm{~b} ; \mathrm{n}=227 ; \log \mathrm{v}=2.7670-0.0452 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=203 \mathrm{mg} ; \mathrm{V}_{50}=$ $460 \mathrm{ft} / \mathrm{sec}$.


Fig. 5.25-Station 3S layout chart.


Fig. 5.26-Station 3S, preshot, at 3218-ft range on the northeast blast line.


Fig. 5.27-Overpressure vs. time at station 3S.
$\qquad$


Fig. 5.28-Dynamic pressure vs. time at station 35 .


Fig. 5.29-Traps 3 S 1 a and b, postshot.
$\qquad$


Fig. 5.30-Installation 3S2, postshot.
 Stondard Error of Estimere
3 Slo
Natural Stones

Mass, mg
Fig. 5.31 -Analysis of natural-stone missiles from trap $3 \mathrm{~S} 1 \mathrm{a}: 1 \mathrm{t}=71 ; \log \mathrm{v}=2.9444-0.1783 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=51.3 \mathrm{mg} \mathrm{V}_{50}=$ $434 \mathrm{ft} / \mathrm{sec}$.
Predicted Velocity (Reflected)
Regression Line
Geometric Means (Mass and velocity)

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Predicted Velocity (Reflected)*

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Predicted Velocity (Reflected)*
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—__ Regression Line

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3SIb
Notural Stones


Mass, mg
Geometric Means (Mass and Velocity)
* Not Applicable

Fig. 5.32-Analysis of natural-stone missiles from trap \(3 \mathrm{~S} 1 \mathrm{~b}: \mathrm{n}=109 ; \log \mathrm{v}=2.6046-0.0635 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.24 ; \mathrm{M}_{50}=219 \mathrm{mg} ; \mathrm{V}_{50}=\) \(286 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.33-Analysis of natural-stone missiles from installation \(3 \mathrm{~S} 2: \mathrm{n}=95 ; \log \mathrm{v}=2.8475-0.0872 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{\mathbf{5 0}}=58.6 \mathrm{mg}\); \(\mathrm{V}_{50}=494 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.34-Station 6S layout chart.


Fig. 5.35-Station 6S, preshot, at 3739-ft range on the northeast blast line.
a)


Fig. 5.36-Overpressure vs. time at station 6 S.


Fig. 5.37 - Dynamic pressure vs. time at station 6 .


Fig. 5.38-Traps 6 sla, and \(\mathbf{b}\), postshot. Note the charred wood. The absorber, however, was undamaged by thermal radiation.


Fig. 5.39-Installation 6S2, postshot. The thermal-shield frame was destroyed.
.

--------- \(\begin{array}{ll}\text { Regression Lindord Error of Estimote }\end{array}\)
(2) Geometric Means (Mass and Velocity)

6SIa
Natural Stones


Fig. 5.40-Analysis of natural-stone missiles from trap \(6 \mathrm{~S} 1 \mathrm{a}: \mathrm{n}=86 ; \log \mathrm{v}=2.3973-0.0897 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{\mathbf{5 0}}=153 \mathrm{mg} ; \mathrm{V}_{\mathbf{5 0}}=\) \(159 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.41-Analysis of natural-stone missiles from trap \(6 \mathrm{~S} 1 \mathrm{~b}: \mathrm{n}=192 ; \log \mathrm{v}=2.4998-0.1248 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=103 \mathrm{mg} ; \mathrm{V}_{50}=\) \(177 \mathrm{ft} / \mathrm{sec}\).
-..-.- Predicted Velocity \(\quad\) - Predicted velocity (Reflected)* —__ Regression Line

Regression Line Estimoterd Error of Estimo
( Geometric Meons (Moss and Velocity


Fig. 5.42-Analysis of natural-stone missiles from installation \(6 \mathrm{~S} 2: \mathrm{n}=259 ; \log \mathrm{v}=2.4595-0.1056 \log \mathrm{~m} ; \mathrm{Egv}_{\mathrm{g}}=1.17 ; \mathrm{M}_{50}=154 \mathrm{mg} ;\) \(\mathrm{V}_{50}=169 \mathrm{ft} / \mathrm{sec}\).

STATION: 7S RANGE: 4115

TERRAIN: HILL
BLAST LINE: NE

MD Military Debris, On Ground


Fig. 5.43-Station 7S layout chart.


Fig. 5.44 -Station 7S, preshot, at 4115 -ft range on the northeast blast line.


Fig. 5.45-Overpressure vs. time at station 7S.


Fig. 5.46 - Traps 7S1a and \(b\), postshot, showing thermal-shield frame still in place.


Fig. 5.47 - Trap 7S2, postshot. A piece of the thermal-shield support can be seen on the right side of the trap.


7 Sla
- Geometric Means (Mass and Velocity)

Notural Stones


Fig. 5.48-Analysis of natural-stone missiles from trap 7S1a: \(n=66 ; \log \mathrm{v}=2.5684-0.1335 \log \mathrm{~m} ; \mathrm{Egv}_{\mathrm{g}}-1.19 ; \mathrm{M}_{50}=79.4 \mathrm{mg} ; \mathrm{V}_{50}=\) \(206 \mathrm{ft} / \mathrm{sec}\).


Natural Stones


Fig. 5.49-Analysis of natural-stone missiles from trap \(7 \mathrm{~S} 1 \mathrm{~b}: \mathrm{n}=111 ; \log \mathrm{v}=2.6620-0.1465 \log \mathrm{~m} ; \mathrm{Egv}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=104 \mathrm{mg} ; \mathrm{V}_{50}=\) \(233 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.50-Analysis of natural-stone missiles from trap 7S2: \(\mathrm{n}=70 ; \log \mathrm{v}=2.6253-0.1394 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.17 ; \mathrm{M}_{50}=76.3 \mathrm{mg} ; \mathrm{V}_{50}=\) \(231 \mathrm{ft} / \mathrm{sec}\).

\section*{STATION: 8 S RANGE: 4980' \\ TERRAIN: DALE BLAST LINE: NE}

MD Military Debris, On Ground
St Steel Spheres, \(14^{\prime \prime}\) Above Ground



Fig. 5.52-Installation 8 S 2 at station \(8 \mathrm{~S}, 4980-\mathrm{ft}\) range on the northeast blast line.


Fig. 5.53 - Overpressure vs. time at station 8 S .


Fig. 5.54-Dynamic pressure vs. time at station 8 S .


Fig. 5.55-Installation 8 S 1 , postshot.


Fig. 5.56 - Traps \(8 S 2 \mathrm{a}\) and b, postshot.
-.-. Predicted Velocity
—...-. - Fredicted Velocity (Refiected)*
-_-------- Regression Line
Standard Error of Estimote
( Geometric Means (Mass and Velocity)


Mass, mg

Fig. 5.57-Analysis of natural-stone missiles from installation \(8 \mathrm{~S} 1: \mathrm{n}=26 ; \log \mathrm{v}=2.5214-0.1906 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{\mathbf{5 0}}=55.2 \mathrm{mg}\); \(V_{50}=155 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.58 -Analysis of natural-stone missiles from trap \(8 S 2 \mathrm{a}: \mathrm{n}=35 ; \log \mathrm{v}=2.4599-0.1496 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=103 \mathrm{mg} ; \mathrm{V}_{50}=\) \(144 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.59-Analysis of natural-stone missiles from trap \(8 \mathrm{~S} 2 \mathrm{~b}: \mathrm{n}=162 ; \log \mathrm{v}=2.4804-0.1491 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=51.1 \mathrm{mg} ; \mathrm{V}_{50}=\) \(168 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.60 - Profile of south blast line, shot Smoky, 0 - to 6000 - ft range. See Table 5.1 for explanation of blast parameters plotted.


Fig. 5.61-Station 4 S layout chart. The concrete stabilized area was used by another project. (See Ref. 2.)


Fig. 5.62 - Installation 4 S 2 , preshot, at \(3406-\mathrm{ft}\) range on the south blast line.


Fig. 5.63 - Overpressure vs. time at station 4 S


Fig. 5.64 - Dynamic pressure vs. time at station 4 S.


Fig. 5.65-Installation 4 S1, postshot. Note effect of thermal scorching on balsa absorber placed end-grain in the trap.


Fig. 5.67-Analysis of natural-stone missiles from installation \(4 \mathrm{~S} 1: \mathrm{n}=43 ; \log \mathrm{v}=2.8446-0.1044 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=55.2 \mathrm{mg} ;\) \(\mathrm{V}_{50}=460 \mathrm{ft} / \mathrm{sec}\).
-...-. - Predicted velocity
—————— Fredicted Velocity (Reflected)"
-------- Regression Line
Standard Error of Estimat
(2) Geometric Meons (Mass and Velocity)

Natural Stones


Fig. 5.68-Analysis of natural-stone missiles from trap 4S2a: \(n=81 ; \log \mathrm{v}=2.5473-0.0890 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.29 ; \mathrm{M}_{50}=272 \mathrm{mg} ; \mathrm{V}_{50}=\) \(214 \mathrm{ft} / \mathrm{sec}\).
———————ll \(\begin{aligned} & \text { Predicted Velocity } \\ & \text { Predicted Velocity } \\ & \text { Regression Line }\end{aligned}\)
\(\begin{array}{ll}\ldots \ldots-\ldots-\ldots & \text { Regression Line } \\ \text { Stondard Error of Estimate }\end{array}\)
Q Geametric Means (Mass and Velocity)
\(800 \quad \begin{aligned} & \text { 4S2b } \\ & \text { Natural Stones }\end{aligned}\)


Mass, mg
Fig. 5.69-Analysis of natural-stone missiles from trap \(4 \mathrm{~S} 2 \mathrm{~b}: \mathrm{n}=135 ; \log \mathrm{v}=2.5831-0.0868 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{g} v}=1.29 ; \mathrm{M}_{50}=197 \mathrm{mg} ; \mathrm{V}_{50}=\) \(242 \mathrm{ft} / \mathrm{sec}\).

STATION: 9S TERRAIN: FLAT
RANGE: 5680'
BLAST LINE: S

MD Military Debris, On Ground
St Steel Spheres, 9"Above Ground
Roman numeral in parenthesis designotes type of missile absorber


Fig. 5.70-Station 9S layout chart.


Fig. 5.71-Traps 9S2a and b, preshot, at 5680 -ft range on the south blast line. Note military debris and large steel spheres in the foreground.


Fig. 5.72- Overpressure vs. time at station 9S.


Fig. 5.73-Traps 9S2a and b, postshot. Note steel spheres on concrete base in front of traps.

Fig. 5.74-Analysis of natural-stone missiles from installation \(9 \mathrm{~S} 1: \mathrm{n}-17 ; \log \mathrm{v}=2.4472-0.1334 \log \mathrm{~m} ; \mathrm{Egv}_{\mathrm{g}}=1.11 ; \mathrm{M}_{\mathbf{5 0}}=45.6 \mathrm{mg} ;\) \(V_{50}=168 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.75 -Analysis of natural-stone missiles from trap \(952 \mathrm{a}: \mathrm{n}=18 ; \log \mathrm{v}=2.4984-0.1726 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.09 ; \mathrm{M}_{50}=49.3 \mathrm{mg} ; \mathrm{V}_{50}=\) \(161 \mathrm{ft} / \mathrm{sec}\).
-_- Predicted Velocity
-.....- Predicted velocity (Reflected)" -------- Regression Line Estord Error of Estimote Standord Error of Estimote
Geometric Means (Moss and Velocity)
Natural Stones


Mass, mg

Fig. 5.76-Analysis of natural-stone missiles from trap \(9 \mathrm{~S} 2 \mathrm{~b}: \mathrm{n}=46 ; \log \mathrm{v}=2.4841-0.1620 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.10 ; \mathrm{M}_{50}=41.8 \mathrm{mg} ; \mathrm{V}_{50}=\) \(167 \mathrm{ft} / \mathrm{sec}\).


Fig. 5.77-Profile of north blast line, shot Smoky. See Table 5.1 for explanation of blast parameters plotted.

\section*{STATION: 5S \\ RANGE: 3722' \\ TERRAIN: DALE BLAST LINE: \(N\)}

Roman numeral in parenthesis designates type of missile absorber


Fig. 5.78-Station 5S layout chart.


Fig. 5.79-Station 5S, preshot, looking toward GZ. This station was at \(3722-\mathrm{ft}\) range on the north blast line and was the only station not in line-of-sight with the point of detonation.


Fig. 5.80 - Overpressure vs. time at station 5 S.


Fig. 5.81-Traps 5S1a and b, postshot.


Fig. 5.82-Installation 5S2, postshot.


Fig. 5.83-Analysis of natural-stone missiles from trap \(5 \mathrm{~S} 1 \mathrm{a}: \mathrm{n}=73 ; \log \mathrm{v}=2.5374-0.2090 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{g} v}=1.09 ; \mathrm{M}_{50}=47.0 \mathrm{mg} ; \mathrm{V}_{50}=\) \(154 \mathrm{ft} / \mathrm{sec}\).


Mass, mg
Fig. 5.84-Analysis of natural-stone missiles from trap \(5 \mathrm{~S} 1 \mathrm{~b}: \mathrm{n}=23 ; \log \mathrm{v}=2.5467-0.2029 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.08 ; \mathrm{M}_{50}=51.1 \mathrm{mg} ; \mathrm{V}_{50}=\) \(159 \mathrm{ft} / \mathrm{sec}\).
——. -... Predicted Velocity
-_-..- Predicted Velocity (Reflected)* ____-_-_ Regression Line

Standard Error of Estimate
(2) Geometric Means (Mass and Velocity)


Fig. 5.85-Analysis of natural-stone missiles from installation \(5 \mathrm{~S} 2: \mathrm{n}=23 ; \log \mathrm{v}=2.5063-0.1995 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.15 ; \mathrm{M}_{50}=97.3 \mathrm{mg}\); \(\mathrm{V}_{50}=129 \mathrm{ft} / \mathrm{sec}\).

\section*{Chapter 6}

\title{
SHOT GALILEO, EXPERIMENTAL PROCEDURE AND RESULTS
}

\subsection*{6.1 GENERAL}

Participation in shot Galileo involved studies of the translation of (1) fragments from windows mounted in houses and in open areas, (2) natural stone, marked gravel, and spheres in open areas, and (3) debris from a concrete-block wall.

Figure 6.1 is a map of Area 1, NTS, showing the location of the missile stations as well as the blast-wave instrumentation used in this shot. On this chart there was an \(8^{\circ}\) difference between the azimuth angles of the blast line and of the line of the missile studies. The estimated yield for this shot was 11 kt (see Table 6.1), producing an overpressure at the near range \((2750 \mathrm{ft})\) of about 8.4 psi and at the distant range ( 4700 ft ) of about 3.8 psi . No blast data were obtained at the intermediate range ( 3750 ft ) for station 4.3 GTS .

An interesting overall view of all stations used in this shot is shown in Fig. 6.2. This photograph was taken from the 500 -ft tower at GZ. Yucca Lake (dry) can be seen in the background. The concrete-block wall was located just left of the rut road at station 7G. At locations 7 GTS and 4.3 GTS , the tool sheds to the right of the main road were made usable for missile studies by cementing absorbing material on the sides that faced GZ. The houses used in this study can be seen at station 3G: the precast concrete on the left and the reinforced concrete block on the right. Both houses had flat tops.

Data for glass-fragment missiles were obtained at certain locations in cooperation with another project that was studying the penetration effects of this type of missile on biological targets (dogs). \({ }^{1}\) The trauma to which a dog was exposed was estimated by placing a trap (or traps) as near the dog installation as possible. A dog was also placed behind the concreteblock wall at station 7G.

The method of presentation used in this chapter is essentially the same as that used in Chaps. 4 and 5. After the description of each installation, or small group of similar installations, the results obtained are discussed and presented graphically. All results are summarized in Table 6.2. In a few cases the missile samples obtained were too small to justify graphical presentations.
6.2 STATION 3G, 4700-FT RANGE

\subsection*{6.2.1 General Discussion and Blast Parameters}

Traps were installed at four locations at this station (Fig. 6.1). The first location was a concrete slab that had been the floor of a rambler house destroyed on a previous shot. Marked gravel and spheres were placed on the slab. At the second location traps were installed behind windows in an open area. Natural-stone missiles were also studied. The other two locations were inside the reinforced-block and the precast-concrete (concrete-slab) houses.
(See List of Symbols.)
\begin{tabular}{|c|c|c|c|c|c|c|c|c|}
\hline \(\mathrm{p}_{0}=12\). & \multicolumn{3}{|r|}{\(\mathrm{c}_{0}=1124 \mathrm{ft} / \mathrm{sec}\left(18.8^{\circ} \mathrm{C}\right)\)} & \multicolumn{2}{|l|}{Estimated yield: \(11 \mathrm{kt}^{(1)}\)} & \multicolumn{3}{|l|}{Terrain: flat desert} \\
\hline Station & Range, ft & \[
\begin{aligned}
& \left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{m}}{ }^{(2)} \\
& \text { psi-sec }
\end{aligned}
\] & \[
\begin{gathered}
\left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{r}},{ }^{(3)} \\
\text { psi-sec }
\end{gathered}
\] & \[
\begin{gathered}
\left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{m}}, \\
\mathrm{sec}
\end{gathered}
\] & \[
\begin{gathered}
\left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{r}},{ }^{(4)} \\
\mathrm{sec}
\end{gathered}
\] & \[
\begin{gathered}
\left(p_{s}\right)_{\mathrm{m}},{ }^{(2)} \\
\mathrm{psi}
\end{gathered}
\] & \[
\begin{gathered}
\left(p_{s}\right)_{c},{ }^{(5)} \\
\mathrm{psi}
\end{gathered}
\] & \[
\begin{gathered}
\left(p_{s}\right)_{\mathrm{r}}{ }^{(6)} \\
\mathrm{psi}
\end{gathered}
\] \\
\hline 3G & 4700 & 1.122 & & 0.756 & & 4.5 & 3.85 & \\
\hline 4.3GTS & 3750 & & 1.355 & & 0.675 & & (5.32) & 5.34 \\
\hline 7G and 7GTS & 2750 & 1.754 & & 0.576 & & 8.7 & 8.38 & \\
\hline
\end{tabular}
\({ }^{(1)}\) Estimation made by assuming a "typical air burst" as described in The Effects of Nuclear Weapons. and by using data for stations 3G and 7G.
\({ }^{(2)}\) Determined from BRL mechanical-gauge records (no gauge at station 4.3GTS).
\({ }^{(3)}\) Overpressure impulse computed by interpolation equation derived from \(\left(I_{p}\right)_{\mathrm{m}}\) values
\[
\log \left(\mathrm{I}_{\mathrm{p}}\right)_{\mathrm{r}}=3.1100-0.8333 \log \mathrm{R}
\]
\({ }^{(4)}\) Overpressure duration computed by interpolation equation derived from \(\left(t_{p}^{+}\right)_{m}\) values
\[
\log \left(t_{p}^{+}\right)_{r}=-1.9844+0.5073 \log R
\]
\({ }^{\text {(5) }}\) Peak overpressure computed for a classical blast wave of impulse \(\left(I_{p}\right)_{m}\) and of duration \(\left(t_{p}^{+}\right)_{m}\). Measured values of impulse and duration were not obtained at station 4.3 GTS ; therefore regression values \(\left(I_{p}\right)_{r}\) and ( \(\left.t_{p}^{+}\right)_{r}\) were used (see text).
\({ }^{(6)}\) Peak overpressure computed by interpolation equation derived from \(\left(p_{s}\right) c\) values for stations 3 G and 7G
\[
\log \left(p_{s}\right)_{r}=5.9122-1.4506 \log R
\]

Note: The line where missile data were obtained was \(150^{\circ}\) azimuth from GZ. Blast data were obtained from a line of \(158^{\circ}\) azimuth.

Blast-wave data obtained for station 3G are plotted in Fig. 6.3. Overpressure vs. time is shown as a solid line for the measured values and as a dashed line for an ideal wave having the same impulse and duration as the measured values (refer to Chap. 3). Agreement between these curves is good except for the early-time periods where, apparently, inertia in the instrumentation delayed the initial rise, caused it to overshoot, and then delayed the return of the recording to a quasistable state. Peak overpressure of the ideal wave ( 3.85 psi ) was used to compute predicted missile velocities. Dynamic pressure was not measured at this station.

The traps used at station 3G, except for trap 3G8b, were the ones used by Project 33.4 in Operation Teapot. \({ }^{2}\) They were different from the others discussed in this report in that the missile-collecting area was square ( 3.516 sq ft ) and the absorber was type I .

\subsection*{6.2.2 Concrete-slab Location, Installations 3G1 to 3G4}

Figure 6.4 illustrates the placement of marked gravel and spheres on the concrete slab. Ninety large ( \(7 / 16^{-}, 1 / 2^{-}\), and \(9 / 16^{-}\)-in.-diameter) steel spheres were placed in trough-like supports 18 in . above the surface at two locations. One weighted* croquet ball was placed on a \(21-\mathrm{in}\). high thin rod support (see Figs. 6.5 and 6.6). A total of 13,715 smaller spheres was placed at the eight locations marked in Fig. 6.4. Green, yellow, and black samples contained 2110 spheres each, and red and clear samples, 1055 each. Each sample consisted of the following spheres in the indicated proportions:
*The mass of the croquet ball was increased in order that its acceleration coefficient would correspond roughly to that of a tumbling man ( \(0.035 \mathrm{sq} \mathrm{ft} / \mathrm{lb})\). The significance of this concept is treated in Refs. 3 and 4 (see Chap. 3).
\begin{tabular}{lrrr}
\(1 / 8\)-in.-diameter nylon (Ny \(1 / 8\) ) & \(5.2 \%\) & \(1 / 4\)-in.-diameter steel (St \(1 / 4\) ) & \(1.4 \%\) \\
\(-1 / 8\)-in.-diameter aluminum (Al \(1 / 8\) ) & \(10.4 \%\) & 36.0 mg (av.) soda glass (Gs) & \(53.5 \%\) \\
\(3 / 16\)-in.-diameter aluminum ( \(\mathrm{Al} ~ 3 / 16\) ) & \(5.2 \%\) & 72.6 mg (av.) soda glass (G1) & \(13.1 \%\)
\end{tabular}
\[
\begin{array}{lr}
1 / 4 \text {-in.-diameter aluminum }\left(\mathrm{Al}^{1 / 4}\right. \text { ) } & 0.7 \% \\
3 / 8 \text {-in.-diameter aluminum }(\mathrm{Al} 3 / 8) & 0.1 \% \\
1 / 8 \text {-in.-diameter steel (St } 1 / 8 \text { ) } & 10.4 \%
\end{array}
\]

Figure 6.5 is a preshot photograph of the concrete-slab location looking away from GZ. The structure on the left was a reinforced bathroom shelter that was left intact after the remainder of the rambler house originally placed on this slab had been destroyed in Operation Teapot.

Figure 6.6 illustrates typical placement of spheres and gravel at this location. The trough-like support held the large steel spheres, whereas the smaller spheres were placed in the tissue-paper bags beneath the steel spheres and the croquet ball.

Figure 6.7 is a closeup photograph of installation 3G3 taken after the detonation. This installation, as well as others on this shot, was in good condition, and no signs of thermal or blast damage were seen.

Because so few missiles were caught at this location, the data were not prepared in the plotted form. The small number caught was undoubtedly the result of their low velocities relative to the threshold values of the absorber. Note the gravel left in front of the trap in Fig. 6.7. Installation 3G1 caught one natural stone and four pieces of gravel (see Table 6.2 for results). Data were obtained for two natural stones, six pieces of gravel, and one \(1 / 8\)-in. steel sphere from installation 3G4 (Table 6.2). No results were obtained for installations 3G2 or 3G3.

\subsection*{6.2.3 Open Area, Installations 3G5 to 3G9}

Figure 6.8 illustrates diagrammatically the plan for missile studies for five installations ( 3 G 5 to 3 G 9 ) that were placed in an open area between the concrete slab and the reinforcedblock house (see Fig. 6.1). Installation 3G5, the only one not behind a window, was meant for the study of natural-stone missiles.

Four of the installations at this location are shown in Fig. 6.9 (installation 3G5 is not shown). The reinforced block house can be seen on the left side of the photograph. The trap installation that was second from the left consisted of an empty box. A dog was later placed in this box. The trap above the box (3G8b) was the small size similar to those used at the other stations.

Figures 6.10 ( 3 G 6 catching window glass) and 6.11 ( 3 G 7 behind plate glass) are postshot photographs of two representative installations. In Fig. 6.11 the impressions of fragments that struck the trap flat can be seen.

Nothing was caught in installation 3G5. Installation 3G6, which was located 4.6 ft behind a window, caught 42 fragments. The velocities and masses for these fragments are plotted in Fig. 6.12. Note two predicted-velocity lines shown on this and subsequent charts for glassfragment data at station 3G. The lower line was computed for an ideal blast wave similar to that depicted in Fig. 6.3. The upper line was computed for an ideal wave whose peak overpressure is equal to the reflected value (assuming normal incidence and perfect reflection) of the peak overpressure of the incident ideal wave. The duration was assumed to be the same as that measured. This "reflected" assumption* results in a prediction line that is compatible with the data from installation 3G6 (Fig. 6.12) since the predicted values are near the top of distribution of the measured ones.

Seven fragments of plate glass were recovered from installation 3G7. The plot shown in Fig. 6.13 indicates that the measured velocities lie between the two prediction lines discussed above.

Trap 3G8b, placed 10.9 ft behind the window and above the dog installation, which was 31.5 in . high, caught 15 fragments. The regression line relating velocity to mass for these

\footnotetext{
*For a more complete discussion of this assumption see Chap. 3.
}
missiles is shown in Fig. 6.14. The slope of the line appears to be "wild" owing to a small range in mass along with a relatively large range in velocity.

Data for 16 fragments of window glass from installation 3G9 are plotted in Fig. 6.15. Because the distance the missiles traveled was greater than that for trap 3G8b, discussed above, the predicted-velocity lines are slightly higher for installation 3G9. The geometric mean of the measured velocities was also somewhat greater for the missiles traversing the longer distance-104 compared with \(96 \mathrm{ft} / \mathrm{sec}\) for installation 3G9 and trap 3G8b, respectively.

\subsection*{6.2.4 Reinforced Concrete-block House, Traps 3G10a to 3G11e}
(a) General. Locations of the seven traps used in the concrete-block house are indicated on the floor plan presented in Fig. 6.16. Elevation views of the stacked traps are shown at the bottom of the figure. The bedroom window facing GZ, which was 3 ft high and 6 ft wide, was 3 ft 7 in . above floor level. There were three panes in the horizontal direction and three in the vertical. Each pane was 23.5 in . wide and 11.5 in . high. Panes in the living-room window were the same size. This window consisted of 20 panes: five in the horizontal direction and four in the vertical. The window, which was 10 ft wide and 4 ft high, was 2 ft 7 in . above floor level.
(b) Bedroom Traps, 3G10a to 3G10c. The traps that were placed in the bedroom are shown in Fig. 6.17. Note the iron straps holding the traps together and the chain used to anchor the stack against the wall.

The missile-collecting area of each trap was divided into nine segmental areas, and the number of missiles caught per square foot was computed for each. These numbers were plotted at the midpoint of the appropriate areas as illustrated in Fig. 6.18. Contour lines were then drawn to connect points of equal spatial density. The heights of the top and the bottom of the bedroom window from which the missiles originated are indicated as dashed lines. The highest density ( 82 missiles per square foot) occurred just below the height of the bottom of the window and the lowest ( 12 missiles per square foot) near the center-height of the window.

Data for the bottom trap of the trio are presented in Fig. 6.19. Predicted-velocity lines were computed for glass in normal orientation (flat) to the wind. However, on this chart predicted velocities are also shown for fragments in edgewise orientation, indicating a surprisingly small effect due to orientation with respect to the wind. \({ }^{4}\)

Figures 6.20 and 6.21 contain the data obtained for the middle and top traps, respectively. Though the number of missiles caught in each trap was different, the velocities measured were quite similar and the masses were only slightly different (missiles in the top trap were somewhat heavier).
(c) Living-room Traps, 3G11a to 3G11e. Figure 6.22 is a postshot view of installation 3G11. Two traps are shown on the floor, a dog trap (empty) above them, and two more traps on top. This assembly was held together and to the wall with angle iron.

A spatial-distribution chart similar to the one described for installation 3 G 10 is presented in Fig. 6.23 for missiles caught at installation 3G11. Unfortunately, no data were obtained in the most interesting region, the central region occupied by the dog trap. The contour lines (dashed) extrapolated to this region are of dubious value. A total of 500 missiles was caught in the lower traps, whereas only 354 were recovered from the upper two. This indicates a tendency for the fragments to fall during transit.

Velocity and mass data for the 854 missiles caught at installation 3G11 are plotted by trap in Figs. 6.24 to 6.27. In general, the velocity points lie between the two predicted-velocity lines, based on blast waves with peak overpressures of 3.85 (see Fig. 6.3) and 8.66 psi (reflected value for incident shock overpressure of 3.85 psi ).

Attention is called to the results of two additional analyses which were made for the data from trap 3G11e (Fig. 6.27). Whether or not the slope of the regression line should be the same for large as for small fragments was determined by making one analysis for missiles with masses less than 219.5 mg and another for those with masses greater than 219.5 mg . The regression slopes found were somewhat different. This may be attributed in part to the variation of threshold velocity with missile mass: the fact that the smaller fragments have higher threshold velocities tends to increase the average velocity of the small missiles that were
caught (Fig. 2.6). Another factor to be considered is that the smaller fragments have slightly higher acceleration coefficients \({ }^{4}\) (note higher predicted velocities for the smaller missiles, e.g., in Fig. 6.27).

\subsection*{6.2.5 Precast-concrete House, Traps 3G12a to 3G13e}

\subsection*{6.2.6 Analysis of Combined Data Obtained in Houses}

Because the experimental conditions for the traps placed in the two houses were quite similar, an analysis of the combined data was made. The results for 2523 fragments of window glass are set forth in Fig. 6.41. Essentially the same features are evident on this chart as were seen on the plots for the individual traps (e.g., Figs. 6.19 and 6.20).

Use was made of the large quantity of data obtained in houses to test the validity of the log-normal assumption for the velocity and mass distributions. The graphical test used is shown in Fig. 6.42. The ordinate of this chart is marked on the right in geometric-standard-
deviation units drawn to a linear scale and on the left in the corresponding percent-of-totalsample units used to plot the experimental data. The abscissa is a logarithmic scale used for both mass and velocity. The straight lines are a graphical representation of computed values of the geometric mean and geometric standard deviations.* Thus a comparison between the sample points (taken at arbitrary intervals) and the lines indicates that the log-normal assumption is reasonable. However, the points for small masses fall below the line. The reason for this may be that some of the small fragments were overlooked since they were difficult to locate in the absorber.

\subsection*{6.3 STATION 4.3GTS, 3750-FT RANGE}

\subsection*{6.3.1 General}

The tool-shed shelter used at this station was constructed and tested by a project in Operation Teapot (1955). The structure survived the original test and was made available in the next operation for the study of secondary missiles. The diagram in Fig. 6.43 indicates the placement of marked gravel and spheres as well as the location and size of the shelter. A 2 -in.-thick layer of type II absorber was cemented \(\dagger\) to the structure on the side facing GZ. The total missile-collecting area was 7 by 7 sq ft , equivalent to more than 16 small traps. A double layer of aluminum foil similar to that used in the other traps for thermal protection was placed over the absorber.

A total of 2328 spheres was placed at this location. Three of these were weighted croquet balls (described in Sec. 6.2.2) and 90 were \(7 / 16^{-}, 1 / 2^{-}\), and \(9 / 16^{-i n}\).-diameter steel spheres (placed as indicated in Fig. 6.43). A total of 2110 small spheres was evenly divided between the two locations indicated. \(\ddagger\) In addition to these, 125 "extra large" soda-glass spheres with an average mass of 243 mg were placed at the 11.4 - ft distance.

The placement of marked gravel indicated in Fig. 6.43 is illustrated in Fig. 6.44. Also depicted are the steel spheres in the trough-like support, the croquet ball on a long thin rod, and the tissue-paper packets on the ground containing smaller spheres.

Since the blast line was not instrumented at the range of this station ( 3750 ft ), interpolated values of peak overpressure and duration were used to compute predicted missile velocities. These interpolated quantities, recorded in Table 6.1 , were 5.34 psi and 0.675 sec for peak overpressure and duration, respectively.

Figure 6.45 is a postshot photograph of station 4.3 GTS . Shreds of aluminum foil can be seen hanging on the absorber, which was found to be in good condition. Some of the impressions made by the missiles that struck this trap are visible.

\subsection*{6.3.2 Marked-gravel Data}

Data for 16 pieces of gravel placed 11.4 ft from the station are presented in Fig. 6.46. Since both the average acceleration coefficient as a function of missile mass \({ }^{4}\) and the distance of travel are known for these stones, the predicted-velocity line should pass through the center of the measured velocity points. For all stones having masses less than 100 mg , however, the measured velocities were higher than those predicted. Missiles with lower velocities were probably present but were not caught because of insufficient impact velocity (refer to thresholdvelocity chart, Fig. 2.8).

Data for gravel missiles originating 29.2 ft from the station are shown in Fig. 6.47. The predicted velocities are higher than in the previous instance (cf. with Fig. 6.46) because the

\footnotetext{
*The straight lines were determined as follows: The geometric mean mass ( \(M_{50}\) ), for example, was plotted at zero geometric-standard-deviation units. The quantities \(\mathrm{M}_{50} \cdot \mathrm{~S}_{\mathrm{gm}}\) and \(M_{50} / S_{g m}\) were plotted at +1 and -1 geometric-standard-deviation units, respectively, where \(S_{g m}\) is the geometric standard deviation of mass.
\(\dagger\) A commercial linoleum cement was used.
\(\ddagger\) The small spheres placed at 29.2 ft were painted blue; those at 11.4 ft were not painted and are labeled in Fig. 6.43 as "clear." The percentages of the various types of spheres used in each sample were the same as those set forth in Sec. 6.2.2.
}
distance of travel was greater. There is little evidence of the threshold-velocity effect that was noted at the smaller distance. The average measured and predicted velocities are in good agreement.

A spatial-distribution chart similar to that described in Sec. 6.2 .4 is presented in Fig. 6.48 for the gravel placed at 29.2 ft . High spatial densities of missiles tend to be near the bottom of the trap due to gravity and near the outside edges due to wind streaming around the trap. It is remarkable that some missiles were caught near the top of the trap, indicating an average trajectory about \(13.5^{\circ}\) from the horizontal.

Spatial-distribution charts were also prepared for the average masses and velocities of the gravel missiles discussed in the preceding paragraph. The region of high mass at the top of the chart (Fig. 6.49) is somewhat surprising. However, examination of the previous figure reveals that the number of missiles on which this "high" was based was comparatively small. The velocity-distribution chart (Fig. 6.50) shows a marked tendency for the missiles that struck high on the trap to have high velocities.

\subsection*{6.3.3 Natural-stone Data}

Velocity and mass data for 586 natural stones trapped at station 4.3GTS are presented in Fig. 6.51. The predicted-velocity line lies reasonably near the top of the distribution of measured velocity points* except for the missiles of higher mass where the predicted-velocity line is too high relative to the measured points.

The spatial-distribution chart for natural stones (Fig. 6.52) indicates that maximum densities occurred about 3 ft from the ground at the left and right edges and in the center of the trap. As the blast winds streamed around the installation, the winds oriented toward the center of the obstacle would be diverted at a shorter distance from the trap than were the winds on either side. For this reason one might expect to find a region of high density in the center of the trap as well as on either edge. In the spatial distribution of gravel (Fig. 6.48), the fact that the high in the center was missing may be explained by the manner in which gravel was placed in front of the trap, as illustrated in Fig. 6.44.

Unlike the situation for gravel, there was a marked tendency for the heavier natural stones to impact at a relatively low level above ground (see Figs. 6.49 and 6.53). Since, for the same blast exposure, small stones acquire higher velocities than do large ones, it is not surprising to find (in Fig. 6.54) a region of high velocity at the top of the trap and a region of low velocity at the bottom. In fact, it is generally true that (see Figs. 6.53 and 6.54) regions of high or low velocity correspond to regions of low or high mass, respectively.

\subsection*{6.3.4 Sphere Data}

Complete statistical data for 18 spheres caught at this location are presented in Table 6.2. The average velocity for the largest sample caught ( 14 small glass spheres) was 29.2 per cent higher than predicted. This may be explained by the fact that the spheres of lower velocities were not caught due to insufficient penetration; i.e., the distribution of missile velocities obtained was distorted due to the inability of the trap to catch missiles whose velocities were below threshold values (see threshold-velocity chart, Fig. 2.11).

\subsection*{6.4 STATION 7GTS, 2750-FT RANGE}

\subsection*{6.4.1 General}

The overpressure vs. time data obtained at the 2750 -ft range for stations 7GTS and 7G are presented in Fig. 6.55. Correspondence between the curves for overpressure (computed for an ideal wave and measured) is generally good, the computed curve yielding a more realistic value of peak overpressure for use in predicting missile velocities. However, the measured

\footnotetext{
*Predicted velocities were made for natural stones on the assumption of optimum distance of travel for maximum velocity. Any other distance of travel would result in missile velocities being lower than the predicted values.
}
dynamic-pressure record (Fig. 6.56) is considerably different from the computed curve for an ideal wave, particularly for the first 0.15 sec .

The experiment at this station was similar to the one at station 4.3GTS. The diagram in Fig. 6.57 indicates positions of placement for gravel and spheres as well as the amount of gravel used. The number (2328) and distribution of spheres were exactly the same as those described in the second paragraph of Sec. 6.3 .1 (see also Fig. 6.44). It is to be noted, however, that the distances of placement were greater at station 7GTS since a stronger blast wave was expected at this location than at station 4.3GTS.

It was not feasible to place an extra thermal shield before the absorber as was done at selected 7G installations placed at the same range; therefore a heavier and more thermalresistant Styrofoam (absorber type III) was used here than at station 7G. As evidenced by the postshot photograph (Fig. 6.58), the absorber suffered no significant damage. As shown in this photograph, two members of one of the trough-like sphere supports are imbedded in the absorber. The dark appearance of the absorber was due mostly to impregnation of a thin outer layer of Styrofoam by dust associated with the blast wave.

\subsection*{6.4.2 Marked-gravel Data}

Results obtained for 42 pieces of gravel placed 15.5 ft from the trap are graphed in Fig. 6.59. The prediction line lies quite close to the upper standard error of estimate line, which is itself about 12 per cent above the regression line. Similar data obtained for gravel placed at 39.5 ft are shown in Fig. 6.60. Again the predicted-velocity line and the upper standard error of estimate line are near each other. The deviation of the prediction curve from the regression line is about the same as noted above for the gravel arriving from the \(15.5-\mathrm{ft}\) distance.

A spatial-distribution chart (Fig. 6.61) was prepared for the 294 pieces of gravel caught which originated 39.5 ft from the trap. Maximum missile densities occurred about 3 ft from the ground, one on the left and the other on the right side (compare Figs. 6.48 and 6.52). Missiles striking the trap 7 ft above ground level had an average trajectory about \(10^{\circ}\) from the horizontal. The spatial-distribution plot for average masses of the gravel missiles (Fig. 6.62) does not show any definite trends. The region of high mass at the upper left is somewhat surprising. It should be remembered that in this region the missile density was low, and thus the average mass ( 948 mg ) producing this "high" was based on relatively few missiles. Comparison of Figs. 6.62 and 6.63 shows, in general, regions of high or low velocity which correspond to regions of low or high mass, respectively. This agrees with observations made for natural stones caught at station 4.3GTS (refer to Sec. 6.3.3 and Figs. 6.53 and 6.54).

\subsection*{6.4.3 Natural-stone Data}

Log velocity vs. log mass is plotted in Fig. 6.64 for 1238 natural stones caught at this location. Two predicted-velocity lines are shown on this and subsequent charts for natural stones caught at station 7G. The upper line was prepared using acceleration coefficients determined for a sample of natural stones from station 4.3GTS. Because it was uncertain whether or not natural stones at other stations in the Galileo shot (Area 1, NTS) were similar to the stones at station 4.3GTS, another predicted-velocity line was prepared using acceleration coefficients for the marked gravel. The greatest difference between predicted velocities indicated by these lines occurs for the missiles of high masses. Both lines were computed for the maximum velocity attained; i.e., distance of displacement was assumed to be that which would result in maximum velocity being attained. For the large natural stones, the predicted-velocity line for gravel agrees with the measured data better than does the other prediction line. However, the reverse could be said for the missiles of low masses.

The spatial-distribution chart of natural stones caught at station 7GTS (Fig. 6.65) indicates that the highest missile densities occurred between 2.5 and 4 ft above ground level. The same three regions of high density noted for natural stones at station 4.3GTS (cf. Fig. 6.52) are present on this chart; however, in the present case the outside "highs" are farther from the edges of the trap than those observed for the station at lower overpressure.

Figure 6.66, the spatial-distribution chart, indicates that there was no definite trend in average mass of the missiles as a function of location in the trap. Owing to the fairly uniform
distribution of missile masses, there is little correspondence between regions of high or low average velocity (Fig. 6.67) and those of low or high mass, respectively.

\subsection*{6.4.4 Sphere Data}

Velocities were obtained for three \(7 / 16\)-in.-diameter steel spheres* at this location. Because the data were not significantly different from those obtained for three similar spheres at station 7G, analysis was made for the combined lot and is presented in Table 6.2. It is interesting that, although velocities evaluated for the six spheres varied from 33 to \(56 \mathrm{ft} / \mathrm{sec}\), the average was 44.5 - just \(0.5 \mathrm{ft} / \mathrm{sec}\) less than that predicted.

Statistical data for 19 "extra large" (Gx with average mass of 243 mg ) and 7 small glass spheres caught at station 7 GTS are listed in Table 6.2. The deviation of the measured from the predicted velocities for the larger glass spheres ( -13 per cent) is about the same as that noted for the marked gravel. The average velocity determined for the seven small spheres, however, was only 1.1 per cent less than that predicted.

\subsection*{6.5 STATION 7G, 2750-FT RANGE}

\subsection*{6.5.1 General}

The blast-wave measurements presented for station 7GTS in Sec. 6.4 (Figs. 6.55 and 6.56) also apply for station 7G, which was located near 7GTS (see Figs. 6.1 and 6.2) and at the same range.

The design chart in Fig. 6.68 indicates the placement of a concrete-block wall, window and plate glass, and marked gravel and spheres. The total number of spheres and amount of gravel used are specified for each location. Each sample of colored spheres consisted of the same proportions of the various types described in Sec. 5.2.2.

\subsection*{6.5.2 Concrete-block Wall, Traps 7G1a to 7G3b}
(a) General. Figure 6.69 is a preshot view of the concrete-block wall and associated traps. Trap installations were located \(10.2,20.2\), and 40.2 ft from the wall. The installation that was 20.2 ft from the wall consisted of a missile trap placed over a dog installation. \({ }^{1}\) Extra thermal shields were installed at the two most distant locations but not at the near position, which was protected from thermal radiation by the wall itself.

Figures 6.70 and 6.71 illustrate the scatter of blocks and fragments from the wall. The absorber at the installation 10.2 ft from the wall (Fig. 6.72) was ruined by the impaction of blocks and large fragments. Installation 7 G 2 , which was 40.2 ft from the wall (Fig. 6.73), was relatively undamaged in spite of numerous blocks that came to rest nearby. Some damage was noted on the right side of trap 7G3b (Fig. 6.74) resulting from impact of a large object. The debris that accumulated before the more distant traps, 7G2a and b and 7G3b (Figs. 6.73 and 6.74), appears to be less fragmented than that in front of the near traps, 7G1a and b (Fig. 6.72).
(b) Block-wall Results. Final resting positions for the larger wall fragments (whole, half, and joined blocks) are plotted in Fig. 6.75. \(\dagger\) One block (not plotted, but indicated at the top of the chart) was found as far as 403 ft downwind and 170 ft left of the center of the wall. Note the absence of blocks behind the trap installations. A study of the downwind displacement, \(d_{x}\), of these wall fragments (illustrated in Fig. 6.76) indicated an approximate log-normal distribution. \(\ddagger\) This analysis yielded a value for the geometric mean of the downwind displacement \(\mathrm{d}_{\mathrm{x} 50}\),

\footnotetext{
*The significance of translation data for spheres of this type was discussed in Secs. 4.9.6 and 5.3.1.
\(\dagger\) It was estimated that the wall originally contained 236 blocks. Without mortar the dimensions of each block were approximately 7.5 by 7.5 by 16 in . The average weight of the blocks left whole after the detonation was 33.9 lb . This weight includes that of the mortar which adhered to the blocks; the total weight of the concrete-block wall before the detonation was estimated to have been more than 4.2 tons.
\(\ddagger\) A description of this type of analysis was presented in Sec. 6.2.6.
}
of 38.34 ft (also plotted on Fig. 6.75), which means that half of the missiles were translated more than this distance and half, less. As shown in Fig. 6.76, about 10 per cent of the 155 whole and multiple blocks was displaced downwind more than 100 ft .

An additional analysis was made using all wall fragments weighing more than 0.1 lb . The plot of the mass distribution of these 1528 fragments (shown in Fig. 6.77) illustrates an approximate log-normal distribution with a geometric mean mass of 1.366 lb . The reason for the abrupt percentage increase between mass points at 31 and 39 lb is that the mass of whole blocks, some with adhering mortar, was between these values.

When all wall fragments were considered, it was found that the downwind displacement, \(d_{x}\), was neither a log-normal distribution (as was found for the larger fragments, Fig. 6.76) nor a linear-normal one. In the plot presented in Fig. 6.78 of \(d_{x}\) vs. per cent of total sample, the experimental points were fitted by "eye" with a smooth curve. The usefulness of this plot will be made clear in the following paragraph.

Dispersion of wall fragments in a direction perpendicular to the blast wind (crosswind or \(\mathrm{d}_{\mathrm{y}}\) ) was studied in the following manner: The grid illustrated in Fig. 6.75 was divided into \(10-\mathrm{ft}\)-wide strips in the \(\mathrm{d}_{\mathrm{x}}\) direction and extended as far as necessary in the \(\mathrm{d}_{\mathrm{y}}\) direction to include all fragments. Assuming that the mean \(\mathrm{d}_{\mathrm{y}}\) displacement of fragments found in each \(10-\mathrm{ft}\) \(d_{x}\) strip to be along a line perpendicular to the center of the wall ( \(\bar{d}_{y}=0\) ), a standard deviation in linear \(d_{y}, S_{d_{y}}\), was computed for each \(10-\mathrm{ft}_{\mathrm{d}}\) interval. There was considerable variability in the computed standard deviations. However, it was found that a plot of \(S_{d_{y}}\) as a function of the square of the corresponding \(d_{x}\) values resulted in a scatter of points through which a straight line could be drawn. This procedure is illustrated in Fig. 6.79. Note that the quantity plotted on the ordinate was devised so that negative \(d_{x}\) values squared would remain negative.

The data represented in Figs. 6.78 and 6.79 were used in the following way to determine the smoothed contour lines shown in Fig. 6.80 which connect points on the grid plane where the spatial densities of wall fragments are the same: Sample percentages were evaluated at each \(10-\mathrm{ft} \mathrm{d}_{\mathrm{x}}\) interval with the chart in Fig. 6.78. The number of fragments within each 10 - ft interval was determined from these figures. The spatial distribution of fragments in each strip was assumed to be gaussian with a mean \(d_{y}\) displacement of zero and standard deviation equal to that determined by the straight line in Fig. 6.79. Thus, by use of normal distribution tables, it was possible to compute spatial density as a function of \(d_{y}\) for each \(10-\mathrm{ft}\) strip. Values of \(d_{y}\) were determined for spatial densities of \(0.3,1,3,10\), and 30 fragments per 100 sq ft , as illustrated in Fig. 6.80. For the smaller densities the \(d_{y}\) dispersion became greater as the downwind distances from the wall increased. The 10 -line, however, shows the opposite effect. The points on this chart were plotted at the center of \(10-\mathrm{ft}\) squares ( 100 sq ft ), and the associated figures represent the number of fragments found in each square. Probably owing to inaccurate extrapolation, the contours to the left of the wall extend to regions where missiles were not found. The fact that the missiles found upwind of the wall were small is demonstrated by the absence of points on the chart (Fig. 6.75) in that area for whole, half, and joined blocks.

The smoothed contour lines described above and illustrated in Fig. 6.80 present a description of the average displacement of fragments to be expected from repeated experiments of a similar nature, even though they fail to describe the measured data in every detail. It is useful to note that there was no significant difference in the mass distributions for missiles displaced a short distance compared to those translated greater distances except for those small fragments displaced upwind which were discussed above.
(c) Trap Results. No data were obtained from the traps at installation 7G1 since the absorber suffered large-scale deformation from the impaction of blocks and block fragments (Fig. 6.72). Natural-stone data were obtained for the remaining three traps placed at greater distances behind the wall. These data are presented in Figs. 6.81 to 6.83. Since considerably more natural stones were caught at the locations that were uninfluenced by the wall, comparison with data presented in Sec. 6.5.3 for installations 7G4 and 7G5 indicates that the traps behind the wall may have experienced some shielding. Velocities obtained for stones from the behind-the-wall traps are generally low compared to those predicted for natural stones or for gravel for station 4.3GTS (refer to Sec. 6.4).

\subsection*{6.5.3 Spheres and Natural Stones, Traps 7G4a to 7G5b}

The two installations in the foreground of Fig. 6.84 were designed to study the translation of spheres and natural stones. The five installations appearing in the background will be discussed in succeeding sections. Figures 6.85 and 6.86 depict the appearance of installations 7G4 and 7G5 after the detonation. The thermal-shield frames were left relatively undamaged, and the absorber was found to be in good condition.

The placement of spheres for these installations was described in Fig. 6.68 and Sec. 6.5.1. Data for 196 spheres that were caught ( 10,730 were placed) are presented in Table 6.2. Since the experimental conditions were approximately the same, corresponding data for spheres from all four traps were combined in every case. Data for each type of sphere are presented separately. Since there was no significant difference between their impact velocities, no distinction was made between the small metal and nylon spheres placed at 15.5 ft and those placed at 39.5 ft . Predicted velocity and percentage deviation from the predicted velocity are listed in Table 6.2 for each distance of translation-even for the cases where the data from two or three distances were combined.

Larger samples were obtained for the glass spheres than for the other types, and the average velocities obtained for those translated 39.5 ft were significantly higher than for those arriving from 15.5 ft . It is interesting that the average velocities measured for these spheres were 10 to 15 per cent lower than those predicted - about the same deviation found for marked gravel at station 7GTS (see Figs. 6.59 and 6.60).

Natural-stone data obtained for the four traps at this location are plotted in Figs. 6.87 to 6.90. The upper* traps at each installation caught more missiles whose velocities were generally higher than did the lower traps. The maximum velocity line predicted for gravel (see Sec. 6.4) generally agrees with the higher missile velocities obtained for the upper traps. However, velocities evaluated from the ground-level traps were all considerably lower than predicted.

\subsection*{6.5.4 Window-glass and Plate-glass Installations, Traps 7G6a to 7G9b}
(a) General. Four installations were used at station 7G to investigate the translation of fragments from windows mounted in open areas (see Figs. 6.68 and 6.84). Three of these used ordinary double-strength window glass placed \(21.2,11.2\), and 6.2 ft from the trap, and the fourth used plate glass at a distance of 11.2 ft . Studies of the penetration of dogs by glass fragments were conducted by another project \({ }^{1}\) at the two 11.2 - ft installations. Natural-stone data were also obtained by all traps in this group.
(b) Installation 7G6. Figure 7G6 is a postshot view of the 7G6 traps. These traps were located 21.2 ft behind the window. Data for 221 fragments caught by the lower trap and 229 by the upper one are displayed in Figs. 6.92 and 6.93, respectively. Note that in each case the predicted-velocity line, which was computed under the assumption of no reflection, satisfactorily defines the upper limit of measured velocities. These results differ markedly from those from station 3G, especially for the traps placed inside houses where most of the velocities were above this line (e.g., see Fig. 6.41).

A few fragments at this installation struck the traps flat. It is quite probable that these missiles were oriented perpendicular to the wind during the entire trajectory from the window to the trap. A separate calibration (see Chap. 2) made for fragments impacting in this manner showed more reliability than did the general calibration for glass fragments. This was largely due to the elimination of the variable of orientation for the missiles that struck flat. Thus, for the reasons stated above, the velocities obtained for fragments that impacted flat could be expected to exhibit less variability than those for missiles that rotated during transit or after impact. Although the velocities for four flat-impact missiles caught at this location (plotted in Fig. 6.94) were fairly consistent with the predicted line, more consistent data were obtained at other locations at station 7G (discussed later in this section).

Data for natural-stone missiles caught at this installation are presented in Figs. 6.95 and 6.96. The number of missiles caught and their average velocities were lower than for the

\footnotetext{
*The designator for upper traps ends with "b" and for lower traps, with "a."
}
traps at stations 7G4 and 7G5 where windows were not present. One reason that fewer missiles were recovered from the traps behind the windows was "over saturation" of the absorber; i.e., velocity could not be determined for an object striking the absorber at the same location where another object had previously impacted. Too, it is possible that the window-frame installation afforded some shielding of the traps from natural-stone missiles originating at greater distances from the traps than that at which the window was placed.
(c) Traps 7G7b and 7G8b. Traps 7G7b and 7G8b, which were placed above dog traps ( 31.5 in . high), were located 11.2 ft behind glass installations — plate glass for trap 7G7b and window glass for trap 7G8b.

At the plate-glass installation, the absorber suffered extensive deformation from the flat impaction of large fragments. A postshot photograph was not made. However, the appearance of trap 7G7b was similar to that of the traps for installation 8P3 (shown in Fig. 4.86). Installation 7G8, which was located behind window glass, is shown in Fig. 6.97.

The data for the plate glass from trap 7G7b were divided into two groups: those for 28 fragments whose orientations in the absorber were random (Fig. 6.98) and those for 4 fragments impacting flat (Fig. 6.99). The larger sample of randomly oriented missiles showed considerable variation in velocity when compared to the smaller sample of missiles striking flat. For the group that impacted flat, the regression line passes very close to all four velocity points and is almost parallel to the predicted-velocity line. The measured velocities were about 8.7 per cent lower than those predicted. As shown by this chart (Fig. 6.99), both measured and predicted velocities are slightly higher for the larger fragments than are the corresponding velocities for the smaller missiles.

Velocities were determined for 127 fragments of window glass caught in trap 7G8b. These data, plotted in Fig. 6.100, show that the higher velocities conform roughly with the predicted line.

Data for natural stones caught in traps 7G7b and 7G8b are plotted in Figs. 6.101 and 6.102, respectively. Traps 7G7b and 7G8b were placed higher above the ground* than other stacked traps that were behind windows ( 7 G 6 b and 7 G 9 b ). It is interesting to note that the traps placed higher above the ground recorded higher velocities for natural stones (compare Figs. 6.101 and 6.102 with Figs. 6.96 and 6.109).
(d) Traps 7G9a and 7G9b. The 7G9 window-glass installation \(\dagger\) is shown in Fig. 6.103. The window and traps were 6.2 ft apart. Figure 6.104 depicts the condition of the traps after the detonation. Plant stems can be seen imbedded in the absorber and collected on the surface in front of the installation.

This installation was identical to 7G6 except for the distance between the window and the trap-21.2 ft for installation 7 G 6 and 6.2 ft for installation 7 G 9 . A comparison of the results obtained from the two installations (see Figs. 6.105 and 6.106 ) indicates that the geometric mean velocities for the fragments traveling the greater distance were somewhat higher. This could be expected from theory. \({ }^{3}\) More missiles were caught at the greater distance, possibly because their higher velocities were more important than the increased spatial dispersion which also increased with distance. This argument depends on the observation that the percentage of a sample of a given missile caught in a trap depends on the average velocity at impact: missiles striking the trap at velocities less than the threshold velocity are not caught.

Another interesting comparison between installations 7G6 and 7G9 is that only 4 of 454 fragments struck the traps flat for the longer distance of translation ( 21.2 ft for installation 7G6), whereas 18 of 403 fragments did so for the shorter distance ( 6.2 ft for installation 7G9). The velocity data for the later missile sample presented in Fig. 6.107 indicates a close correspondence, in general, with the predicted velocities. The velocities measured for the larger fragments, however, were somewhat higher than predicted.

\footnotetext{
*At station 7G the "b" traps above the "A," or dog traps, were 31.5 in. above ground and those above "a" traps were 15 in . above ground.
\(\dagger\) The window panes were painted different colors for the purpose of identification of the source. However, a separate analysis for different colored fragments was not made for this report.
}

Results obtained for natural stones at traps 7G9a and 7G9b are graphed in Figs. 6.108 and 6.109. Velocities for the ground-level trap were generally lower than for the other Trap.

\subsection*{6.5.5 Marked-gravel and Natural-stone Installation 7G10} blast data. The method for obtaining computed values of peak overpressure, \(\left(p_{s}\right)_{c}\), was discussed in Chap. 3. For station 4.3 GTS , where blast measurements were not available, \(\left(p_{s}\right)_{c}\) was obtained from interpolated values of overpressure impulse, \(\left(I_{p}\right)_{r}\), and overpressure duration, \(\left(\mathrm{t}_{\mathrm{p}}^{+}\right)_{\mathrm{r}}\). The computed values of maximum overpressure were \(3.85,5.32\), and 8.38 psi for ranges of 4700,3750 , and 2750 ft , respectively.

Assuming that shot Galileo* could be characterized as the "typical air burst" described in Effects of Nuclear Weapons, \({ }^{5}\) the yield was estimated to be 11 kt (using the blast data tabulated for stations 3 G and 7G).

\subsection*{6.7.2 Tabulated Results}

A summary of all data obtained for shot Galileo except that for the concrete-block wall (Sec. 6.5.2) is given in Table 6.2. \(\dagger\)

Since more than one type of missile was caught at many of the installations, the same trap may be listed at several locations in the table.

\subsection*{6.7.3 Station 3G, 4700-ft Range}

An attempt was made at this station to record the velocities of marked gravel, natural stones, and various types of spheres. However, only 1 sphere ( \(1 / 8\)-in.-diameter steel), 10 pieces of gravel, and 3 natural stones were trapped. The "catch" was low because of insufficient impact velocity to cause effective penetration. \(\ddagger\) It was observed that the stones that remained in the absorber did so because they happened to strike the trap with a sharp point or edge forward. The average velocities evaluated for samples biased in this manner were too high since the absorbers were calibrated for random orientations at impact.

The translation of fragments from windows was investigated at station 3G by mounting windows in open regions and by using conventional windows in two houses. Double-strength window glass ( \(1 / 8 \mathrm{in}\). thick) was used at all locations except for one outside the installation; \(1 / 4\)-in.-thick plate glass was used at this location. Impact velocities were obtained at station \(3 G\) for a total of 2603 glass fragments: 2523 of these were caught in the houses by 14 traps placed, facing GZ, about 10 ft from windows; 73 were caught in 3 traps behind window installations in open areas; and 7 were from the plate-glass installation mounted in an open area.

It was noted that the double-strength glass mounted in open areas produced much larger ( \(\mathrm{M}_{50}\) ranged from 1.44 to 3.69 g ) but fewer missiles than did the same type of glass used in the conventional manner in houses ( \(\mathrm{M}_{50}\) ranged from 0.227 to 0.557 g ). The impact velocities, however, were generally higher in the houses than in the open regions.

For both types of installations the measured velocities were higher than those which could be explained by the usual method for predicting velocities of secondary missiles. It was found that velocities for most of the fragments were between the values predicted by the usual method and those predicted assuming the maximum overpressure to be equal to the reflected value of the incident maximum overpressure.

\subsection*{6.7.4 Station 4.3GTS, 3750-ft Range}

This station consisted of one large trap constructed by cementing a 2 -in. layer of absorber on the GZ side of a cubical structure. The missile-collecting area was 7 ft wide and 7 ft high (see Fig. 6.45). Marked gravel and spheres were placed 11.4 and 29.2 ft in front of the trap.

Velocities were obtained for 765 missiles at this location - 161 pieces of marked gravel, 586 natural stones, and 18 spheres. In general, the measured and predicted velocities were in good agreement for the gravel and natural stones. The average velocity of the largest sample of spheres ( 14 soda-glass spheres) caught, however, was 29 per cent higher than the predicted velocity. This discrepancy was probably due to the relatively high threshold velocity of the absorber, i.e., spheres of lower velocity may have struck the trap but were not caught because of insufficient penetration.

\subsection*{6.7.5 Station 7GTS, 2750-ft Range}

The missile trap used at this station was similar to station 4.3GTS. Marked gravel and spheres were placed 15.5 and 39.5 ft from the trap. Impact velocities were determined for 336

\footnotetext{
*Burst height was 500 ft .
\(\dagger\) See Table 2.1 for description of absorber types.
\(\ddagger\) See discussion of threshold velocities in Sec. 2.5.
}
table 6.2-SUMMARY of results, shot galileo
(See List of Symbols:)
Regression Equation: \(\log v=a+b \log m\)


See Table 2.1 for description of absorber types.
Combination of window glass in houses from 3G10a, b, and c, 3G1Ia, b, d, and e, 3G12a, b, and c, and 3G13a, b, d, and e.
\({ }^{(2)}\) Grams.
Combination of \(\mathrm{Gr}, \mathrm{d}=15.5\), from 7GTS, 7 G 10 a , b .



\(\square\)
pieces of gravel, 1238 natural stones, and 29 spheres - a total of 1603 missiles. In general, the correspondence of measured velocities with those predicted was good; e.g., the geometric mean velocity for 294 pieces of gravel displaced 39.5 ft was \(166 \mathrm{ft} / \mathrm{sec}-12.6\) per cent lower than the predicted velocity.

\subsection*{6.7.6 Station 7G, 2750-ft Range}

Experimental studies at station 7G involved the translation of (1) debris from a concreteblock wall, 40 ft long and 64 in . high, (2) marked spheres and gravel, (3) fragments from windows mounted in open areas, and (4) natural stones.

Trap installations were placed \(10.2,20.2\), and 40.2 ft behind the concrete-block wall mentioned above. The absorber in the traps at the \(10.2-\mathrm{ft}\) installation was ruined by the impaction of blocks and block fragments. Natural stones (no block fragments) were retrieved from the traps at the other installations. Final resting positions were determined for 1528 wall fragments whose weights ranged from 0.1 to more than 100 lb (multiple blocks). The greatest downwind displacement measured for a whole block (about 34 lb ) was 403 ft . Fifty per cent of the whole and multiple blocks was found more than 38 ft from the original position of the wall. Spatial-distribution charts were prepared which illustrate the dispersion of the wall fragments crosswind as well as downwind.

Velocities were determined for 1016 fragments from four windows (one plate glass) mounted in open areas 6.2 to 21.2 ft from the trap installations. The higher velocities measured were adequately explained by the velocities predicted, using the incident maximum overpressure ( 8.38 psi ). Velocities for 26 fragments that struck the absorber flat were, in general, only slightly less than those predicted.

Data for 1139 natural stones were obtained from 15 traps at station. 7G. Two traps caught 190 pieces of gravel. Most of the measured velocities for stone (including gravel) were about the same as, or less than, those predicted.

One hundred and ninety-six spheres were caught by four traps. The largest sample of a particular type of sphere consisted of 47 small soda-glass spheres with an average mass of 41 mg. The average measured velocity for these spheres was \(166 \mathrm{ft} / \mathrm{sec}, 10.3\) per cent less than the predicted velocity of \(185 \mathrm{ft} / \mathrm{sec}\).

\section*{REFERENCES}
1. V. C. Goldizen, D. R. Richmond, T. L. Chiffelle, I. G. Bowen, and C. S. White, Missile Studies with a Biological Target, Operation Plumbbob Report, WT-1470, Jan. 23, 1961.
2. I. G. Bowen, A. F. Strehler, and M. B. Wetherbe, Distribution and Density of Missiles from Nuclear Explosions, Operation Teapot Report, WT-1168, December 1956.
3. I. G. Bowen, R. W. Albright, E. R. Fletcher, and C. S. White, A Model Designed to Predict the Motion of Objects Translated by Classical Blast Waves, USAEC Report CEX-58.9, June 29, 1961.
4. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determinations of Aerodynamic-drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.
5. Samuel Glasstone (Ed.), The Effects of Nuclear Weapons, Superintendent of Documents, U. S. Government Printing Office, Washington 25, D. C., June 1957.


Fig. 6.1-Station locations for shot Galileo in Area 1 , NTS.


Fig. 6.2-Photograph of Galileo installations taken from the 500 -ft tower at GZ. Dry lake bed can be seen in the background.


Fig. 6.3-Overpressure vs. time at station 3G.


Fig. 6.4-Layout chart for installations 3G1, 3G2, 3G3, and 3G4. The level of the missile bases is 3 in . below slab height; the bathroom shelter is 8 ft high. Roman numeral in parentheses designates type of missile absorber.


Fig. 6.5-Preshot view of installations 3G1, 3G2, 3G3, and 3G4 (from right to left). The concrete pad and bathroom shelter were remains of a rambler house that was used in Operation Teapot.


Fig. 6.6-Missiles that were set out 22.5 ft in front of installations 3 G 2 and 3 G 3 . Large steel spheres can be seen on the trough-like support on the left. The support on the right holds a weighted croquet ball. Piles of marked gravel and packets of spheres are on the concrete surface.


Fig. 6.7-Installation 3G3, postshot. Note marked gravel scattered in front of the trap.


Fig. 6.8-Layout chart for installations 3G5, 3G6, 3G7, 3G8b, and 3G9. Trap 3G8b is stacked above 3G8A.


Fig. 6.9-Preshot view of installations \(3 G 6,3 G 7,3 G 8 b\), and \(3 G 9\). The reinforced-block house in which trap installations \(3 G 10\) and \(3 G 11\) were located and a corner of the precast-concrete house in which 3 G 12 and 3 G 13 were located are also shown.


Fig. 6.10-Installation 3G6, postshot, placed 4.6 ft behind window-glass installation. Torn aluminum foil can be seen on the face of the trap.


Fig. 6.11-Installation 3G7, postshot, placed 9.6 ft behind plate-glass installation. Note dents made in absorber by missiles that were not captured.


Fig. 6.12-Analysis of window -glass missiles from installation \(3 \mathrm{G} 6: \mathrm{d}=4.6 \mathrm{ft} ; \mathrm{n}=42 ; \log \mathrm{v}=2.0540+0.0097 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=\) \(1440 \mathrm{mg} ; \mathrm{V}_{50}=122 \mathrm{ft} / \mathrm{sec}\).



Fig. 6.13-Analysis of plate-glass missiles from installation 3G7: \(d=9.6 \mathrm{ft} ; \mathrm{n}=7 ; \log \mathrm{v}=2.2374-0.0631 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.26 ; \mathrm{M}_{50}=\) \(2965 \mathrm{mg} ; \mathrm{V}_{50}=104 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.14-Analysis of window-glass missiles from trap \(3 \mathrm{G} 8 \mathrm{~b}: \mathrm{d}=10.9 \mathrm{ft} ; \mathrm{n}=15 ; \log \mathrm{v}=2.7679-0.2300 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gV}}=1.32 ; \mathrm{M}_{50}=\) \(2620 \mathrm{mg} ; \mathrm{V}_{50}=96 \mathrm{ft} / \mathrm{sec}\).
-_. - Predicted Velocity
-...-...- Predicted Velocity (Reflected)

Window Glass


Mass, mg
Fig. 6.15-Analysis of window-glass missiles from installation \(3 \mathrm{G} 9: \mathrm{d}=19.5 \mathrm{ft} ; \mathrm{n}=16 ; \log \mathrm{v}=2.0453-0.0079 \log \mathrm{~m} ; \mathbf{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{\mathbf{5 0}}=\) \(3692 \mathrm{mg} ; \mathrm{V}_{50}=104 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.16-Floor plan of reinforced concrete-block house, 4700-ft range. Traps \(3 G 10 a, b\) and \(c\), and \(3 G 11 a, b, d\) and e all have type \(I\) absorbers. The bedroom window opposite traps \(3 \mathrm{G10a}, \mathrm{~b}\) and c is 3 ft 7 in . above the floor, has nine 11.5by \(23.5-\mathrm{in}\). panes, and is 6 by 3 ft . The living-room window opposite traps 3G11a, b , d and e is 2 ft 7 in . above the floor, has twenty 11.5-by 23.5-in. panes, and is 10 by 4 ft .


Fig. 6.17-Traps 3G10a, b and c, postshot, placed in front bedroom of the reinforced concrete-block house.


Fig. 6.18-Spatial distribution of window-glass missiles in installation 3G10 traps. Numbers
indicate missiles per square foot.
- Regression Line

3G100
Window Gloss in House Regression Line
(2) Geometric Means (Mass ond Velocity)


Fig. 6.19-Analysis of window-glass missiles from trap 3G10a: \(d=10.0 \mathrm{ft} ; \mathrm{n}=70 ; \log \mathrm{v}=2.1173+0.0172 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=\) \(333 \mathrm{mg} ; \mathrm{V}_{50}=145 \mathrm{ft} / \mathrm{sec}\). Additional analysis was made for mass region above 219.5 mg .
\(\qquad\)


Fig. 6.20-Analysis of window-glass missiles from trap \(3 \mathrm{G} 10 \mathrm{~b}: \mathrm{d}=10.0 \mathrm{ft} ; \mathrm{n}=240 ; \log \mathrm{v}=2.1278+0.0150 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=\) \(314 \mathrm{mg} ; \mathrm{V}_{50}=146 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.21-Analysis of window-glass missiles from trap \(3 \mathrm{G} 10 \mathrm{c}: \mathrm{d}=10.0 \mathrm{ft} ; \mathrm{n}=134 ; \log \mathrm{v}=2.0297+0.0497 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=\) \(404 \mathrm{mg} ; V_{50}=144 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.22-Traps 3G11a, b, d and e, postshot, placed in living room of the reinforced concrete-block house. The box between the upper and lower traps contained a dog (Project 33.4). Note window frame that was dislodged by the blast wave.


Fig. 6.23-Spatial distribution of window-glass missiles in installation 3 G 11 traps. Numbers indicate missiles per square foot.


Mass, mg
Fig. 6.24-Analysis of window-glass missiles from trap 3G11a: \(d=10.0 \mathrm{ft} ; \mathrm{n}=164 ; \log \mathrm{v}=2.1153+0.0136 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.21 ; \mathrm{M}_{50}=\) \(303 \mathrm{mg} ; \mathrm{V}_{50}=141 \mathrm{ft} / \mathrm{sec}\).


Mass, mg

Fig. 6.25-Analysis of window-glass missiles from trap 3G11b: \(\mathrm{d}=10.0 \mathrm{ft} ; \mathrm{n}=336 ; \log \mathrm{v}=2.0923+0.0104 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=\) \(236 \mathrm{mg} ; \mathrm{V}_{50}=131 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.26-Analysis of window-glass missiles from trap \(3 G 11 d: d=10.0 \mathrm{ft} ; \mathrm{n}=278 ; \log \mathrm{v}=2.0628+0.0337 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=\) \(355 \mathrm{mg} ; \mathrm{V}_{50}=141 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.27-Analyses of window-glass missiles from trap 3G11e. Three analyses were made: one for total mass-range; one for smaller masses; and one for larger masses. \(\mathrm{d}=10 \mathrm{ft}\).


Fig. 6.28-Floor plan of precast-concrete house, 4700 -ft range. Traps \(3 \mathrm{G} 12 \mathrm{a}, \mathrm{b}\) and c and \(3 \mathrm{G} 13 \mathrm{a}, \mathrm{b}, \mathrm{d}\) and e all have type I absorbers. Trap 3 G 13 C is a dog trap. The bedroom window opposite traps \(3 \mathrm{G} 12 \mathrm{a}, \mathrm{b}\) and c has nine \(11.5-\mathrm{by} 23.5-\mathrm{in}\). panes, is 6 by 3 ft , and is 3 ft 6 in . above the floor. The living-room window opposite traps \(3 \mathrm{G} 13 \mathrm{a}, \mathrm{b}\), d and e, has twenty \(11.5-\mathrm{by} 23.5-\mathrm{in}\). panes, is 8 by 5 ft , and is 1 ft 6 in . above the floor. The trap elevations are the same as those in the reinforced concrete-block house. (See Fig. 6.16.)


Fig. 6.29-Traps 3G12a, b and c, postshot, placed in the front bedroom of the precast-concrete house.


Fig. 6.30-Spatial distribution of window-glass missiles in installation 3 G 12 traps. Numbers indicate missiles per square foot.


Fig. 6.31-Analysis of window-glass missiles from trap 3G12a: \(\mathrm{d}=10.7 \mathrm{ft} ; \mathrm{n}=74 ; \log \mathrm{v}=2.0902+0.0127 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=\) \(299 \mathrm{mg} ; \mathrm{V}_{50}=132 \mathrm{ft} / \mathrm{sec}\).
-


Fig. 6.32-Analysis of window-glass missiles from trap 3G12b: \(d=10.7 \mathrm{ft} ; \mathrm{n}=228 ; \log \mathrm{v}=2.0791+0.0320 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{g} v}=1.19 ; \mathrm{M}_{50}=\) \(383 \mathrm{mg} ; \mathrm{V}_{50}=145 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.33-Analysis of window-glass missiles from trap \(3 G 12 \mathrm{c}: \mathrm{d}=10.7 \mathrm{ft} ; \mathrm{n}=123 ; \log \mathrm{v}=2.1055+0.0201 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.24 ; \mathrm{M}_{50}=\) \(495 \mathrm{mg} ; \mathrm{V}_{50}=144 \mathrm{ft} / \mathrm{sec}\).
d

-
Fig. 6.34-Traps 3G13a, b, d and e, postshot, placed in the living room of the precast-concrete house. The box between upper and lower traps contained a dog; (Project 33.4).


Fig. 6.35-Postshot view of the living room of the precastconcrete house where traps \(3 \mathrm{G} 13 \mathrm{a}, \mathrm{b}\), d, and e were located. The window looks toward GZ. The door shown in the corner was blown from its original closed position on the right side of the room. The debris on the floor consisted of mostly fragments of window glass.


Fig. 6.36-Spatial distribution of window-glass missiles in installation 3G13 traps. Numbers indicate missiles per square foot.

3GI3a


Window Glass in House


Fig. 6.37-Analysis of window-glass missiles from trap 3G13a: \(d=10.7 \mathrm{ft} ; \mathrm{n}=196 ; \log \mathrm{v}=2.0528+0.0405 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=\) \(281 \mathrm{mg} ; \mathrm{V}_{50}=133 \mathrm{ft} / \mathrm{sec}\).


Mass, mg
Fig. 6.38-Analysis of window-glass missiles from trap \(3 \mathrm{G} 13 \mathrm{~b}: \mathrm{d}=10.7 \mathrm{ft} ; \mathrm{n}=150 ; \log \mathrm{v}=2.0638+0.0268 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=\) \(252 \mathrm{mg} ; \mathrm{V}_{50}=134 \mathrm{ft} / \mathrm{sec}\).

\section*{-. - - Predicted Velocity}


Predicted Velocity (Reflected)
Regression Line Standard Error of Estimate
Geometric Means (Mass and Velocity)
Window Glass in House


Mass, mg
Fig. 6.39 -Analysis of window-glass missiles from trap 3G13d: \(d=10.7 \mathrm{ft} ; \mathrm{n}=207 ; \log \mathrm{v}=2.1197+0.0124 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=\) \(328 \mathrm{mg} ; \mathrm{V}_{\mathbf{5 0}}=\mathbf{1 4 2} \mathrm{ft} / \mathrm{sec}\).


Predicted Velocity
Predicted Velocity (Reflected)
Regression Line
Standard Error of Estimate
3G13e
( Geometric Means (Mass and Velocity)
Window Glass in House


Fig. 6.40-Analysis of window-glass missiles from trap \(3 G 13 \mathrm{e}: \mathrm{d}=10.7 \mathrm{ft} ; \mathrm{n}=154 ; \log \mathrm{v}=2.0971+0.0221 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=\) \(557 \mathrm{mg} ; \mathrm{V}_{50}=144 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.41-Analysis of window-glass missiles from 14 traps (stations \(3 G 10,3 G 11,3 G 12\), and \(3 G 13\) ): \(d\) (average) \(=10.3 \mathrm{ft} ; \mathrm{n}=2523\); \(\log v=2.0913+0.0216 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.20 ; \mathrm{M}_{50}=321 \mathrm{mg} ; \mathrm{V}_{50}=140 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.42-Glass-fragment data for 14 traps in houses showing per cent of total missiles with masses or velocities less than a given value.
\(\mathrm{n}=2523\).

STATION: 4.3GTS
RANGE: 3750'
BLAST LINE: S \(30^{\circ} \mathrm{E}\) of GZ

\(A=\) Weighted Croquet Ball, 40 \(\frac{11}{2}\) Above Ground
Fig. 6.43-Layout chart for station 4.3GTS.


Fig. 6.44-Preshot photograph of station 4.3GTS planted missiles. Large steel spheres are on trough-like support; weighted croquet ball, on rod support; marked gravel, in piles on the ground; small spheres, in packets between gravel piles.


Fig. 6.45-Postshot photograph of station 4.3GTS installation. A 2 -in.-thick layer of type II missile absorber was cemented to the side facing GZ of a tool-shed shelter.
—.-. Predicted velocity
_ Predicted Velocity (Reflected)
_-_-_--- Regression Line


Fig. 6.46-Analysis of gravel missiles from station 4.3GTS: \(d=11.4 \mathrm{ft} ; \mathrm{n}=16 ; \log \mathrm{v}=2.5852-0.2323 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=182 \mathrm{mg} ;\) \(V_{50}=115 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.47-Analysis of gravel missiles from station 4.3GTS: \(d=29.2 \mathrm{ft} ; \mathrm{n}=145 ; \log \mathrm{v}=2.5803-0.2227 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=288 \mathrm{mg}\); \(\mathrm{V}_{50}=108 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.48-Spatial distribution of gravel missiles, \(d=29.2 \mathrm{ft}\), recovered from station 4.3GTS. The numbers refer to the number of missiles per square foot passing through the front surface of the trap.


Fig. 6.49-Spatial distribution of the average masses (in mg ) of gravel missiles, \(d=29.2 \mathrm{ft}\), recovered from station 4.3 GTS . The average mass of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.50-Spatial distribution of the average velocities (in \(\mathrm{ft} / \mathrm{sec}\) ) of gravel missiles, \(\mathrm{d}=29.2 \mathrm{ft}\), recovered from station 4.3 GTS. The average velocity of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.51-Analysis of natural-stone missiles from station 4.3GTS: \(n=586 ; \log v=2.4969-0.1691 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{50}=35.2 \mathrm{mg} ;\) \(V_{50}=172 \mathrm{ft} / \mathrm{sec}\).
\(\geqslant\)


Fig. 6.52-Spatial distribution of natural-stone missiles recovered from station 4.3GTS. Numbers indicate missiles per square foot.


Fig. 6.53-Spatial distribution of the average masses (in mg ) of natural-stone missiles recovered from station 4.3GTS. The average mass of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.54-Spatial distribution of the average velocities ( \(\mathrm{in} \mathbf{f t} / \mathrm{sec}\) ) of natural-stone missiles recovered from station 4.3 GTS . The average velocity of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.55-Overpressure vs. time for stations 7GTS and 7G.


Fig. 6.56-Dynamic pressure vs. time for stations 7GTS and 7G.

STATION: 7GTS
RANGE: 2750'
BLAST LINE: S \(30^{\circ} \mathrm{E}\) of GZ

\(A=\) Weighted Croquet Ball, \(40 \frac{1^{\prime \prime}}{2}\) Above Ground
Fig. 6.57-Layout chart for station 7GTS.


Fig. 6.58-Station 7GTS, postshot. This station used a 2 -in. -thick layer of type III absorber cemented to the side of the tool shed which faced GZ.
\(\begin{array}{ll}\text {-.------- } & \text { Regression Line } \\ \text { Standard Error of Estimate }\end{array}\)
7GTS


Mass, mg
Fig. 6.59 -Analysis of gravel missiles from station \(7 \mathrm{GTS}: \mathrm{d}=15.5 \mathrm{ft} ; \mathrm{n}=42 ; \log \mathrm{v}=2.4966-0.1318 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{\mathbf{5 d}}=236 \mathrm{mg} ;\)
\(\mathrm{V}_{\mathbf{5 0}}=153 \mathrm{ft} / \mathrm{sec}\).
\begin{tabular}{ll}
\(\ldots-\cdots\) & Predicted Velocity \\
Predicted Velocity (Reflected)
\end{tabular}

7GTS
( Geometric Means (Moss and velocity)


Fig. 6.60-Analysis of gravel missiles from station 7GTS: \(d=39.5 \mathrm{ft} ; \mathrm{n}=294 ; \log \mathrm{v}=2.5533-0.1397 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=239 \mathrm{mg} ;\) \(V_{50}=166 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.61-Spatial distribution of gravel missiles, \(d=39.5 \mathrm{ft}\), recovered from station 7GTS. Numbers indicate missiles per square foot.


Fig. 6.62-Spatial distribution of the average masses (in mg). of gravel missiles, \(d=39.5 \mathrm{ft}\), recovered from station 7GTS. The average mass of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.63-Spatial distribution of the average velocities (in \(\mathrm{ft} / \mathrm{sec}\) ) of gravel missiles, \(\mathrm{d}=39.5 \mathrm{ft}\), recovered from station 7GTS. The average velocity of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.64-Analysis of natural-stone missiles from station 7GTS: \(n=1238 ; \log \mathrm{v}=2.5314-0.1201 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.17 ; \mathrm{M}_{50}=40.5 \mathrm{mg}\); \(V_{50}=218 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.65-Spatial distribution of natural-stone missiles recovered from station 7GTS. Numbers indicate missiles per square foot.


Fig. 6.66-Spatial distribution of the average masses (in mg) of natural-stone missiles recovered from station 7GTS. The average mass of missiles caught within a particular area segment was plotted at the center of the segment.


Fig. 6.67-Spatial distribution of the average velocities (in \(\mathrm{ft} / \mathrm{sec}\) ) of natural-stone missiles recovered from station 7GTS. The average velocity of missiles caught within a particular area segment was plotted at the center of the segment.

\section*{76 Station, Range \(2750^{\prime}\)}

\section*{- Window glass, of thick, mounted \\ b-Plate glass, \(\frac{1}{4}\) thick, mounted}
- 2110 black spheres; 25 oo \(7_{8 \prime \prime}^{\prime \prime}\) and 2500 I" steel spheres
d-1055 red, 2110 yellow, 1055 bive spheres; \(\frac{\frac{1}{2}_{2}^{2}}{}\) steel spheres, 40
- 2110 yellow spheres, \(\frac{1}{2}\) " steel spheres, 40
\(1-\frac{1}{3} n^{3}\) yellow grovel
\(-\frac{1}{4} 1^{3}\) blue grovel
n -Thermal shield


Fig. 6.68-Layout chart for station 7G. The concrete-block wall was \(64-\mathrm{in}\). high and \(7.5-\mathrm{in}\). wide. The small letter suffix by the trap designators indicates level of the stacked traps: " \(a\) " for ground-level and " \(b\) " for one above another trap.


Fig. 6.69 - Concrete-block wall ( 64 in . high, 40 ft long, and 7.5 in . thick) at station 7G, preshot.
\(a \quad\)


Fig. 6.70-Photograph illustrating the scatter of blocks from the wall at station 7G. Traps shown are: 7Gla and b (foreground), 10.2 ft from the wall; 7G2a and b (right), 40.2 ft from the wall; and 7 G 3 b (behind 7G1a and b), 20.2 ft from the wall.


Fig. 6.71-Blocks from the concrete-block wall and traps 7Gla and b, 7G2a and b, and 7G3b, postshot.
\(\qquad\)


Fig. 6.72-Traps 7G1a and \(b\) (foreground) and 7 G 2 a and b (background). The absorbers in the installation 7 G 1 traps, 10.2 ft behind the wall, were ruined by the impact of the large heavy blocks.


Fig. 6.73-Traps 7G2a and b, 40.2 ft behind the concrete-block wall, postshot.

5



Fig. 6.74-Trap 7G3b, placed above the dog installation (Project 33.4), 20.2 ft from the concrete-block wall.


Fig. 6.75-Final positions of whole, half, and joined blocks from the concrete-block wall.


Fig. 6.76 - Downwind distribution of the whole, half, and joined blocks from the concrete-block wall.


Mass, Ibs
Fig. 6.77 - The mass distribution of all fragments with masses over 0.1 lb from the concrete-block wall.


Fig. 6.78-The downwind distribution of all fragments with masses over 0.1 lb from the concrete-block wall.


Fig. 6.79-Standard deviation of crosswind displacement \(\left(\mathrm{S}_{\mathrm{d}_{\mathrm{y}}}\right)\) vs. downwind displacement squared \(\left(d_{x}^{2} \cdot d_{x} /\left|d_{x}\right|\right)\) for all fragments with masses over 0.1 lb from the concrete-block wall.


Fig. 6.80 - Spatial distribution of all fragments with masses over 0.1 lb from the concrete-block wall

7G20

- Geomefric Means (Mass and Velocity)
-- - - Predicted Velocity (Max for NS)
- - - Predi Predicted Velocity (Max for Gr)
--------- Stondard Error of Estimote
\& Geometric Means (Mass and Velocity)



Mass, mg

Fig. 6.82-Analysis of natural-stone missiles from trap \(7 \mathrm{G} 2 \mathrm{~b}: \mathrm{n}=38 ; \log \mathrm{v}=2.5761-0.1926 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{\mathbf{5 0}}=35.9 \mathrm{mg} ; \mathrm{V}_{50}=\) \(189 \mathrm{ft} / \mathrm{sec}\).

Q Geometric Means (Mass and Velocity)


Fig. 6.83-Analysis of natural-stone missiles from trap \(7 \mathrm{G} 3 \mathrm{~b}: \mathrm{n}=31 ; \log \mathrm{v}=2.5407-0.1592 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.09 ; \mathrm{M}_{50}=34.6 \mathrm{mg} ; \mathrm{V}_{50}=\) \(198 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.84-Station 7G installation from traps 7G4a and b (in the foreground on the right) through traps 7G10a and \(b\) (in the badkground), preshot.


Fig. 6.85-Installation 7G4, postshot, showing frame that held aluminum foil for extra protection. These traps caught spheres and natural stones.


Fig. 6.86 - Installation 7G5, postshot, showing remains of frame that held aluminum foil for extra thermal protection. These traps caught spheres and natural stones.

7G4o


Fig. 6.87-Analysis of natural-stone missiles from trap 7G4a: \(n=73 ; \log v=2.5126-0.1730 \log \mathbf{m} ; \mathrm{E}_{\mathrm{gv}}=1.11 ; \mathrm{M}_{50}=53.1 \mathrm{mg} ; \mathrm{V}_{50}=\) \(164 \mathrm{ft} / \mathrm{sec}\).


Mass, mg
Fig. 6.88-Analysis of natural-stone missiles from trap 7G4b: \(n=244 ; \log v=2.5393-0.1660 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=36.9 \mathrm{mg} ; \mathrm{V}_{50}=\) \(190 \mathrm{ft} / \mathrm{sec}\).


Mass, mg
Fig. 6.89-Analysis of natural-stone missiles from trap 7G5a: \(n=54 ; \log v=2.5215-0.1893 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.11 ; \mathrm{M}_{50}=7.1 \mathrm{mg} ; \mathrm{V}_{50}=\) \(148 \mathrm{ft} / \mathrm{sec}\).
—. -. - Predicted Velocity (Max for NS
—....... Predicted Velocity (Max for Gr ) ___-_-_ Regression Line Stondard Error of Estimoie - Geometric Meons (Mass and Velocity)


Mass, mg
Fig. 6.90-Analysis of natural-stone missiles from trap 7G5b: \(n=236 ; \log v=2.5609-0.1826 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=49.8 \mathrm{mg} ; \mathrm{V}_{50}=\) \(178 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.91-Traps 7G6a and b, postshot, placed 21.2 ft behind window installation.
\(\theta\)

447


Fig. 6.92-Analysis of window-glass missiles from trap 7G6a: \(d=21.2 \mathrm{ft} ; \mathrm{n}=221 ; \log \mathrm{v}=2.2117-0.0248 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.19 ; \mathrm{M}_{50}=\) \(323 \mathrm{mg} ; \mathrm{V}_{50}=141 \mathrm{ft} / \mathrm{sec}\). The analysis of window glass impacting flat at trap 7 G 6 a is presented in Fig. 6.94.


Fig. 6.93-Analysis of window-glass missiles from trap 7G6b: \(d=21.2 \mathrm{ft} ; \mathrm{n}=229 ; \log \mathrm{v}=2.2502-0.0234 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{g} v}=1.15 ; \mathrm{M}_{50}=\) \(355 \mathrm{mg} ; \mathrm{V}_{50}=155 \mathrm{ft} / \mathrm{sec}\). The analysis of window glass impacting flat at trap 7G6b is presented in Fig. 6.94.
-_. - - Predicted Vocity
—_ Predicted Velociny Rellested
--------- Standard Error of Estimole*


Mass, mg
Fig. 6.94-Analysis of window -glass missiles that arrived flat at traps 7G6a and 7G6b (two from 7G6a and two from 7G6b). Masses of unrecovered missiles were determined from area of depression in absorber made by the glass fragment: \(d=21.2 \mathrm{ft} ; \mathrm{n}=4 ; \log \mathrm{v}=1.7958+\) \(0.1253 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.04 ; \mathrm{M}_{50}=10,387 \mathrm{mg} ; \mathrm{V}_{50}=199 \mathrm{ft} / \mathrm{sec}\).
-.-.-.- Predicted velocity (Max for NSS __- Predicted Velocity (Mox for GT) -------- Stondord Error of Estimate - Geometric Moons (Mass and Velocity)

7660

Mass, mg
Fig. 6.95 - Analysis of natural-stone missiles from trap 7G6a: \(n=26 ; \log v=2.5112-0.1729 \log m ; E_{g v}=1.11 ; M_{50}=116 \mathrm{mg} ; V_{50}=\) \(143 \mathrm{ft} / \mathrm{sec}\).
————— Predicted Velocity (Max for NS) -......- Fredicted Velocity (Max for \(\mathrm{G}_{\mathrm{r}}\) ) _-_-_-_-_ Regression Line
- Standard Error of Estimate - Standard Error of Estimate


Fig. 6.96-Analysis of natural-stone missiles from trap \(7 \mathrm{G} 6 \mathrm{~b}: \mathrm{n}=30 ; \log \mathrm{v}=2.4875-0.1441 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.12 ; \mathrm{M}_{50}=112 \mathrm{mg} ; \mathrm{V}_{50}=\) \(156 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.97-Trap 7G8b, postshot, above dog installation (Project 33.4) and 11.2 ft behind window.


Fig. 6.98-Analysis of plate-glass missiles from trap 7G7b: \(d=11.2 \mathrm{ft} ; \mathrm{n}=28 ; \log \mathrm{v}=2.2144-0.0459 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.22 ; \mathrm{M}_{50}=1189 \mathrm{mg}\); \(\mathrm{V}_{50}=118 \mathrm{ft} / \mathrm{sec}\). The analysis of plate glass impacting flat at trap 7G7b is presented in Fig. 6.99.

 Flat Plate Gloss

Mass, g
Fig. 6.99-Analysis of plate-glass missiles that arrived flat at trap 7G7b: \(d=11.2 \mathrm{ft} ; \mathrm{n}=4 ; \log \mathrm{v}=2.0829+0.0161 \log \mathrm{~m}\) (mg); \(\mathrm{E}_{\mathrm{gv}}=\) \(1.004 ; \mathrm{M}_{50}=151,000 \mathrm{mg} ; \mathrm{V}_{50}=147 \mathrm{ft} / \mathrm{sec}\). Since \(\mathrm{E}_{\mathrm{gv}}=1.004\), the standard-error-of-estimate lines fall almost upon the regression line and cannot be shown. These missiles were not recovered, but their masses were determined from areas of impression left in absorber.
,


Predicted Velocity
Predicted Velocity (Reflected)* Regression Line Standard Error of Estimate
(8) Geometric Means (Mass and Velocity)

Window Glass


Moss, mg
Fig. 6.100 - Analysis of window -glass missiles from trap 7G8b: \(d=11.2 \mathrm{ft} ; \mathrm{n}=127 ; \log \mathrm{v}=2.2133-0.0220 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=\) \(604 \mathrm{mg} ; \mathrm{V}_{50}=142 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.101—Analysis of natural-stone missiles from trap 7G7b: \(n=101 ; \log v=2.5475-0.1608 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=36.3 \mathrm{mg} ; \mathrm{V}_{50}=\)
\(198 \mathrm{ft} / \mathrm{sec}\).
\begin{tabular}{|c|c|}
\hline & Predicted Velocity (Max for NS) \\
\hline -. - & Predicted Velocity (Max for Gr) \\
\hline & Regression Line Stondard Error of Estimote \\
\hline \(\otimes\) & Geometric Meons (Mass and Velocity \\
\hline
\end{tabular}


Fig. 6.102-Analysis of natural-stone missiles from trap 7G8b: \(n=29 ; \log v=2.4335-0.0954 \log m ; E_{g v}=1.16 ; M_{50}=67.2 \mathrm{mg} V_{50}=\) \(182 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.103-Irstallation 7G9, preshot. Window is 6.2 ft from traps


Fig. 6.104-Installation 7G9 traps, postshot, placed 6.2 ft behind window.


Fig. 6.105-Analysis of window-glass missiles from trap 7G9a: \(\mathrm{d}=6.2 \mathrm{ft} ; \mathrm{n}=192 ; \log \mathrm{v}=2.1827-0.0211 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.17 ; \mathrm{M}_{50}=\) \(442 \mathrm{mg} ; \mathrm{V}_{50}=134 \mathrm{ft} / \mathrm{sec}\). The analysis of window glass impacting flat at trap 7 G 9 a is presented in Fig. 6.107.
\(h\)



Fig. 6.106-Analysis of window-glass missiles from trap 7G9b: \(d=6.2 \mathrm{ft} ; \mathrm{n}=193 ; \log \mathrm{v}=2.2084-0.0304 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.18 ; \mathrm{M}_{50}=\) \(454 \mathrm{mg} ; \mathrm{V}_{50}=134 \mathrm{ft} / \mathrm{sec}\). The analysis of window glass impacting flat at trap 7 G 9 b is presented in Fig. 6.107.

-_...-. - Predicted velocity (Reflected)*
_-_-_-_-_ Regression Line
Slandard Error of Estimot


Fig. 6.107-Analysis of window-glass missiles that arrived flat at traps 7G9a and 7G9b ( 11 from 7G9a and 7 from 7 G 9 b ): \(\mathrm{d}=6.2 \mathrm{ft}\); \(\mathrm{n}=\)
\(18 ; \log v=1.9735+0.0675 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.05 ; \mathrm{M}_{50}=6223 \mathrm{mg} ; \mathrm{V}_{50}=170 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.108-Analysis of natural-stone missiles from trap 7G9a: \(n=42 ; \log v=2.4539-0.1479 \log m ; E_{g v}=1.12 ; M_{50}=140 \mathrm{mg} ; V_{50}=\) \(137 \mathrm{ft} / \mathrm{sec}\).


Mass, mg
Fig. 6.109-Analysis of natural-stone missiles from trap 7G9b: \(n=25 ; \log v=2.4198-0.1138 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.22 ; \mathrm{M}_{50}=141 \mathrm{mg} ; \mathrm{V}_{50}=\) \(150 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.110-Installation 7C10 traps, postshot. These traps caught marked gravel and natural stones

7GIOd
Gravel ( \(d=15.5\) )


Fig. 6.111-Analysis of gravel missiles from trap 7G10a: \(d=15.5 \mathrm{ft} ; \mathrm{n}=51 ; \log \mathrm{v}=2.7053-0.2634 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=302 \mathrm{mg} ;\) \(V_{50}=113 \mathrm{ft} / \mathrm{sec}\).


Q Geometric Means (Mass and Vetocity)


Fig. 6.112-Analysis of gravel missiles from trap \(7 \mathrm{G} 10 \mathrm{~b}: \mathrm{d}=15.5 \mathrm{ft} ; \mathrm{n}=117\); \(\log \mathrm{v}=2.5716-0.1828 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.13 ; \mathrm{M}_{50}=255 \mathrm{mg} ;\) \(V_{50}=135 \mathrm{ft} / \mathrm{sec}\).
-_ - - Predicted Velocity
—...-..-. Predicted Velocity (Reflected)*
-.-.---.-- Stondard Error of Estimate


Mass, mg.
\(V_{50}=149 \mathrm{ft} / \mathrm{sec}\).


469

7610a


Fig. 6.114-Analysis of natural-stone missiles from trap \(7 \mathrm{G} 10 \mathrm{a}: \mathrm{n}=51 ; \log \mathrm{v}=2.4807-0.1590 \log \mathrm{~m} ; \mathrm{E}_{\mathrm{gv}}=1.14 ; \mathrm{M}_{50}=73.3 \mathrm{mg} ; \mathrm{V}_{50}=\) \(153 \mathrm{ft} / \mathrm{sec}\).


Fig. 6.115-Analysis of natural-stone missiles from trap 7G10b: \(n=133 ; \log v=2.5913-0.1471 \log \mathbf{m} ; \mathrm{E}_{\mathrm{gv}}=1.16 ; \mathrm{M}_{50}=56.6 \mathrm{mg} ; \mathrm{V}_{50}=\) \(183 \mathrm{ft} / \mathrm{sec}\).

\section*{Chapter 7}

\section*{DISCUSSION AND SUMMARY}

\subsection*{7.1 MISSILE STATIONS AND BLAST-WAVE PARAMETERS}

Missile studies were made in open areas at 20 different ranges from GZ, in 8 shelters at 5 ranges, and in 2 houses at the same range. These stations were located on three shots. The code names and estimated yields for these shots were Priscilla, 38 kt ; Smoky, 44.5 kt ; and Galileo, 11 kt .

Attempts were made to measure overpressure and dynamic pressure as functions of time at most of the stations located in open areas. Measured overpressure impulse and duration were used to compute the equivalent ideal, or classical, blast-wave parameters in each instance (see Chap. 3). Maximum overpressures of the equivalent ideal waves are plotted in Fig. 7.1 as a function of range from GZ (small circles). Measured maximum overpressures vs. range are plotted as small triangles on the same charts. The points shown on the Priscilla chart for the ideal wave display ar approximate linear relation (note the regression equations recorded on each chart). Because of distorted wave forms, the maximum overpressure measured for each of the three precursor stations for shot Priscilla (10P, 15P, and 20P) is lower than computed for an ideal wave with the same impulse and duration. Initial overshoot of the mechanical gauges resulted in the maximum overpressures measured at the nonprecursor stations ( 6 P and 8 P ) being higher than those for the ideal wave.

Because of the irregular nature of the terrain, the data for shot Smoky, shown in the center chart of Fig. 7.1, display greater variability than those for Priscilla. Another factor that could have contributed to the variability in the blast data is that the Smoky stations were located in three general directions from GZ, viz., south, north, and northeast. Examination of the measured overpressure vs. time records presented in Chap. 5 reveals that the wave shapes recorded at all stations except the most distant one (9S) were distorted, particularly in the first portion of the wave. As a result, the measured values of maximum overpressure are generally lower than those computed for the ideal wave. The dales on the northeast blast line were shallow in comparison to the dale of the north line (see profile charts in Figs. 5.4 and 5.77 ). Figure 7.1 reveals that, with respect to the mean values defined by the regression line, the ideal-wave overpressures associated with the shallow dales are high but that the overpressure for the deeper dale ( \(5 S\) ) is low. It is of interest to note that the maximum idealwave overpressures for the four stations on the relatively flat south line are near the regres-sion-line values even though one of these stations (4S) was inside the precursor region.

Overpressure records were obtained at only two of the three ranges where stations were located on shot Galileo (see top chart in Fig. 7.1). Neither of these records shows any indication of precursor effects (see Figs. 6.3 and 6.55). Evidence of this is that the maximum overpressures measured are higher than for the ideal waves.

Blast-wave parameters associated with the ideal wave (equivalent in impulse and duration to the measured wave) were used to compute theoretical or predicted velocities for missiles caught at each of the stations designated by code number in Fig. 7.1. The computational procedure (discussed in Chap. 3) was based on material previously reported. \({ }^{1,2}\)

\subsection*{7.2 SUMMARY OF TRAPS AND MISSILES}

The numbër of various sizes of traps used on each shot is listed in: Table 7.1. The traps labeled "small": and "medium" consisted of absorbing material placed in suitable box-like housing; the areas of absorber exposed to the blast wave were 2.745 and 3.516 sq ft , respectively (see Chap. 2). The "large" traps consisted of 2-in. layers of plastic absorber, cemented

Table 7.1-SUMMARY OF TRAPS
\begin{tabular}{lcccccc} 
\\
\hline & & No. of traps
\end{tabular}
to the walls of structures, the missile-collecting area in each case depending upon the surface available. In the three shots 162 traps, having a total exposed area for the collection of missiles of 671 sq ft , were used. Only 9 of the 162 traps were made unusable by thermal, pressure, or missile effects, and 12 others underwent the blast experience without trapping any missiles. Seven of the latter group were inside closed shelters on shot Priscilla.

Table 7.2 - SUMMARY OF OBJECTS PLACED IN FRONT OF TRAPS
\begin{tabular}{|c|c|c|c|c|c|c|c|}
\hline \multirow[b]{2}{*}{Shot} & \multicolumn{3}{|l|}{Area of glass, sq ft} & \multirow[b]{2}{*}{\[
\begin{aligned}
& \text { Gr,* } \\
& \text { cu ft }
\end{aligned}
\]} & \multirow[b]{2}{*}{MD,* pieces} & \multirow[t]{2}{*}{Large steel spheres} & \multirow[b]{2}{*}{Other spheres} \\
\hline & WGH** & . WG* & PG* & & & & \\
\hline Priscilla & 0 & \(426.7 \dagger\) & 106.7 & 10.5 & 5775 & 270 & 66,468 \\
\hline Smoky & 0 & 0 & 0 & 0 & 4400 & 405 & 0 \\
\hline Galileo & 108.8 & 160.0 & 53.3 & 2.9 & 0 & 450 & 28,742 \\
\hline Total. & 108.8 & \(586.7 \dagger\) & 160.0 & 13.4 & 10175 & 1125 & 95,210 \\
\hline
\end{tabular}
*See list of symbols.
\(\dagger\) Does not include about 2000 sq ft of glass used in the pig-pen studies. See Secs. 4.3, 4.5, 4.7, and 4.8.

Table 7.3-NUMBER OF MISSILES FOR WHICH VELOCTTIES WERE DETERMINED
\begin{tabular}{lccrcccccc}
\hline & & & & & \multicolumn{5}{c}{\begin{tabular}{c} 
Large \\
steel
\end{tabular}} \\
\multicolumn{1}{c}{ Shot } & WGH* & WG* \(^{*}\) & PG* & NS* & Gr* \(^{*}\) & MD* & spheres & spheres & Total \\
\hline Priscilla & & 3728 & 88 & 1756 & 799 & 32 & 12 & 700 & 7115 \\
Smoky & & & & 2876 & & 2 & 5 & & 2883 \\
Galileo & 2523 & 1057 & 39 & 2966 & 697 & & 11 & 233 & 7526 \\
Total & 2523 & 4785 & 127 & 7598 & 1496 & 34 & 28 & 933 & 17524 \\
\hline
\end{tabular}
*See list of symbols.
Various objects and missile-producing material placed preshot in front of the traps are listed by shot in Table 7.2. If the same material were placed at more than one distance from a trap, the missiles placed at each location were painted a different color for later identification. The "large steel spheres" listed in the table were \(7 / 16,1 / 2\), and \(9 / 16\) in. in diameter.

Trapped missiles for which velocities were determined are listed in Table 7.3 by missile type for each shot. It is interesting to note that 52 per cent of the trapped missiles was stone
(natural stone and gravel) and 42 per cent was glass fragments (window glass \(\leftarrow\) both in open . . areas and in houses - and plate glass). In addition to the missiles trapped (listed in Table 7.3), total displacements were measured for 145 large stones, 8 concrete blocks, and 8 bricks in shot Priscilla and for 1527 fragments from a concrete-block wall in shot Galileo.

\subsection*{7.3 GLASS FRAGMENTS FROM WINDOWS}

It was found that ordinary windows in houses produced large numbers of fragment missiles in comparison with isolated windows mounted in open areas. As a crude indicator of this effect, 23 missiles per square foot of glass area were caught in the two Galileo houses, whereas only 6.6 fragments per square foot of glass were trappedin the open-area installations on the same shot.

The fragments trapped in houses, and to a lesser extent those in open areas at the lower pressures, had impact velocities higher than could be explained theoretically using the parameters of the incident blast wave. The highest velocities measured could be explained, however, by arbitrarily assuming that the blast wave accelerating the missiles had a maximum overpressure equal to the reflected value (normal incidence) of the free-field maximum overpressure (see Secs. 3.5 and 6.2 .3). The fortuitousness of this computational procedure is apparent when the rather complicated phenomenon of missile production in houses is considered. First, the blast wave is reflected from windows as well as from the walls that contain the windows. Then, assuming that the windows fail but that the walls do not, the large volume of the house is filled with air streaming through relatively small window areas. Since the air flow through a window is divergent upon entering the house, the initially high air velocities rapidly decrease with distance.

It was observed that the steel window frames used in houses and in open areas were usually slightly bent in the direction of the blast wave. One frame in a house was actually blown free of its mount (see Fig. 6.22). It is doubtful that the frames would have been bent if they had not contained glass. Thus one might suppose that defractive loading contributed not only to fragmentation of the glass but also to the acquisition of an initial velocity by the window panes before fragmentation was complete (see discussion of this subject in Sec. 4.16.3). The latter effect would be more pronounced for situations where the duration of the defractive loading was greatest or where the time required for fragmentation was longest. Thus the velocities obtained for fragments from windows in open areas were higher than expected for stations where the blast waves were relatively weak but were more consistent with the predicted velocities where the blast waves were stronger.

In comparing the glass-fragment data obtained at all stations, a correspondence was noted between the geometric mean mass of the fragments caught in a trap and the geometric mean velocity. The samples containing the smaller fragments generally were the ones with the higher mean velocities. The variation of acceleration coefficient between small and large glass fragments is not large enough to explain the effect noted. An explanation is quite simple, however, if it is assumed that a relatively strong blast wave not only accelerates the fragments to higher velocities but also fragments the window glass into smaller pieces. Thus the fragments caught in the houses had smaller masses than those caught at the same range in open areas, but the reverse was true of their velocities.

It is significant that none of the fragments caught in houses impacted with the flat surface against the absorber but that 0.5 per cent of the window-glass fragments and 12.6 per cent of the plate-glass fragments caught in open areas impacted in this manner. Several factors could influence the rotation of a fragment during its travel.from the window to the trap. One is missile size* - larger fragments have higher moments of inertia and therefore-greater resistance to forces tending to cause rotation. Another phenomenon inducing rotation is turbulence of the wind, which is likely to be more pronounced inside houses than in open areas. Still another, but more subtle, phenomenon is the mechanism of breakage of window glass. Results obtained

\footnotetext{
*The largest fragments were plate glass. The window-glass fragments in open areas tended to be larger than those in houses.
}
from another study \({ }^{3}\) for low (marginal) blast pressures indicate that fragments from the center of the pane break free before those from the perimeter and therefore acquire correspondingly higher velocities. This sequence of events would not only result in an initial torque tending to cause rotation of many of the fragments but would also help explain the rather large variation in velocities measured in individual samples.

\subsection*{7.4 NATURAL STONES AND GRAVEL}

In spite of irregularities in size and shape inherent in stones, those trapped in this series of experiments furnished the best experimental evidence with which to test the validity of the theoretical model for the prediction of missile velocities (see Chap. 3). The superiority of the data for stones over those for spheres resulted from the greater abundance and general reliability of the stone data. Data for gravel that was marked for identification and placed at certain distances in front of the traps could be compared directly with theory. Since the distance of translation of the natural stones was unknown, only the maximum velocities measured could be compared with the predicted maximum velocities.

In general the data obtained for stones were consistent with theory, based on a blast wave of the ideal form, except for stations where the wave was markedly altered by precursor or hill-and-dale effects. Velocities measured for stones in the precursor region were generally higher than predicted. Both hill and dale stations at the shorter ranges yielded natural stones with velocities above the predicted values.* This was also true of a hill station at a relatively large range (Sec. 5.2.6). At the greater ranges one station placed in a long flat dale (Sec. 5.2.7) yielded stone missiles with velocities that were consistent with theory, but other velocity measurements made in more pronounced dales (Secs. 5.2 .5 and 5.4 ) were appreciably lower than those predicted.

Because of the abundance of the stone data and the interest in translational-velocity estimates for man, the theory and aerodynamic measurements documented in Refs. 1 and 2 were used to determine the theoretical relation between the maximum velocities measured for stone and those which would have been measured for an object with an acceleration coefficient equal to that for man. Empirically determined stone velocities are used to estimate the maximum velocity for man. Thus, to some extent, at least, the variations existing between the translational effects of atypical blast waves and those predicted from theory are taken into account. The theoretical ratios of the maximum velocities of man weighing 70, 100, and 165 lb to the maximum velocities for \(100-\mathrm{mg}\) stones are plotted in Fig. 7.2 for each station placed in open areas. These plots indicate that the ratio of the velocity of man to that of stone increases with overpressure as well as with yield.

The following is an example of the utilization of the information in Fig. 7.2 to estimate the velocity of a man \(\dagger\) weighing 165 lb : For station 4 S (shot Smoky) the appropriate velocity ratio read on the chart is about 0.22 . From Figs. \(5.67,5.68\), and 5.69 , it is found that \(100-\mathrm{mg}\) stones at this station had maximum velocities between 400 and \(500 \mathrm{ft} / \mathrm{sec}\). Thus the maximum velocity for a \(165-\mathrm{lb}\) man is estimated to be between 88 and \(110 \mathrm{ft} / \mathrm{sec}\). It is appropriate to note that the maximum velocity for this size man predicted for this blast situation, but as suming an ideal-wave form, is only \(66 \mathrm{ft} / \mathrm{sec}\).

\subsection*{7.5 SPHERES}

Sphere studies were made at most of the stations located in open areas and in a shelter with open entryway (see Sec. 4.13). Since the spheres used had approximately the same acceleration coefficients as man, the data obtained in the shelter has a special significance. The
*Because of the translational power of the blast wave at these ranges, stone missiles originating from a hill location may have been caught by a trap placed in the downwind dale. \(\dagger\) The field data for spheres were also used to estimate the velocity of man. See the following section.
shelter was located 900 ft from the GZ of a \(700-\mathrm{ft}\) air burst with a yield of about 38 kt (shot Priscilla). Velocities measured varied from 45 to \(159 \mathrm{ft} / \mathrm{sec}\).

Data for man-equivalent spheres; were also obtained at three stations in open areas. At station 7 G , located in the near-ideal blast region where the maximum overpressure was 8.4 psi, velocities were obtained for 11 steel spheres with diameters of \(7 / 16,1 / 2\), and \(9 / 16\) in. (see Secs. 6.4.4 and 6.5.3 and Table 6.2). These spheres have acceleration coefficients about the same as those of a \(70-100-\), and \(165-\mathrm{lb}\) man, respectively, for random orientations with respect to the wind. The averages of the measured velocities for the three sphere samples ranged from 32 to \(44 \mathrm{ft} / \mathrm{sec}\), and the deviations from the predicted velocities \({ }^{1}\) vary from 0 to 12.7 per cent.

Two \(1 / 2\)-in. steel spheres were caught at station 10 P , which was located in the precursor region on shot Priscilla (see Sec. 4.10.4 and Table 4.6). The average velocity for these spheres was \(198 \mathrm{ft} / \mathrm{sec}, 143\) per cent greater than the value predicted ( \(81 \mathrm{ft} / \mathrm{sec}\) ), based on an ideal-wave form. \({ }^{1}\) Although this deviation seems excessively high, it is about the same as those for a few of the higher velocity stones caught in the same trap (see Figs. 4.123 and 4.124.)

Data were also obtained for five \(7 / 16\)-in. steel spheres at station 4 S , which was located in the precursor region on shot Smoky (see Sec. 5.3.1 and Table 5.2). The average of the measured velocities for these spheres was \(75 \mathrm{ft} / \mathrm{sec}\), about 4 per cent less than the value predicted on the basis of an ideal blast wave. Natural stones caught at the same installation had measured velocities as much as 100 per cent greater than those predicted. Thus the sphere data from station 4 S is not consistent with the stone data at the same station or with the sphere data from station 10P. The reason for this inconsistency is not known. However, one might speculate that the spheres at station 4 S were dislodged from their mount by the earth shock prior to the arrival of the blast wave (see Figs. 5.2 and 5.62). The spheres at station 10P were suspended from a wire frame in aluminum-foil bags and thus would have been more difficult to dislodge.

The sphere velocities of the smaller spheres measured at the stations located in open areas where the blast wave was near ideal were generally in agreement with the predicted velocities provided the sample sizes were sufficiently large to make a valid comparison.* An exception to the above was encountered at some of the Priscilla stations where it was indicated that the velocity determinations were erroneous due to softening of the surface of the absorber by thermal effects. Installations on later shots were given additional thermal protection where appropriate.

\subsection*{7.6 MILITARY DEBRIS}

Velocities were evaluated for only 34 pieces of military debris or fragments of steel. These missiles represent about 0.33 per cent of the total number of pieces of debris placed in front of traps in open areas on shots Priscilla and Smoky. The largest samples were obtained at the precursor stations 10P and 15P on shot Priscilla (see Figs. 4.118 and 4.131). The military-debris data obtained at these stations were similar to that obtained for gravel in that deviations of the measured from the predicted velocities were about the same for both types of missiles at each of the stations. The velocity and mass ranges for the military debris were 110 to \(373 \mathrm{ft} / \mathrm{sec}\) and 4.495 to 289 g at station 10 P and 195 to \(301 \mathrm{ft} / \mathrm{sec}\) and 9.042 to 86 g at station 15 P .

\subsection*{7.7 SPALLATION MISSILES}

Missile traps were placed in seven underground shelters with closed entryways at ranges from 860 to 1360 ft from GZ on shot Priscilla (see Sec. 4.14). The purpose of the investigation was to measure the velocity of pieces of concrete which might spall from the shelter walls due
*Although the spheres of a particular type were relatively uniform in size and weight, the measured velocities for a given blast situation varied considerably.
to underground shock effects. No missiles were caught in any of the shelters, and postshot examination of the walls indicated that appreciable spallation had not occurred.

\subsection*{7.8 LARGE STONES, CONCRETE BLOCKS, AND BRICKS}

Total displacement, rather than velocity, was measured for 145 large stones, 8 concrete blocks, and 8 bricks placed on shot Priscilla and for 1528 fragments from a concrete wall on shot Galileo.

Groups of stones and masses varying from 0.15 to 20 kg were placed at seven ranges from GZ on shot Priscilla (see Sec. 4.15). Two concrete blocks and two bricks were also included with each group of stones. The displacement of the stones placed outside the precursor region varied from 0 to 54 ft , the smaller stones tending to travel farther than the larger ones. For the stones that were inside the precursor region, the minimum displacement measured was 235 ft and the maximum, 1814 ft . There was no significant relation between stone mass and distance of travel.

A \(40-\mathrm{ft}\) concrete-block wall was built 2750 ft from GZ on shot Galileo (see Sec. 6.5.2). A broad side of the wall was oriented toward GZ. The blast wave at this location was near ideal in form and had a peak overpressure of about 8.4 psi . One block was found as far as 403 ft from the original position of the wall; the geometric mean, or median, distance of travel for whole blocks and multiple blocks joined with mortar was 38 ft . Final positions were measured for a total of 1528 wall fragments that weighed more than 0.1 lb .

\section*{REFERENCES}
1. I. G. Bowen, R. W. Albright, E. R. Fletcher, and C. S. White, A Model Designed to Predict the Motion of Objects Translated by Classical Blast Waves, USAEC Report CEX-58.9, June 29, 1961.
2. E. R. Fletcher, R. W. Albright, V. C. Goldizen, and I. G. Bowen, Determination of Aerody-namic-drag Parameters of Small Irregular Objects by Means of Drop Tests, USAEC Report CEX-59.14, October 1961.
3. E. R. Fletcher, R. V. Taborelli, I. G. Bowen, and C. S. White, Window Glass Breakage and Translational Effects Due to Sonic Boom, USAEC Report CEX-60.4 (in preparation).


Fig. 7.1-Overpressure vs. range plots for missile stations on shots Priscilla, Smoky, and Galileo.



Fig. 7.2-Theoretical relations between maximum velocities of man and stone computed for each of the missile stations in open areas.

\section*{Appendix}

\section*{RELATION BETWEEN THE MEAN, THE GEOMETRIC MEAN, AND THE GEOMETRIC STANDARD DEVIATION FOR A LOG-NORMAL DISTRIBUTION*}

The density function of a normally distributed variable \(u\) can be stated as
\[
\begin{equation*}
\mathrm{p}(\mathrm{u})=\left\{\exp \left[-(\mathrm{u}-\overline{\mathrm{u}})^{2} /\left(2 \mathrm{~S}_{\mathrm{u}}^{2}\right)\right]\right\} /\left(\mathrm{S}_{\mathrm{u}} \sqrt{2 \pi}\right) \tag{A.1}
\end{equation*}
\]
where \(\bar{u}\) is the mean value of \(u\) and \(S_{u}\) is the standard deviation of \(u\). Now, if \(u=\log x\), the distribution is \(\log\) normal in the variable \(x\). To find \(\bar{x}\), the mean or expectation value of \(x\), multiply \(x\left(x=10^{u}\right)\) by the probability function, Eq. A.1, and integrate from \(-\infty\) to \(+\infty\).
\[
\begin{align*}
\overline{\mathbf{x}} & =\mathbf{E}(\mathbf{x})=\mathbf{E}\left(10^{u}\right)=\int_{-\infty}^{+\infty}\left[10^{u} /\left(S_{u} \sqrt{2 \pi}\right)\right]\left[\exp \left[-(u-\bar{u})^{2} /\left(2 S_{u}^{2}\right)\right] d u\right. \\
& =\int_{-\infty}^{\infty}\left[1 /\left(S_{u} \sqrt{2 \pi}\right)\right] \exp \left\{-\left[(u-\bar{u})^{2} /\left(2 S_{u}^{2}\right)\right]+u \ln 10\right\} d u \tag{A.2}
\end{align*}
\]

The next step involves rearrangement of the exponent to attain the same basic form of the exponent in Eq. A.1.
\[
\begin{align*}
\overline{\mathrm{x}}= & \exp \left\{\bar{u} \ln 10+\left[\left(\mathrm{S}_{\mathrm{u}}^{2} \ln ^{2} 10\right) / 2\right]\right\} \cdot \\
& {\left[1 /\left(\mathrm{S}_{\mathrm{u}} \sqrt{2 \pi}\right)\right] \int_{-\infty}^{+\infty} \exp \left\{-\left[\mathrm{u}-\left(\bar{u}+\mathrm{S}_{\mathrm{u}}^{2} \ln 10\right)\right]^{2} /\left(2 \mathrm{~S}_{\mathrm{u}}^{2}\right)\right\} d u } \tag{A.3}
\end{align*}
\]

The integrand is a normal probability distribution in \(u\), the quantity ( \(\bar{u}+S_{u}^{2} \ln 10\) ) being the mean value. Except for the exponent outside the integrand (which does not contain the variable \(u\) ), the form is now the same as Eq. A.1. Integration of the normal probability function from \(-\infty\) to \(+\infty\) gives a value of 1 ; i.e., the probability is 1.0 that all values of \(u\) are between \(-\infty\) and \(+\infty\). Thus Eq. A. 3 reduces to
\[
\begin{align*}
\overline{\mathrm{x}} & =\exp \left\{\overline{\mathrm{u}} \ln 10+\left[\left(\mathrm{S}_{\mathrm{u}}^{2} 10\right) / 2\right]\right\} \\
& =10^{\bar{u}} \exp \left[\left(\mathrm{~S}_{\mathrm{u}} \ln 10\right)^{2} / 2\right] \tag{A.4}
\end{align*}
\]

\footnotetext{
*Log is used to designate logarithms to the base 10 , and \(\ln\), to the base e.
}

Since \(u=\log x, \bar{u}=\overline{\log x}\). By definition of the geometric mean \(\left(x_{50}\right), \log x_{50}=\overline{\log x}\). Thus \(10^{\bar{u}}=\) \(10^{\overline{\log x}}=10^{\overline{\log x_{50}}}=x_{50}\). Thus Eq. A. 4 can be written
\[
\begin{equation*}
\bar{x} / x_{50}=\exp \left[(\operatorname{Su} \ln 10)^{2} / 2\right] \tag{A.5}
\end{equation*}
\]

As defined above, \(S_{u}\) is the standard deviation in \(u\). Since \(u=\log x\), the standard deviation in \(u\) is also the standard deviation in \(\log x\), or \(S_{u}=S_{1 x}\). By definition the standard deviation in \(\log x\) is the logarithm of the geometric standard deviation in \(x, S_{1 x}=\log S_{g x}\). Thus \(S_{u}=\log S_{g x}\). By using the latter relation in Eq. A. 5 and the fact that \(\log S_{g x} \ln 10=\ln S_{g x}\), the following is obtained:
\[
\begin{equation*}
\bar{x} / x_{50}=\exp \left[\left(\log S_{g x} \ln 10\right)^{2} / 2\right]=\exp \left[\left(\ln S_{g x}\right)^{2} / 2\right] \tag{A.6}
\end{equation*}
\]

The relation expressed by Eq. A. 6 was used in the interpretation of mass and velocity samples obtained in the field study (see Sec. 2.6 and Fig. 2.12).```


[^0]:    Lovelace Foundation for Medical Education and Research Albuquerque, New Mexico
    February 1962

[^1]:    *The reasons for this are that (1) the times of exposure to thermal effects are relatively short and (2) the absorbing materials with the required mechanical properties are usually good thermal insulators (low conductance) with low heat capacities.
    $\dagger$ Manufactured by Dow Chemical Co.
    $\ddagger$ This is the same material that was used by Project 33.4 during Operation Teapot. ${ }^{1}$

[^2]:    * Libby Owens Ford, B quality.
    $\dagger$ Franklin Glass Corp.
    $\ddagger$ The fragment missiles could be given an impact orientation by lightly imbedding an edge of the fragment in the sabot.

[^3]:    *At the time this work was done an electronic computer was not available.

[^4]:    * The average density of window and plate glass was $2.42 \mathrm{~g} / \mathrm{cm}^{3}$.

[^5]:    *The techniques used are described in Sec. 2.3.3 of Ref. 1.
    $\dagger$ This drag coefficient for spheres is valid within large ranges of Reynolds numbers if the flow can be considered to be incompressible.

[^6]:    * This distance increases with overpressure, i.e., small at low pressures and large at high pressures.

[^7]:    *Missile traps, trap anchors, and window mounts are described in Chap. 2.
    $\dagger$ The method used to determine the distance that missiles were placed in front of the traps is discussed in Chap. 3.

[^8]:    *The analysis includes data for nine natural stones caught at station $6 P P$, which was at the same range as station 6 P .
    $\dagger$ This topic was discussed in Sec. 3.6.
    $\ddagger$ Studies at these stations were made in cooperation with Project 4.1 (see Sec. 4.3).
    $\S$ Total mass can be obtained by multiplying the number of missiles, $n$, by the average mass, $\overline{\mathrm{M}}$, found in Table 4.6.

    TThe procedure for computing the regression equations is outlined in Table 4.5.

[^9]:    *See Sec. 3.3.

[^10]:    *For testing purposes, a 1 -ft-square 2-in.-thick piece of type III absorber was cemented to an unused portion of the shelter wall near the installation of type IV absorber. The exact position is indicated in Fig. 4.152 by the black cement visible on the wall on the left side in the photograph. After the shot and test material was found on the floor of the shelter. Even though the sample was protected with aluminum foil, there were signs of heat distortion and compression. This result may have been due in part to the fact that the material was blown from its original position on the wall.

[^11]:    *A more complete listing of statistical parameters can be found in Table 4.6.

[^12]:    - 

[^13]:    * Detonated on a $700-\mathrm{ft}$ tower in Area 2C, Nevada Test Site.

[^14]:    * Statistical and analytical procedures were discussed in Chap. 2.
    $\dagger$ Since maximum overpressure is computed from measured impulse and duration, this statement has significance if it is assumed that impulse is more accurately determined than the duration.

