

**AERIAL GAMMA RAY AND MAGNETIC SURVEY**

**RATON BASIN PROJECT**

**SHIPROCK & GALLUP QUADRANGLES, ARIZONA/NEW MEXICO,**

**AND**

**ALBUQUERQUE QUADRANGLE, NEW MEXICO**

**CAUTION**

**This is a time release report.  
Do not release any part of this  
publication before**

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**FINAL REPORT**

**VOLUME I**



Prepared by:

**geometrics**  
Sunnyvale, California

JUNE 1979

Work Performed Under  
Bendix Field Engineering Corporation  
Grand Junction Operations, Grand Junction, Colorado  
Subcontract 78-182-L  
and  
Bendix Contract EY-76-C-13-1664

Prepared for the  
Department of Energy  
Grand Junction Office  
Grand Junction, Colorado 81501

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## ABSTRACT

From October 26 to December 1, 1978, geoMetrics, Inc., collected 9576 line miles of high sensitivity airborne radiometric and magnetic data in New Mexico and Arizona within three 1° x 2° NTMS quadrangles (Albuquerque, Gallup and Shiprock). These maps represent one half of the Raton Basin Project.

All radiometric and magnetic data were fully reduced and interpreted by geoMetrics, and are presented as 4 volumes (one Volume I and 3 Volume II's) in this report.

These three quadrangles lie primarily within the Colorado Plateau Physiographic Province. The San Juan Basin is the dominant geologic structure, occupying one third of the survey area. It is bordered by; the Defiance, Zuni, and Nacimiento Uplifts; Black Mesa Basin; and Rio Grande Rift. Two large Tertiary volcanic fields lie in the southeast. Major faulting occurs in the Rio Grande Rift and on the boundaries of the uplifts. The majority of exposed rocks are non-marine sediments of Mesozoic age.

Major uranium ore deposits occur in Mesozoic rocks within the Grants Mineral Belt (Ambrosia Lake, Laguna, Gallup, and Shiprock Districts) and Monument Valley. Scattered minor occurrences of uranium are found in Phanerozoic and Precambrian rocks throughout the area.

A total of 358 groups of statistically anomalous samples (Albuquerque, 99; Gallup, 129; Shiprock, 130) meet the criteria for valid anomalies and are discussed in the individual interpretation sections. Although a portion of the anomalies are clearly related to known uranium mines and milling operations, most lie in geologic units known either to (1) contain only minor deposits or, (2) contain no uranium deposits at all.



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## INTRODUCTION

Under the U.S. Department of Energy's (DoE), National Uranium Resource Evaluation (NURE) Program, geoMetrics, Inc., conducted a high sensitivity airborne radiometric and magnetic survey of the Shiprock, Gallup and Albuquerque 1:250,000 quadrangles, within the States of Arizona and New Mexico (see Figure 1). The objectives of the (DoE)/NURE program, of which this project is a small part, may be summarized as follows:

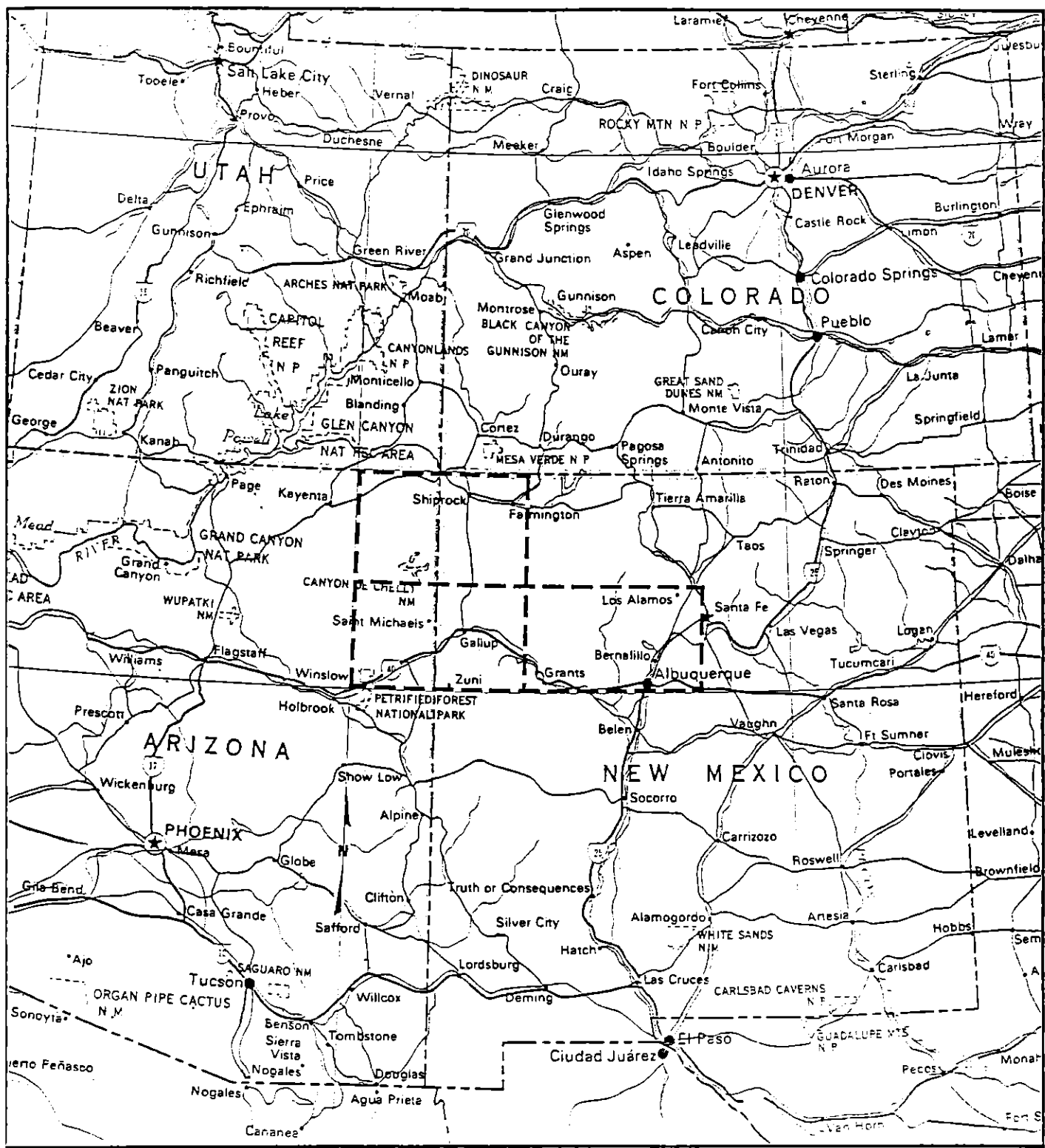
"To develop and compile geologic and other information with which to assess the magnitude and distribution of uranium resources and to determine areas favorable for the occurrence of uranium in the United States." (DoE)

As an integral part of the DoE/NURE Program, the National Airborne Radiometric Program is designed to provide cost effective, semiquantitative reconnaissance radio element distribution information to aid in the assessment of regional distribution of uraniferous materials within the United States.

All Airborne data collected during the course of this project were done so utilizing two Aerospatiale SA315B "Lama" helicopters (U.S. Registry No.s N47319 and N49531), herein designated Lama I and II, and an S2F Grumman Tracker (U.S. Registry No. N9AG). The S2F used 4096 cubic inches of NaI crystal and the Lamas used 2304 cubic inches of NaI crystal. All three aircraft utilized high sensitivity proton magnetometers (0.25 gamma).

This final report is organized in two logical sections: (a) Volume I, containing the survey description, specifications, data processing methods, interpretation methods, regional geologic review and regional survey results and (b) one volume, Volume II, for each of the three quadrangles covered by the Raton Basin Project Area. Each Volume II contains a detailed geologic summary, interpretation report, standard deviation maps, pseudo-contour maps, interpretation map, flight line and geologic base map, individual corrected profiles, computer map unit histograms and statistical tables.

All data processing, statistical analyses, and interpretation were performed at the geoMetrics computer facility, Sunnyvale, California. After processing, the corrected data were statistically evaluated to define those areas which were radiometrically anomalous relative to other areas within each computer map unit. Standard deviation maps and radiometric and magnetic profile data were first evaluated



110° W

105° W

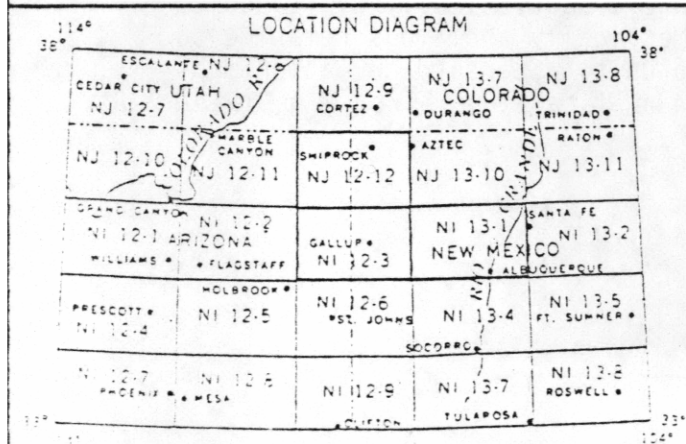
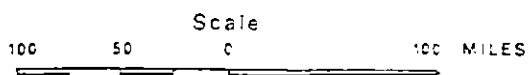


FIGURE 1  
LOCATION MAP  
ALBUQUERQUE, GALLUP AND  
SHIPROCK QUADRANGLES



individually and then integrated into a final interpretation map for each NTMS quadrangle.

Corrected profiles of all radiometric variables (total count, potassium, uranium, thorium, uranium/thorium, uranium/potassium, and thorium/potassium ratios), magnetic data, radar altimeter data, barometric altimeter data, air temperature, and airborne bismuth contributions are presented as profiles in Volume II of this report. Single record and averaged data are presented on microfiche at 1.0 second sample intervals, corrected for Compton Scatter, referenced to 400 foot mean terrain clearance, at Standard Temperature and Pressure (STP) and corrected for atmospheric bismuth in Appendix C of Volume I. Digital magnetic tapes are available from geoMetrics containing raw spectral data, single record data, magnetic data, and statistical analysis results.

## SUMMARY OF SURVEY RESULTS

### REGIONAL GEOLOGY

Of the three 1° x 2° map sheets covered by this portion of the Raton Basin Project, the Gallup and Shiprock Sheets lie entirely within the Colorado Plateau Physiographic Province. The western half of the Albuquerque quadrangle is also within the Plateau, while its eastern half contains portions of the Southern Rocky Mountains, and Basin and Range Provinces.

The San Juan Basin is the dominant geologic structure, occupying nearly one third of the report area (see Figure 2). Within the Basin, sediments are deepest on the east edge of the Shiprock quadrangle and thin to the west and south on the uplifts.

There are ten other major structures which lie on the edge of the San Juan Basin in the survey area. In the west there are: the Four Corners Platform, the Defiance Uplift and Black Mesa Basin. To the south lie the Gallup Sag and the Zuni Uplift. In the southeast, in the Albuquerque quadrangle, there are: the Nacimiento Uplift, the Puerco Platform, the Jemez and Mount Taylor Volcanic fields, the Rio Grand Rift, and the Sandia Uplift.

Major faulting in the area occurs primarily as complex normal faulting in the Rio Grande Rift and bordering faults along the uplifts.

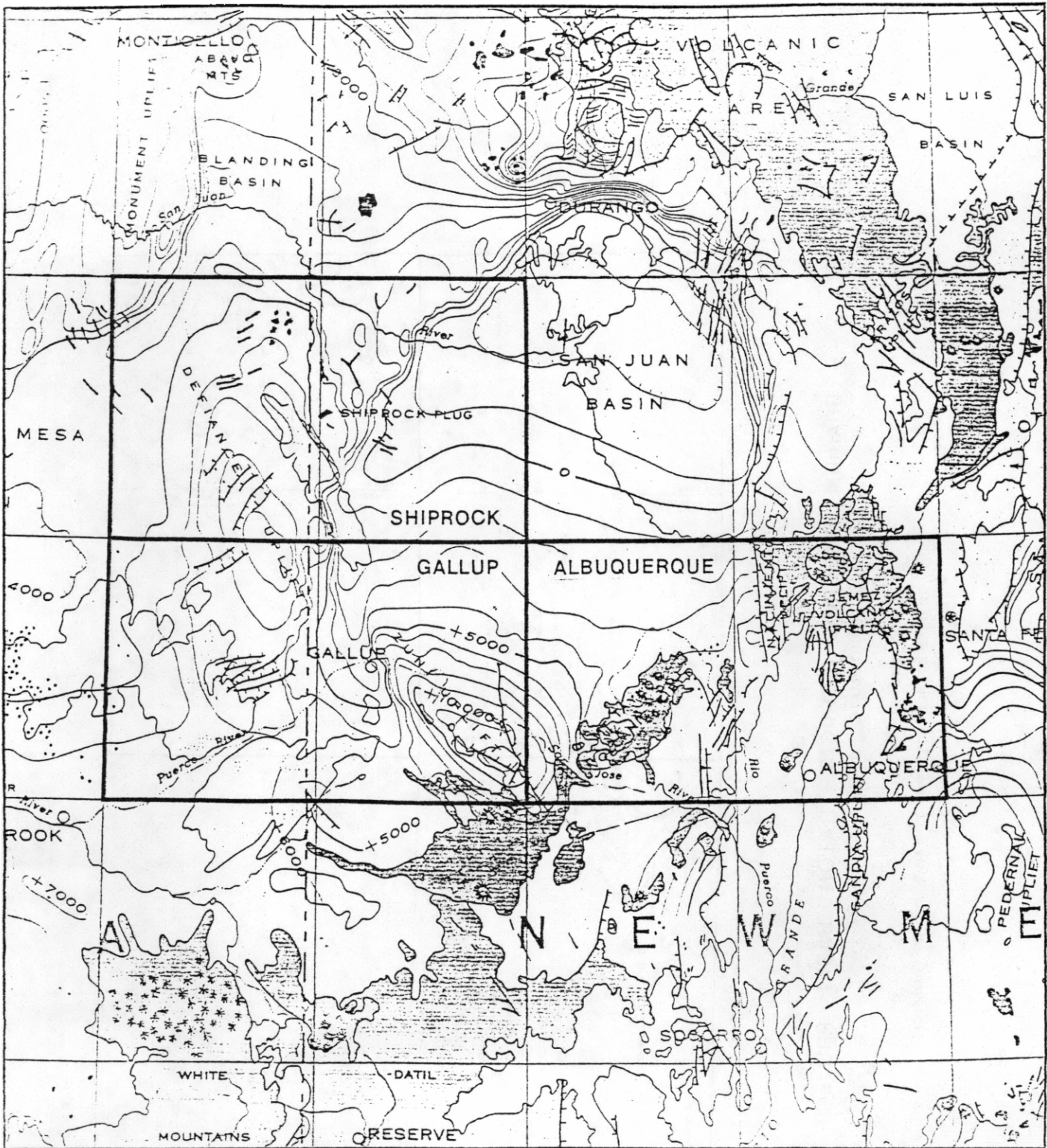
Two thirds of the outcropping rocks in the survey area consist of continental sediments of Mesozoic age. Cenozoic nonmarine sediments and Tertiary volcanics of the Mount Taylor and Jemez Volcanic field (in the Albuquerque sheet) cover another quarter of the area. Minor outcrops of Paleozoic sediments and Precambrian rocks in the uplifts cover the remainder of the quadrangles.

Major uranium ore bodies exist as epigenetic deposits in the Jurassic Morrison Formation. The deposits are primarily found in fluvial sandstones of the Westwater Canyon and Brushy Basin members of the Morrison in the Shiprock District and the Gallup and Ambrosia Lake Districts of the Grants Mineral Belt (see Figure 3). Major uranium deposits are found in the Jackpile member of the Morrison in the Laguna District. In the northwest corner of the Shiprock quadrangle, the major uranium producer is the Triassic Chinle Formation. Scattered occurrences and minor deposits are found in Mesozoic, Paleozoic and Precambrian rocks throughout the three quadrangles (Ridgley, and others 1978).

110°W

108°W

106°W



37°N  
36°N  
35°N

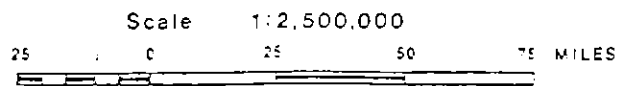
110°W

108°W

106°W

After  
USGS and AAPG  
Tectonic Map of the United States  
by  
Cohee and others (1962)

FIGURE 2  
TECTONIC STRUCTURE MAP  
ALBUQUERQUE, GALLUP AND  
SHIPROCK QUADRANGLES



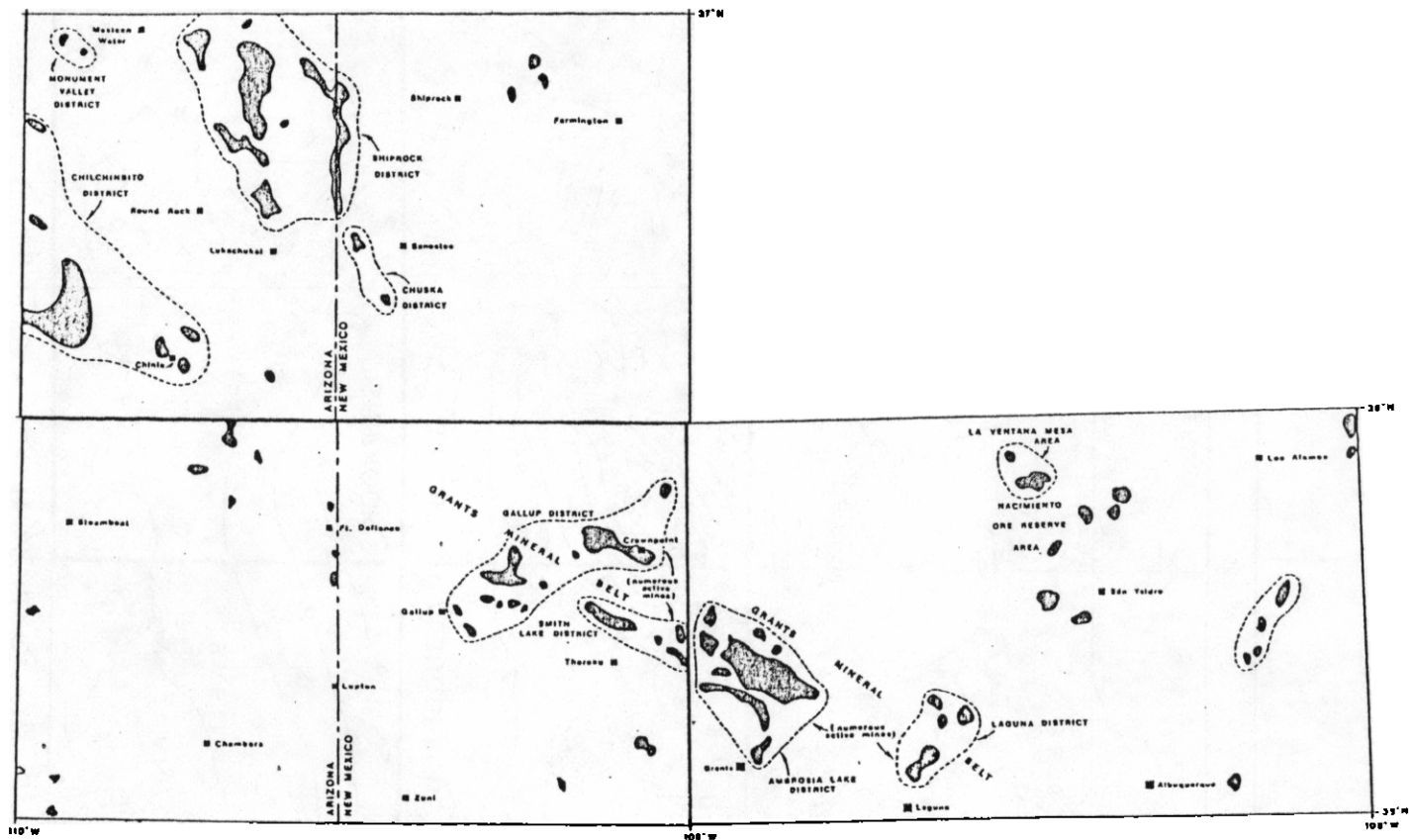


FIGURE 3  
 URANIUM MINES AND DISTRICTS - RATON BASIN PROJECT  
 APPROXIMATE SCALE 1 : 800,000

- MINING DISTRICTS
- ▨ GROUPS OF NUMEROUS MINES, CLAIMS, AND/OR PROSPECTS
- TOWNS



## INTERPRETATION

### Albuquerque

A total of 99 groups of statistically anomalous uranium samples were defined as "anomalies" and displayed in Volume II. Several anomalies are associated with the extensive uranium mining and milling operations being conducted at Ambrosia Lake. Others are associated with geologic units known to contain uranium occurrences. The greatest number of anomalies occur in the Cretaceous Menefee Formation in the San Juan Basin. Magnetic data generally within the major geologic structures, except where overlying Tertiary volcanics obscure the deeper magnetic sources.

### Gallup

For this quadrangle, 129 groups of anomalous uranium samples meet the requirements for valid "anomalies" and are displayed in Volume II. Several anomalies are associated with the major uranium deposits in the Smith Lake and Gallup Districts of the Grants Mineral Belt. Other anomalies correlate with geologic units known to contain uranium occurrences. The greatest number of anomalous samples occur in scattered Quaternary units and in the Triassic Chinle Formation near the quadrangle's western edge.

### Shiprock

Of the 130 valid "anomalies" found in this quadrangle, only a small proportion of the anomalies were associated with the known uranium producing districts. Most of the other anomalies were associated with geologic units not known for uranium occurrences; with the exception of the Menefee Formation which contained the most anomalous samples. The highest average uranium count rates were observed in the Tertiary volcanic units. The single largest statistical anomaly is associated with a uranium mine. Magnetic data generally define major structures but only marginally reflect structural features in the overlying sediments.

## OPERATIONS

### PRODUCTION SUMMARY-RATON BASIN

For the three NTMS Quadrangles comprising this portion of the "Raton Basin" Project a total of 3140 line miles, excluding reflights and overlaps and missing data were flown by the Tracker and 6436 line miles were flown by the helicopters. The production summary presented below and the detailed daily production in Appendix B describes half of the total project. This project covered three other 1° x 2° quadrangles which will be covered in a separate report.

Prior to the start of the survey operations, the three aircraft were calibrated in the DoE test pads Dynamic Test Range. The S2F Tracker and Lama I were calibrated in June, 1978 and Lama II was calibrated in July, 1978. Data collection within this portion of the project was initiated by S2F on October 26, 1978 from Gallup, New Mexico. Lama I began survey operations on November 7 from Albuquerque, New Mexico. Lama II began operations from Gallup, New Mexico. The S2F completed its portion on November 1. Lama I and II both finished on December 1 (see Figure 4).

Throughout the course of the overall project, the average ground speed maintained by the helicopters was 70 mph. The Tracker's speed ranged from 130 to 140 mph.

Overall, in excess of 95% of the data collected were within the specification limits (a sample altitude statistical distribution is shown in Figure 5).

### DATA COLLECTION PROCEDURES

#### Operating Parameters/Sampling Procedures

This survey was conducted using data collection parameters summarized below:

1. Data sampling was performed by a time-based system using 1.0 second sample intervals. All sensor data with analog output were digitally sampled at each scan based upon the clock timing rate of 1.0 seconds. The data so collected are the instantaneous values of the altimeter, temperature, pressure, and magnetometer parameters determined at the time of the data scan, but represent a count time of 1.0 seconds for the gamma ray spectrometer data.
2. The helicopter's objective ground speed was 70 mph and the Tracker's was 135 mph. Neither one of the objective speeds was exceeded unless dictated by safety.

-6-

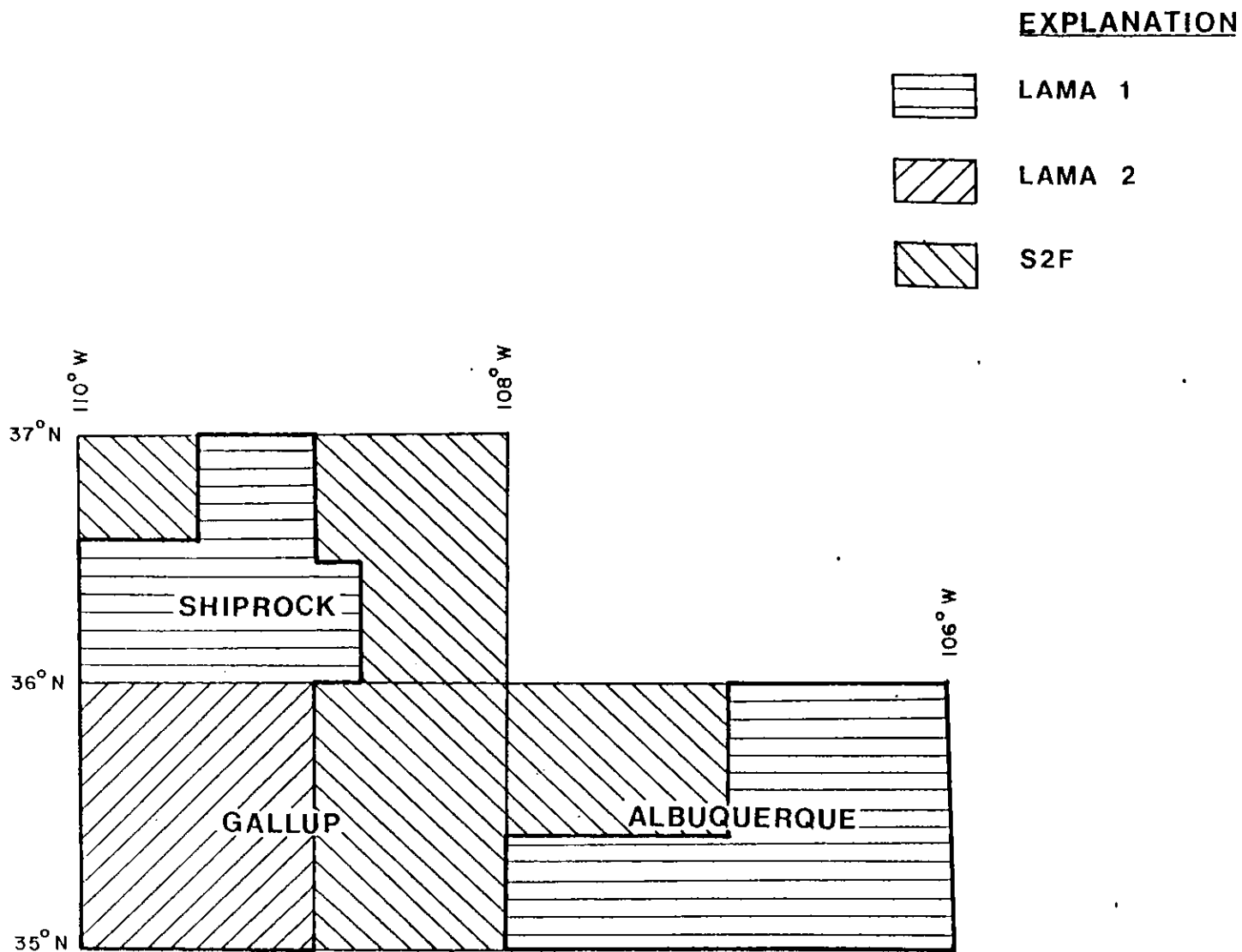
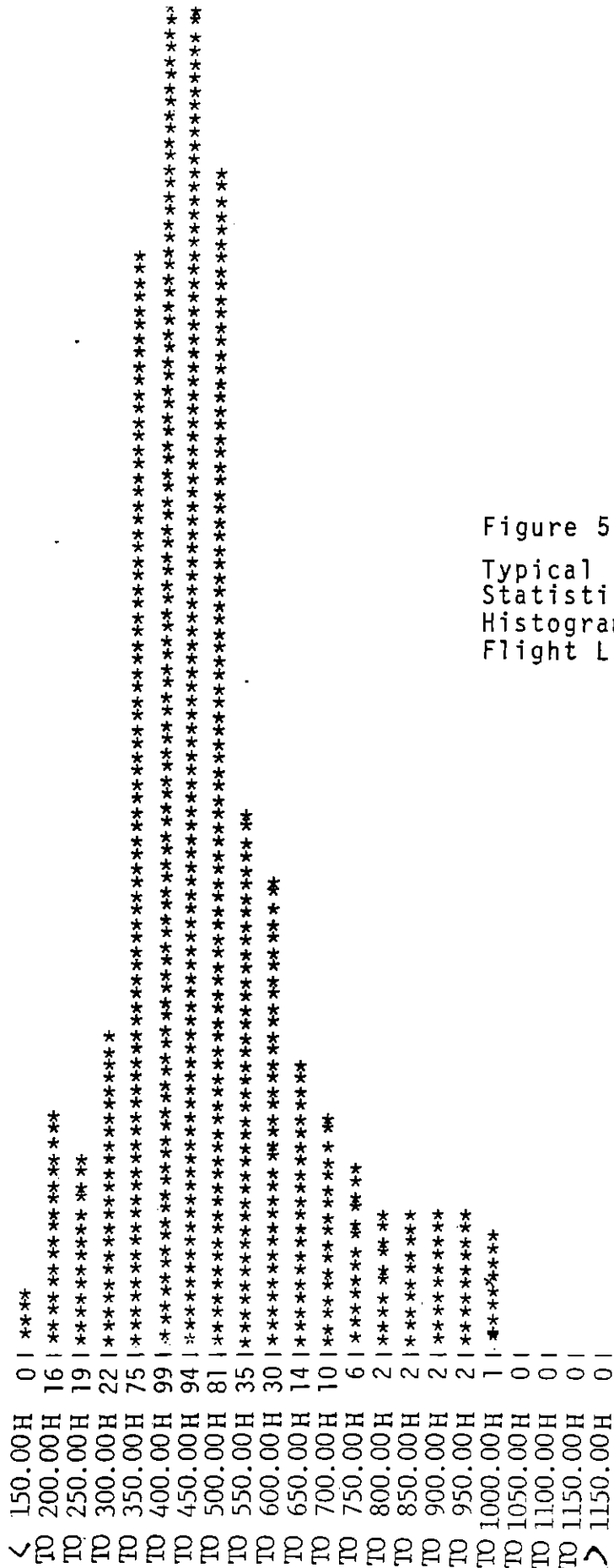


FIGURE 4  
DIAGRAM OF FIXED WING AND HELICOPTER  
SURVEY AREA

NUMBER OF OCCURRENCES



THE MINIMUM RADAR ALTMTR IS 147.500 FEET  
 THE MAXIMUM RADAR ALTMTR IS 975.000 FEET  
 THE AVERAGE RADAR ALTMTR IS 424.336 FEET  
 THE STANDARD DEVIATION IS 123.4900 FEET

(GROUND CLEARANCE IN FEET)

Figure 5  
 Typical Radar Altimeter  
 Statistical Summary  
 Histogram for Single  
 Flight Line

3. For both Lamas, the downward looking crystal volume was 2,048 cubic inches providing an objective V/V (crystal volume in cubic inches divided by ground speed in miles per hour) of 29.3 at 70 m.p.h. The S2F used 3584 cubic inches of downward looking crystal yielding the objective V/V ratio of 26.5.
4. The upward looking crystal volume was 256 cubic inches for the Lamas and 512 cubic inches for the S2F.

#### Navigation/Flight Path Recovery

Profiles were flown east-west at 3 miles (4.8 kilometers) spacing in the Gallup, Shiprock and Albuquerque quadrangles. North-south tie lines were flown at 12 miles (19.2 kilometers) spacing in all these quadrangles.

Navigation was accomplished using a combination of visual and doppler navigation techniques in the fixed wing and visual in the rotary wing aircraft. Flight lines were drawn on 1:250,000 scale NTMS quadrangle sheets for the S2F and on 1:24,000 quadrangles for the Lama. The pilot/navigator utilized these maps to provide visual navigation features. Flight lines were generally started and ended visually for both the Lama and the S2F. While doppler was used to fly a straight line between end points in the S2F, visual methods were continued for the Lama.

Simultaneously, a 35 mm tracking camera was used to record actual flight position. This camera's fiducial numbering system was directly synchronized to the digital recording system such that a one-to-one correlation between position and data could be made. Upon completion of a data collection flight, the 35 mm film was processed and actual flight path positions located on the appropriate scale map sheets. At the boundaries of the fixed wing/rotary wing areas, flight lines were flown by both the S2F and the Lamas with overlaps of 1 mile over the best available terrain (see Figure 4).

#### Infield System Calibration

Due to the complex nature of both the systems and the required data interpretation, much emphasis was placed on infield calibration of each data collection system. The objective of this calibration was to ensure continuous high quality of the data collected. The daily calibration procedures used are set forth in the following summary check list:

#### A. Pre Flight

1. Use cesium sources (same positioning on crystals every day), peak each Photomultiplier tube/crystal using the digital split-window detector of the GR-800.
2. Run full cesium spectrum on analog recorder for both down and up looking crystals. Calculate the cesium resolution (see sample in Figure 8). Run spectrum out past the K40 peak on down crystals for centering evaluation of K40 peak.
3. Use thorium sources (same position every day) check upper end of spectrum in both up and down crystals using the digital split window of the GR-800.
4. Run full thorium spectrum of down crystals on analog recorder. Check for centering of K40 and Th peaks in spectrum.

#### B. During Flight

1. Run test line at survey altitude (400 ft), for approximately five miles, prior to production data collection (record both analog and digital).
2. Prior to production data collection, the above data are evaluated to ensure  $\pm 20\%$  limits on total count compared to first test flight from that base of operations.
3. During production data collection, monitor radon analog data for unusual increases. Visually correlate these with temperature and barometric pressure.
4. Upon completion of production data collection, re-fly test line at survey altitude (400 ft). Record both analog and digital.

#### C. Post Flight

1. Verify test line total count within  $\pm 20\%$  of first test line at that base of operations.
2. Using cesium sources (same position as pre-flight), run full cesium spectrum for both down and up crystals (allow it to record through the K40 peak in the down crystals).
3. Calculate the resolution of down and up crystal pack.
4. Determine shift, if any, in K40 peak position.

### Field Digital Data Verification

At the completion of each flight, the raw digital data tapes were checked for data quality and completions on geoMetrics' G-725. The G-725 system is a totally portable mini computer (and peripherals) consisting of; an Interdata 516, two 9 track tape drives, a CRT, a Line printer, and a floppy disc. Any digital problems encountered were immediately evaluated by the electronics operators and data man, thus assuring optimum data quality.

## DATA COLLECTION SYSTEM

### ROTARY WING AIRCRAFT

The helicopters used for the survey are both Aerospatiale SA315B LAMAS, Registry Nos. N47319 (Code System No. 05) and N49531 (Code System No. 06). The SA315B LAMA was originally designed and built by Societe Nationale Industrielle Aerospatial of France to meet the requirements of the Indian Armed Forces for a medium-sized helicopter capable of working in the Himalayas. Since the craft was initially designed to haul heavy payloads in rugged mountainous terrain, its overall performance and safety features make it ideal for low level, rotary-wing airborne geophysical survey work. There is virtually no other medium-sized, rotary-wing aircraft which can carry the adequate payload at the necessary constant low airspeeds and tight terrain clearances and still maintain a wide envelope of safety, all while operating economically. Performance data for the SA315B LAMA (both general and in its present geophysical survey configuration) are given below:

Type: Turbine-driven general purpose helicopter.

Rotor System: Three-blade main and antitorque rotors. All metal main rotor blades, of constant chord, are on articulated hinges, with hydraulic drag-hinge dampers.

Rotor Drive: Main rotor driven through planetary gearbox, with free-wheel for autorotation. Take-off drive for tail rotor at lower end of main gearbox, from where a torque shaft runs to a small gearbox which supports the tail rotor and houses the pitch-change mechanism. Cyclic and collective pitch controls are powered.

Fuselage: Glazed cabin has light metal frame. Center and rear of fuselage have a triangulated steel-tube framework.

Landing Gear: Skid type, with removable wheels for ground maneuvering. Pneumatic floats for normal operation from water, and emergency flotation gear, inflatable in the air, are available.

Power Plant: One 870 shp Turbomeca Artouste IIIB turboshaft engine, derated to 550 shp. Fuel tank in fuselage center-section, with capacity of 151.3 U. S. gallons (useable) (573 litres).

Accommodation: Glazed cabin seats pilot and passenger side by side in front and three passengers behind. Provision for external sling for loads of up to 2,204 lbs. (1,000 kg). Can be equipped for rescue (hoist capacity 265 lb.; 120 kg), liaison, observation, training,



agricultural, photographic, ambulance and other duties. As an ambulance, can accommodate two stretchers and a medical attendant internally.

Dimensions, External:	Main rotor diameter	36 ft., 1-3/4 in.
	Tail rotor diameter	6 ft., 3-1/4 in.
	Main rotor blade chord (constant)	13.8 in.
	Length overall, both rotors turning	42 ft., 4-3/4 in.
	Length of fuselage	33 ft., 8 in.
	Height overall	10 ft., 1-3/4 in.
	Skid track	7 ft., 9-3/4 in.

Performance (Sea Level Standard Conditions)

		Internal		External	
		Average	Maximum	Average	Maximum
At Gross Weight	lb	3,310	4,300	4,200	5,070
Empty Weight	lb	2,216	2,216	2,216	2,216
Useful Load	lb	1,094	2,084	1,984	2,854
Sling Load (max)	lb				2,500
Cruise Speed	mph	118	118	55-75	55-75
Top Speed, Vne	mph	130	130	-	-
Useable Fuel US	gal	146	146	46	46
Service Ceiling	ft	(23,000)*	17,710	18,370	10,800
HIGE Ceiling	ft	(23,000)*	16,730	17,600	9,220
HOGC Ceiling	ft	(23,000)*	15,170	16,100	5,000
Rate of Climb SL	fpm	1,580	1,080	1,120	730
Max. Range, SL	mi	308	308	31**	31**

( ) Maximum certified altitude - 23,000 ft.

\*\* Mission radius - includes: 10 minutes fuel reserve  
3 minutes SL Hover  
Return with no load

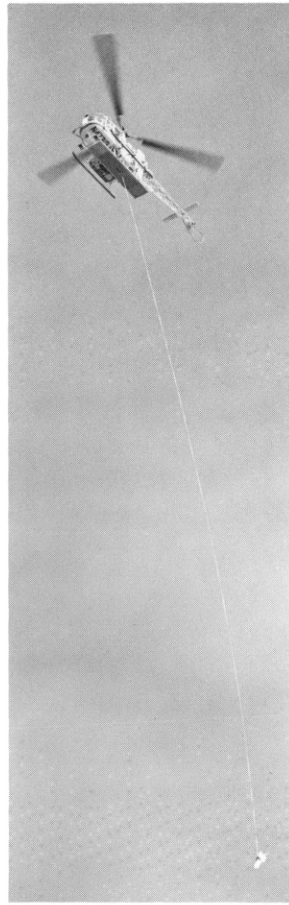
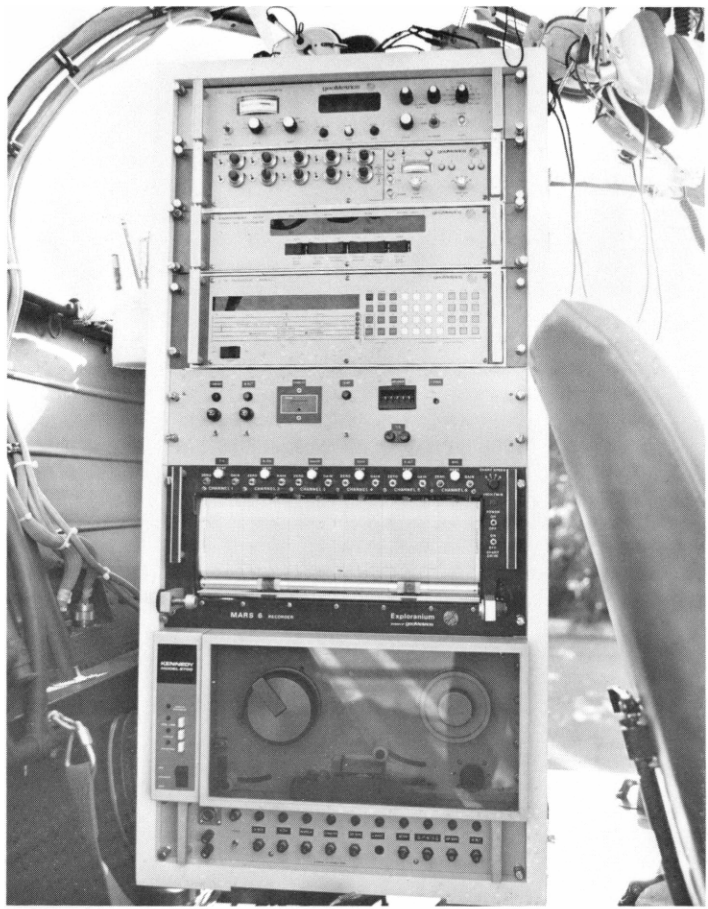
## Present geophysical Configuration

Lama Weight Empty	2,193 lbs.
Maximum Fuel	900 lbs.
Geophysical Electronics	850 lbs.
Pilot	165 lbs.
Navigator	<u>175 lbs.</u>
Total	4,458 lbs.

## Electronics

The major components of the airborne data collection system are summarized below (shown pictorially in Figure 6 and schematically in Figure 8 later in this report):

1. Gamma Ray Spectrometer, geoMetrics GR-800, utilizing a dual 256 channel capability to provide spectral data in the 0.4 to 3.0 MeV range for both the downward looking and the upward looking crystal packages and coverage in the 3.0 to 6.0 MeV range for cosmic background.
2. Crystal Detector, geoMetrics Model NaI-100/CS consisting of 2048 cubic inches in the downward looking configuration and 256 cubic inches appropriately shielded in an upward looking configuration.
3. A geoMetrics Digital Data Acquisition System, Model G-714 with "read-after-write" data verification, recording the following on magnetic tape:
  - a. 512 channels of gamma ray spectrometer data
  - b. Total magnetic intensity
  - c. Fiducial number from data system/camera
  - d. Manually inserted information, i.e. date, survey area, and flight line number
  - e. Altitude from radar altimeter and barometric altimeter (by analog-to-digital conversion)
  - f. Time in days, hours, minutes and seconds



**HELICOPTER GEOPHYSICAL SURVEY SYSTEM  
(Aerospatiale SA 315B Lama)**

Ideally suited to contour flying, this exploration platform is employed on a Midwest E. R. D. A. survey along the front range of the Rocky Mountains in central Colorado. [Far left]: A single shock-mounted instrument rack includes GeoMetrics Model G-803 Proton Magnetometer (top of rack), Model GR-900-2 Detector Interface console, Model GR-800D Multichannel Gamma Ray Spectrometer and G-714 Digital Data Acquisition System. A specially designed Intervalometer console is located above the Exploranium MARS-6 six-channel Analog Recorder and the Kennedy Model 9700 Magnetic Tape Deck. A fused Power Distribution Panel is shown at the bottom of the rack. Operator's seat is folded up to the left of the instrument rack. [Left]: Magnetometer "bird" sensor is towed from a 100 ft. nylon sleeved signal cable. [Bottom left]: The Lama was outfitted at GeoMetrics manufacturing facilities in Sunnyvale, California. [Below]: A center platform, held secure by the cargo hook, contains both a Model DET-1024 and DET-1024/256 R exSquare™ detector for a total volume of 2,048 cu. in. downward-looking and 256 cu. in. upward-looking. A Bonzer altimeter and Automax flight-path recovery camera are also included on the center platform. The entire instrumentation system, including detectors and hardware weighs approximately 700 lbs. (318 kg) installed.



Figure 6

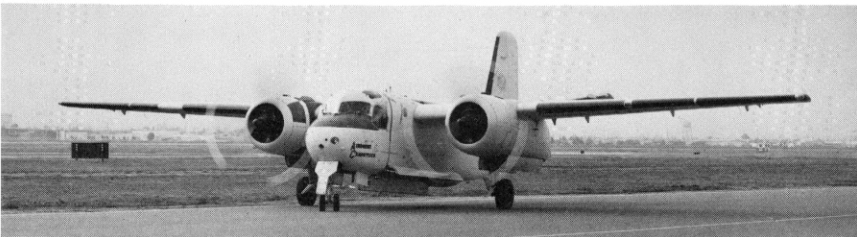
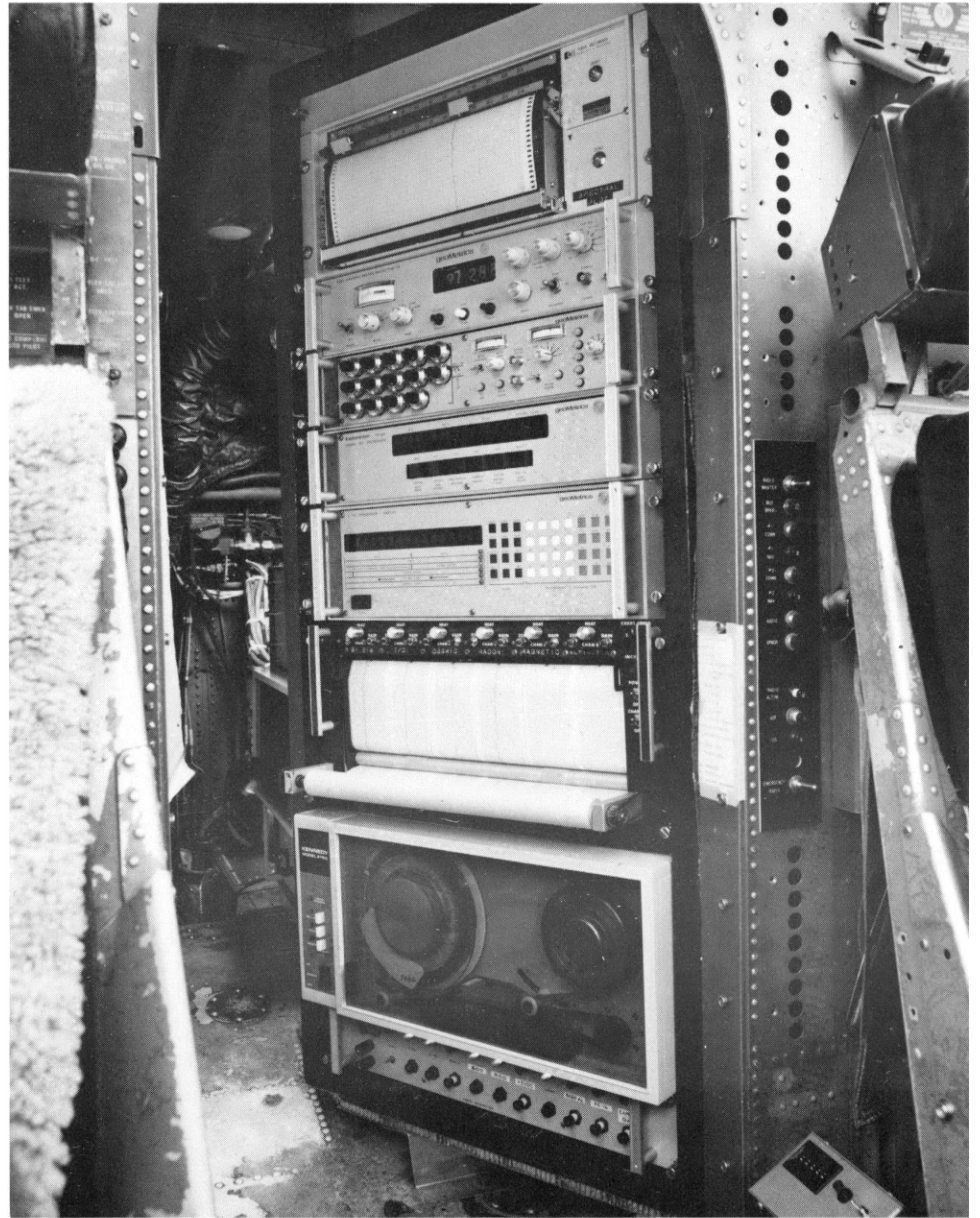


- g. Outside air temperature
- 4. Magnetometer, geoMetrics Airborne Model G-803, capable of 0.125 gamma sensitivity, but operated at 0.25 gamma sensitivity.
- 5. Radar Altimeter, Sperry Model AA200 with recording output and display and minimum altitude range of 0 to 2,500 feet.
- 6. Rosemont Barometric Altimeter with recording output and display.
- 7. Recording Thermometer for monitoring outside air temperature.
- 8. Tracking Camera. Automax 35 mm framing camera with wide angle lens to provide flight path recovery data.
- 9. Analog Recorder geoMetrics (MARS 6) to record the following data:
  - a. Bi214 using a window about the 1.76 MeV peak from the downward looking system.
  - b. Bi air background using a window about the 1.76 MeV peak from the upward looking system.
  - c. Magnetometer
  - d. Radar Altitude
  - e. Total count for downward looking system (0.4 to 3.0 MeV)
  - f. Outside air temperature
  - g. Event and time markers
- 10. HP 7155 single channel analog recorder during pre and post flight calibrations, this recorder is used to plot a full analog spectra for both the down and up crystal systems via the GR-800. Thus, a hard copy record of the data used for resolution, drift, etc., checks are available at all times. This approach provides instant verification of system parameters (refer to Figure 9 later in this report).

## FIXED WING AIRCRAFT

The fixed wing aircraft was a Grumman G-89, S2F Tracker, serial number 3, U.S. Registration N9AG, System code No. 04 (see Figure 7). This aircraft was originally designed and built by Grumman Aircraft Corporation for the U.S. Navy as a highly stable platform for carrying electronic instrumentation in search of submarines from carrier bases and/or short landing fields. Since it was designed for magnetic surveillance, it is a "magnetically clean" aircraft and thus ideal for collecting magnetic data. Overall, the aircraft's performance and safety features make it ideal for low level, fixed-wing airborne geophysical survey work. There is virtually no other fixed-wing aircraft which can carry the adequate payload at the necessary constant low airspeeds and tight terrain clearances and still maintain a wide envelope of safety. Performance data for the S2F in its present geophysical survey configuration are given below:

Aircraft empty		15,123 lbs
Electronic equipment		1,600 lbs
Main fuel usable		3,108 lbs
Auxiliary fuel usable		900 lbs
Pilot		175 lbs
Electronic operator		<u>175 lbs</u>
Maximum gross weight for geophysical survey operation		21,081 lbs
Maximum allowable aircraft gross weight		24,500 lbs
Minimum control speed	85 KIAS at	24,500 lbs
Safe single engine speed	100 KIAS at	24,500 lbs
Single engine rate of climb at	120 KIAS 390 FPM at	23,000 lbs
Rate of climb (two engines)	2,000 FPM at	5,000 ft
	120 KIAS at 23,000 lbs.	1,200 FPM at 10,000 ft
(KIAS = Knots Indicated Air Speed)		
Cruise Configuration Stalling Speed at Gross Weight		21,000 lbs
	0° Bank - 80 KIAS 45° Bank - 96 KIAS	



Left: Grumman S2F Survey Aircraft. Upper right: Geophysical instruments: G-803 Magnetometer, GR-800 Spectrometer, G-714 Data System & Recorders. Upper left: NaI exSquare™ Crystal detectors—3,072 cu.in. (50.4 l) down, 512 cu.in. (8.4 l) up. Camera: Automax G2.

Figure 7





Usable Fuel	518 U.S. Gallons	3180 lbs Mains
	150 U.S. Gallons	900 lbs Auxiliary

400 lbs per hour at 1000 feet altitude and 120 KIAS at 23,000 lbs. gross weight duration 10 hours plus, due to burn off and lower gross weight.

### Electronics

The major components of the airborne data collection system are summarized below (shown pictorially in Figure 7 and schematically in Figure 8):

1. Gamma Ray Spectrometer, geoMetrics GR-800, utilizing a dual 256 channel capability to provide spectral data in the 0.4 to 3.0 MeV range for both the downward looking and the upward looking crystal packages and coverage in the 3.0 to 6.0 MeV range for cosmic background.
2. Crystal Detector, geoMetrics model NaI-1000/CS consisting of 3584 cubic inches in the downward looking configuration and 512 cubic inches appropriately shielded in an upward looking configuration.
3. A geoMetrics digital Data Acquisition System, model G-714 with "read-after-write" data verification, recording the following on magnetic tape:
  - a. 512 channels of gamma ray spectrometer data
  - b. Total magnetic intensity
  - c. Fiducial number from data system/camera
  - d. Manually inserted information, i.e., data, survey area, and flight number
  - e. Altitude from radar and barometric altimeters (by analog-to-digital conversion)
  - f. Time in days, hours, minutes and seconds
  - g. Outside air temperature
4. Magnetometer, geoMetrics Airborne model G-803, capable of 0.125 gamma sensitivity, but operated at 0.25 gamma sensitivity.

5. Radar Altimeter, Bonzer with a linear recording output, displaying an altitude range of 0 to 2500 feet.
6. Rosemont Barometric Altimeter with recording output and display.
7. Recording Thermometer for monitoring outside air temperature.
8. Tracking Camera, Automax 35 mm framing camera with wide angle lens to provide flight path recovery data
9. Analog Recorder geoMetrics MARS 6 to record the following data:
  - a. Bi<sub>214</sub> using a window about the 1.76 MeV peak from the downward looking system
  - b. Bi air background using a window about the 1.76 MeV peak from upward looking system.
  - c. Magnetometer
  - d. Total count for downward looking system (0.4 to 3.0 MeV)
  - e. Event and time markers
10. HP 7128, two channel analog recorder to record the following data:
  - a. Outside air temperature
  - b. Barometric altimeter
  - c. Event and time markers
  - d. During system calibrations, this recorder is used to plot full analog spectra for both the down and up crystal systems via the Gr-800. Thus, a hard copy of the data used for resolutions, drift, and other checks is available at all times (refer to Figure 9). This approach provides instant verification of system parameters.

FIGURE 8

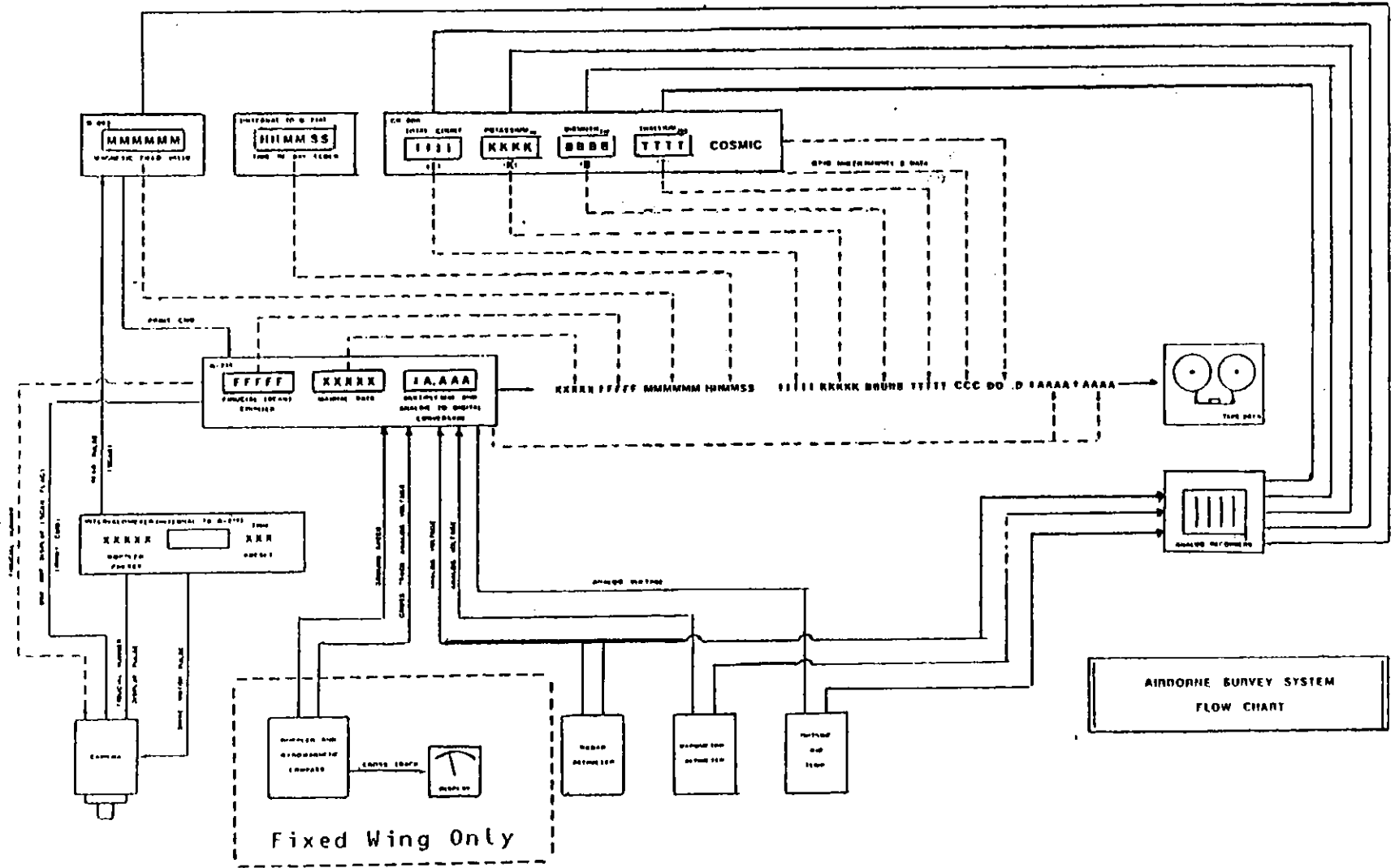




Figure 9

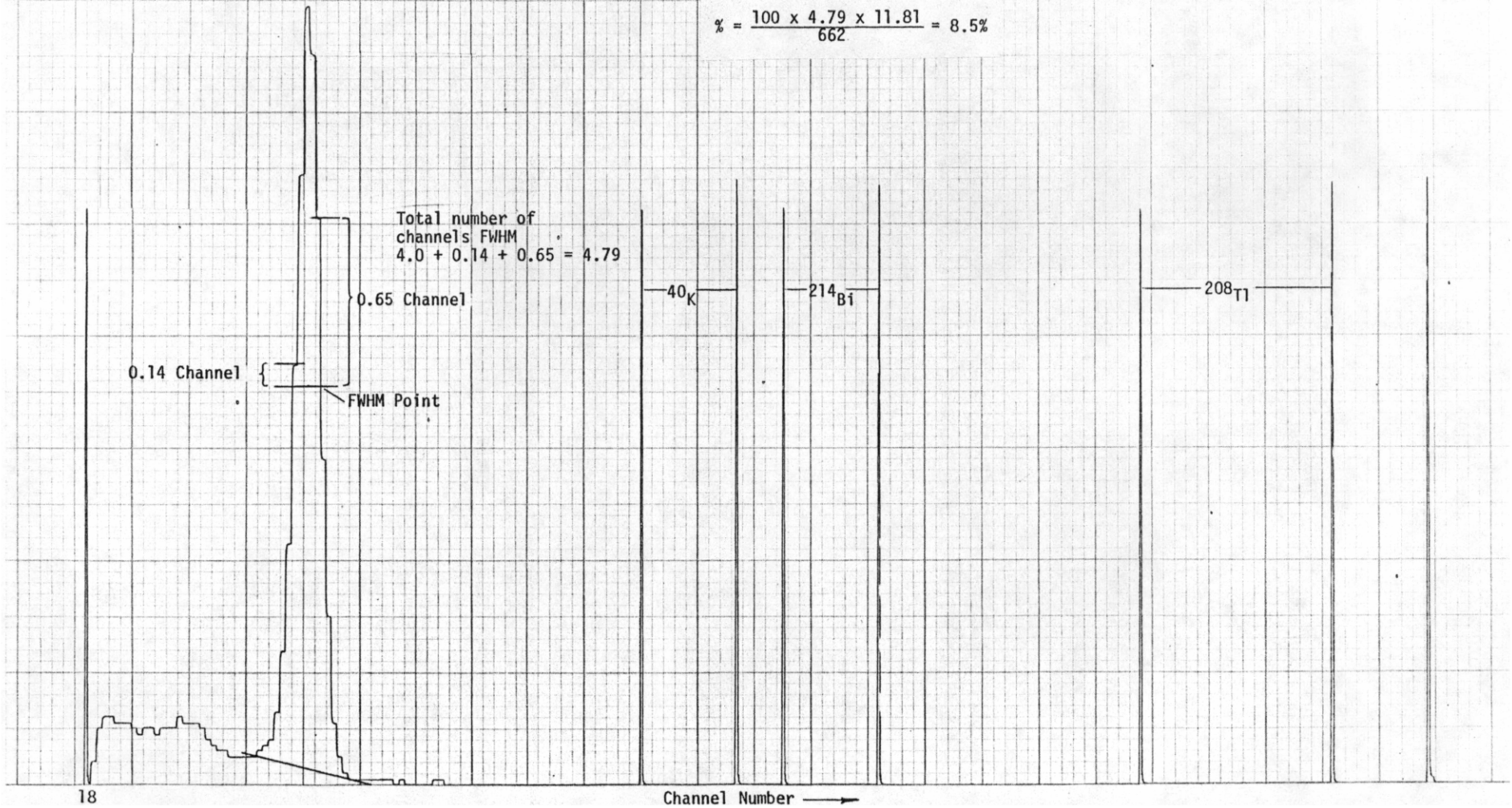
GR-800D ANALOG SPECTRUM PLOT  
DET-1024 Crystal Detector (1,024 in<sup>3</sup>)  
<sup>137</sup>Cs Source 11.81 Kev/Ch

20K c.p.s. Full Scale

Resolution Calculation

$$\% = \frac{100 \times \text{FWHM} \times \text{Kev/Ch}}{662 \text{ Kev}}$$

$$\% = \frac{100 \times 4.79 \times 11.81}{662} = 8.5\%$$





## SYSTEM CALIBRATION

### AIRCRAFT AND COSMIC BACKGROUND

Full spectral data are collected at five (5) altitudes over water (Lama: 14,000 feet, Tracker 15,000 feet; 12,000 feet; 10,000 feet; 8,000 feet and 6,000 feet) in an area where the existence of no airborne Bi214 can be assured (off shore over the Pacific Ocean). This results in separate spectra as shown schematically in Figure 10. We define  $S(12,000)$  to be the spectra at 12,000 feet from 0.4 MeV to 3.0 MeV with  $S(8,000)$  the same spectra at a lower altitude (8,000) and  $C(h)$  the total count between 3.0 and 6.0 MeV at respective altitudes. Since the aircraft background is constant, the difference between any two altitudes separated sufficiently - typically, 2,000 feet-yields the cosmic spectral curve shape as shown schematically in Figure 10. Thus

$$\begin{aligned} S(12,000) - S(8,000) &= \Delta S \\ \text{and} \\ \Sigma C_{12}(h_i) - \Sigma C_8(h_i) &= \Delta C \end{aligned}$$

This cosmic spectral curve is scaled back to 12,000 feet as follows:

$$\frac{C_{12}(h_i) \times \Delta S}{\Delta C} = C(12,000) \text{ the Cosmic Spectrum (Shape and magnitude at 12,000 feet)}$$

The aircraft background is derived as follows:

$$S(12,000) - C(12,000) = \text{A/C Background}$$

Since data were collected at five altitudes, this procedure was repeated for each combination of altitudes and results averaged. Typical aircraft and cosmic spectra are shown in Figure 11 and 12 respectively.

### SYSTEM CONSTANTS

System constants were determined by occupation of the DoE Walker Field Test Pads. (See Ward, 1978, and Stromswold, 1978, for complete descriptions of the building and monitoring of the pads). The five test pads contained varying concentrations of K, U, and T as presented by BFEC:

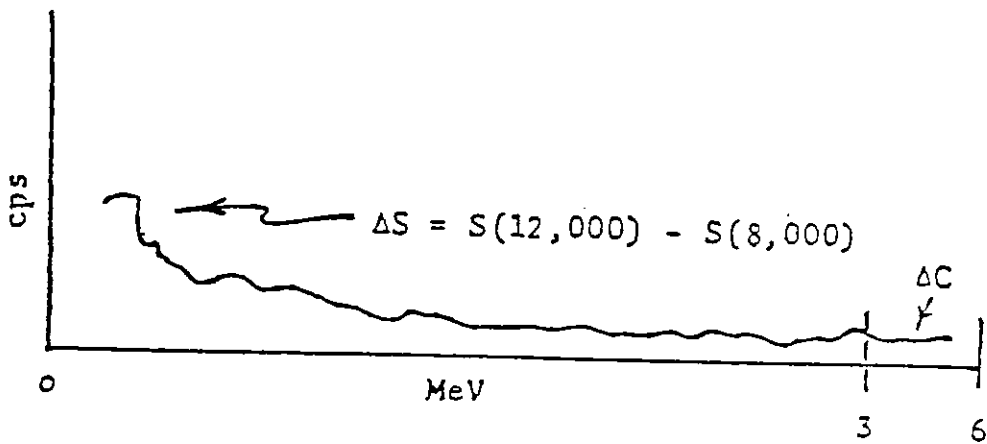
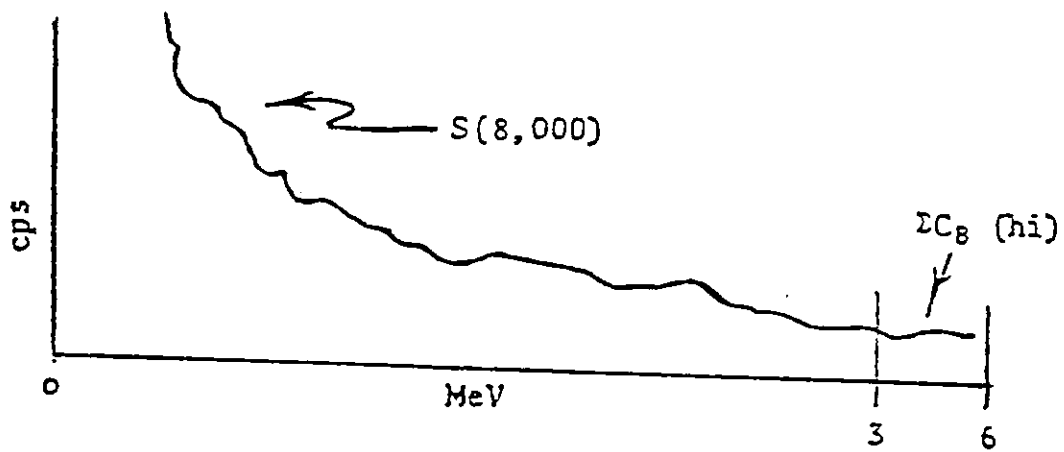
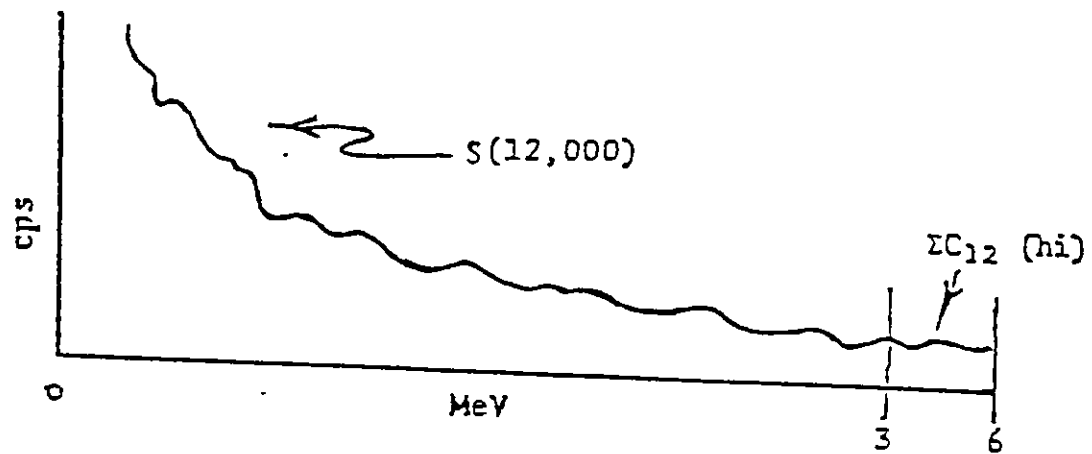


Figure 10 - Multiple altitude spectra schematic







DEIVED AIRCR-LOL BACKGROUND SPECTRUM PACIFIC OCEAN DATA

DERIVED AIRCRAFT BACKGROUND SPECTRUM FROM PACIFIC OCEAN DATA

DOWNWARD-LOOKING CRYSTAL SPECTRUM FOR LINE AC BGD, DATED 072577

TC (0-6 MEV) 184.07 TC (0.4-3.0 MEV) 141.17 COSMIC (3-6 MEV) 0.00  
 U (1.12 MEV) 9.91 K (1.46 MEV) 14.54 U (1.76 MEV) 4.36 T (2.62 MEV) 4.29

AIRCRAFT BACKGROUND  
 ROTARY WING AIRCRAFT  
 DOWNWARD LOOKING CRYSTAL  
 2048 CUBIC INCHES  
 DATE: 25 JULY 1977

CH 0 (0.000 MEV)	0.000 CPS X
CH 1 (0.012 MEV)	0.000 CPS X
CH 2 (0.024 MEV)	0.000 CPS X
CH 3 (0.035 MEV)	0.000 CPS X
CH 4 (0.047 MEV)	0.000 CPS X
CH 5 (0.059 MEV)	0.000 CPS X
CH 6 (0.071 MEV)	0.000 CPS X
CH 7 (0.083 MEV)	0.000 CPS X
CH 8 (0.095 MEV)	0.000 CPS X
CH 9 (0.106 MEV)	0.000 CPS X
CH 10 (0.118 MEV)	0.000 CPS X
CH 11 (0.130 MEV)	0.000 CPS X
CH 12 (0.142 MEV)	0.000 CPS X
CH 13 (0.154 MEV)	0.000 CPS X
CH 14 (0.165 MEV)	0.000 CPS X
CH 15 (0.177 MEV)	0.000 CPS X
CH 16 (0.189 MEV)	0.000 CPS X
CH 17 (0.201 MEV)	0.000 CPS X
CH 18 (0.213 MEV)	-0.025 CPS X
CH 19 (0.225 MEV)	-0.020 CPS X
CH 20 (0.236 MEV)	0.000 CPS X
CH 21 (0.248 MEV)	1.401 CPS XXXXX
CH 22 (0.260 MEV)	3.792 CPS XXXXXXXXXXXX
CH 23 (0.272 MEV)	4.250 CPS XXXXXXXXXXXX
CH 24 (0.284 MEV)	4.334 CPS XXXXXXXXXXXX
CH 25 (0.295 MEV)	3.748 CPS XXXXXXXXXXXX
CH 26 (0.307 MEV)	3.897 CPS XXXXXXXXXXXX
CH 27 (0.319 MEV)	3.818 CPS XXXXXXXXXXXX
CH 28 (0.331 MEV)	4.236 CPS XXXXXXXXXXXX
CH 29 (0.343 MEV)	3.433 CPS XXXXXXXXXXXX
CH 30 (0.355 MEV)	2.996 CPS XXXXXXXXXXXX
CH 31 (0.366 MEV)	2.559 CPS XXXXXXXXXXXX
CH 32 (0.378 MEV)	2.269 CPS XXXXXXXXXXXX
CH 33 (0.390 MEV)	2.102 CPS XXXXXXXX
CH 34 (0.402 MEV)	2.051 CPS XXXXXXXX TOTAL COUNT
CH 35 (0.414 MEV)	2.151 CPS XXXXXXXX
CH 36 (0.426 MEV)	2.114 CPS XXXXXXXX
CH 37 (0.437 MEV)	1.976 CPS XXXXXXXX
CH 38 (0.449 MEV)	2.290 CPS XXXXXXXX
CH 39 (0.461 MEV)	2.188 CPS XXXXXXXX
CH 40 (0.473 MEV)	2.226 CPS XXXXXXXX
CH 41 (0.485 MEV)	1.983 CPS XXXXXXXX
CH 42 (0.496 MEV)	2.155 CPS XXXXXXXX
CH 43 (0.508 MEV)	2.158 CPS XXXXXXXX
CH 44 (0.520 MEV)	2.267 CPS XXXXXXXX
CH 45 (0.532 MEV)	2.217 CPS XXXXXXXX
CH 46 (0.544 MEV)	1.997 CPS XXXXXXXX
CH 47 (0.556 MEV)	2.447 CPS XXXXXXXX
CH 48 (0.567 MEV)	2.540 CPS XXXXXXXX
CH 49 (0.579 MEV)	2.586 CPS XXXXXXXX
CH 50 (0.591 MEV)	2.708 CPS XXXXXXXX
CH 51 (0.603 MEV)	2.481 CPS XXXXXXXX
CH 52 (0.615 MEV)	2.372 CPS XXXXXXXX
CH 53 (0.626 MEV)	2.866 CPS XXXXXXXX
CH 54 (0.638 MEV)	1.682 CPS XXXXXXXX
CH 55 (0.650 MEV)	1.681 CPS XXXXXXXX
CH 56 (0.662 MEV)	1.490 CPS XXXXXXXX
CH 57 (0.674 MEV)	1.474 CPS XXXXXXXX
CH 58 (0.686 MEV)	1.447 CPS XXXXXXXX
CH 59 (0.697 MEV)	1.431 CPS XXXXXXXX
CH 60 (0.709 MEV)	1.476 CPS XXXXXXXX
CH 61 (0.721 MEV)	1.453 CPS XXXXXXXX
CH 62 (0.733 MEV)	1.467 CPS XXXXXXXX
CH 63 (0.745 MEV)	1.579 CPS XXXXXXXX
CH 64 (0.756 MEV)	1.497 CPS XXXXXXXX
CH 65 (0.768 MEV)	1.548 CPS XXXXXXXX
CH 66 (0.780 MEV)	1.421 CPS XXXXXXXX
CH 67 (0.792 MEV)	1.282 CPS XXXXXXXX
CH 68 (0.804 MEV)	1.155 CPS XXXXXXXX
CH 69 (0.816 MEV)	1.246 CPS XXXXXXXX
CH 70 (0.827 MEV)	1.245 CPS XXXXXXXX
CH 71 (0.839 MEV)	1.161 CPS XXXXXXXX
CH 72 (0.851 MEV)	1.253 CPS XXXXXXXX
CH 73 (0.863 MEV)	1.231 CPS XXXXXXXX
CH 74 (0.875 MEV)	1.452 CPS XXXXXXXX
CH 75 (0.887 MEV)	1.452 CPS XXXXXXXX
CH 76 (0.898 MEV)	1.543 CPS XXXXXXXX
CH 77 (0.910 MEV)	1.444 CPS XXXXXXXX
CH 78 (0.922 MEV)	1.284 CPS XXXXXXXX
CH 79 (0.934 MEV)	1.289 CPS XXXXXXXX
CH 80 (0.946 MEV)	1.150 CPS XXXXXXXX
CH 81 (0.957 MEV)	1.144 CPS XXXXXXXX
CH 82 (0.969 MEV)	1.085 CPS XXXXXXXX
CH 83 (0.981 MEV)	1.061 CPS XXXXXXXX
CH 84 (0.993 MEV)	0.941 CPS XXXXXXXX
CH 85 (1.005 MEV)	0.919 CPS XXXXXXXX
CH 86 (1.017 MEV)	0.822 CPS XXXXXXXX
CH 87 (1.028 MEV)	0.816 CPS XXXXXXXX
CH 88 (1.040 MEV)	0.853 CPS XXXXXXXX
CH 89 (1.052 MEV)	0.901 CPS XXX BISMUTH 214
CH 90 (1.064 MEV)	0.822 CPS XXX
CH 91 (1.076 MEV)	0.867 CPS XXX
CH 92 (1.087 MEV)	0.968 CPS XXX
CH 93 (1.099 MEV)	0.851 CPS XXX
CH 94 (1.111 MEV)	0.905 CPS XXX
CH 95 (1.123 MEV)	0.847 CPS XXX
CH 96 (1.135 MEV)	0.861 CPS XXX
CH 97 (1.147 MEV)	0.800 CPS XXX
CH 98 (1.158 MEV)	0.727 CPS XXX
CH 99 (1.170 MEV)	0.751 CPS XXX BISMUTH 214
CH 100 (1.182 MEV)	0.697 CPS XXX
CH 101 (1.194 MEV)	0.663 CPS XXX
CH 102 (1.206 MEV)	0.657 CPS XXX
CH 103 (1.217 MEV)	0.633 CPS XXX
CH 104 (1.229 MEV)	0.719 CPS XXX
CH 105 (1.241 MEV)	0.671 CPS XXX
CH 106 (1.253 MEV)	0.475 CPS XXX
CH 107 (1.265 MEV)	0.601 CPS XXX
CH 108 (1.277 MEV)	0.661 CPS XXX
CH 109 (1.288 MEV)	0.669 CPS XXX
CH 110 (1.300 MEV)	0.606 CPS XXX
CH 111 (1.312 MEV)	0.630 CPS XXX
CH 112 (1.324 MEV)	0.652 CPS XXX
CH 113 (1.336 MEV)	0.644 CPS XXX
CH 114 (1.347 MEV)	0.652 CPS XXX
CH 115 (1.359 MEV)	0.791 CPS XXX POTASSIUM 40
CH 116 (1.371 MEV)	0.787 CPS XXX
CH 117 (1.383 MEV)	0.834 CPS XXX
CH 118 (1.395 MEV)	0.984 CPS XXX
CH 119 (1.407 MEV)	1.072 CPS XXX
CH 120 (1.418 MEV)	1.124 CPS XXX
CH 121 (1.430 MEV)	1.088 CPS XXX
CH 122 (1.442 MEV)	1.210 CPS XXX
CH 123 (1.454 MEV)	1.231 CPS XXX
CH 124 (1.466 MEV)	1.207 CPS XXX
CH 125 (1.477 MEV)	0.995 CPS XXX
CH 126 (1.489 MEV)	0.967 CPS XXX
CH 127 (1.501 MEV)	0.624 CPS XXX
CH 128 (1.513 MEV)	0.635 CPS XXX
CH 129 (1.525 MEV)	0.512 CPS XXX
CH 130 (1.537 MEV)	0.488 CPS XXX
CH 131 (1.548 MEV)	0.409 CPS XXX
CH 132 (1.560 MEV)	0.360 CPS XXX POTASSIUM 40
CH 133 (1.572 MEV)	0.339 CPS XXX
CH 134 (1.584 MEV)	0.438 CPS XXX
CH 135 (1.596 MEV)	0.310 CPS XXX
CH 136 (1.608 MEV)	0.259 CPS XXX
CH 137 (1.619 MEV)	0.250 CPS XXX
CH 138 (1.631 MEV)	0.353 CPS XXX
CH 139 (1.643 MEV)	0.353 CPS XXX
CH 140 (1.655 MEV)	0.332 CPS XXX
CH 141 (1.667 MEV)	0.326 CPS XXX BISMUTH 214
CH 142 (1.678 MEV)	0.267 CPS XXX
CH 143 (1.690 MEV)	0.275 CPS XXX
CH 144 (1.702 MEV)	0.245 CPS XXX
CH 145 (1.714 MEV)	0.347 CPS XXX
CH 146 (1.726 MEV)	0.352 CPS XXX
CH 147 (1.738 MEV)	0.293 CPS XXX
CH 148 (1.749 MEV)	0.359 CPS XXX
CH 149 (1.761 MEV)	0.279 CPS XXX
CH 150 (1.773 MEV)	0.334 CPS XXX
CH 151 (1.785 MEV)	0.245 CPS XXX
CH 152 (1.797 MEV)	0.255 CPS XXX
CH 153 (1.808 MEV)	0.174 CPS XXX
CH 154 (1.820 MEV)	0.228 CPS XXX
CH 155 (1.832 MEV)	0.188 CPS XXX
CH 156 (1.844 MEV)	0.115 CPS XXX
CH 157 (1.856 MEV)	0.085 CPS XXX BISMUTH 214
CH 158 (1.868 MEV)	0.147 CPS XXX
CH 159 (1.879 MEV)	0.147 CPS XXX
CH 160 (1.891 MEV)	0.139 CPS XXX
CH 161 (1.903 MEV)	0.109 CPS XXX
CH 162 (1.915 MEV)	0.091 CPS XXX
CH 163 (1.927 MEV)	0.151 CPS XXX
CH 164 (1.938 MEV)	0.088 CPS XXX
CH 165 (1.950 MEV)	0.136 CPS XXX
CH 166 (1.962 MEV)	0.157 CPS XXX
CH 167 (1.974 MEV)	0.119 CPS XXX
CH 168 (1.986 MEV)	0.109 CPS XXX
CH 169 (1.998 MEV)	0.113 CPS XXX
CH 170 (2.009 MEV)	0.106 CPS XXX
CH 171 (2.021 MEV)	0.147 CPS XXX
CH 172 (2.033 MEV)	0.137 CPS XXX
CH 173 (2.045 MEV)	0.171 CPS XXX
CH 174 (2.057 MEV)	0.154 CPS XXX
CH 175 (2.068 MEV)	0.188 CPS XXX
CH 176 (2.080 MEV)	0.162 CPS XXX
CH 177 (2.092 MEV)	0.104 CPS XXX
CH 178 (2.104 MEV)	0.138 CPS XXX
CH 179 (2.116 MEV)	0.137 CPS XXX
CH 180 (2.128 MEV)	0.119 CPS XXX
CH 181 (2.139 MEV)	0.169 CPS XXX
CH 182 (2.151 MEV)	0.148 CPS XXX
CH 183 (2.163 MEV)	0.101 CPS XXX
CH 184 (2.175 MEV)	0.114 CPS XXX
CH 185 (2.187 MEV)	0.088 CPS XXX
CH 186 (2.199 MEV)	0.101 CPS XXX
CH 187 (2.210 MEV)	0.085 CPS XXX
CH 188 (2.222 MEV)	0.130 CPS XXX
CH 189 (2.234 MEV)	0.117 CPS XXX
CH 190 (2.246 MEV)	0.113 CPS XXX
CH 191 (2.258 MEV)	0.116 CPS XXX
CH 192 (2.269 MEV)	0.088 CPS XXX
CH 193 (2.281 MEV)	0.097 CPS XXX
CH 194 (2.293 MEV)	0.095 CPS XXX
CH 195 (2.305 MEV)	0.087 CPS XXX
CH 196 (2.317 MEV)	0.059 CPS XXX
CH 197 (2.329 MEV)	0.015 CPS XXX
CH 198 (2.340 MEV)	0.041 CPS XXX
CH 199 (2.352 MEV)	0.070 CPS XXX
CH 200 (2.364 MEV)	0.087 CPS XXX
CH 201 (2.376 MEV)	0.085 CPS XXX
CH 202 (2.388 MEV)	0.084 CPS XXX
CH 203 (2.399 MEV)	0.064 CPS XXX
CH 204 (2.411 MEV)	0.123 CPS XXX THALLIUM 208
CH 205 (2.423 MEV)	0.076 CPS XXX
CH 206 (2.435 MEV)	0.116 CPS XXX
CH 207 (2.447 MEV)	0.147 CPS XXX
CH 208 (2.459 MEV)	0.108 CPS XXX
CH 209 (2.470 MEV)	0.120 CPS XXX
CH 210 (2.482 MEV)	0.092 CPS XXX
CH 211 (2.494 MEV)	0.127 CPS XXX
CH 212 (2.506 MEV)	0.169 CPS XXX
CH 213 (2.518 MEV)	0.206 CPS XXX
CH 214 (2.529 MEV)	0.262 CPS XXX
CH 215 (2.541 MEV)	0.184 CPS XXX
CH 216 (2.553 MEV)	0.206 CPS XXX
CH 217 (2.565 MEV)	0.195 CPS XXX
CH 218 (2.577 MEV)	0.173 CPS XXX
CH 219 (2.589 MEV)	0.201 CPS XXX
CH 220 (2.600 MEV)	0.309 CPS XXX
CH 221 (2.612 MEV)	0.238 CPS XXX
CH 222 (2.624 MEV)	0.187 CPS XXX
CH 223 (2.636 MEV)	0.171 CPS XXX
CH 224 (2.648 MEV)	0.177 CPS XXX
CH 225 (2.660 MEV)	0.089 CPS XXX
CH 226 (2.671 MEV)	0.122 CPS XXX
CH 227 (2.683 MEV)	0.124 CPS XXX
CH 228 (2.695 MEV)	0.119 CPS XXX
CH 229 (2.707 MEV)	0.090 CPS XXX
CH 230 (2.719 MEV)	0.027 CPS XXX
CH 231 (2.730 MEV)	0.012 CPS XXX
CH 232 (2.742 MEV)	-0.026 CPS XXX
CH 233 (2.754 MEV)	-0.024 CPS XXX
CH 234 (2.766 MEV)	0.038 CPS XXX
CH 235 (2.778 MEV)	0.083 CPS XXX
CH 236 (2.790 MEV)	0.060 CPS XXX
CH 237 (2.801 MEV)	0.038 CPS XXX THALLIUM 208
CH 238 (2.813 MEV)	0.023 CPS XXX
CH 239 (2.825 MEV)	0.008 CPS XXX
CH 240 (2.837 MEV)	0.078 CPS XXX
CH 241 (2.849 MEV)	0.027 CPS XXX
CH 242 (2.860 MEV)	0.047 CPS XXX
CH 243 (2.872 MEV)	0.079 CPS XXX
CH 244 (2.884 MEV)	0.084 CPS XXX
CH 245 (2.896 MEV)	0.025 CPS XXX
CH 246 (2.908 MEV)	0.025 CPS XXX
CH 247 (2.920 MEV)	-0.015 CPS XXX
CH 248 (2.931 MEV)	0.037 CPS XXX
CH 249 (2.943 MEV)	-0.065 CPS XXX
CH 250 (2.955 MEV)	0.042 CPS XXX
CH 251 (2.967 MEV)	0.002 CPS XXX
CH 252 (2.979 MEV)	-0.018 CPS XXX
CH 253 (2.990 MEV)	0.031 CPS XXX
CH 254 (3.002 MEV)	-0.106 CPS XXX TOTAL COUNT
CH 255 (3.014 MEV)	0.000 CPS XXX



<u>PAD</u>	<u>K</u>	<u>U</u>	<u>T</u>
Matrix	1.45%	2.19 ppm	6.26 ppm
K	5.14%	5.09 ppm	8.48 ppm
U	2.03%	30.29 ppm	9.19 ppm
T	2.01%	5.14 ppm	45.33 ppm
Mixed	4.11%	20.39 ppm	17.52 ppm

Since the measurements were taken over a relatively short time period (a few hours), it was assumed that the matrix pad measurements contain not only the effects of the matrix pad itself, but also aircraft background (which is a constant), cosmic background (constant over the time period of interest), and all other local background (e.g. BiAir, etc.) effects. (The matrix pad is constructed with only the basic concrete mix without the additional elemental minerals). Thus, by subtracting the matrix pad count rates from the count rates in the four pads, we have eliminated aircraft and cosmic background and BiAir effects for the four pads. The pad concentrations are then modified in a similar fashion by the subtraction of the matrix pad concentrations. The differential concentrations in the pads are given in the table below.

<u>PAD</u>	<u>K</u>	<u>U</u>	<u>T</u>
K-Matrix	3.7%	2.9 ppm	2.2 ppm
U-Matrix	0.6%	28.5 ppm	2.9 ppm
T-Matrix	0.6%	3.0 ppm	39.0 ppm
Mixed-Matrix	2.7%	18.8 ppm	11.3 ppm

Considering the above, we can define a functional relationship between the differential concentrations and the residual count rates which will provide a method of determining the calibration constants for the spectrometer system. These calibration constants are the six (6) stripping coefficients which account for the interactions occurring between the elemental channels in the system (Compton scatter coefficients, etc.).

On the basis of an ideal situation, one would anticipate that some of these interactions should be negligible. This is not totally the case, since we are dealing with a system which has less than infinite resolving power (i.e. the energies are smeared to some extent).

Thus, energy peaks within a spectrum of a given element are Gaussian shaped rather than pure line spectra. Additionally, we are dealing with finite spectral windows, multiple peaked spectra, and pulse pileup; all tend to couple each window's response to the other.

Keeping in mind that we are dealing with the count rates corresponding to the concentrations presented in the last table, we define the following:

$KC_i$  = uncorrected system count rate for the K channel

$UC_i$  = uncorrected system count rate for the U channel

$TC_i$  = uncorrected system count rate for the T channel

$K_i$  = the percent differential concentration of potassium

$U_i$  = ppm differential concentration of uranium

$T_i$  = ppm differential concentration of thorium

where "i" refers to the ith pad.

We also define the following:

$\zeta_{kk}$  = sensitivity of  $KC_i$  to concentrations of  $K_i$

$\zeta_{ku}$  = sensitivity of  $KC_i$  to concentrations of  $U_i$

$\zeta_{kt}$  = sensitivity of  $KC_i$  to concentrations of  $T_i$

$\zeta_{uk}$  = sensitivity of  $UC_i$  to concentrations of  $K_i$

$\zeta_{uu}$  = sensitivity of  $UC_i$  to concentrations of  $U_i$

$\zeta_{ut}$  = sensitivity of  $UC_i$  to concentrations of  $T_i$

$\zeta_{tk}$  = sensitivity of  $TC_i$  to concentrations of  $K_i$

$\zeta_{tu}$  = sensitivity of  $TC_i$  to concentrations of  $U_i$

$\zeta_{tt}$  = sensitivity of  $TC_i$  to concentrations of  $T_i$

Using the above definitions, we now construct the functional relationship by means of the following nine (9) equations in sets of three (3) per pad.

$$\begin{array}{l}
\text{K pad} \quad KC_k = \zeta_{kk}K + \zeta_{ku}U + \zeta_{kt}T \\
\quad \quad UC_k = \zeta_{uk}K + \zeta_{uu}U + \zeta_{ut}T \\
\quad \quad TC_k = \zeta_{tk}K + \zeta_{tu}U + \zeta_{tt}T \\
\text{U pad} \quad KC_u = \zeta_{kk}K + \zeta_{ku}U + \zeta_{kt}T \\
\quad \quad UC_u = \zeta_{uk}K + \zeta_{uu}U + \zeta_{ut}T \\
\quad \quad TC_u = \zeta_{tk}K + \zeta_{tu}U + \zeta_{tt}T \\
\text{T pad} \quad KC_t = \zeta_{kk}K + \zeta_{ku}U + \zeta_{kt}T \\
\quad \quad UC_t = \zeta_{uk}K + \zeta_{uu}U + \zeta_{ut}T \\
\quad \quad TC_t = \zeta_{tk}K + \zeta_{tu}U + \zeta_{tt}T
\end{array}$$

Separating these equation into consistent groups, we get for the uncorrected count rates in the K channel

$$\begin{array}{l}
(\text{K pad}) \quad KC_k = \zeta_{kk}K_k + \zeta_{ku}U_k + \zeta_{kt}T_k \\
(\text{U pad}) \quad KC_u = \zeta_{kk}K_u + \zeta_{ku}U_u + \zeta_{kt}T_u \\
(\text{T pad}) \quad KC_t = \zeta_{kk}K_t + \zeta_{ku}U_t + \zeta_{kt}T_t
\end{array}$$

The equations can be expressed in matrix notation

$$\begin{bmatrix} KC_k \\ KC_u \\ KC_t \end{bmatrix} = \begin{bmatrix} K_k & U_k & T_k \\ K_u & U_u & T_u \\ K_t & U_t & T_t \end{bmatrix} \cdot \begin{bmatrix} \zeta_{kk} \\ \zeta_{ku} \\ \zeta_{kt} \end{bmatrix}$$

Where the k, u and t subscripts represent the K, U and T pads.

In a similar manner we can write two other matrix equations for  $UC_i$  and  $TC_i$  respectively.

$$\begin{bmatrix} UC_k \\ UC_u \\ UC_t \end{bmatrix} = \begin{bmatrix} K_k & U_k & T_k \\ K_u & U_u & T_u \\ K_t & U_t & T_t \end{bmatrix} \cdot \begin{bmatrix} \zeta_{uk} \\ \zeta_{uu} \\ \zeta_{ut} \end{bmatrix}$$

$$\begin{bmatrix} TC_k \\ TC_u \\ TC_t \end{bmatrix} = \begin{bmatrix} K_k & U_k & T_k \\ K_u & U_u & T_u \\ K_t & U_t & T_t \end{bmatrix} \cdot \begin{bmatrix} \zeta_{tk} \\ \zeta_{tu} \\ \zeta_{tt} \end{bmatrix}$$

Collecting the above, these equations can be expressed in matrix form as

$$\begin{bmatrix} KC_k & UC_k & TC_k \\ KC_u & UC_u & TC_u \\ KC_t & UC_t & TC_t \end{bmatrix} = \begin{bmatrix} K_t & U_k & T_k \\ K_u & U_u & T_u \\ K_t & U_t & T_t \end{bmatrix} \cdot \begin{bmatrix} \zeta_{kk} & \zeta_{uk} & \zeta_{tk} \\ \zeta_{ku} & \zeta_{uu} & \zeta_{tu} \\ \zeta_{kt} & \zeta_{ut} & \zeta_{tt} \end{bmatrix}$$

or

$$\bar{A} = \bar{B} \cdot \bar{\zeta}$$

where  $\bar{A}$  is the residual count rate matrix,  $\bar{B}$  is the matrix of the known differential concentrations and  $\bar{\zeta}$  the sensitivity matrix.

Rearranging the above equations we have

$$\bar{B} = \bar{A} \cdot \bar{\zeta}^{-1}$$

We now define

$$\bar{\zeta}^{-1} = \bar{\Delta}$$

Eliminating  $\bar{\zeta}$ , we get

$$\bar{B} = \bar{A} \cdot \bar{\Delta}$$

We can now solve for  $\bar{\Delta}$  by matrix inversion.

Therefore, the differential concentrations in the mixed pad can be derived from the k,u,t pads to check the computed  $\bar{\Delta}$ .

$$\begin{bmatrix} K_m \\ U_m \\ T_m \end{bmatrix} = \begin{bmatrix} \Delta_{kk} & \Delta_{ku} & \Delta_{kt} \\ \Delta_{uk} & \Delta_{uu} & \Delta_{ut} \\ \Delta_{tk} & \Delta_{tu} & \Delta_{tt} \end{bmatrix} \cdot \begin{bmatrix} KC_m \\ UC_m \\ TC_m \end{bmatrix}$$

where the subscript m refers to the mixed pad. Expanding this in algebraic form we obtain the following set of equations:



$$K_m = \Delta_{kk}(KC_m + \frac{\Delta_{ku}UC_m}{\Delta_{kk}} + \frac{\Delta_{kt}}{\Delta_{kk}} TC_m)$$

$$U_m = \Delta_{uu}(UC_m + \frac{\Delta_{ut}TC_m}{\Delta_{kk}} + \frac{\Delta_{uk}}{\Delta_{uu}} KC_m)$$

$$T_m = \Delta_{tt}(TC_m + \frac{\Delta_{tu}UC_m}{\Delta_{tt}} + \frac{\Delta_{tk}}{\Delta_{tt}} KC_m)$$

The terms in parentheses in the above 3 equations are the "corrected stripped count rates" for the system, and the stripping coefficients are as follows:

$$S_{ku} = \frac{\Delta_{ku}}{\Delta_{kk}} \quad (\text{effect of uranium on potassium})$$

$$S_{kt} = \frac{\Delta_{kt}}{\Delta_{kk}} \quad (\text{effect of thorium on potassium})$$

$$S_{ut} = \frac{\Delta_{ut}}{\Delta_{uu}} \quad (\text{effect of thorium on uranium})$$

$$S_{uk} = \frac{\Delta_{uk}}{\Delta_{uu}} \quad (\text{effect of potassium on uranium})$$

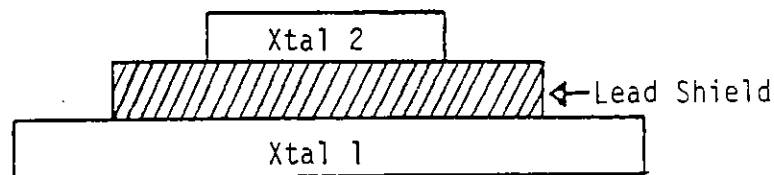
$$S_{tu} = \frac{\Delta_{tu}}{\Delta_{tt}} \quad (\text{effect of uranium on thorium})$$

$$S_{tk} = \frac{\Delta_{tk}}{\Delta_{tt}} \quad (\text{effect of potassium on thorium})$$

These stripping coefficients are defined in terms of  $S_{ij}$  in order to eliminate confusion with  $\alpha$ ,  $\beta$ , and  $\gamma$ , which are sometimes defined slightly differently.

#### ATMOSPHERIC RADON CORRECTION

Consider the crystal configuration shown below:



Let 1 and 2 designate the down and up crystal respectively. The down crystal sees radiation rates of  $I_1$  composed of the air signal  $I_a$  and the ground signal  $I_g$  plus aircraft and cosmic background.

$$\text{Therefore } I_1 = I_g + I_a + A_1 + C_1$$

Similarly, the up crystal sees the air signal and ground signal (both somewhat attenuated) plus an aircraft and cosmic background.

$$\text{Therefore } I_2 = \lambda I_g + m I_a + A_2 + C_2$$

Where  $m$  is the response to the air signal and  $\lambda$  is the % of the ground signal getting through to the up detector.

Using the test pad data, the factor  $\lambda$  can be determined. Consider the two previous equations. When we subtract the matrix pad data from the K, U, and T pad data, we have essentially set  $A_1$ ,  $A_2$ ,  $C_1$ , and  $C_2$  and  $I_a$  equal to zero.

$$\begin{aligned} \text{Therefore } I_1 &= I_g \\ I_2 &= \lambda I_g \\ \lambda &= \left( \frac{I_2}{I_1} \right) \end{aligned}$$

Instead of using the count rates we can use the resultant sensitivities  $1/\Delta_{uu}$  to determine  $\lambda$  for the elemental channel U.

$$\lambda_u = \frac{1/\Delta_{uu} \text{ (up)}}{1/\Delta_{uu} \text{ (down)}}$$

It should be noted that due to "shine around" (since the shielding is not an infinite plane, the upward looking crystal responds to the surrounding terrain) on the test pads, as altitude increases, should decrease, thus  $\lambda = f(h)$ .

Only the factor  $m$  remains to be determined. This unfortunately cannot be determined from test pad data. It can however be determined by flying over water (e.g. use of the Lake Mead over-water data).

Consider the equations for  $I_1$  and  $I_2$  again

$$I_1 = I_g + I_a + A_1 + C_1$$

$$I_2 = \lambda I_g + mI_a + A_2 + C_2$$

Over water  $I_g = 0$

We have  $A_1$ ,  $A_2$ ,  $C_1$ , and  $C_2$  defined.

Removing the aircraft and cosmic background from the over water data and we are left with

$$I_1 = I_a$$

$$I_2 = mI_a$$

Since  $m$  is the shielding factor response to the air signal, we should have an air signal to "shield". Thus  $m$  is best determined if there is radon present.

Both up and down counting rates are corrected for aircraft and cosmic background and so we can solve the following two equations for  $I_a$ .

$$I_1 = I_g + I_a$$

$$I_2 = \lambda I_g + mI_a$$

$$mI_a = I_2 - \lambda I_g$$

but  $I_g = I_1 - I_a$

then  $I_a (m - \lambda) = I_2 - \lambda I_1$

or 
$$I_a = \frac{I_2 - \lambda I_1}{m - \lambda} = \text{Bi Air}$$

and  $I_a$  is then the Bi Air contribution from the surrounding air. This is then subtracted from the down looking U count resulting in corrected data.

## FIXED WING/ROTARY WING DATA NORMALIZATION

As required in the Raton Basin Project, the rotary wing data were normalized to the fixed wing data to provide continuity within NTMS data sets. Normalization was accomplished by multiplying the rotary wing reduced averaged record K, U, T, and total count values by an appropriate constant derived from data obtained on the Walker Field Calibration Pads, Lake Mead Dynamic Test Range, and flight line overlaps/intersections in the adjoining project area

To obtain the normalization constant the following technique was implemented:

1. The fixed wing/rotary wing ratio of K, U, T, and total count cps for the Walker Field Calibration Pads were calculated and tabulated.
2. The fixed wing/rotary wing ratio of K, U, T, and total count cps for quasi-coincident fixed wing/rotary wing samples (spatially within  $50\pm$  feet) were calculated for all four flights at each of the eight altitudes flown over the land portion of the Lake Mead Dynamic Test Range. Tabulation of these results included the plotting of histograms, scatter plots and associated statistical parameters.
3. Flight line overlaps/intersections occurring within individual NTMS sheets were subjected to the same procedure as in 2 above.

From results of the above, the proper normalization constant was selected and input to the processing scheme. In the case of all three (3) portions of the Powder River R & D Project a multiplicative factor of 1.5 was applied to the rotary wing average record data (K, U, T, and total count) to normalize it to the fixed wing data.

## DATA PROCESSING

### DATA PREPARATION

The following sections summarize the techniques used for reduction and processing of the airborne data.

#### Field Tape Verification and Edit

The field data tapes containing the airborne data are read into the computer to verify the recording and data quality. Data recovery is essentially 100% from the field tapes. During this phase, statistics are generated summarizing the altitude (radar and barometric), ground speed and air temperature for each flight line. Simultaneously, the spectral peaks are evaluated for shifts using a centroid calculation and the particular window's peak channel. The data are also checked for correct scan lengths and proper justification of data fields within each scan and live time calculations are made. During this process, the desired window data fields are extracted from each spectrum and rewritten as a reformatted copy tape.

The reformatted tape data are then edited, checked and corrected. The data for each flight line are then read (with aborted or unnecessary flight line data edited out) and each data variable is checked for consistency, data spikes, gradients, etc. Every correction suggested by the computer is evaluated by the data processing personnel prior to actual correction. Upon completion of the phase, the data on the output tape are "clean" and ready for subsequent correction of the radiometrics and tying of the magnetics.

#### Flight Line Location

A single frame 35 mm camera is used for obtaining position recovery information. The photo locations are spotted or transferred to a suitable base map and are digitized. The fiducial numbers of the spotted points along each line are entered during the digitizing process. A computer program is used to check the consistency of these data using calculated intersections from tie line to tie line and from traverse to traverse. This program allows easy detection of entry errors as well as potential flight path recovery errors.

A computer program then calculates the map location for each intersection and the beginning and end of each line based on the fiducial numbers and the control line/tie grid. A computer plot is made of these locations to check against the field plot and correct editing

information. These flight lines are then overlain on the geologic base map and each map unit is digitized such that each sample falls within a single unit. This resulting location information is then merged with the geophysical data using the fiducial numbers as common reference.

#### RADIOMETRIC DATA REDUCTION

Reduction of these data was carried out utilizing system calibration constants as derived from high altitude over water flights, Lake Mead Dynamic Test Range, and the Walker Field Test Pads. The data reduction sequence used may be summarized as follows: (see Figure 13 for Flow Diagram)

1. Spectrum stabilization
2. Dead time correction
3. Aircraft and Cosmic background correction
4. Compton stripping
5. Radon correction
6. Altitude correction
7. Data plots
8. Statistical analysis

Processing of the data was performed using the window energies given below:

Total count - 0.4 to 3.0 MeV

K - 1.37 to 1.57 MeV

U - 1.66 to 1.87 MeV (downward and upward looking system)

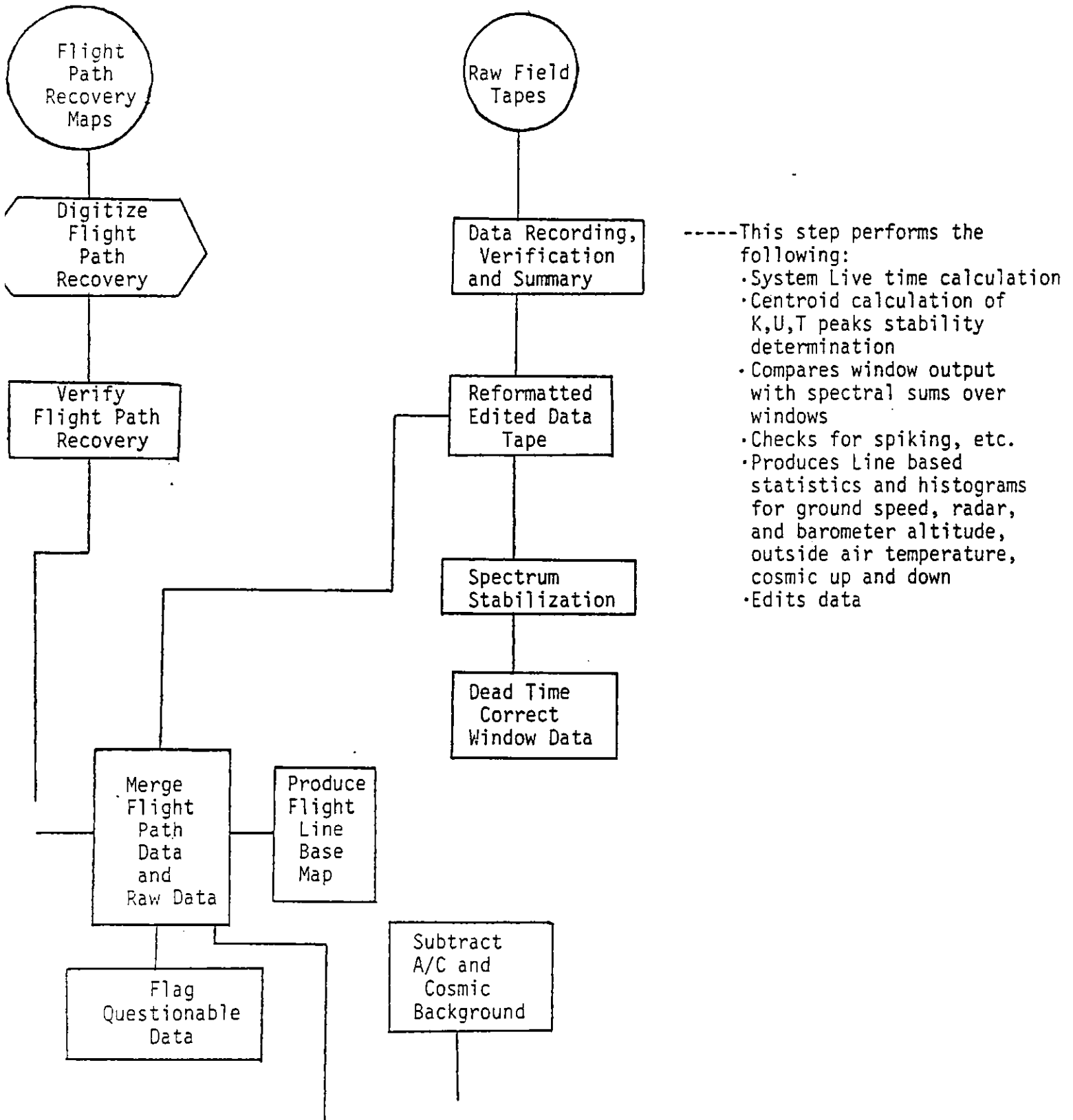
T - 2.41 to 2.81 MeV

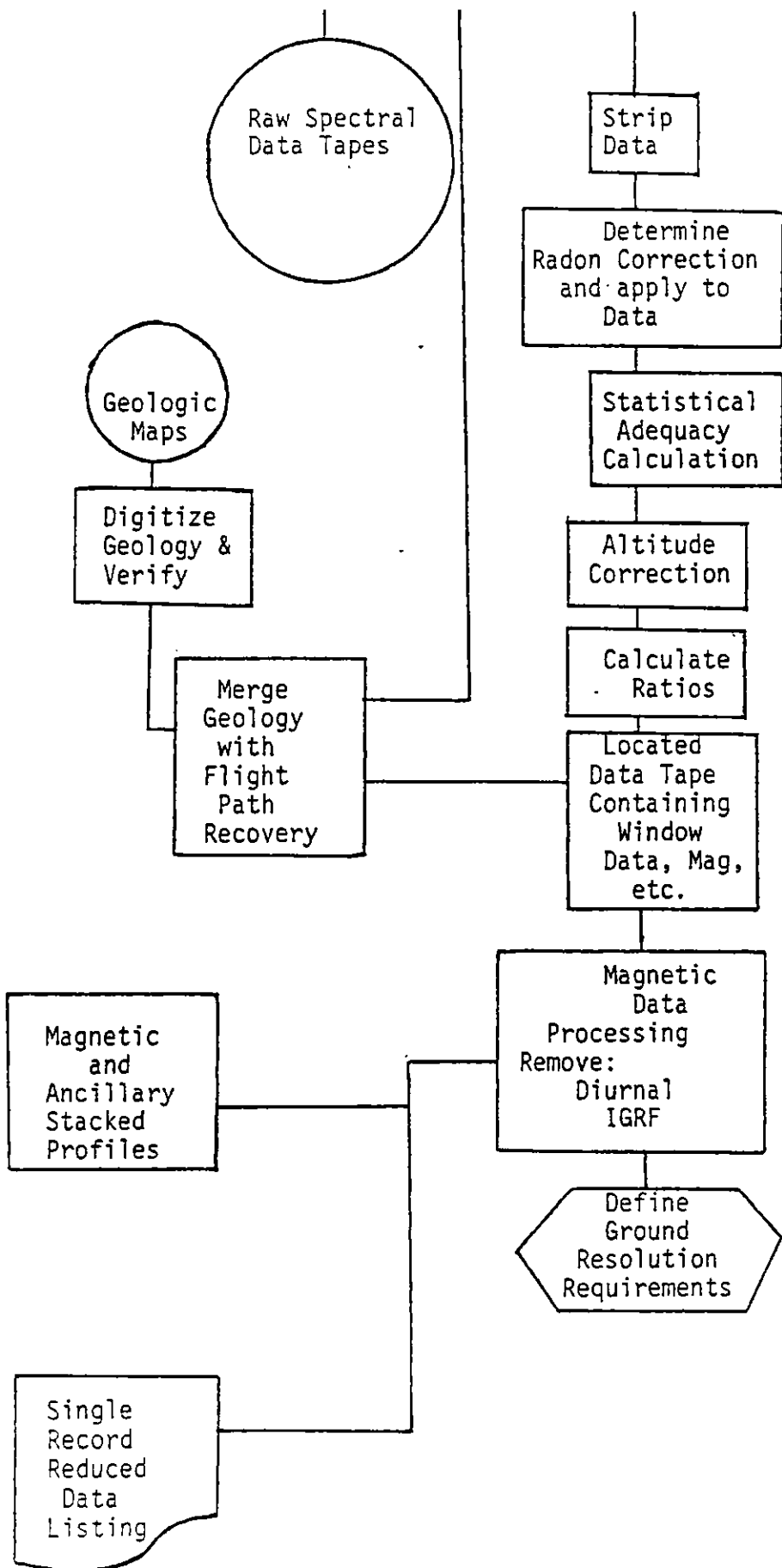
Cosmic - 3 to 6 MeV (downward and upward looking system)

Aircraft and Cosmic background for both Lamas and the Tracker over these windows described above are summarized below:

DATA PROCESSING FLOW DIAGRAM

FIGURE 13



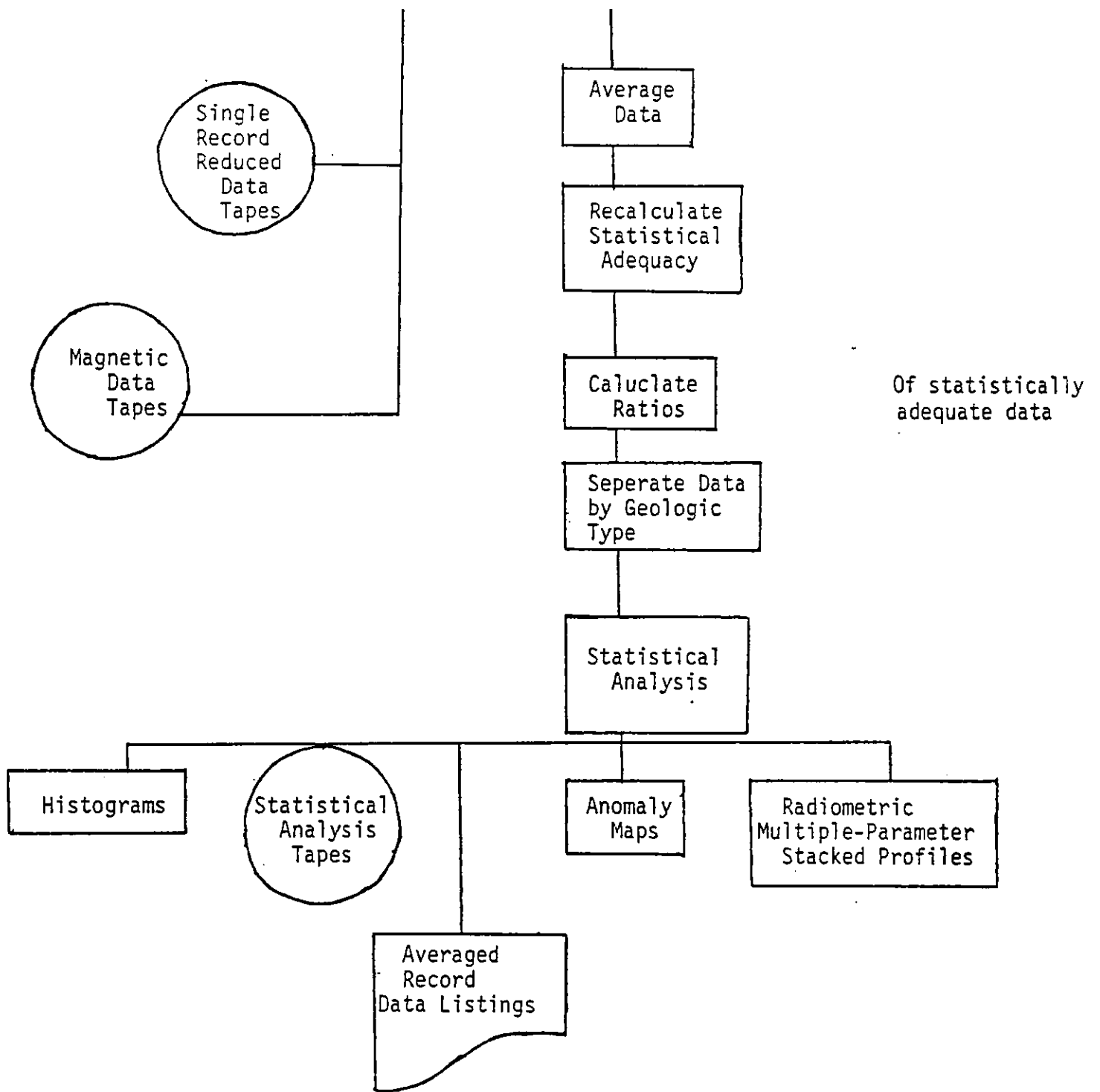


Down crystals stripped with  $S_{ut} = f(h)$  with all other factors constants. Up crystals stripped via down coupling factor.

Use crystal coupling equation to strip and determine radon to be subtracted from down looking crystal.

For statistically adequate data





		<u>S2F</u>		<u>LAMA I</u>		<u>LAMA II</u>	
		Aircraft	Cosmic*	Aircraft	Cosmic*	Aircraft	Cosmic*
TC	(cps)	212.04	3.115	150.83	3.023	102.30	3.316
K	(cps)	22.83	0.177	19.10	0.165	16.54	0.1772
U <sub>dn</sub>	(cps)	8.90	0.145	5.06	0.142	3.59	0.151
U <sub>up</sub>	(cps)	1.75	0.152	0.36	0.114	0.49	0.162
T	(cps)	7.76	0.189	4.25	0.191	2.56	0.2006

\*Cosmic background values are in cps per 1.0 cps in the 3-6 MeV window.

Compton corrections to the down data were made using the following constants:

<u>S<sub>ij</sub></u>	<u>S2F</u>	<u>LAMA I</u>	<u>LAMA II</u>
S <sub>ku</sub>	0.8613	0.8195	0.8258
S <sub>kt</sub>	0.1588	0.1815	0.1686
S <sub>ut</sub>	0.2986	0.2838	0.2621
S <sub>uk</sub>	0.0	0.0	0.0
S <sub>tu</sub>	0.05312	0.06926	0.07267
S <sub>tk</sub>	0.0	0.0	0.0

The ij subscripts represent the influence of the j<sup>th</sup> window on the i<sup>th</sup> window.

All parameters except for S<sub>ut</sub> are considered constants. S<sub>ut</sub> was considered an altitude dependent parameter utilizing the following expression (after Grasty, 1975).

$$S_{ut} = S_{ut_0} + 0.0076h, \text{ where } h \text{ is the altitude in hundreds of feet.}$$

Altitude attenuation coefficients used are defined as follows:

ALTITUDE ATTENUATION COEFFICIENTS			
	S2F	LAMA I	LAMA II
TC (per foot)	.0022065	.002129	.001663
K (per foot)	.003001	.002897	.002479
U (per foot)	.002710	.002640	.002260
T (per foot)	.002274	.002198	.001876

All radiometric data presented in the strip charts have been normalized to 400 feet mean terrain clearance at STP using the expression

$$\exp \mu_j \left[ \frac{273.15}{760} \times \frac{P}{T} \right] \left[ h - 400 \right]$$

where h is the height in feet,  $\mu_j$  is the appropriate altitude attenuation coefficient, P is in mm of Hg, and T is in degrees Kelvin. In cases where the altitude exceeds 1,000 feet, the correction coefficients were limited to the 1,000 foot value.

Bi Air calculations are made using the following expressions:

$$Bi_{Air} = \frac{U_{up} - (R_{us} + \frac{C'_{uk}}{C'_{uu}} R_{ks} + \frac{C'_{ut}}{C'_{uu}} R_{ts})}{m - \lambda}$$

Where  $U_{up}$  = count rate from upward detectors

$\lambda$  = crystal coupling constant

m = crystal geometric factor

$C'_{uk}$ ,  $C'_{ut}$ ,  $C'_{uu}$ , = stripping coefficients relating down data to up data

$R_{us}$  = stripped uranium count rate - down system

$R_{ks}$  = stripped potassium count rate - down system

$R_{ts}$  = stripped thorium count rate - down system

The numerical values for the constant  $\lambda$ ,  $m$ ,  $C_{uk}$ , and  $C'_{uu}$  are given below:

	S2F	LAMA I	LAMA II
$\lambda$	0.0488	0.0686	0.0556
$m$	0.189	0.168	0.195
$C'_{uk}$	0.0	0.0	0.0
$C'_{uu}$	0.03586	0.06078	0.05782
$C'_{ut}$	0.01687	0.02081	0.01765
$\mu\lambda$	- .000233	- .000233	- .000018
$\mu m$	- .000034	- .000034	- .000011

$\mu\lambda$  &  $\mu m$  are attitude dependent as follows:

$$\lambda = \lambda - \mu\lambda \cdot x \cdot h, \text{ where } h \text{ is in feet}$$

$$m = m - \mu m \cdot x \cdot h, \text{ where } h \text{ is in feet}$$

These Bi Air data are filtered and the filtered results are then removed, on a point by point basis, from the corrected uranium window data.

## MAGNETIC DATA REDUCTION

The magnetic data reduction processes are: correction for diurnal variation, tying to a common magnetic datum, and subtraction of the regional magnetic field as defined by the International Geomagnetic Reference Field (IGRF). During data acquisition, the magnetic field is monitored by a ground-based diurnal magnetometer that samples every four seconds at a sensitivity of one-quarter gamma. These data are recorded on magnetic tape along with the time for synchronization with the airborne data.

The diurnal data are edited to keep only those readings taken during flight time and then remove spikes and man-made magnetic events. After editing, these data are displayed in profile form to ensure that all corrections necessary have been made. Next, the data are synchronized in time with the airborne data, interpolated, and subtracted from the airborne magnetic data.

The diurnally corrected magnetic data are then processed by a tying program that compares the magnetic differences at intersections of flight lines and tie lines. This program calculates individual magnetic field biases for each flight tie line based on tie line intersections. This allows miss-ties to be minimized throughout the survey. These biases usually represent, after diurnal correction, systematic magnetic changes caused by such things as heading error, changes in location of the ground-based magnetometer, or changes in the airborne equipment. The biases are manually evaluated and selectively applied.

## STATISTICAL ANALYSIS

The results of the radiometric data reduction phase are single record samples (1.0 second interval). These data are then evaluated for statistical adequacy prior to altitude correction to ensure they are significant within the context of the anticipated errors in count statistics. These data are then averaged and input to the statistical Hypothesis Testing procedures.

### Statistical Adequacy Test

The statistical adequacy test is made to determine whether the corrected data sample is sufficiently greater than the "noise" to represent the "signal" of interest.

We can define three separate criteria for detection thresholds (ref. Currie, Analytical Chemistry, Volume 40, No. 3, March 1968) of which only one is directly applicable to our case; this is the

"critical level". This is the level at which the decision is made that a signal is "detected". We thus define this critical level as that level at which the data are statistically adequate.

Setting the actual levels in counts per second, "a priori " for each elemental window is difficult at best since the full effect of all parameters affecting the counts is not known to a sufficient degree of certainty. If the corrections to the data are a significant portion of the count rate, most of the error (exclusive of systematic errors due to electronics, etc.) in the corrected data can be ascribed to random errors within the applied corrections. The corrections are basically the results of counting radioactive decay products (gamma rays) and are therefore assumed to follow the classical Poisson distribution. The following assumptions concerning these corrections are:

1. In the best case, the error in each correction is additive.
2. The sum of these corrections also follows a Poisson distribution.
3. The uncertainty in the correction itself, is equal to the square root of the correction applied.
4. This uncertainty is directly reflected in the corrected single record count rate.

With these assumptions in mind, the criterion for determining the statistical adequacy of a given data sample may be defined as follows:

"If a corrected single record data sample exceeds 1.5 times the square root of the summed correction applied to that data sample, then that data sample is statistically adequate."

Since any calculation using statistically inadequate data (such as ratios) is also inadequate, the adequacy of each element of the single sample record data is tested prior to the calculation. This is done during the course of the processing by retaining all corrections applied to each data sample and determining its adequacy as explained above.

Not only are the results of this statistical adequacy test used to insure that calculated ratios will be meaningful but they are also utilized to determine the optimum interval over which the data should be averaged (e.g. 5 seconds or 7 seconds, etc.) to improve the overall data statistical adequacy. In the case of this project,

the resulting averaging sample interval was 7 seconds. This resulted in 98% or better of the uranium data to be statistically adequate, exclusive of those data which were outside of altitude specifications (the overall altitude specification was maintained at the 98% level) and excluding the known water saturated map units and water bodies.

### Hypothesis Testing

For this processing it is assumed that correlations between radiometric variables and computer map units can be described by normal (Gaussian) and/or log-normal distributions. The averaged data are treated in a standardized manner is described below.

Each sample with its six variables (K, U, T, and three ratios) is grouped by its corresponding computer map unit. Statistically inadequate data and samples with out-of-specification radar altitudes are excluded from the testing. A modified Chi-Square testing scheme is utilized to evaluate the following two hypotheses:

1. The count rate distribution for a specified computer map unit can be best represented by a normal distribution.

or

2. The count rate distribution for a specified computer map unit can be best represented by a log-normal distribution.

No Chi-Square tests were performed on units having less than 20 statistically adequate samples. In addition to the Chi-Square Test, all units are plotted as histograms and compared with the results of the hypothesis testing to clarify any ambiguities. For some units the best estimate of central tendency for a particular variable was the unit median and not the arithmetic mean. The lower mode was used as the measure of central tendency for polymodal distributions.

Each radiometric parameter for a given computer map unit is then classified as either a normal or log-normal distribution. The measure of central tendency and dispersion for each of these distributions are then utilized as a basis for determining which data are anomalous within a given unit. A sample of such a histogram is presented in Figure 14.

MAP UNIT : QA

TOTAL NUMBER OF SAMPLES 2195

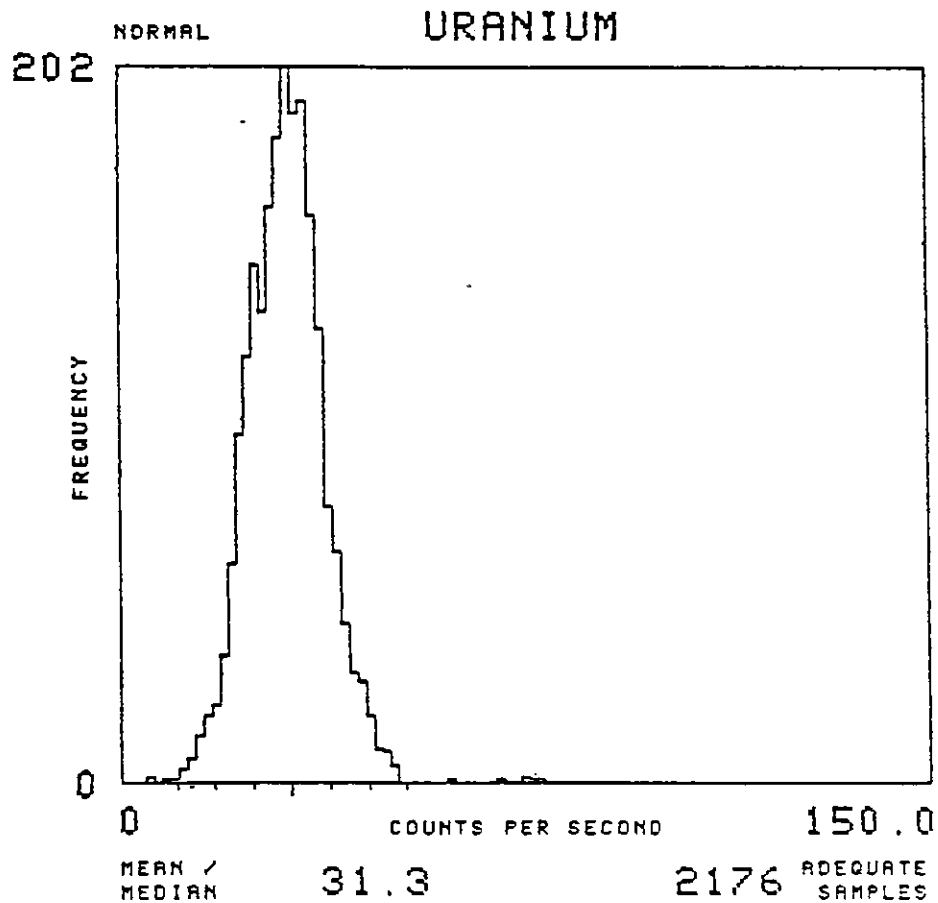


Figure 14 Sample Computer Map Unit Histogram



## DATA PRESENTATION

### GENERAL

The majority of the actual presented data are contained in Volume II. These include the uranium anomaly/interpretation maps and pseudo-contour maps of potassium, uranium, thorium, and magnetic data and are integrated as part of the text in the interpretation section. In addition to these data, Volume II contains data presented in the form of radiometric profiles, flight path recovery maps, standard deviation maps, and histograms. Microfiche data are contained in Appendix C of this volume. Data tapes are available separately.

### RADIOMETRIC PROFILES

Stacked profiles were prepared from the averaged data for each traverse and tie line. These stacked profiles, plotted at a linear scale of 1:250,000, contain the following parameters: corrected Total Count, corrected Potassium, corrected Uranium, corrected Thorium, U/TH, U/K, and TH/K ratios, Bi Air, radar altimeter, and magnetometer data. Each of the stacked profile sheets contains a plot of the flight path superimposed on a geologic strip map. Included along these profiles are the fiducial numbers which correspond to flight path position as displayed on the flight path recovery maps. Each of the stacked profiles represents the data contained on the specific flight line within the boundaries of the specified NTMS Quadrangle sheet.

Radiometric traces on the stacked profiles contain an indicator showing those data which are statistically inadequate. These statistically inadequate data are marked by a small vertical tick at the sample location. The altitude profile has been limited in display to 1,000 feet. A dashed line at the 700 foot level is presented to show those data which do not meet the altitude specifications. The vertical scale of each variable remains constant on all stacked profiles. When overranging occurs, the trace is stepped and the step labeled showing the actual value. A pictorial representation of such a stepping profile is shown in Figure 15. At the end of each stacked profile, a statistical summary of the minimum value, maximum value, mean, and standard deviation for that variable is presented.

Contained in Volume II of this report is an equivalent set of stacked profiles for each quadrangle, photographically reduced to an approximate scale of 1:500,000.

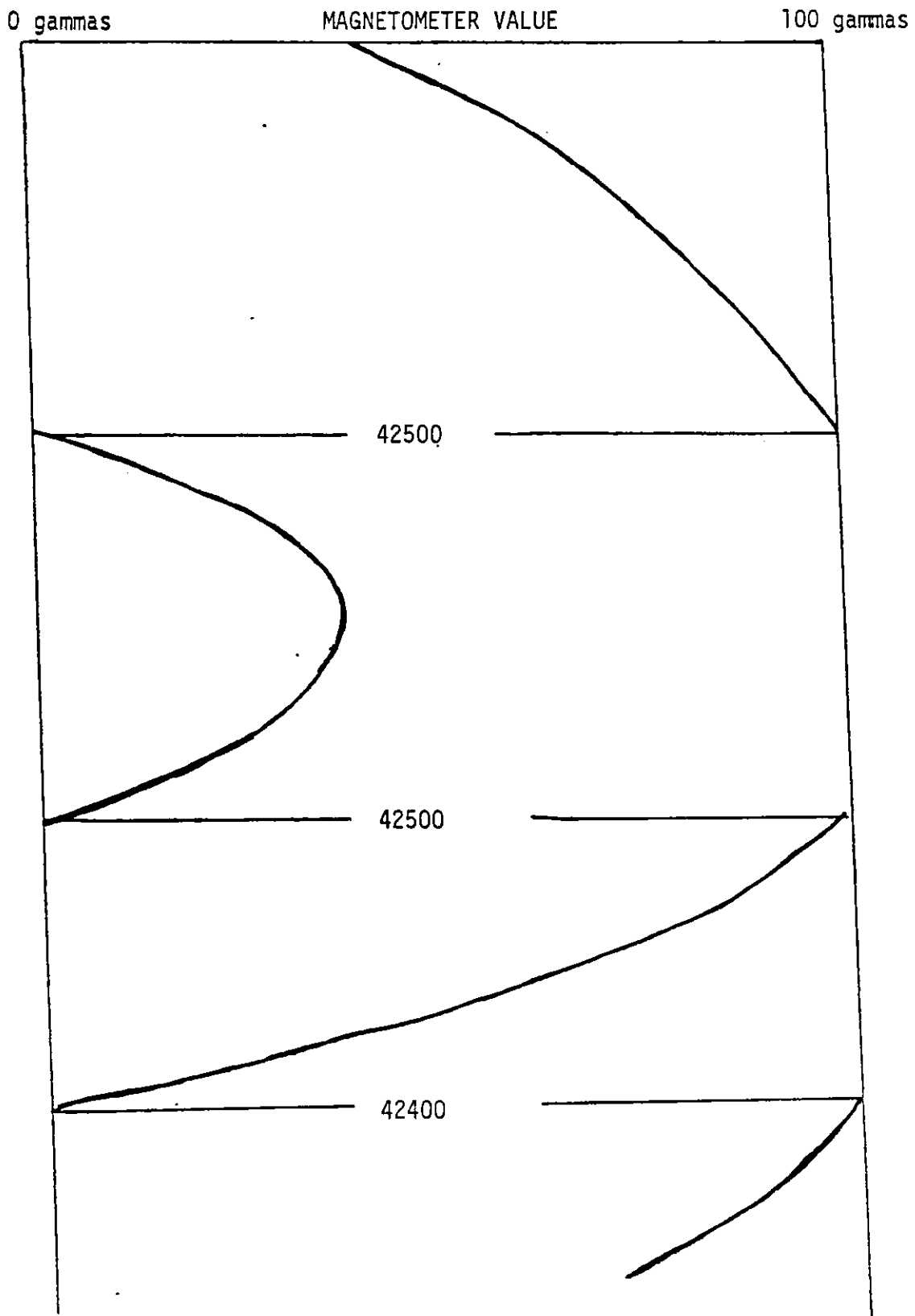


Figure 15 Plotter Step Value Labeling

### MAGNETIC PROFILES

A set of profiles containing the magnetic data (corrected, with IGRF removed), barometric altimeter data, radar altimeter data, diurnal monitor data, and temperature data are available at a linear scale of 1:250,000. Each of the stacked profiles contains a plot of the flight path superimposed on the geology over which the aircraft flew. Reduced scale, 1:500,000 copies of these are presented in Volume II of this report.

### FLIGHT PATH MAPS

For each of the NTMS quadrangle sheets covered by this survey, a flight path position map is available at a scale of 1:250,000. The actual flight path has been superimposed on the geologic quadrangle maps. Flight lines and tie lines are annotated along with fiducial numbers of located positions. Reduced scale, 1:500,000, copies of these can be found in Volume II of this report.

### STANDARD DEVIATION MAPS

Gamma ray standard deviation maps have been prepared for each NTMS quadrangle included in this survey. The six maps generated represent the following parameters: potassium, uranium, thorium, and U/TH, U/K and TH/K ratios. The data contained in each map represent only those data which are considered statistically adequate. This automatically excludes all data collected over water or data which falls outside of altitude specifications (i.e. altitude greater than 700 and less than 200 feet). The symbolism of each of the six maps is identical. The center of each circle represents the central averaged sample since the data had been averaged over a 7 second interval. The small boxes adjacent to each of the circles represents one standard deviation from the mean for that specific data sample. In order to determine whether the data shown are represented by positive or negative standard deviations, consider each map with north pointing away from the viewer. For east/west lines (traverse lines) positive standard deviations lie above or to the north of the traverse line with negative standard deviation below or to the south. On the north/south lines (tie lines) positive standard deviations are to the left of the viewer (west) with negative standard deviations to the right (east).

These maps were generated at a scale of 1:250,000 for each NTMS sheet. In addition, these anomaly maps are presented in Volume II of this report at a reduced scale of approximately 1:500,000.

## HISTOGRAMS

Computer generated histograms, showing the count rate distribution for each of the six gamma ray parameters measured and calculated as a function of computer map unit are presented in Volume II of this report. Information contained on these histograms includes the distribution, the standard deviation as calculated about the mean, and the total number of samples from which the distribution was derived.

## DATA LISTINGS

Single record reduced and averaged record (statistical analysis) data listings have been prepared on microfiche. The microfiche are contained in Volume I of this report as Appendix C. Each of the single record and averaged record data listings are presented for the data contained in a single quadrangle. The data contained in the single record data listings are summarized below:

1. Fiducial number
2. System/Quality (SAKUT) - The first digit identifies the system used to collect the sample. The remaining digits define the results of statistical adequacy testing for altitude, potassium, uranium, and thorium. A value of 0 indicated that the data are statistically adequate. A value of 1 indicates that the data are statistically inadequate. All data collected in excess of 700 feet and less than 200 feet are considered statistically inadequate.
3. Time - time presented in hours, minutes, and seconds
4. Altitude - altitude presented in feet above terrain
5. LAT/LONG - Latitude and Longitude presented in terms of decimal degrees
6. Magnetic field expressed in residual gammas
7. Geology - code representing geologic units
8. K, U, T - count rate of corrected K, U, T data
9. U/TH, U/K, TH/K - calculated ratios of the three parameters

10. Total count - corrected total count data (0.4 to 3.0 MeV)
11. COS - downward looking cosmic count rate in the 3-6 MeV channel
12. Uair - atmospheric Bi-214 count rate
13. Temperature - outside air temperature in degrees centigrade
14. Press - barometric pressure in inches of mercury

The averaged record (statistical analysis) data listings are summarized below:

1. Fiducial number
2. System/Quality (SAKUT) - The first digit identifies the system used to collect the sample. The remaining digits define the results of statistical adequacy testing for altitude, potassium, uranium, and thorium. A value of 0 indicates that the data are statistically adequate. A value of 1 indicates that the data are statistically inadequate. All data collected in excess of 700 feet and less than 200 feet are considered statistically inadequate.
3. LAT/LONG - Latitude and longitude presented in terms of decimal degrees
4. Magnetic field expressed in residual gammas
5. Geology - code representing geologic formations
6. K, U, T - count rate of corrected K, U, T data and the number of (+) standard deviations from the mean
7. U/TH, U/K, Th/K - calculated ratios of the three parameters, and the number of (+) standard deviations from the mean
8. Total count - corrected total count data (0.4 to 3.0 MeV)
9. COS - downward looking cosmic count rate in the 3-6 MeV channel
10. Uair - atmospheric Bi-214 count rate

## DATA TAPES

Data tape files have been generated for each of the 1:250,000 NTMS quadrangle sheets. The tapes are IBM compatible and recorded on 9 track EBCDIC at 800 bpi. Four separate sets of data tapes are presented: raw spectral data tapes, single record reduced data tapes, statistical analysis tapes, and magnetic data tapes. Detailed descriptions of the data tape formats are presented in Appendix A.

## DATA INTERPRETATION METHODS

### GENERAL

The stated objective of the NURE Program is the evaluation of the uranium potential of the United States. In support of this goal, high sensitivity airborne radiometric and magnetic surveys have been implemented to obtain reconnaissance information pertaining to regional distribution of uraniumiferous materials. Within this context, data interpretation has been oriented toward regional detection and description of anomalously high concentrations of uranium.

By far the most significant natural sources of gamma radiation in the geologic environment are the radioactive decay series of potassium 40 (K40), thorium 232 (Th232) and uranium 238 (U238) of which 0.7% is uranium 235. Potassium 40 is the largest contributor to natural radioactivity, accounting for nearly 98%, as it is the most abundant gamma ray emitter-.012% of all potassium in nature. (Refer to GSA Memoir 97 for abundances of uranium, thorium, and potassium.)

Potassium 40 is directly identified by the airborne spectrometer from a single clear peak at 1.46 mev (million electron volts) in its gamma ray spectrum. However, thorium 232 and uranium 238 do not have any clear, distinct peaks at sufficiently high energies to allow direct detection from airborne systems. Instead, daughter products which do have distinct peaks are measured as representing the abundance of the parent element. For thorium 232, the daughter nuclide thallium 208 (Tl208) has a distinct peak at 2.62 mev while uranium 238 has a daughter, bismuth 214 (Bi214) possessing a clear peak at 1.76 mev (see Figure 8 for a composite decay series spectrum). Consequently the fundamental assumption implicit to airborne uranium and thorium measurements is that the measured daughter products are in radioactive equilibrium - the number of atoms of disintegrating daughter nuclides are equal to the number being formed (see Adams and Gasparini, 1970).

An airborne gamma ray measurement is the sum of photons counted during a specified time interval from a multitude of gamma ray sources which include the three geologic emitters that are being sought plus other interfering sources. These others include, but are not limited to higher energy cosmic rays, aircraft and instruments, contributions from overlapping decay series and airborne radon 222. (See Burson, 1974 and McSharry, 1973 for a more complete discussion of airborne radiometric measurements, and Radiometric Data Reduction in this volume for a complete description of data correction procedures.)

When correlating ground data (geochemical, geological, etc.) with the corrected data derived from raw airborne measurements, the interpreter must remember what an individual airborne gamma ray sample physically measures. First, the terrestrial component of the gamma radiation measured by the airborne detector emanated primarily from the upper 18 inches of material on the earth's surface (Gregory and Horwood, 1963). The airborne measurement cannot "see" any deeper into the underlying rock material and is essentially a measurement of the soil's or exposed (weathered) rock's radioactivity. Secondly, since each airborne sample is an accumulation of gamma rays measured on a moving platform over a fixed period of time, the individual sample represents a large areal extent of surficial material. For this survey, with specifications of 400 feet mean terrain clearance and an average ground speed of 70 to 135 miles per hour, a one second sample corresponds to an oval approximately 700 to 1200 feet long by 600 feet wide (assuming an infinite, uniformly distributed source). Accordingly, averaged samples represent tremendous volumes of surficial materials.

#### METHODOLOGY

As described previously, the gamma ray data were located by computer map units, histograms were produced and statistical analyses performed. The basic unit for interpretation then is the averaged sample and its attendant deviations about a particular map unit's mean.

The uranium anomaly/interpretation map displays all central averaged samples that meet the following criteria:

1. The averaged uranium sample must be greater than or equal to 1 standard deviation above its map unit mean.
2. The sample must have a U/T ratio greater than or equal to 1 standard deviation above its unit mean.
3. Each U/T ratio defined in (2) must have a corresponding thorium value lying at least greater than minus one (-1) standard deviation below the mean. If the thorium sample is less than one standard deviation below the mean, the U/T ratio is considered questionable.

All the averaged samples displayed on the map are then examined for clusters, trends, and comparisons with all other available data.



Minimum requirements in the subsequent interpretation discussions of each quadrangle for anomalies listed in the uranium anomaly summary are defined as follows:

Two (2) consecutive averaged U samples lying two or more standard deviations above the mean or three (3) consecutive averaged U samples, two of which are one (1) or more standard deviations and the third of which is two (2) or more standard deviations above the mean.

Statistical anomalies which meet the above criteria can result from several factors or circumstances including: (1) true concentration of uraniumiferous minerals, (2) differential surface cover (soils and/or vegetation) within a lithologic unit, (3) local weather conditions such as rain and snow, (4) extreme facies variation within a mapped unit, and (5) differential weathering of rocks within mapped units. Obviously an averaged sample which lies on the boundary between two map units is not truly reflecting either one, but is rather an average of both. Thus, for two markedly different units, such a sample would be anomalous relative to one of the units and not be a true indication of radioactive differences within the unit.

The potassium, thorium, uranium, and residual magnetic data were plotted as separate pseudo-contour maps and overlain on the geologic base map and standard deviation maps. Regional trends of each variable and average counting rates could thus be easily and quickly determined and compared with the associated geological, magnetic, and statistical trends. Only the long wavelengths within each variable would show any line-to-line continuity on the pseudo-contour maps and thus, only regional trends will appear.

Each quadrangle's stacked profiles were also overlain on the corresponding geologic and standard deviation maps and anomaly map to further delineate trends and to allow a more detailed analysis of individual anomalies. Since the interpretation was concentrated on detection of anomalous uranium, subtle trends present in the potassium and thorium channels and ratios were only examined in a cursory manner. Even during such a brief examination of the profiles, it was evident that the spectrometer system was highly sensitive to changes in surface materials, even in areas of low counting rates such as glacial drift. Thus radiometrics have a real potential for performing general surficial mapping "geochemical analysis" on a geologic unit (or soils) basis in addition to merely radioactive mineral "anomaly hunting".

Mean values of percent potassium (%K), equivalent uranium (eU), and equivalent thorium (eT) incorporated into the text are based on the radiometric system's sensitivity as defined by calibrations on the

DoE's Lake Mead Dynamic Test Range. Normalized equivalent sensitivities at 400 feet altitude are:

<u>Radioelement</u>	<u>Equivalent Percent / Ppm</u>	<u>Counts/Second</u>
K	1% K	90.3
U	1 ppm eU	10.0
T	1 ppm eT	6.4

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## BIBLIOGRAPHY

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APPENDIX A  
TAPE FORMATS  
GALLUP AND SHIPROCK QUADRANGLES  
ARIZONA AND NEW MEXICO  
ALBUQUERQUE  
NEW MEXICO

THE STANDARD DEVIATION VALUES LISTED IN THE STATISTICAL TABLE ARE INCORRECT. TRUE STANDARD DEVIATION VALUES CAN BE SCALED OFF OF THE HISTOGRAMS. ANY QUESTIONS CONCERNING THIS PROBLEM MAY BE DIRECTED TO ANY STAFF GEOPHYSICIST AT EG&G GEOMETRICS, (408) 734-4616, EXT. 500



## Appendix A

### SINGLE RECORD REDUCED DATA TAPE

REFERENCE: PARAGRAPHS 4.7.2 AND 6.1.5, BFEC 1200-C

The SINGLE RECORD REDUCED DATA TAPE is unlabeled nine track, 800 BPI, NRZI. All data recorded as EBCDIC characters. Each tape contains but one file of header, data, and trailer records for no more than one NTMS quadrangle. The maximum record length is 5472 characters.

The tape is organized such that each flight line of data is preceded by a header record and followed by a trailer record. If a flight line is not complete on a given physical tape, no trailer record follows its last data record on the first tape, nor does a header record precede its first data record on the second tape.

#### Header Record

The header record is 144 characters long with six defined data fields. These fields are:

1. Type of tape. A 32-character field with the text "SINGLE RECORD REDUCED DATA" left justified.
2. Project identification. A 32-character field with, for example, the text "NTMS NL 16-1,2 RAWLINS" left justified. With the exception of special projects, such as the Lake Mead Dynamic Test Range, all project identification fields begin with "NTMS" followed by the sheet number. Additional information may be abbreviated.
3. Subcontractor name. A 10-character field with the text "GEOMETRICS."
4. System identification. A 6-character field with the aircraft registration number right justified.
5. Flight line number. A 6-character field with the flight line number right justified.
6. Date flown. A 6-character field with the date, expressed as YYJJJ, right justified. YY are the last two digits of the calendar year and JJJ is the Julian date. When reflights re-

quire the insertion of the data from multiple days' flying, the date used is that of the original flight.

The remaining 52 characters of the header record are blank filled. A length of 144 characters was chosen to allow for future expansion and because 144 is divisible by the number of characters per word of many popular computers.

### Data Record

Each data record may contain up to 38 data scans (logical records), with each scan 144 characters long. Therefore, the minimum physical length of a data record is 144 characters and the maximum physical length is 5472 characters.

The data scan has eighteen defined data fields.

1. Record identification number	F10.2	1- 10
2. Latitude in degrees	F10.4	11- 20
3. Longitude in degrees	F10.4	21- 30
4. Residual magnetic field in gammas	F15.2	31- 45
5. Terrain clearance in feet	F 5.0	46- 50
6. Surface geologic map unit	A 8	51- 58
7. System/Quality flag code (SAKUT)	A 6	59- 64
8. Cosmic count rate, in cps	F 8.1	65- 72
9. Atmospheric Bi-214 count rate, in cps	F 8.1	73- 80
10. Gross count rate (0.4-3.0 MeV), in cps	F 9.1	81- 89
11. Thorium (TL-208) count rate, in cps	F 9.1	90- 98
12. Uranium (Bi-214) count rate, in cps	F 9.1	99-107
13. Potassium (K-40) count rate, in cps	F 9.1	108-116
14. Uranium/Thorium count rate ratio	F 6.3	117-122
15. Uranium/Potassium count rate ratio	F 6.3	123-128
16. Thorium/Potassium count rate ratio	F 6.3	129-134
17. Outside air temperature, in degrees C	F 5.1	135-139
18. Barometric pressure, in inches of mercury	F 5.2	140-144

### Trailer Record

A trailer record follows the last data record for each flight line. This record is always 5472 characters long, all of which are the digit nine.

## STATISTICAL ANALYSIS TAPE

REFERENCE: PARAGRAPHS 4.7.3 AND 6.1.5, BFEC 1200-B

The STATISTICAL ANALYSIS TAPE is unlabeled nine track, 800 BPI, NRZI. All data recorded as EBCDIC characters. The maximum record length is 5472 characters. Each tape contains but one file of data for no more than one NTMS Quadrangle.

For each NTMS Quadrangle, the first record(s) on the tape contain summary information for all the geologic map units within the quadrangle. This summary information is followed by averaged record data for each survey flight line.

The tape is organized such that the summary geologic information and each flight line of data are preceded by a header record and followed by a trailer record. If a flight line is not complete on a given physical tape, no trailer record follows its last data record on the first tape, nor does a header record precede its first data record on the second tape.

### Header Record

The header record is 144 characters long with four defined fields for the summary geologic information and six defined fields for the averaged record data. The fields in common are:

1. Type of tape. A 32-character field with the text "STATISTICAL ANALYSIS" left justified.
2. Project identification. A 32-character field with, for example, the text "NTMS NL 16-1,2 RAWLINS" left justified. All project identification fields begin with "NTMS" followed by the sheet number. Additional information may be abbreviated.
3. Subcontractor name. A 10-character field with the text "GEOMETRICS."
4. System identification. A 6-character field with the aircraft registration number right justified.

The additional fields for the averaged record data are:

5. Flight line number. A 6-character field with the flight line number right justified.
6. Date flown. A 6-character field with the date, expressed as YYJJJ, right justified. YY are the last two digits of the calendar year and JJJ is the Julian date. When reflights require the insertion of data from multiple days' flying, the date used is that of the original flight.

Undefined fields of the header record are blank filled. A length of 144 is divisible by the number of characters per word of many popular computers.

### Trailer Record

A trailer record follows the last data record for the summary geologic information and the averaged record data for each flight line. This record is always 5472 characters long, all of which are the digit nine.

### Summary Geologic Information Record

Each summary geologic Information Record may contain up to 38 geologic map units (logical records), with each logical record 144 characters long. Therefore, the minimum physical length of the summary geologic information record is 144 characters and maximum physical length is 5472 characters.

The summary geologic information logical record has nineteen defined data fields.

1. Geologic map unit	A10,2X	1- 12
2. Potassium distribution	A 2	13- 14
3. Potassium measure of central tendency	F10.4	15- 24
4. Potassium standard deviation	F10.4	25- 34
5. Uranium distribution type	A 2	35- 36
6. Uranium measure of central tendency	F10.4	37- 46
7. Uranium standard deviation	F10.4	47- 56
8. Thorium distribution type	A 2	57- 58
9. Thorium measure of central tendency	F10.4	59- 68
10. Thorium standard deviation	F10.4	69- 78
11. Uranium/Thorium distribution type	A 2	79- 80
12. Uranium/Thorium measure of central tendency	F10.4	81- 90
13. Uranium/Thorium standard deviation	F10.4	91-100
14. Uranium/Thorium distribution type	A 2	101-102
15. Uranium/Potassium measure of central tendency	F10.4	103-112

16. Uranium/Potassium standard deviation	F10.4	113-122
17. Thorium/Potassium distribution type	A 2	123-124
18. Thorium/Potassium measure of central tendency	F10.4	125-134
19. Thorium/Potassium standard deviation	F10.4	135-144

### Data Record

Each record of averaged record data may contain up to 38 data scans (logical records), with each scan 144 characters long. Therefore, the minimum physical length of a data record is 144 characters and the maximum physical length is 5472 characters.

The data scan has the following defined data fields:

1. Record identification number	F10.2	1- 10
2. Latitude in degrees	F10.4	11- 20
3. Longitude in degrees	F10.4	21- 30
4. Residual magnetic field in gammas	F15.2	31- 45
5. Surface geologic map unit	5X,A8	46- 58
6. System/Quality flag code (SAKUT)	A 6	59- 64
7. Gross count rate (0.4-3.0 MeV), in cps	F 7.1	65- 71
8. Atmospheric Bi-214 count rate, in cps	F 7.1	72- 78
9. Thorium (Tl-208) count rate, in cps	F 7.1	79- 85

## MAGNETIC DATA TAPE

REFERENCE: PARAGRAPHS 4.7.4 AND 6.1.5, BFEC 1200-B

The MAGNETIC DATA TAPE is unlabeled nine track, 800 BPI, NRZI. All data are recorded as EBCDIC characters. Each tape contains but one file of header, data, and trailer records for no more than one NTMS quadrangle. The maximum record length is 4800 characters.

The tape is organized such that each flight line of data is preceded by a header record and followed by a trailer record. If a flight line is not complete on a given physical tape, no trailer record follows its last data record on the first tape, nor does a header record precede its first data record on the second tape.

### Header Record

The header record is 120 characters long with defined data fields. These fields are:

1. Type of tape. A 32-character field with the text "MAGNETIC DATA" left justified.
2. Project identification. A 32-character field with, for example, the text "NTMS NL 16-1,2 RAWLINS" left justified. All project identification fields begin with "NTMS" followed by the sheet number. Additional information may be abbreviated.
3. Subcontractor name. A 10-character field with the text "GEOMETRICS."
4. System identification. A 6-character field with the aircraft registration number right justified.
5. Flight line number. A 6-character field with the flight line number right justified.
6. Date flown. A 6-character field with the date, expressed as YYJJJ, right justified. YY are the last two digits of the calendar year and JJJ is the Julian date. When reflights require the insertion of data from multiple days' flying, the date used is that of the original flight.

The remaining 28 characters of the header record are blank filled. A length of 120 characters was chosen to allow for future expansion and because 120 is divisible by the number of characters per word of many popular computers.

### Data Record

Each data record may contain up to 40 data scans (logical records), with each scan 120 characters long. Therefore, the minimum physical length of a data record is 120 characters and the maximum physical length is 4800 characters.

The data scan has eleven defined data fields.

1. Record identification number	F10.2	1- 10
2. Latitude in degrees	F10.4	11- 20
3. Longitude in degrees	F10.4	21- 30
4. Time in day (hour, minutes, seconds)	3I2	31- 36
5. Terrain clearance in feet	F 9.0	37- 45
6. Barometric pressure in inches of mercury	F 5.2	46- 50
7. Surface geologic map unit	A10	51- 60
8. Total magnetic field in gammas	F10.2	61- 70
9. Residual magnetic field in gammas	F10.2	71- 80
10. System identification code	I 2	81- 82
11. Optional data	28X	83-110
12. Base station magnetic field in gammas	F10.2	111-120

### Trailer Record

A trailer record follows the last data record for each flight line. This record is always 4800 characters long, all of which are the digit nine.

## RAW SPECTRAL DATA TAPE

REFERENCE: PARAGRAPHS 4.7.1 AND 6.1.5, BFEC 1200-B

The RAW SPECTRAL DATA is unlabeled nine track, 800 BPI, NRZI. All data are recorded as EBCDIC characters. Each tape contains but one file of header, data, and trailer records for no more than one NTMS quadrangle. The maximum record length is 5472 characters.

The tape is organized such that each flight line of data is preceded by a header record and followed by trailer record. If a flight line is not complete on a given physical tape, no trailer record follows its last data record on the first tape, nor does a header record precede its first data record on the second tape.

### Header Record

The header record is 144 characters long with seven defined data fields. These fields are:

1. Type of tape. A 32-character field with the text "RAW SPECTRAL DATA" left justified.
2. Project identification. A 32-character field with, for example, the text "NTMS NL 16-1,2 RAWLINS" left justified. With the exception of special projects, such as the Walker Field Test Pads and Lake Mead Dynamic Test Range, all project identification fields begin with "NTMS" followed by the sheet number. Additional information may be abbreviated.
3. Subcontractor name. A 10-character field with the text "GEOMETRICS."
4. System identification. A 6-character field with the aircraft registration number right justified.
5. Flight line number. A 6-character field with the flight line number right justified.
6. Data flown: A 6-character field with the date, expressed as YYJJJ, right justified. YY are the last two digits of the calendar year and JJJ is the Julian date.



7. Sample period. A 6-character field describing the spectrometer accumulation time. Examples are: 1.0 SEC, 0.5 SEC, etc.

The remaining 46 characters of the header record are blank filled. A length of 144 characters was chosen to allow for future expansion and because 144 is divisible by the number of characters per word of many popular computers.

### Data Record

Each data record may contain up to four data scans (logical records), with each scan 1368 characters long. Therefore, the minimum physical length of a data record is 1368 characters and the maximum physical length is 5472 characters.

The data scan has fifteen defined data fields.

1. Record identification	F10.2	1- 10
2. Latitude in degrees	F10.4	11- 20
3. Longitude in degrees	F10.4	21- 30
4. Time of day (HHMMSS)	312	31- 36
5. Total magnetic field in gammas	F 9.2	37- 45
6. Terrain clearance in feet	F 5.0	46- 50
7. Barometric pressure in inches mercury	F 5.2	51- 55
8. Outside temperature in degrees C	F 5.1	56- 60
9. Quality flag code (altitude)	I 4	61- 64
10. Raw count data - 4 detector	255I3	65- 829
11. Live time - 4 detector-in seconds	F10.5	830- 839
12. Raw count data - 2 detector	255I2	840-1349
13. Live time - 2 detector-in seconds	F10.5	1350-1359
14. Cosmic - 4 detector	I 5	1360-1364
15. Cosmic - 2 detector	I 4	1365-1368

If a scan is not within the recovered path locations, the latitude and longitude, data fields 2 and 3, set to 0.0000.

The quality flag code, data field 9, is made equal to 0000 if the radar altimeter is within specifications and equal to 1000 if the radar altimeter is not within specifications.

The raw count data, fields 10 and 12, are presented for channels 0 through 254, corresponding to energies from 0 to 3 MeV for both the downward looking (4 ) and upward looking (2 ) detector arrays. The accumulation periods for the 4 and 2 detectors are identical, so each scan has data for both detectors. The counts in each channel are observed, with no corrections for ADC dead time nor conversion to counts per second. Energy per channel is 11.82 KeV. Since the

spectrometer does not respond to energies below 200 KeV, the counts in channels 0 through 17 (varies with system) will always be zero.

The live times, data field 11 and 13, are calculated by subtracting the product of the gross counts (0 to 6 MeV) and ADC dead time (8 sec) from the actual accumulation period for the data scan. This procedure is valid because the successive approximation ADC used has a fixed conversion time of 8 sec regardless of pulse amplitude.

The cosmic counts, data fields 14 and 15, are observed with no corrections for ADC dead time nor conversion to counts per second.

The data scan logical record length of 1368 characters was chosen to allow recording of all spectrometer channels for both 4 and 2 detectors with little chance of individual channel overflow given accumulation times of approximately one second. If overflow does occur, the overflow value is represented modulo 1000 (4 detector) or modulo 100 (2 detector) with leading zeros not suppressed. The specific value of 1368 characters was chosen because it is divisible by the number of characters per word of many popular computers.

#### Trailer Record

Trailer record follows the last data record for each flight line. This record is always 5472 characters long, all of which are the digit nine.

APPENDIX B  
PRODUCTION SUMMARY  
GALLUP AND SHIPROCK QUADRANGLES  
ARIZONA AND NEW MEXICO  
ALBUQUERQUE  
NEW MEXICO

THE STANDARD DEVIATION VALUES LISTED IN THE STATISTICAL TABLE ARE INCORRECT. TRUE STANDARD DEVIATION VALUES CAN BE SCALED OFF OF THE HISTOGRAMS. ANY QUESTIONS CONCERNING THIS PROBLEM MAY BE DIRECTED TO ANY STAFF GEOPHYSICIST AT EG&G GEOMETRICS, (408) 734-4616, EXT. 500

APPENDIX B

DAILY PRODUCTION SUMMARY - OCTOBER 1978

S2f TRACKER N9AG

RATON BASIN PROJECT

10-20-78	Ferry from Santa Fe to Gallup New Mexico
10-21-78	Nil Production - Weather - Rain
10-22-78	" " " "
10-23-78	" " " "
10-24-78	" " " Low Cloud
10-25-78	" " Waiting for Ground to Dry

Gallup Quadrangle

10-26-78	678 Line Miles
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Gallup and Albuquerque Quadrangles

10-27-78	678 Line Miles
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10-28-78	565 " "
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Shiprock Quadrangle

10-29-78	616 Line Miles
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10-30-78	Nil Production - Weather - Low Cloud
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10-31-78	803 Line Miles
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Total Line Miles for October 1978 = 3340 miles

DAILY PRODUCTION SUMMARY - NOVEMBER 1978

S2f TRACKER N9AG

RATON BASIN PROJECT

11-01-78

Reflights Only

DAILY PRODUCTION SUMMARY - NOVEMBER 1978

LAMA I N47319

RATON BASIN PROJECT

11-06-78                      Ferry to Albuquerque

Albuquerque Quadrangle

11-07-78                      384 miles

11-08-78                      335 miles

11-09-78                      477 miles

11-10-78                      Weather - 1 Day

11-11-78                      Weather - 1 Day

11-12-78                      Weather - 1 Day

11-13-78                      387 miles

11-14-78                      Weather - 1 Day

11-15-78                      Weather - 1 Day

11-16-78                      Weather - 1 Day

11-17-78                      596 miles

11-18-78                      364 miles

11-19-78                      Ferry to Gallup

Shiprock Quadrangle

11-20-78                      560 miles

11-21-78                      474 miles

11-22-78                      222 miles

11-23-78                      Weather  
through 11/29/78

11-30-78                      250 miles

12-01-78                      194 miles

Total Miles Flown During Above Period - 4243 miles

DAILY PRODUCTION SUMMARY - NOVEMBER 1978

LAMA II N49531

Gallup Quadrangle

11-22-78	306 miles. Ferry to Gallup
11-23-78	276 miles
11-24-78	Weather - 1 Day
11-25-78	Weather - 1 Day
11-26-78	Weather - 1 Day
11-27-78	342 miles
11-28-78	507 miles
11-29-78	228 miles
11-30-78	480 miles

Albuquerque Quadrangle

12-01-78	56 miles
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Total Miles Flown For Above Period - 2195 miles



APPENDIX C  
MICROFICHE OF DATA  
GALLUP AND SHIPROCK QUADRANGLES  
ARIZONA AND NEW MEXICO  
ALBUQUERQUE  
NEW MEXICO



