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**Effects of LWR Coolant Environments on Fatigue Lives of
Austenitic Stainless Steels***

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EFFECTS OF LWR COOLANT ENVIRONMENTS ON FATIGUE LIVES OF AUSTENITIC STAINLESS STEELS

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ABSTRACT

The ASME Boiler and Pressure Vessel Code fatigue design curves for structural materials do not explicitly address the effects of reactor coolant environments on fatigue life. Recent test data indicate a significant decrease in fatigue life of pressure vessel and piping materials in light water reactor (LWR) environments. Fatigue tests have been conducted on Types 304 and 316NG stainless steel in air and LWR environments to evaluate the effects of various material and loading variables, e.g., steel type, strain rate, dissolved oxygen (DO) in water, and strain range, on fatigue lives of these steels. The results confirm the significant decrease in fatigue life in water. The environmentally assisted decrease in fatigue life depends both on strain rate and DO content in water. A decrease in strain rate from 0.4 to 0.004%/s decreases fatigue life by a factor of ≈ 8 . However, unlike carbon and low-alloy steels, environmental effects are more pronounced in low-DO than in high-DO water. At $\approx 0.004\%/s$ strain rate, reduction in fatigue life in water containing <10 ppb DO is greater by a factor of ≈ 2 than in water containing ≥ 200 ppb DO. Experimental results have been compared with estimates of fatigue life based on the statistical model. The formation and growth of fatigue cracks in austenitic stainless steels in air and LWR environments are discussed.

INTRODUCTION

The ASME Boiler and Pressure Vessel Code Section III, Subsection NB [1], which contains rules for the construction of Class 1 components for nuclear power plant, recognizes fatigue as a possible mode of failure in pressure vessel steels and piping materials. Cyclic loadings on a structural component occur because of changes in mechanical and thermal loadings as the system goes from one load set (e.g., pressure, temperature, moment, and force loading) to any other load set. For each pair of load sets, an individual fatigue usage factor

is determined by the ratio of the number of cycles anticipated during the lifetime of the component to the allowable cycles. Figure I-90 of Appendix I to Section III of the Code specifies fatigue design curves that define the allowable number of cycles as a function of applied stress amplitude. The cumulative usage factor (CUF) is the sum of the individual usage factors. The ASME Code Section III requires that the CUF at each location must not exceed 1.

Subsection NB-3121, of Section III of the Code states that the data on which the fatigue design curves (Figs. I-9.0) are based did not include tests in the presence of corrosive environments that might accelerate fatigue failure. Article B-2131 in Appendix B to Section III states that the owner's design specifications should provide information on any reduction to fatigue design curves necessitated by environmental conditions. Recent fatigue strain-vs.-life (S-N) data illustrate potentially significant effects of light water reactor (LWR) coolant environments on the fatigue resistance of pressure vessel and piping materials [2-7]. Therefore, the margins in the ASME Code may be less conservative than originally intended.

A program is being conducted at Argonne National Laboratory (ANL) to provide data and models for predicting environmental effects on fatigue design curves and an assessment of the validity of fatigue damage summation in piping and vessel steels under load histories typical of LWR components. Based on the existing fatigue S-N data, interim fatigue design curves that address environmental effects on fatigue life of carbon and low-alloy steels and austenitic stainless steels (SSs) have been proposed [8]. Statistical models have been developed for estimating the effects of various material and loading conditions on fatigue lives of materials used in the construction of nuclear power plant components [9,10].

This paper presents fatigue data on austenitic SSs under conditions where information is lacking in the existing S-N data base.

Table 1. Chemical composition (wt.%) of austenitic stainless steels used for fatigue tests

Material	Heat	Source	C	P	S	Si	Cr	Ni	Mn	Mo	Cu	N
Type 316NG ^a	D432804	Vendor	0.011	0.020	0.001	0.52	17.55	13.00	1.76	2.49	0.10	0.108
		ANL	0.013	0.020	0.002	0.49	17.54	13.69	1.69	2.45	0.10	0.105
Type 304 ^b	30956	Vendor	0.060	0.019	0.007	0.48	18.99	8.00	1.54	0.44	—	0.100

^aASME SA312 seamless stainless steel pipe (hot finished), 610 mm O.D. and 30.9 mm wall, fabricated by Sumitomo Metal Industries, Ltd. Solution-annealed at 1038–1093°C for 0.5 h and water-quenched.

^bSolution-annealed at 1050°C for 0.5 h.

Fatigue tests have been conducted on Types 304 and 316NG SS in air and LWR environments to evaluate the effects of material and loading variables such as steel type, strain rate, DO, and strain range, on the fatigue lives of these steels. Experimental results have been compared with estimates of fatigue life based on the statistical model. The formation and growth of fatigue cracks in austenitic stainless steels in air and LWR environments are discussed.

EXPERIMENTAL

Fatigue tests have been conducted on Types 316NG and 304 SS to establish the effects of LWR coolant environments on fatigue lives of these steels. The chemical composition of the two steels is given in Table 1. Smooth cylindrical specimens with 9.5-mm diameter and 19-mm gage length were used for the fatigue tests. Specimen gage length was given a 1- μ m surface finish in the axial direction to prevent circumferential scratches that might act as sites for crack initiation.

All tests were conducted at 288°C with fully reversed axial loading (i.e., $R = -1$) and a triangular or sawtooth waveform. The strain rate for the triangular wave and fast-loading half of the sawtooth wave was 0.4%/s. Tests in water were conducted in a small autoclave under stroke control where the specimen strain was controlled between two locations outside the autoclave. Tests in air were performed under strain control with an axial extensometer; specimen strain between the two locations used in the water tests was also recorded. Information from the air tests was used to determine the stroke required to maintain constant strain in the specimen gage length for tests in water; the stroke is gradually increased during the test to account for cyclic hardening of the material and to maintain constant strain in the specimen gage section. Details regarding the test facility and procedure have been described elsewhere [2]. The fatigue test data from the present study and those obtained previously [6] on Type 316NG in air and high-DO water are listed in Tables 2 and 3.

AIR ENVIRONMENT

The existing fatigue S-N data indicate that the fatigue lives of austenitic SSs in air is independent of temperature in the range from room temperature to 427°C. Fatigue lives of Types 304 and 316 SS are comparable and those of Type 316NG are superior. The effects of strain rate on fatigue life can not be established from the existing S-N data. Limited results suggest that some heats are sensitive to strain rate; fatigue life may decrease up to 30% with decreasing strain rate.

Statistical models have been developed for estimating the effects of material and loading conditions on the fatigue lives of

austenitic SSs [9,10]. These models are based on the JNUFAD* data base for "Fatigue Strength of Nuclear Plant Component" from Japan, the data compiled by Jaske and O'Donnell for developing fatigue design criteria for pressure vessel alloys [11], and the tests conducted by Conway et al. [12] and Keller [13], and at ANL [6]. In air, the fatigue life N of Types 304 and 316 SS is expressed as

$$\ln(N) = 6.690 - 1.980 \ln(\epsilon_a - 0.12) \quad (1a)$$

and that of Type 316NG SS as

$$\ln(N) = 7.072 - 1.980 \ln(\epsilon_a - 0.12), \quad (1b)$$

where applied strain amplitude ϵ_a is in %.

The fatigue lives of Types 304, 316, and 316NG SSs in air at various temperatures are compared with the values estimated from Eqs. 1a and 1b in Fig. 1. At temperatures of 25–450°C, the fatigue lives of Types 304 and 316 SS in air show no dependence on temperature. For Type 316NG, the results at room temperature show good agreement with estimates from the statistical model or the Jaske and O'Donnell curve, but the data at 288°C are lower than the estimated values at strain amplitudes < 0.2%.

Also, note that the ASME mean curve is not consistent with the existing fatigue S-N data for austenitic SSs. At strain amplitudes < 0.5%, the mean curve predicts significantly longer fatigue lives than those observed experimentally; the fatigue lives at 288°C and $\approx 0.25\%$ strain range fall very close to the ASME Code design curve. When the Code fatigue S-N curve for austenitic SSs was extended to 10^8 cycles, account was taken of this discrepancy, but no change was made to the curve for < 10^6 cycles. The statistical model shows good agreement with the Jaske and O'Donnell average curve.

The fatigue S-N data from the present study and those obtained previously on Type 316NG [6] in air are shown in Fig. 2. Also shown are the S-N curves estimated from the statistical model, best-fit curve proposed by Jaske and O'Donnell, and the ASME mean curve. The results show good agreement with estimates from the statistical model or Jaske and O'Donnell curve. However, for Type 316NG SS, the data at 288°C and strain ranges < 0.4% are lower than the predicted values.

The cyclic strain hardening of Type 316NG tested in air at room temperature and 288°C and strain ranges of 0.25–1.0% is shown Fig. 3. At both temperatures, the steel exhibits rapid hardening during

* Private communication from M. Higuchi, Ishikawajima-Harima Heavy Industries Co., Japan, to M. Prager of the Pressure Vessel Research Council, 1992. The old data base "FADAL" has been revised and renamed "JNUFAD."

Table 2. Fatigue test results for Type 316NG austenitic stainless steel

Test No.	Environ.	Dissolved Oxygen ^a (ppb)	pH at RT	Conductivity (mS/cm)	Tensile Strain Rate (%/s)	Comp. Strain Rate (%/s)	Stress Range (MPa)	Strain Range (%)	Life N ₂₅ (Cycles)
<u>Room Temp.</u>									
1394	Air	—	—	—	5.0E-1	5.0E-1	694.7	1.51	4,649
1391	Air	—	—	—	5.0E-1	5.0E-1	554.8	1.00	13,561
1390	Air	—	—	—	5.0E-1	5.0E-1	518.1	0.75	25,736
1396	Air	—	—	—	5.0E-1	5.0E-1	506.7	0.76	30,000
1420	Air	—	—	—	4.9E-1	4.9E-1	495.3	0.49	54,249
1392	Air	—	—	—	5.0E-1	5.0E-1	475.9	0.51	60,741
1393	Air	—	—	—	5.0E-1	5.0E-1	464.7	0.41	127,386
1395	Air	—	—	—	5.0E-1	5.0E-1	456.7	0.35	183,979
1397	Air	—	—	—	5.0E-1	5.0E-1	446.0	0.30	347,991
1398	Air	—	—	—	5.0E-1	5.0E-1	436.7	0.27	666,000
1399	Air	—	—	—	5.0E-1	5.0E-1	431.8	0.25	>1,900,000
1400	Air	—	—	—	5.0E-1	5.0E-1	427.4	0.25	1,775,000
<u>288°C</u>									
1408	Air	—	—	—	5.0E-1	5.0E-1	416.6	0.76	21,548
1790	Air	—	—	—	5.0E-3	5.0E-1	452.8	0.75	16,765
1409	Air	—	—	—	5.0E-1	5.0E-1	377.2	0.50	53,144
1410	Air	—	—	—	5.0E-1	5.0E-1	377.6	0.50	51,194
1792	Air	—	—	—	5.0E-3	5.0E-1	413.4	0.51	35,710
1407	Air	—	—	—	5.0E-1	5.0E-1	364.4	0.40	82,691
1430	Air	—	—	—	5.0E-1	5.0E-1	348.3	0.30	168,852
1435	Air	—	—	—	5.0E-1	5.0E-1	342.0	0.25	314,352
1480	Air	—	—	—	4.9E-1	4.9E-1	340.1	0.25	319,308
1485	Air	—	—	—	5.1E-1	5.1E-1	340.4	0.25	369,206
<u>320°C</u>									
1405	Air	—	—	—	5.0E-1	5.0E-1	426.0	0.75	20,425
1404	Air	—	—	—	5.0E-1	5.0E-1	387.4	0.50	47,011
1406	Air	—	—	—	5.0E-1	5.0E-1	371.6	0.40	82,691
<u>288°C</u>									
1426	Hi DO	>200	—	—	8.0E-1	8.0E-1	405.1	0.80	12,069
1427	Hi DO	>200	—	—	8.2E-2	8.2E-2	421.7	0.82	6,679
1428	Hi DO	>200	—	—	7.4E-3	7.4E-3	441.4	0.74	5,897
1797	Hi DO	750	5.9	0.076	5.0E-3	5.0E-1	437.3	0.78	4,520
1414	Hi DO	>200	—	—	5.0E-1	5.0E-1	375.3	0.50	26,230
1418	Hi DO	>200	—	—	5.0E-1	5.0E-1	375.5	0.50	25,714
1423	Hi DO	>200	—	—	5.0E-2	5.0E-2	378.8	0.50	17,812
1425	Hi DO	>200	—	—	4.9E-3	4.9E-3	393.2	0.49	13,684
1431	Hi DO	>200	—	—	2.9E-1	2.9E-1	356.5	0.29	116,754
1434	Hi DO	>200	—	—	2.9E-2	2.9E-2	350.0	0.29	40,643
1436	Hi DO	>200	—	—	2.5E-2	2.5E-2	354.0	0.25	>1,719,851
1512	Hi DO	>200	—	—	2.4E-1	2.4E-1	361.2	0.24	2,633,954
1796	PWR	5	6.4	20.20	5.0E-1	5.0E-1	403.6	0.80	12,500
1812	PWR	2	6.5	20.00	5.0E-2	5.0E-1	413.9	0.80	6,375
1791	PWR	4	6.5	19.23	5.0E-3	5.0E-1	441.9	0.77	3,040
1793	PWR	4	6.4	19.23	5.0E-3	5.0E-1	434.3	0.80	3,020
1794	PWR	4	6.4	20.00	5.0E-3	5.0E-1	390.9	0.50	7,370

^aPWR = simulated PWR water containing 2 ppm lithium and 1000 ppm boron.

Table 3. Fatigue test results for Type 304 austenitic stainless steel at 288°C

Test No.	Environ.	Dissolved Oxygen ^a (ppb)	pH at RT	Conductivity (mS/cm)	Tensile Strain Rate (%/s)	Comp. Strain Rate (%/s)	Stress Range (MPa)	Strain Range (%)	Life N ₂₅ (Cycles)
1801	Air	—	—	—	4.0E-1	4.0E-1	419.2	0.76	24,500
1805	Air	—	—	—	4.0E-3	4.0E-1	467.9	0.76	14,410
1804	Air	—	—	—	4.0E-1	4.0E-1	382.8	0.51	61,680
1806	PWR	4	6.0	18.87	4.0E-1	4.0E-1	428.9	0.73	11,500
1810	PWR	5	6.4	18.89	4.0E-2	4.0E-1	447.6	0.77	5,800
1808	PWR	4	6.4	18.87	4.0E-3	4.0E-1	468.3	0.77	2,850
1807	PWR	4	6.5	18.87	4.0E-1	4.0E-1	374.6	0.51	25,900

^a PWR = simulated PWR water containing 2 ppm lithium and 1000 ppm boron.

Figure 1. Fatigue S-N behavior for Types 304, 316, and 316NG austenitic stainless steels in air at various temperatures

the first 50–100 cycles of fatigue life. The extent of hardening increases with applied strain range and is greater at room temperature than at 288°C. The initial hardening is followed by softening and a saturation stage at 288°C and by continuously softening at room temperature. Type 304 SS shows identical cyclic hardening behavior. The effect of strain rate on cyclic hardening of the two steels is shown in Fig. 4. Cyclic stresses increase with decreasing strain rate. For

Type 316NG, the extent of softening after the initial hardening period is less at slow strain rates.

The cyclic stress vs. strain curves for Types 304, 316, and 316NG SS at room temperature and 288°C are shown in Fig. 5; cyclic stress corresponds to the value at half life and at a strain rate of 0.4%/s. For the various steels, cyclic stresses increase in the following order;

Figure 2. Total strain range vs. fatigue-life data for Types 316 NG and 304 SS in air

Figure 3. Effect of strain range on cyclic strain-hardening behavior of Type 316NG stainless steel in air at room temperature and 288°C

Figure 4. Effect of strain rate on cyclic strain-hardening behavior of Types 316NG and 304 stainless steel in air at 288°C

Figure 5.
Cyclic stress-strain curve for Types 304, 316, and 316NG
stainless steels in air at room temperature and 288°C

The cyclic stress vs. strain curves for Types 304, 316, and 316NG SS at room temperature and 288°C are shown in Fig. 5; cyclic stress corresponds to the value at half life and at a strain rate of 0.4%/s. For the various steels, cyclic stresses increase in the following order; Types 316NG, 304, and 316. Furthermore, cyclic stresses are 20–30% lower at 288–430°C than at room temperature. At room temperature, the total strain amplitude ϵ_a (%) can be expressed in terms of the cyclic stress amplitude σ_a (MPa) for Type 316 SS by

$$\epsilon_a = \frac{\sigma_a}{1950} + \left(\frac{\sigma_a}{588.5} \right)^{1.94}, \quad (2a)$$

for Type 304 SS by

$$\epsilon_a = \frac{\sigma_a}{1950} + \left(\frac{\sigma_a}{503.2} \right)^{2.19}, \quad (2b)$$

and for Type 316NG by

$$\epsilon_a = \frac{\sigma_a}{1950} + \left(\frac{\sigma_a}{503.2} \right)^{2.19}. \quad (2c)$$

At 288–430°C, the cyclic stress vs. strain curve can be expressed for Type 316 SS by

$$\epsilon_a = \frac{\sigma_a}{1760} + \left(\frac{\sigma_a}{496.8} \right)^{2.19}, \quad (2d)$$

for Type 304 SS by

$$\epsilon_a = \frac{\sigma_a}{1760} + \left(\frac{\sigma_a}{373.9} \right)^{2.31}, \quad (2e)$$

and for Type 316NG by

$$\epsilon_a = \frac{\sigma_a}{1760} + \left(\frac{\sigma_a}{330.1} \right)^{3.24}. \quad (2f)$$

The cyclic stress vs. strain curves for Type 316 SS at 400°C (from JNUFAD data base) and 0.1, 0.01, and 0.001%/s strain rate, are shown in Fig. 6. The results indicate that at 288–430°C, cyclic stresses are increased with decreasing strain rate.

Figure 6. Effect of strain rate on cyclic stress-strain curve for Type 316 stainless steel in air

WATER ENVIRONMENT

Fatigue Life

The fatigue S-N data on austenitic SSs indicate a significant decrease in fatigue life in water. Also, the effect of environment on fatigue life increases with decrease in strain rate. However, the existing data base is inadequate to establish the effects of temperature or DO content on the fatigue lives of these steels; most of the data have been obtained in water containing 0.2 ppm or higher DO levels and at temperatures in the range of 288–320°C. Based on the available fatigue S-N data, statistical models have also been developed for estimating the fatigue lives of austenitic SSs in LWR environments [9,10]. The primary sources of relevant S-N data for austenitic SSs are the JNUFAD data base and the tests conducted by GE in a test loop at the Dresden 1 reactor [14,15] and at ANL [6]. In LWR environments, the fatigue life N of Types 304 and 316 SS is expressed as

$$\ln(N) = 6.331 - 1.980 \ln(\epsilon_a - 0.12) + 0.134 \dot{\epsilon}^* \quad (3a)$$

and that of Type 316NG as

$$\ln(N) = 6.713 - 1.980 \ln(\epsilon_a - 0.12) + 0.134 \dot{\epsilon}^*, \quad (3b)$$

where ϵ_a is applied strain amplitude in % and $\dot{\epsilon}^*$ is transformed strain rate defined as

$$\begin{aligned} \dot{\epsilon}^* &= 0 & (\dot{\epsilon} > 1 \text{ %/s}) \\ \dot{\epsilon}^* &= \ln(\dot{\epsilon}) & (0.001 \leq \dot{\epsilon} \leq 1 \text{ %/s}) \\ \dot{\epsilon}^* &= \ln(0.001) & (\dot{\epsilon} < 0.001 \text{ %/s}). \end{aligned} \quad (3c)$$

The statistical model is recommended for predicting fatigue lives $\leq 10^6$ cycles. The lower bound value of 0.001%/s on the strain rate effect was based on the results for carbon and low-alloy steels [2,3]. Note

that because of inadequate data base, the statistical models for austenitic SSs do not consider the effect of DO on fatigue life.

The fatigue S-N data for Types 316NG and 304 SS in water at 288°C are shown in Fig. 7; the ASME Code fatigue design curve is also shown in the figure. The results indicate significant decrease in fatigue life in water; the reduction in life depends both on strain rate and DO content in water. The fatigue lives of Types 316NG and 304 SS in air, simulated PWR, and high-DO water are plotted as a function of strain rate in Fig. 8. In all environments, the fatigue lives of these steels decrease with decreasing strain rate. The effect of strain rate is the smallest in air and largest in a low-DO PWR environment. In a simulated PWR environment, a decrease in strain rate from 0.4 to 0.004%/s decreases fatigue life by a factor of ≈ 8 . The decrease in life is lower at low strain ranges, e.g., a factor of ≈ 8 at 0.75% and ≈ 5 at 0.3% strain range.

The results also indicate that environmental effects on the fatigue lives of austenitic SSs are more pronounced in low-DO than in

Figure 7. Total strain range vs. fatigue-life data for Types 316 NG and 304 SS in air and water

Figure 8. Effect of strain rate on the fatigue lives of austenitic SSs in air, simulated PWR, and high-DO-water environments

The results also indicate that environmental effects on the fatigue lives of austenitic SSs are more pronounced in low-DO than in high-DO water. At slow strain rates, e.g., $\approx 0.004\%/s$, the reduction in fatigue life in simulated PWR environment (<10 ppb DO) is greater by a factor of ≈ 2 than in high-DO water (≥ 200 ppb DO). Such a dependence of fatigue life on DO content is quite different than that for ferritic steels. For carbon and low-alloy steels, environmental effects on fatigue life increase with increasing DO content above a minimum threshold value of 0.05 ppm. Environmental effects on fatigue life are modest at DO levels below 0.05 ppm, i.e., fatigue life is lower by a factor of < 2 than that in air. In view of these results, the statistical models for austenitic SSs (Eqs. 3a-3c) will be updated to incorporate the effects of DO and strain rate on fatigue life.

The cyclic strain-hardening behavior of Type 316NG stainless steel in air and PWR water at 288°C is shown in Fig. 9. The water environment has no effect on the cyclic stress response of austenitic SSs.

Metallographic Examination

A detailed examination of the fatigue test specimens was conducted to investigate the role of high-temperature oxygenated water on fatigue cracking. In general, the specimens tested in air show slight discoloration, whereas the specimens tested in oxygenated water developed a gray/black corrosion scale. Photomicrographs of the gage surface of Type 316NG specimens tested in air, simulated PWR, and high-DO water environments are shown in Fig. 10. Specimens tested in water have crystalline oxides and a thin gray corrosion scale. X-ray diffraction analyses of specimens tested in water indicate that the corrosion scale primarily consists of magnetite (lodestone Fe_3O_4 or ferroferric oxide FeFe_2O_4), chromium oxide (CrO), and maghemite ($\gamma\text{-Fe}_2\text{O}_3$). In addition to these phases, a specimen tested in high-DO water also contained hematite (ferric oxide or $\alpha\text{-Fe}_2\text{O}_3$).

Figure 11 shows photomicrographs of the fracture surface at approximately the same crack length for Type 316NG specimens tested in air, high-DO water, and low-DO PWR water. Fatigue striations can be seen clearly on all specimens. The spacing between striations indicate that crack growth rates increase in the following sequence: air, high-DO water, and low-DO PWR water.

The crack frequency, i.e., number of cracks per unit length of the specimen gage surface, was measured along longitudinal sections of specimens tested in different environments. Preliminary results indicate that crack frequency is lower in high-DO water than in air or the simulated PWR environment. For Type 316NG SS tested at 288°C , $\approx 0.75\%$ strain range, and $0.005\%/s$ strain rate, the number of cracks (longer than $20\ \mu\text{m}$) along a 7-mm gage length was 16, 14, and 8 in air, simulated PWR, and high-DO water, respectively.

The formation and growth of surface cracks appear to be different in simulated PWR water than in air, although the crack frequency is the same in all environments (Fig. 12). Metallographic examination of the test specimens indicates that in all environments, cracks primarily form within persistent slip bands (PSBs). During cyclic straining, strain localization in PSBs results in the formation of

Figure 9. Cyclic strain-hardening behavior of Type 316NG stainless steel in air and PWR water at 288°C

(a)

(b)

(c)

Figure 10. SEM photomicrographs of gage surface of Type 316NG SS specimens tested in (a) air, (b) simulated PWR, and (c) high-DO-water environments

(a)

(b)

(c)

Figure 11. Photomicrographs of the fracture surface of Type 316NG SS specimens tested at 288°C and 0.75% strain range in (a) air, (b) high-DO water, and (c) simulated low-DO PWR water

Strain Rate: 0.4%/s

Strain Rate: 0.004%/s

Air

Simulated PWR Water

Water with ≈ 0.7 ppm Dissolved Oxygen

Figure 12. Micrographs of surface cracks along longitudinal sections of Type 316 NG stainless steel in air and water environments

extrusions and intrusions at the surface; ultimately microcracks develop in these PSBs with continued cycling. Once a microcrack is formed, it continues to grow along its slip plane or a PSB as Mode II (shear) crack in Stage I growth. The orientation of the crack is usually 45° to the stress axis. The Stage I crack may extend across several grains before the increasing stress intensity of the crack promotes slip on systems other than the primary slip. Because slip is no longer confined to planes at 45° to the stress axis, the crack begins to propagate as a Mode I (tensile) crack, normal to the stress axis as Stage II growth. This behavior was observed in all specimens tested in air and in most instances for specimens tested in high-DO water.

In a simulated PWR environment (<10 ppb DO), the surface cracks appear to grow entirely as Mode I tensile cracks normal to the stress axis; this suggests that factors other than mechanical fatigue control the growth of surface microcracks in PWR water. The enhanced crack growth rates of pressure vessel and piping materials in LWR environments have been attributed to either the slip dissolution [16,17] or hydrogen-induced cracking [18] mechanisms. Enhanced crack growth by slip dissolution requires that a protective oxide film is thermodynamically stable, thus ensuring that a crack will propagate with a high aspect ratio without degrading into a blunt pit and that a strain increment occurs to rupture that film, thereby exposing the underlying matrix to the environment. Once the passive oxide film is ruptured, crack extension is controlled by dissolution of freshly exposed surfaces and the oxidation characteristics. The hydrogen-induced cracking of LASs is explained as follows: hydrogen produced by the oxidation reaction at or near the crack tip is partly absorbed into the metal; the absorbed hydrogen diffuses ahead of the crack tip and interacts with MnS inclusions, leading to the formation of cleavage cracks at the inclusion matrix interface; and linkage of the cleavage cracks results in discontinuous crack extension that is additional to that caused by mechanical fatigue.

The formation and growth of surface cracks in PWR water are consistent with both the slip dissolution or hydrogen-induced cracking mechanisms. However, lower fatigue lives in low-DO water than in high-DO water can not be reconciled in terms of the slip dissolution mechanism. If slip dissolution alone was responsible for enhanced crack growth rates in water, similar growth behavior (growth of surface cracks entirely as Mode I tensile cracks) and similar reduction in fatigue lives should also be observed in high-DO water. It is likely that hydrogen-induced cracking is important in low-DO water PWR environments. Fatigue tests are in progress on austenitic SSs to characterize the formation and growth of surface cracks in LWR environments.

CONCLUSIONS

The existing fatigue S-N data for austenitic stainless steels in air and water environments have been summarized. The results indicate that the fatigue lives of Types 304 and 316 SS are comparable and those of Type 316NG are superior. In air, the fatigue lives of austenitic SSs are independent of temperature in the range from room temperature to 427°C. Limited results suggest that some heats are sensitive to strain rate; fatigue life may decrease up to 30% with decreasing strain rate. The results indicate that the current ASME

mean curve is not consistent with the existing fatigue S-N data for austenitic SSs. At strain amplitudes <0.5%, the mean curve predicts significantly longer fatigue lives than those observed experimentally; the fatigue lives at 288°C and ≈0.25% strain range fall very close to the ASME Code design curve.

Fatigue tests have been conducted on Types 316NG and 304 SS to establish the effects of LWR coolant environments on fatigue lives of these steels. The results confirm the significant decrease in fatigue life in water; the decrease in life depends both on strain rate and DO content in water. Environmental effects on fatigue life are the same for Types 304 and 316NG austenitic SS. In a simulated PWR environment, a decrease in strain rate from 0.4 to 0.004%/s decreases fatigue life by a factor of ≈8. However, unlike carbon and low-alloy steels, environmental effects are more pronounced in low-DO than in high-DO water. At ≈0.004%/s strain rate, reduction in fatigue life in water containing <10 ppb DO is greater by a factor of ≈2 than in water containing ≥200 ppb DO. Experimental results have been compared with estimates of fatigue life based on the statistical model.

A detailed examination of the fatigue test specimens was conducted to investigate the role of high-temperature oxygenated water on fatigue cracking. The formation and growth of surface cracks appear to be different in simulated PWR water than in air. In all environments, cracks primarily form within PSBs. In air and for most cases in high-DO water, surface cracks initially grow along its slip plane as shear cracks in Stage I growth along planes at 45° to the stress axis. In a simulated PWR environment (<10 ppb DO), the surface cracks appear to grow entirely as tensile cracks in Stage II growth normal to the stress axis. Additional tests are in progress to characterize the formation and growth of surface cracks in LWR environments. Fatigue tests are also being conducted to define a threshold strain amplitude below which environmental effects on fatigue life do not occur, and the threshold strain rate below which effects of environment saturate.

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